



Fig. 1. The coordinate system

cone [7-10] so that the transverse curvature terms have to be retained in the governing differential equations. The result is that the differential equations are non-similar, which means the differential equations are no longer reducible to ordinary differential equations.

In this paper, the problem of natural convection flow over a frustrum of a cone will be treated for the case of specified wall heat flux. Similar to reference [8], the governing differential equations are first linearized by the quasilinearization method. The resulting equations are then solved by a finite-difference method outlined in detail in reference 11. Referring to Figure 1, the flow over a frustrum of a cone will approach to the flow over a full cone as x_0 approaches to zero. It should be pointed out that the differential equations in this analysis will be reduced to the similarity equations of Lin [6] if the transverse curvature terms are omitted.

Analysis

The governing differential equations for the solution of natural convection flow past a slender vertical frustrum of a cone with constant wall heat fluxes (Fig. 1) can be written in terms of dimensionless quantities as:

$$\frac{\partial(\bar{r}\bar{u})}{\partial\bar{x}} + \frac{\partial(\bar{r}\bar{v})}{\partial\bar{y}} = 0 \quad (1)$$

$$\bar{u} \frac{\partial\bar{u}}{\partial\bar{x}} + \bar{v} \frac{\partial\bar{u}}{\partial\bar{y}} = \frac{1}{\bar{r}} \frac{\partial}{\partial\bar{y}} \left\{ \bar{r} \frac{\partial\bar{u}}{\partial\bar{y}} \right\} + \theta \quad (2)$$

$$\bar{u} \frac{\partial\theta}{\partial\bar{x}} + \bar{v} \frac{\partial\theta}{\partial\bar{y}} = \frac{1}{\text{Pr}} \frac{1}{\bar{r}} \frac{\partial}{\partial\bar{y}} \left\{ \bar{r} \frac{\partial\theta}{\partial\bar{y}} \right\} \quad (3)$$

The boundary conditions are:

$$\bar{y} = 0: \bar{u} = 0; \bar{v} = 0; \frac{\partial\theta}{\partial\bar{y}} = -1$$

$$\bar{y} = \alpha: \bar{u} = 0; \theta = 0.$$

The dimensionless quantities in Eqs. (1-3) are related to their corresponding physical variables through the following definitions:

$$\bar{x} = \frac{x-x_0}{L}; \bar{y} = \frac{y}{L} \sqrt{\text{Re}_L}; \bar{u} = \frac{u}{u_c}; \bar{v} = \frac{v}{u_c} \sqrt{\text{Re}_L} \quad (4)$$

$$\bar{r}_0 = \frac{r_0(x)}{L} \sqrt{\text{Re}_L}; \theta = \frac{T-T_\infty}{\left(\frac{q_0 L}{k \sqrt{\text{Re}_L}} \right)}; \bar{r} = \frac{r}{L} \sqrt{\text{Re}_L}$$

where

$$u_c = \left\{ g_e \beta \cos \alpha \left[\frac{q_0 L}{k} \right] (L\nu)^{1/2} \right\}^{2/5} \quad (5)$$

$$\text{Re}_L = \frac{u_c L}{\nu}.$$

If the characteristic length, L , is chosen as the distance x_0 (Fig. 1), then the Reynolds number becomes:

$$\text{Re}_L = \frac{u_c x_0}{\nu} = \left\{ \frac{g_e \beta \cos \alpha (q_0 x_0 / k) x_0^3}{\nu^2} \right\}^{2/5} = \text{Gr}_{x_0}^{2/5} \quad (6)$$

where Gr_{x_0} is the Grashof number based on x_0 .

Next, the following transformation will be introduced:

$$\xi = \bar{x}; \eta = \frac{\bar{y}}{\bar{x}^{1/5}} \quad (7)$$

$$f(\xi, \eta) = \frac{\psi}{\bar{x}^{4/5} r_0}; \quad g(\xi, \eta) = \frac{\theta}{\bar{x}^{1/5}}$$

where the stream function ψ is defined by:

$$\bar{r}\bar{u} = \frac{\partial\psi}{\partial\bar{y}} \quad \text{and} \quad \bar{r}\bar{v} = -\frac{\partial\psi}{\partial\bar{x}} \quad (8)$$

and, for cones,

$$r_0 = x \sin \alpha.$$

Equations (1-3) then become:

$$\begin{aligned} & \left(\frac{\bar{r}_0}{\bar{r}}\right) \left\{ \left(\frac{\bar{r}}{\bar{r}_0}\right) \left[\left(\frac{\bar{r}_0}{\bar{r}}\right) f' \right] \right\}' + \\ & + \left(\frac{\bar{r}_0}{\bar{r}}\right) \left\{ \left(\frac{\bar{r}_0}{\bar{r}}\right) f' \right\}' \left\{ \left(R + \frac{4}{5}\right) f - \frac{1}{5} \eta f' \right\} \\ & - \left(\frac{\bar{r}_0}{\bar{r}}\right)^2 f' \left\{ \left(R + \frac{3}{5} - R \frac{\bar{r}_0}{\bar{r}}\right) f' - \frac{1}{5} \eta f'' \right\} + g \\ & = \xi \left\{ \left(\frac{\bar{r}_0}{\bar{r}}\right)^2 f' \frac{\partial f'}{\partial \xi} - \left(\frac{\bar{r}_0}{\bar{r}}\right) \left[\left(\frac{\bar{r}_0}{\bar{r}}\right) f' \right]' \frac{\partial f}{\partial \xi} \right\} \end{aligned} \quad (9)$$

$$\begin{aligned} & \frac{1}{Pr} \left\{ \left(\frac{\bar{r}}{\bar{r}_0}\right) g' \right\}' + \left(R + \frac{4}{5}\right) fg' - \frac{1}{5} f' g \\ & = \xi \left(f' \frac{\partial g}{\partial \xi} - g' \frac{\partial f}{\partial \xi} \right) \end{aligned} \quad (10)$$

subject to the boundary conditions:

$$\begin{aligned} \eta = 0: & f(\xi, 0) = f'(\xi, 0) = 0; g(\xi, 0) = -1 \\ \eta = \infty: & f'(\xi, \infty) = 0; g(\xi, \infty) = 0 \end{aligned}$$

where primes in Eqs. (9) and (10) represent differentiation with respect to η , and

$$R = \frac{\bar{x}}{\bar{r}_0} \frac{d\bar{r}_0}{d\bar{x}} = \frac{\bar{x}}{1 + \bar{x}} = \frac{\xi}{1 + \xi} \quad (11)$$

The ratio r/r_0 represents the effect of transverse curvature. For flows far downstream, r is very close to r_0 and the effects of the transverse curvature are negligible. Furthermore, the parameter R approaches to 1 when x becomes large. Equations (9) and (10) become similar and can be written as:

$$f''' + \frac{9}{5} ff'' - \frac{3}{5} (f')^2 + g = 0 \quad (12)$$

$$\frac{1}{Pr} g'' + \frac{9}{5} fg' - \frac{1}{5} f' g = 0 \quad (13)$$

which can be reduced to the form given by Lin [6] through the transformation:

$$\eta^* = A \eta, f^* = B f, \theta^* = C g \quad (14)$$

and

$$A = -C = \left(\frac{9}{20}\right)^{1/5}; B = A^4. \quad (15)$$

The starred quantities in Eq. (14) refer to the same variables in Lin [6]'s work.

For cones, we have

$$\begin{aligned} r &= r_0 + y \cos \alpha \\ r_0 &= x \sin \alpha. \end{aligned}$$

In terms of the similarity variables defined in Eq. (7), the ratio \bar{r}/\bar{r}_0 can therefore be written as:

$$\frac{\bar{r}}{\bar{r}_0} = 1 + \gamma \frac{\xi^{1/5}}{1 + \xi} \eta \quad (16)$$

where the TVG (transverse curvature) parameter γ is defined as:

$$\gamma = \frac{\text{Cot} \alpha}{Gr_{x_0}^{1/5}} \quad (17)$$

Substituting \bar{r}/\bar{r}_0 from Eq. (16) into Eqs. (9) and (10), we get:

$$\begin{aligned} & f''' - \frac{\gamma^*}{(\)^2} f'' + \frac{\gamma^*{}^2}{(\)^2} f' + \frac{1}{(\)} \times \\ & \times \left\{ \left(R + \frac{4}{5}\right) ff'' - \left[\left(R + \frac{3}{5}\right) - \frac{R}{(\)} \right] (f')^2 \right\} \\ & - \frac{\gamma^*}{(\)^2} f' \left\{ \left(R + \frac{4}{5}\right) f - \frac{1}{5} \eta f' \right\} + (\) g \\ & = \xi \left\{ \frac{f'}{(\)} \frac{\partial f'}{\partial \xi} - \frac{(\) f'' - \gamma^* f'}{(\)^2} \frac{\partial f}{\partial \xi} \right\} \end{aligned} \quad (18)$$

$$\begin{aligned} & g'' + \frac{\gamma^*}{(\)} g' + \left(R + \frac{4}{5}\right) \frac{Pr}{(\)} fg' - \frac{1}{5} \frac{Pr}{(\)} f' g \\ & = \xi \frac{Pr}{(\)} \left\{ f' \frac{\partial g}{\partial \xi} - g' \frac{\partial f}{\partial \xi} \right\} \end{aligned} \quad (19)$$

subject to the boundary conditions:

$$\begin{aligned} \eta = 0: & f(\xi, 0) = f'(\xi, 0) = 0; g(\xi, 0) = -1 \\ \eta = \infty: & f'(\xi, \infty) = 0; g(\xi, \infty) = 0 \end{aligned}$$

where the notation "()" in Eqs. (18) and (19) represents the two terms on the right-hand side of Eq. (16) and

$$\gamma^* = \gamma \frac{\xi^{1/5}}{1 + \xi} \quad (20)$$

Numerical Solutions

To solve Eqs. (18) and (19), they are first written as a first-order system. The derivatives are then approximated by centered-difference gradients and averages centered at the midpoints of the net rectangles defined by:

$$\begin{aligned} \xi_0 &= 0, \quad \xi_n = \xi_{n-1} + k_n, \quad n = 1, 2, \dots, N \\ \eta_0 &= 0, \quad \eta_j = \eta_{j-1} + h_j, \quad j = 1, 2, \dots, J \end{aligned} \quad (21)$$

$$\eta_J = \eta_\alpha$$

as shown in Figure 2. A non-uniform grid h_j defined by

$$h_j = K h_{j-1} \quad (22)$$

where the ratio of adjacent intervals, K , is a constant. The distance from the surface to the j th station is then given by:

$$\eta_j = h_j \frac{K^j - 1}{K - 1}, \quad j = 1, 2, \dots, J. \quad (23)$$

The ξ -direction grid k_n is arbitrary. Linearization is achieved by the method of quasilinearization and the resulting system of algebraic equations are then solved by an efficient block-tridiagonal factorization technique. Details of the method of solution are identical to the one used in reference 11 and are therefore omitted here.

In engineering applications, it is the surface temperature $T_w(x)$ that is of interest for the case in which the wall heat flux is specified. From the definitions of Eqs. (4) and (7),

$$\begin{aligned} [T_w(x) - T_\alpha]_{\text{TVC}} &= \\ &= \frac{(q_0 x_0 / k) \xi^{1/5}}{\left\{ \frac{g_e \beta \cos \alpha (q_0 x_0 / k) x_0^3}{\nu^2} \right\}^{1/5}} [g(\xi, 0)]_{\text{TVC}} \end{aligned} \quad (24)$$

where the subscript "TVC" means the effect of TVC is included. If the cone angle α is large, the effect of TVC is negligible and Eqs. (9) and (10) become:

$$\begin{aligned} f'''' + \left(R + \frac{4}{5} \right) f f'' - \frac{3}{5} (f')^2 + g \\ = \xi \left(f' \frac{\partial f'}{\partial \xi} - f'' \frac{\partial f}{\partial \xi} \right) \end{aligned} \quad (25)$$

$$\begin{aligned} g'' + \left(R + \frac{4}{5} \right) \text{Pr} f g' - \frac{1}{5} \text{Pr} f' g \\ = \xi \text{Pr} \left(f' \frac{\partial g}{\partial \xi} - g' \frac{\partial f}{\partial \xi} \right) \end{aligned} \quad (26)$$

which are also non-similar. The expression for the wall temperature is given by:

$$\begin{aligned} [T_w(x) - T_\alpha]_{\text{NO-TVC}} &= \\ &= \frac{(q_0 x_0 / k) \xi^{1/5}}{\left\{ \frac{g_e \beta \cos \alpha (q_0 x_0 / k) x_0^3}{\nu^2} \right\}^{1/5}} [g(\xi, 0)]_{\text{NO-TVC}} \end{aligned} \quad (27)$$

The effect of TVC as a result of slenderness can therefore be characterized by the following ratio:

$$R_{\text{TVC}} = \frac{[T_w(x) - T_\alpha]_{\text{TVC}}}{[T_w(x) - T_\alpha]_{\text{NO-TVC}}} = \frac{[g(\xi, 0)]_{\text{TVC}}}{[g(\xi, 0)]_{\text{NO-TVC}}} \quad (28)$$

If the TVC effect is negligible, the ratio R_{TVC} will be approximately 1. The ratio R_{TVC} is greater or smaller than 1 depending on whether the wall temperature T_w is increased or decreased, respectively, as a result of TVC.

To calculate the ratio R_{TVC} , Eqs. (25) and (26) are first solved, from which we get $[g(\xi, 0)]_{\text{NO-TVC}}$ as a function of ξ . Then (18) and (19) are solved, from which we obtain $[g(\xi, 0)]_{\text{TVC}}$ as a function of ξ . The ratio of the two sets of solutions then give the ratio R_{TVC} as a function of ξ . Table 1 gives the ratio R_{TVC} as a function of ξ for $\text{Pr} = 1.00$ and for a few values of TVC parameter, γ , namely 0.25, 0.5, 0.75 and 1.00, respectively. The following conclusions are drawn:

1. For all the cases considered, the effect of TVC is to decrease the wall temperature $T_w(x)$ for a given value of the TVC parameter γ . The decrease is seen to

Table 2. R_{TVC} for various Prandtl numbers (for $\gamma = 0.75$)

ξ	R_{TVC}			
	Pr = 0.1	Pr = 1.0	Pr = 10	Pr = 100
0.00	1.0000	1.0000	1.0000	1.0000
0.25	0.8293	0.8821	0.9151	0.9447
0.75	0.6801	0.7989	0.8703	0.9180
1.75	0.7082	0.8270	0.8948	0.9343
3.75	0.7734	0.8676	0.9205	0.9513
7.75	0.8303	0.9067	0.9473	0.9684
15.75	0.8873	0.9403	0.9666	0.9804
31.75	0.9273	0.9624	0.9800	0.9884
63.75	0.9553	0.9781	0.9882	0.9931
127.75	0.9729	0.9866	0.9930	0.9961

reach 25% for the largest value of the TVC parameter ($\gamma = 1$) presented in Table 1.

2. The ratio R_{TVC} first decreases with ξ , reaches a minimum at a certain location and then increases with ξ . Ultimately, R_{TVC} will approach to 1 at very large value of ξ which means the solutions will approach to the similarity solution of Eqs. (12) and (13). This is physically reasonable since, as ξ increases, the boundary layer thickness becomes small as compared with the radius of the cone, r_0 , and as a result the TVC effect becomes less important.

3. Larger values of the TVC parameter γ corresponds to more profound TVC effect. From the definition of γ , a large γ means either a smaller cone angle α or a smaller x_0 . In both cases, the boundary layer thickness becomes closer to the cone radius r_0 and an increase in the TVC effect results. Since an increase in the wall heat flux, q_0 , results in a decrease in the value of the TVC parameter γ ,

the TVC effect is seen to be inversely proportional to the wall heat flux (raised to the power of 1/5).

4. The effect of TVC is different for different values of Prandtl numbers. As an illustration, the wall temperatures are tabulated in Table 2 for one value of the TVC parameter and for a few values of Prandtl numbers. It is seen that the effect of TVC is more pronounced for smaller Prandtl numbers.

Similar to reference [8], we conclude the paper by showing how the solution for the frustrum of a cone as presented above can be used approximately for a full cone. Consider, for example, the problem of natural convection of air at 80°F over the frustrum of a cone ($L = 1$ ft, $x_0 = 2$ in. and $\alpha = 5^\circ$) with constant heat flux of 1000 Btu/hr-ft². For this case, the TVC parameter γ is found to be approximately 0.38. If we decrease x_0 from 2 in. to 0.5 in., the corresponding value of γ will be 1.15. For such a small value of x_0 (Fig.1), the solution for the frustrum of a cone becomes a good approximation of the boundary layer flow over a full cone with $L = 12.5$ in., especially in view of the fact that the boundary layer equations no longer apply near the tip of a full cone. This approximation is of significance since, from the mathematical point of view, a zero value of x_0 will give rise to a mathematical singularity in the transverse curvature factor \bar{r}/\bar{r}_0 in the differential equation, Eqs. (9) and (10). It should be noted that the same difficulty was by-passed in a similar manner in the analysis of the heat transfer over a slender cone by Kuiken [7]. In his work, Kuiken used in effect the inverse of the distance measured from the tip along the cone surface as the flow-direction co-

Table 1. Selected solutions of R_{TVC} for Pr = 1.00

ξ	$[g(\xi, 0)]_{NO-TVC}$	R_{TVC}			
		$\gamma = 0.25$	$\gamma = 0.50$	$\gamma = 0.75$	$\gamma = 1.00$
0.00	1.8729	1.0000	1.0000	1.0000	1.0000
0.25	1.8282	0.9547	0.9157	0.8821	0.8467
0.75	1.7841	0.9199	0.8542	0.7989	0.7493
1.75	1.7325	0.9307	0.8743	0.8270	0.7871
3.75	1.6930	0.9494	0.9058	0.8676	0.8334
7.75	1.6649	0.9653	0.9345	0.9067	0.8817
15.75	1.6498	0.9787	0.9589	0.9403	0.9227
31.75	1.6406	0.9868	0.9743	0.9624	0.9511
63.75	1.6368	0.9925	0.9852	0.9781	0.9710
127.75	1.6339	0.9955	0.9909	0.9866	0.9823
∞	1.6327	1.0000	1.0000	1.0000	1.0000

ordinate (ξ). As a result, his solutions started from the downstream side and integrated step-by-step in the opposite direction of the flow. Therefore, the tip of the cone in Kuiken's analysis corresponds to ξ equals to infinity. Since his integration stops at a finite value of ξ , the solution never reaches the tip of the cone. By this way, singularity at the tip of the cone was avoided.

References

1. Merk, H.J.; Prins, J.A.: Thermal Convection in Laminar Boundary Layer. Appl. Sci. Res. 4A (1953) 11-24, 195-206
2. Hering, R.G.; Grosh, R.J.: Laminar Free Convection from a Non-Isothermal Cone. Int'l J. of Heat and Mass Transfer 5 (1962) 1059-1067
3. Hering, R.G.: Laminar Free Convection From a Non-Isothermal Cone at Low Prandtl Numbers. Int. J. of Heat and Mass Transfer 8 (1965) 1333-1337
4. Sparrow, E.M.; Guinle, L.D.F.: Deviation From Classical Free Convection Boundary Layer Theory at Low Prandtl Numbers. Int. J. of Heat and Mass Transfer 11 (1968) 1403-1415
5. Roy, S.: Free Convection From a Vertical Cone at High Prandtl Numbers. J. of Heat Transfer, Trans. ASME 96 (1974) 115-117
6. Lin, F.N.: Laminar Free Convection From a Vertical Cone With Uniform Surface Heat Flux. Letters in Heat and Mass Transfer 3 (1976) 49-58
7. Kuiken, H.K.: Axisymmetric Free Convection Boundary Layer Flow Past Slender Bodies. Int. J. of Heat and Mass Transfer 11 (1968) 1141-1153
8. Na, T.Y.; Chiou, J.P.: Laminar Natural Convection Over a Slender Vertical Frustum of a Cone. (to appear in Wärme- und Stoffübertragung 1979)
9. Cebeci, T.; Qasim, J.; Na, T.Y.: Free Convective Heat Transfer From Slender Cylinders Subject to Uniform Wall Heat Flux. Letters in Heat and Mass Transfer 1, No. 2 (1974) 159-162
10. Cebeci, T.: Laminar Free Convection Heat Transfer From the Outer Surface of a Vertical Slender Circular Cylinder. Proc. of the 5th International Heat Transfer Conference 3, No. NCl. 4 (1974) 15-19
11. Na, T.Y.: Numerical Solution of Natural Convection Flow Past a Non-Isothermal Vertical Flat Plate. Appl. Sci. Res. 33 (1978) 519-543

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