

# The Strain Amplification Sensor: A 3D-Printable Stand-alone Strain Gauge for Low-Cost Monitoring

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## Abstract

Current *in-situ* strain sensing techniques focus on determining accurate, time-histories of strain utilizing fairly complex sensing, compensation, data processing, and powering arrangements. A simpler and lower-cost strain sensing approach would open up more opportunities to use strain measurements to support engineering decision making. This work explores a fully mechanical, ultra-low cost strain sensor printed using additive manufacturing techniques. The accuracy of current additive manufacturing techniques are discussed, and the performance of the sensor in terms of accuracy, measurement repeatability, and batch-to-batch manufacturing variability are studied. An example of using the proposed sensor to measure transient thermal weld stresses is presented. Overall, the key challenge to such a sensor is shown to be the accuracy of pin and slot print features and the resulting slip and friction introduced into the sensor. A properly calibrated design printed with current state-of-the-art machines is shown to be capable of resolving strain changes on the order of one micro strain with good repeatability.

*Keywords:* Strain Gauge, Additive manufacturing, 3D printing, Structural Reliability, Weld Distortion

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## 1. Introduction

<sup>2</sup> As structural health monitoring becomes increasingly commonplace, in-  
<sup>3</sup> terest in sensing the response of structures in new conditions is growing. One  
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4 challenge for current sensing systems is very short-term monitoring. If the  
5 monitoring window is only a few hours or days long, then the set-up over-  
6 head for conventional strain-measuring systems significantly impacts their  
7 practicality. In certain applications, such as post-damage event support on  
8 oceangoing vessels or offshore oil installations, having sensors that are intrin-  
9 sically safe in explosive atmospheres is an additional challenge. This work  
10 proposes and tests a fully mechanical, ultra-low cost strain sensor printed us-  
11 ing additive manufacturing, or 3-D printing, techniques. In many ways this  
12 sensor is a simpler variant of a laboratory extensometers produced at the  
13 fraction of a cost of a laboratory device. It provides instantaneous indication  
14 of current strain directly in human-readable form. The objective of such a  
15 device is short-term strain sensing problems where understanding the current  
16 strain state is more important than obtaining a complete time history. Exam-  
17 ples of such uses include measuring welding-induced thermal stresses during  
18 construction, or rapidly instrumenting a structure after a damage event to  
19 understand strains local to the damage. This work discusses the development  
20 of this gauge, and the current limitations of additive manufacturing for this  
21 type of device.

22 The ability to sense structural responses has dramatically expanded in the  
23 past two decades. Conventional strain gauges, fiber optic, capacitive sensors  
24 and active sensing techniques have all been proposed. For example, Lim[1]  
25 has shown that distributed fiber optic sensors are promising when used to  
26 monitor the cross section deformation of pipes. S. Laflamme [2] designed  
27 a soft capacitive sensor to localize damage on large civil structures which  
28 measures strain induced capacitance change. Remote wind turbine blade  
29 monitoring [3] demonstrated that wired conventional strain gauges could aid  
30 in damage detection. Today a wide variety of sensor technologies exist, how-  
31 ever, the majority of the research focus to date has been on improving sensor  
32 capabilities, not on reducing sensor cost.

33 Discussion on cost reduction has focused on two primary avenues to date.  
34 The first is to minimize the installation costs by switching from a wired sys-  
35 tem to a wireless sensing system. On shipboard applications, running wires  
36 through watertight compartments is difficult and expensive. Even for bridges  
37 wired systems are expensive to implement [4]. Wireless systems can signif-  
38 icantly reduce these costs, so long as a means of providing energy to the  
39 sensors is possible [5]. H. Choi [6] proposed a cost effective wireless trans-  
40 mission that uses multi-hop data transmission between nodes to mitigate  
41 the energy used in transmitting data. Extracting energy from the struc-

42 ture has also been explored, for example [7] proposed a vibration harvesting  
43 method for long term monitoring of bridges. For short-term applications,  
44 battery powered systems have been proposed. Lynch et al. demonstrated  
45 a battery-powered wireless system that could be rapidly installed for strain  
46 and acceleration monitoring on a small patrol boat [8].

47 The second approach to reducing cost is to simplify the installation of  
48 strain gauges on the structure. By pre-configuring, wiring, and packaging  
49 the gauges into a compact unit that only needs to be attached to the struc-  
50 ture by simple lugs or bolts installation time is reduced. Such quick-attaching  
51 strain sensors are made by BDI for bridge sensors [9] or bolt-on strain sensors  
52 for silo weight estimates reduce installation time by removing surface prepara-  
53 tion, mounting, and temperature-compensation, and bridge wiring for the  
54 strain gauges. Both wireless and quick-attaching strain systems reduce the  
55 cost of installing a full monitoring system. However, such systems still re-  
56 quire electrical power, signal conditioning, and data acquisition systems to  
57 determine and display strain values. For short-term monitoring (hours to  
58 days) of the current strain in a handful of locations, could a more rapid and  
59 simpler solution be imagined?

60 Recent advances in additive manufacturing, or 3D printing, technology  
61 have enabled construction with increasing repeatability and accuracy [10]  
62 [11]. These developments are predicted to extend the utility of additive  
63 manufacturing from rapid prototyping to production-scale fabrication. Of  
64 the various additive manufacturing techniques available today such as di-  
65 rect deposition and laser sintered powders, the stereolithography technique  
66 is noted for its growing ability to produce precise parts. In stereolithogra-  
67 phy, an ultraviolet light source paints the shape to be fabricated in a vat  
68 of photo-polymer resin. When the light strikes the resin, the resin solidi-  
69 fies, and through repeated tracing of slices of the outline of a part, the part  
70 slowly emerges from the resin. In the past ten years there have been sig-  
71 nificant advances in photoinitiated polymerization [12], which have provided  
72 stereolithography the highest fabrication accuracy [13]. Recent work [14] has  
73 produced 3D monolithic structures with embedded electronics and printed  
74 interconnects using stereolithography technology. With the increasing ac-  
75 curacy of the stereolithography, the possibility of printing a mechanism that  
76 could amplify strain to human-readable motion appeared worth exploring.  
77 As such a device would be entirely mechanical and plastic, it would not re-  
78 quire power sources, data logging or other systems traditionally needed for  
79 strain sensing. The low cost of printing such a mechanism would also mean

80 that the device could be used in a disposable manner. This combination of  
81 low cost and simple installation could allow real-time strain sensing in appli-  
82 cations where traditional system struggle to be practical or cost-effective. In  
83 this paper, all 3D printed prototypes were produced by the stereolithography  
84 process.

85 A patented stand-alone mechanical strain gauge, the Strain Amplification  
86 Sensor (SAS) that optically records strain in real time is presented. First,  
87 the design considerations and overview of the development of the 3D-printed  
88 manufacturing technique are discussed, followed by sensor calibration, re-  
89 peatability, manufacturing variability, and a laboratory weld test. In the  
90 weld test the SAS is evaluated for its ability to provide real time deforma-  
91 tion measurement of transient displacement arising from the thermal input to  
92 the weld. Finally, future work extending the SAS technology and conclusions  
93 are discussed.

## 94 **2. Design**

95 The SAS is a 3D printable assembly using only mechanical methods to  
96 record strain, figure 1. The SAS records the average strain between two  
97 mounting points in both tension and compression through a sensor arm that  
98 activates a series of three amplifying lever arms. Because it is purely me-  
99 chanical, it could be made intrinsically safe for explosive environments by  
100 selecting an appropriate print material. In real time the SAS responds to  
101 strain observed on the measurement material and displays the reading on  
102 the sensor face. This allows easy, human-readable strain reporting during  
103 monitoring without the use of data acquisition system and computers. For  
104 fabrication monitoring and incident response, this simplicity is a key advan-  
105 tage.

106 By being 3D printable, the SAS can easily be tuned to different sensitivi-  
107 ties and detection ranges. Present testing is based upon magnetic attachment  
108 of the SAS. Magnetic attachment affords the user rapid installation and re-  
109 moval. 3D printing each sensor allows for the tunability with respect to both  
110 sensitivity and mounting configuration. The SAS' base can be modified and  
111 adapted to the contours of any surface or to mount between two nearby po-  
112 sitions on a structure that are not necessarily continuous. The base of the  
113 sensor can also be attached more permanently, or to non-magnetic substrates  
114 by using any epoxy compatible with the printed plastic. Smaller base sizes  
115 would also be possible with epoxy, though the overall length of the device

116 is related to the amplification factor of the device, and shrinking the overall  
 117 size of the device would lead to lower sensitivity.

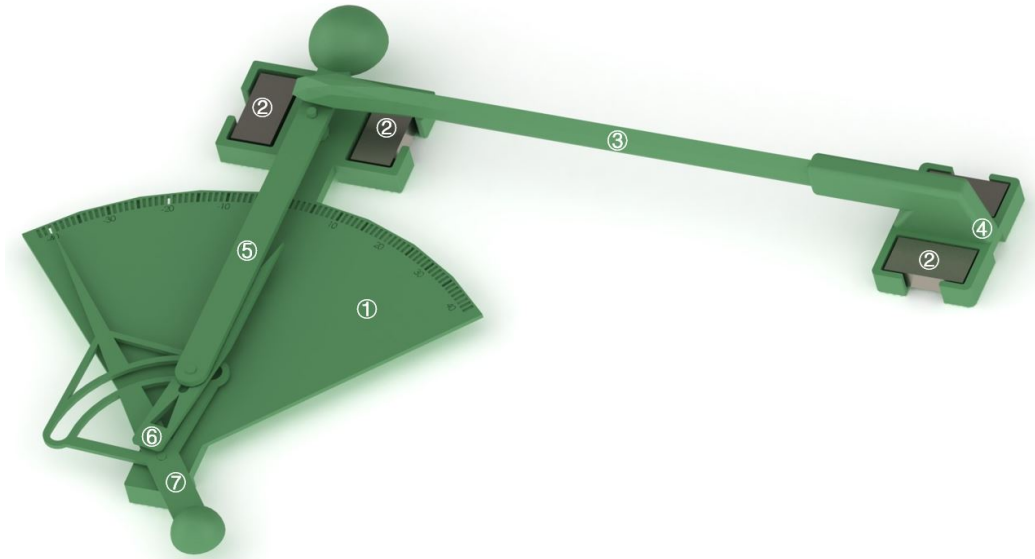


Figure 1: 3D model of SAS (① Sector Base Plate ② Magnet ③ Long Sensor Arm ④ Sensor Arm Base Plate ⑤ Actuator Arm ⑥ Pointing lever ⑦ Measurement Pointer)

118 The principle dimensions of the version of SAS tested are presented in  
 119 table 1. The sensor arm is designed to span long enough to amplify the strain  
 120 to a visual displacement on the sensor face, while the base can be modified  
 and adapted to the contour of the surface it is mounted on.

Length Overall	123mm
Width	84mm
Height	14mm

Table 1: SAS principle dimensions of the version evaluated

121 SAS operates by measuring the change in displacement between two  
 122 mounting locations. A rigid, cantilevered beam extends from one side to  
 123 the other. As the material being measured deforms, the rigid bar places  
 124 force on the other side of the SAS assembly. This force is translated to  
 125 motion of a mechanical system and the mechanical reaction divided by the  
 126

127 known distance over which the cantilevered bar spans produces the basis for  
128 strain measurement.

129 To transform displacement due to strain into a visually observable phe-  
130 nomena, significant amplification of the underlying motion needs to take  
131 place. SAS achieves this in two ways. Firstly, for each of its three levers,  
132 the fulcrum is placed closer to the lever end on which the excitation is being  
133 received, making the opposing end of the lever move over a larger distance  
134 proportional to the relative distance between the end points and the fulcrum.  
135 Second, the attachment location of the second lever arm to the third and fi-  
136 nal lever arm induces opposing relative motion between the second lever arm  
137 and the third. This interaction further amplifies the motion providing a to-  
138 tal theoretical amplification about 2150 of times. Slippage in the joints from  
139 imperfect manufacturing reduced this ratio to roughly 1,800 on the actual  
140 devices.

141 As the distance between the two base plates increases or decreases, the  
142 lever amplification system is activated. Figure 2 demonstrates the lever sys-  
143 tem reaction to a decreasing distance between the two base plates. This  
144 would correspond to compressive stress in the material being measured. The  
145 arrows indicate the moving directions of each individual part. The actua-  
146 tor arm and pointing lever move towards the system while the measurement  
147 pointer moves away from system. Oppositely, when the distance between  
148 two base plates increases, the actuator arm and pointer move away from the  
149 system while measurement pointer moves towards the system.

150 With 3D printing technology still maturing, much time was spent realizing  
151 something close to the CAD models in material form. Resolution capabilities  
152 for stereolithography were found to be significantly poorer than those adver-  
153 tised across the industry. In addition, feature accuracy was found to decrease  
154 with smaller feature sizes. Stereolithogrphahy prints in layers, which makes  
155 printing rounded shapes such as pins and holes somewhat challenging. The  
156 majority of the prototyping time was spent iterating through variations on  
157 the true dimensions until CAD input dimensions produced parts that were  
158 within the required dimensional tolerance.

159 The limitations of stereolithography primarily impacted the performance  
160 of mechanical joints in SAS. SAS uses two mechanical joint types, pin to hole,  
161 and pin to slot, as shown in Figure 3. Both connection types require snug  
162 but smooth interfaces. If SAS' mechanism generates too much internal stress  
163 from friction or interference, it seizes. And failure to transmit motion through  
164 connections dramatically reduces the sensitivity and accuracy of the design.

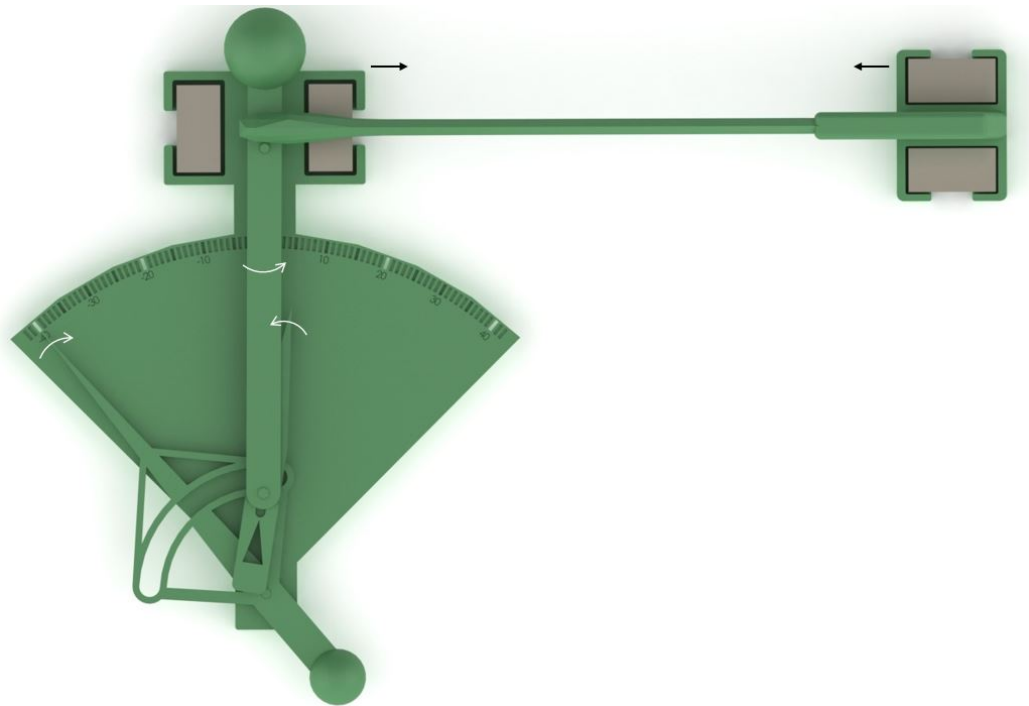


Figure 2: SAS movement illustration, initially in tension and showing movement as compression begins

165 Prior to print iteration, SAS was unable to detect a change from tension to  
166 compression or vice versa less than  $6\ \mu\text{m}$ ; after prototype iteration which  
167 mostly focused on optimizing mechanical joints dimensions, this transition  
168 gap became  $0.4\ \mu\text{m}$ .

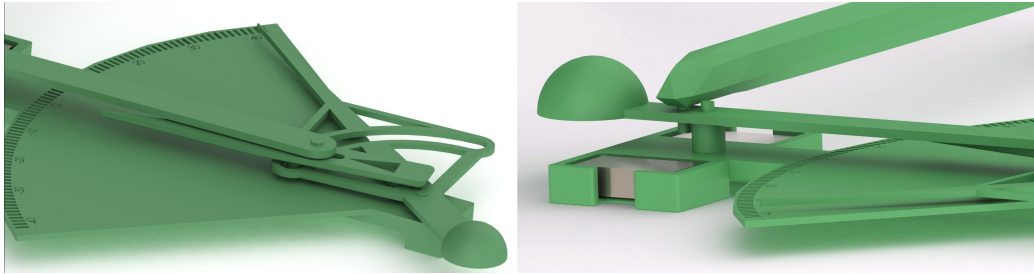


Figure 3: Pin and slot connections and pin and hole connections

169 It was found that a significant source of error in early designs was caused  
170 by the deflection of the long sensor arm under gravity. Deflection occurred  
171 both along its length and local to the driving pin at its end. Deflection at  
172 the end introduced a moment into the pin connection would cause internal  
173 stresses in the lever mechanism. To mitigate the deflection, topology op-  
174 timization was conducted on the sensor arm to reduce its deflection from  
175 vertical at the end located by the pin used to drive the lever mechanism.  
176 To further reduce internal stresses, each of the levers was balanced about its  
177 point of rotation. Balance was accomplished by adding counter weights in  
178 the form of half spheres for longer levers and placing lightening holes on the  
179 longer side of shorter levers.

### 180 3. Testing and Evaluation

#### 181 3.1. Evaluation of the 3D Printed Assembly

182 After several rounds of design iteration, a design and printing approach  
183 which produced a workable sensor emerged. However, given the novelty of  
184 additive manufacturing for this design, and the range of material choices  
185 available for such devices, the repeatability, batch-to-batch variability, and  
186 overall performance of the sensor needed to be investigated. A test program  
187 spanning more than 30 unique prototypes was used to perform this evalua-  
188 tion. All sensors were manufactured remotely by ProtoLabs, a commercial  
189 3D print company. 3D printing material remains largely non-standardized



190 and vendor specific. To characterize the material SAS was manufactured  
 191 from, the ASTM test results on the material provided by the vendor are  
 192 provided in Table 2. During the development of the sensor, several print  
 193 materials were tried. The material differed in the amount of internal fric-  
 194 tion they would create with identical or similar part designs, however, such  
 195 properties are not yet standardized. This added to the challenge in devel-  
 196 oping the prototype gauge. The 3D printer used was the 3D System Viper.  
 197 The printer’s specifications can be seen in Table 3. The cost for 3D printing  
 198 one set of SAS varied significantly throughout the project, but average near  
 199 400 dollars. Much of the variability seemed to be driven by the growing  
 200 demand for printing dental and medical implants with the same 3D print-  
 201 ing technique and machines. At this price point, the device is not yet truly  
 202 disposable however 3D printing costs are expected to continue to fall as the  
 203 technology becomes more established.

<i>ASTM Method</i>	<i>Property Description</i>	<i>Metric</i>	<i>English</i>
D638M	Tensile Modulus	2,100 MPa	305,000 psi
D638M	Tensile Strength at Break	44.9 MPa	6,500 psi
D638M	Elongation to Break	6.1%	6.1%
D790M	Flexural Strength	74.3 MPa	10,770 psi
D790M	Flexural Modulus	2,200 MPa	329,000 psi
D2240	Hardness (Shore D)	85	85
D256A	Izod Impact (Notched)	0.23 J/cm	0.46 ft lb/in
D570-98	Water Absorption	0.7%	0.7%
E831-05	C.T.E. -40°C-0°C (-40°F 32°F)	74.1 $\mu\text{m}/\text{m}\cdot^\circ\text{C}$	41.2 $\mu\text{in}/\text{in}\cdot^\circ\text{F}$
E831-05	C.T.E. 0°C-50°C (32°F 122°F)	96.3 $\mu\text{m}/\text{m}\cdot^\circ\text{C}$	53.6 $\mu\text{in}/\text{in}\cdot^\circ\text{F}$
E831-05	C.T.E. 50°C-100°C (122°F 212°F)	141.8 $\mu\text{m}/\text{m}\cdot^\circ\text{C}$	78.9 $\mu\text{in}/\text{in}\cdot^\circ\text{F}$
E831-05	C.T.E. 100°C-150°C (212°F 302°F)	182 $\mu\text{m}/\text{m}\cdot^\circ\text{C}$	101.3 $\mu\text{in}/\text{in}\cdot^\circ\text{F}$
D150-98	Dielectric Constant 60 Hz	3.16	3.16
D150-98	Dielectric Constant 1 KHz	3.12	3.12
D150-98	Dielectric Constant 1 MHz	2.94	2.94
D149-97a	Dielectric Strength	14.89 kV/mm	378 V/mil
E1545-00	Tg	49°C	120°F
D648	HDT @ 0.46 MPa (66 psi)	59°C	138°F
D648	HDT @ 1.82 MPa (264 psi)	50°C	122°F

Table 2: 3D Print Material Mechanical and Thermal/Electrical Properties (From row1(D638M) to row8(D570-98) are mechanical properties. From row9(E831-05) to row19(D648) are Thermal/Electrical Properties.)

<i>Equipment</i>	<i>Max Build Extents</i>	<i>Layer Thickness</i>	<i>Min Feature Size</i>
3D Systems Viper	5" × 5" × 2.5"	.001"	.002"

Table 3: 3D Printer Technical Specifications

204 *3.2. Test Apparatuses*

205 The objectives of the battery of test that were performed included: char-  
 206 acterizing the SAS's behavior by determining its response to incrementally  
 207 increasing strain, which serves as calibration, determining the repeatability  
 208 of the measurements recorded by SAS and their accuracy, and finally, de-  
 209 termining the differences in SAS's performance between different 3D printed  
 210 batches.

211 Before testing the entire SAS assembly, the amplification mechanism was  
 212 isolated and evaluated. Using a P-603 Piezo Movement Actuator from Physik  
 213 Instrument (PI), material strain displacement was simulated from both ten-  
 214 sion and compression, Figure 4. This test bed served as the basis for cali-  
 215 bration, repeatability testing, and manufacture deviation evaluation between  
 216 batches.

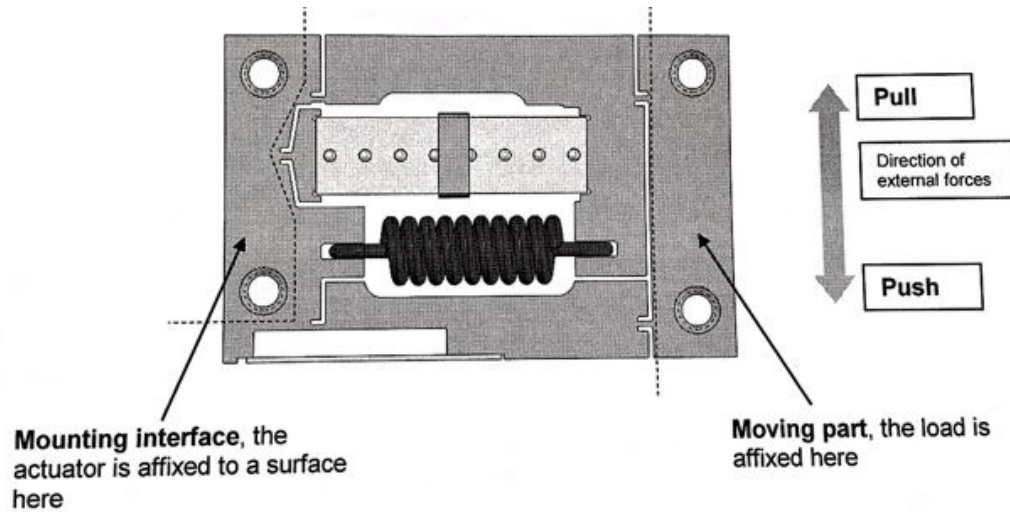


Figure 4: *Plan View of P-603 Piezo Actuator*. 2014. Technical Note of P-603 PiezoMove OEM Flexure-Guided, Lever-Amplified Actuators. Physik Instrument(PI). Germany

217 From the plan view depicted in Figure 4 of the P-603 piezo actuator,

218 the basis for driving SAS can be gathered. Fixing the left screw holes to a  
219 surface, the moving part on the right side pulls or pushes the SAS' sensor arm  
220 to simulate material tension or compression. An aluminum base plate was  
221 used to connect the SAS mechanism to the left side while another aluminum  
222 base was used to fix the driving side to a 3D printed driving bar on the  
223 right side. Figure 6 shows the SAS mounted. A 3D printed driving bar was  
224 chosen as opposed to extending the aluminum base to drive the mechanism to  
225 ensure proper simulation of the internal stresses between all joints, including  
226 the cantilevered beam.

227 *LABVIEW* was used to activate the piezo motor, through a E-709 Digital  
228 Piezo Controller. By commanding "MOV 1 1" or "MVR 1 1" to the write  
229 buffer, the piezo actuator can move to a specific position or move continuously  
230 with specific step and user-specified repeat time. This served as the basis for  
231 command inputs for all the piezo testing.

232 Figure 5 is the flowchart of the testing process based on piezo actuator.  
233 *LABVIEW* receives input commands and transfer them to piezo controller  
234 which can control piezo actuator's movement. The piezo controller and piezo  
235 actuator comprise a feedback system to implement the precise movement  
236 requested. A feedback-driven servo is integrated in the digital piezo controller  
237 to minimize the error between the command signal and the feedback signal  
238 from the position sensor embedded in the motion device.

239 After evaluating the amplification mechanism alone on the piezo actuator,  
240 an assembled SAS was tested on a pull tester. An aluminum test specimen  
241 was placed in the jaws of the pull tester device and SAS along with a conven-  
242 tional piezo electric wheatstone bridge strain gauge were place in the center of  
243 the specimen, Figure 7. This test set up allowed for testing of the entire SAS  
244 mechanism and sensor arm system in tensile stress, and direct comparison  
245 to conventional strain testing.

### 246 3.3. SAS Calibration

247 To calibrate the SAS, the piezo motor was given commands to move the  
248 mechanism at increments on the order of  $< 1\mu m$ . Time between each step  
249 was varied and the sensor's response to variation in time between induced  
250 motion was evaluated.

251 While testing with several different steps, it was discovered that if the  
252 step was too small or too large, SAS would react in an unpredictable man-  
253 ner. That is, for the version of SAS tested, if step sizes were smaller than  
254  $0.4\mu m$ , negligible movement resulted from SAS and over many of such steps,

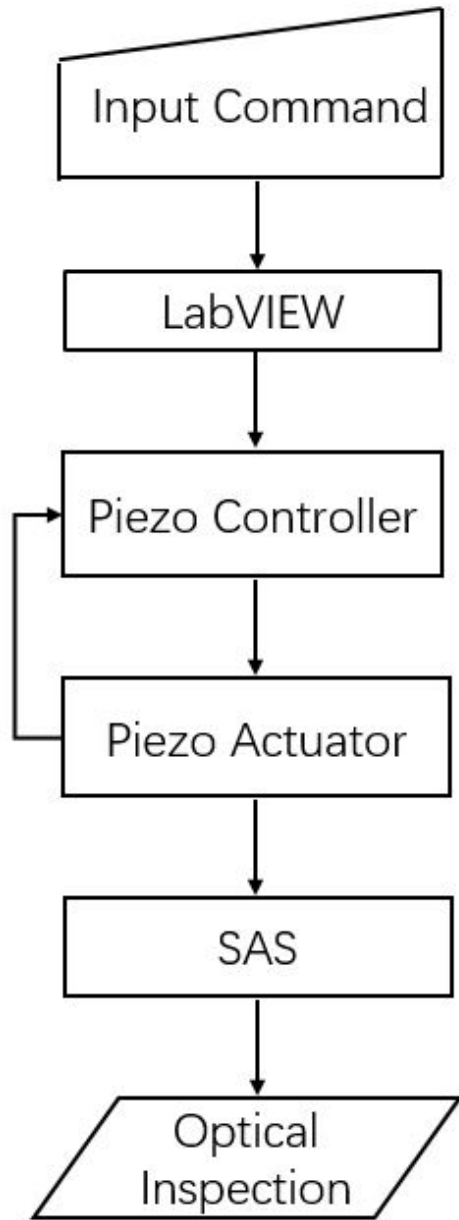


Figure 5: Flowchart for Piezo Actuator Based Test



Figure 6: SAS with Aluminum Base on Piezo Actuator (① Short Sensor Arm ② Aluminum Base)

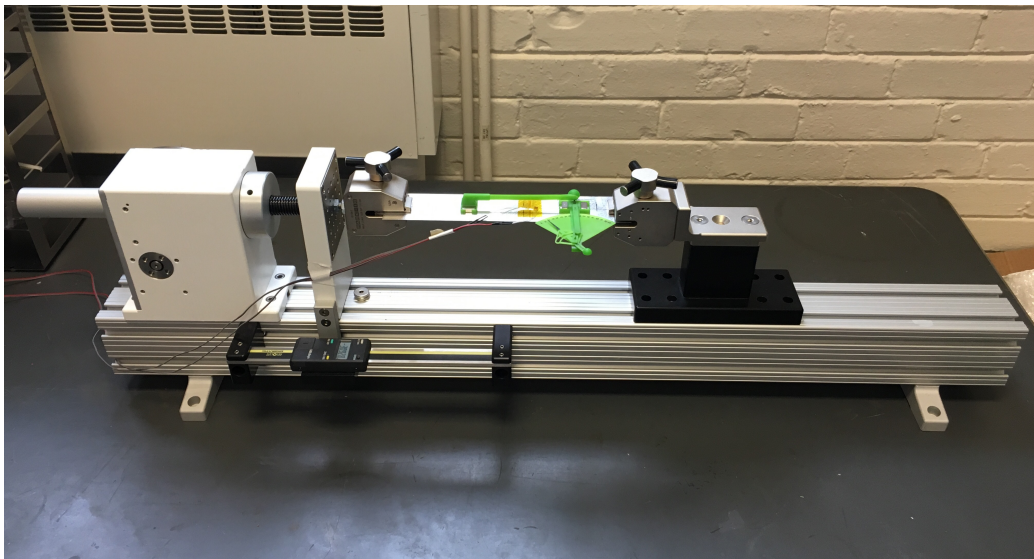


Figure 7: SAS on Pull Tester with Conventional Strain Gauges for Comparison

255 significant variability in cumulative motion would result. For this sensor con-  
256 figuration,  $0.4\mu\text{m}$  is less than the minimum detectable movement. Inability  
257 to control the rate of motor motion resulted in SAS variability for larger step  
258 sizes. The motor motion rate is akin to impulsive loading and not representa-  
259 tive of the types of strain the sensor was designed for. For larger steps,  $1\mu\text{m}$   
260 -  $2.5\mu\text{m}$ , the high rate of the applied displacement would result in dynamic,  
261 not quasi-static response in the gauge. Exposed to this impulse, the SAS  
262 would first pass the correct position and rebound back to a reading of lower  
263 accuracy. With this in mind, calibration step sizes were evaluated between  
264  $0.2\mu\text{m}$  to  $2.5\mu\text{m}$ , with  $0.6\mu\text{m}$  found to be the best step size.

265 The types of applications envisioned for SAS have slowly-varying loads,  
266 with load cycles on the order of several seconds or minutes. Thus, the dy-  
267 namic response of the gauge was not considered during design. However, SAS  
268 is a mechanical system and when operating on a perfectly flat plate and in-  
269 stalled with proper alignment it has one degree of freedom: translation of the  
270 long sensor arm towards or away from the mechanism side. Therefore there  
271 is inherently a natural frequency to the device which is entirely dependent  
272 upon the principle dimensions and build material. The rebounding effect  
273 and sensitivity described above are specific to the SAS configuration being  
274 test. Should SAS be revised for different ranges of sensitivities, rebounding  
275 characterization would need to be re-evaluated. When measuring impulsive  
276 or dynamic loads with a system like the one proposed here, consideration of  
277 the dynamic properties of the gauge is required.

278 Using  $0.6\mu\text{m}$  as the movement step, five complete passes through tension  
279 and compression ranges were conducted, covering  $40\mu\text{m}$  of total displace-  
280 ment. During each step, the pointer rotation angle was recorded. The accu-  
281 racy of this reading is 0.1 degrees, which can be achieved by visual inspection  
282 using markings that were included on the device by the 3D printing process.  
283 An increasing slope between pointer angle and displacement is expected when  
284 displacement is less than  $20\mu\text{m}$  since from 0 degree to 40 degree internal fric-  
285 tion is decreasing. While a decreasing slope is expected when displacement is  
286 greater than  $20\mu\text{m}$  with inner friction increasing again. Plotting the data and  
287 fitting the data with 4<sup>th</sup> order polynomial function, we acquired the relation  
288 between displacement and pointer angle, Figure 8. Then dividing displace-  
289 ment by sensor arm's length, a relationship between rotation angle and strain  
290 is produced. This polynomial provides the basis for measurement of strain  
291 using SAS. For an observed pointer angle and known distance between the  
292 two sensor bases (or length of the sensor arm) with strain being:

$$\eta = \frac{\Delta l}{l} \quad (1)$$

293 where  $\Delta l$  is the material length change undergoing tension or compression  
294 and  $l$  is the original length between two SAS base plates, we can solve for  
295 the  $\Delta l$  given an observed SAS pointer angle reading. The gauge has rotation  
296 angles directly printed on the plate beneath the pointer. Once a calibration  
297 plot is established for a mechanism design, visual angular readings taken in  
298 service can be converted into strain by the plot or the polynomial directly.

299 The calibration conducted with piezoelectric actuator differs slightly from  
300 the use of the gauge with the long sensing arm. Uncertainty in the lever  
301 mechanism that increases strain so that it provides a visual reading is the  
302 dominant uncertainty in calibration, and is captured with the piezoelectric  
303 technique. Errors associated with the magnetic base slipping, or off-center  
304 forces from the long arm are not included, however, these are expected to be  
305 smaller and are often mounting location dependent. Cross-talk errors from  
306 off-axis strain have not yet been characterized. Additionally, some human  
307 error from reading the gauge is likely to occur. Under laboratory conditions,  
308 the gauge could be read to 0.1 degrees of accuracy. However, errors owing  
309 to reading in field conditions, or comparison of readings between different  
310 engineers have not yet been characterized.

### 311 3.4. Repeatability

312 The repeatability of the results obtained during calibration was examined.  
313 Over time, the SAS' mechanical system could be subjected to wear and the  
314 measurements could deviate from those at the beginning of its life-cycle. To  
315 simulate cyclic deterioration, a total of 50 cyclic stress cycles were induced  
316 on the piezo motor test bed. At every  $10^{th}$  cycle the measurement accuracy  
317 of SAS was evaluated. Figure 9 displays the measurements from SAS at  
318 every  $10^{th}$  cycle. Error regarding each SAS measurement in  $i^{th}$  cycle was  
319 calculated by dividing the difference between the SAS measured and input  
320 displacements by the input displacements. With displacement increasing,  
321 the percentage error tends to decrease and the increase in error comes from  
322 SAS' static friction "sticking" which increases error locally. In order to assess  
323 error in every  $10^{th}$  cycle, errors associated with each SAS measurement were  
324 added up and averaged by total input displacement number. Table 4 shows  
325 the average error for  $i^{th}$  cycle. The errors are close indicating that SAS  
326 measurement is stable.

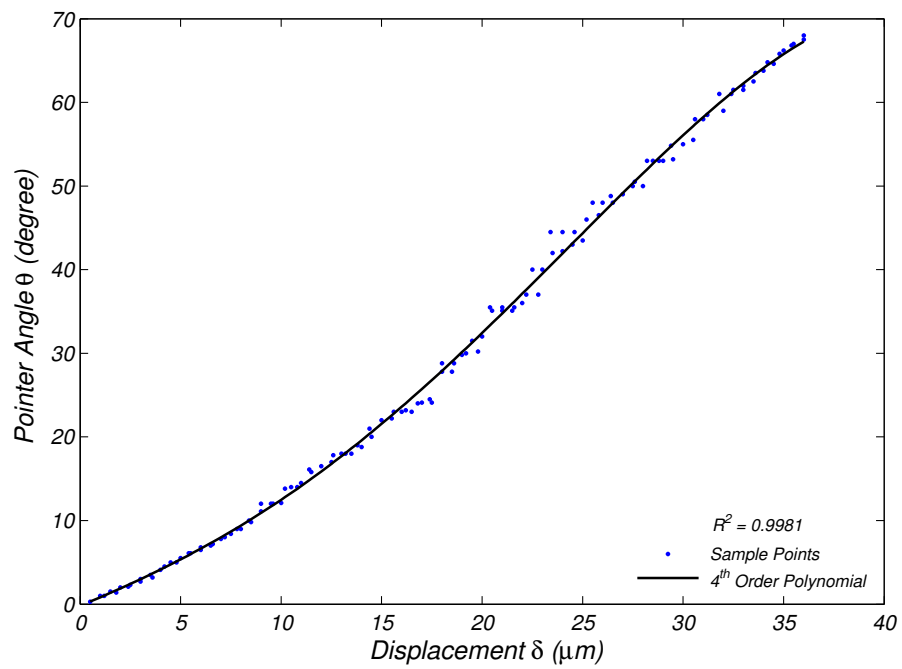


Figure 8: Calibration of the SAS mechanism on the piezo motor at  $0.6\mu\text{m}$  step sizes (Fitting Function:  $\theta = -4E^{-5}\delta^4 + 0.0016\delta^3 + 0.0138\delta^2 + 1.0056\delta - 0.1977$ ,  $R^2 = 0.9981$ )



327 For all of the tests performed below including calibration SAS experienced  
 328 some sticking in the mechanism's range of movement. Sticking is defined as  
 329 the mechanism failing to react to a change in input displacement for one  
 330 increment. This may come from rough edges in pin-hole or pin-slot joints  
 331 and one displacement increment is not enough to push through a patch of  
 332 roughness. Thus the error regarding to this input displacement(local error)  
 333 would increase. Slightly tapping on SAS would help it respond and applying  
 334 another increment of displacement would also result in a reaction that repre-  
 335 sents the total input displacement over the past two inputs. This effectively  
 336 meant SAS would "catch up" and again correctly represent the input dis-  
 337 placement. The frequency of the sticking occurrence over all test data points  
 338 was approximately 5%.

339 To look at the change in accuracy with changing cycles excluding the  
 340 sticking effect, a new figure was prepared with the points at which the SAS  
 341 stuck removed, Figure 10. We can see that all of the 5 cycles' percentage error  
 342 have similar trends decreasing with increasing displacement. This means  
 343 the absolute error for each displacement is close which indicates that SAS  
 344 measurement is stable. The percentage errors after  $32\mu\text{m}$  in  $30^{\text{th}}$  cycle and  
 345  $50^{\text{th}}$  cycle are increasing which is because SAS was stuck in the previous  
 346 movement and the error is thus increasing.

$i^{\text{th}}$ Cycle	10	20	30	40	50
Average Percentage Error(%)	2.75	2.97	2.81	2.86	3.10

Table 4: Average Percentage Error

347 As can be seen in Figure 11, the SAS measurement standard deviation of  
 348  $i^{\text{th}}$  cycle tends to decrease as the number of cyclic range passes increases. This  
 349 is likely because there is something of a break-in period for the 3D printed  
 350 plastic. Rough edges are smoothed with repeated passes. It is expected that  
 351 over time the standard deviation will become asymptotic before increasing  
 352 at some point when the mechanism begins to deteriorate. For short-term  
 353 monitoring applications however, such results indicate that an acceptable  
 354 life can be achieved with current additive manufacturing materials.

355 (The information in Figures 11 and 12 may be better presented as a bar  
 356 charts or tables)

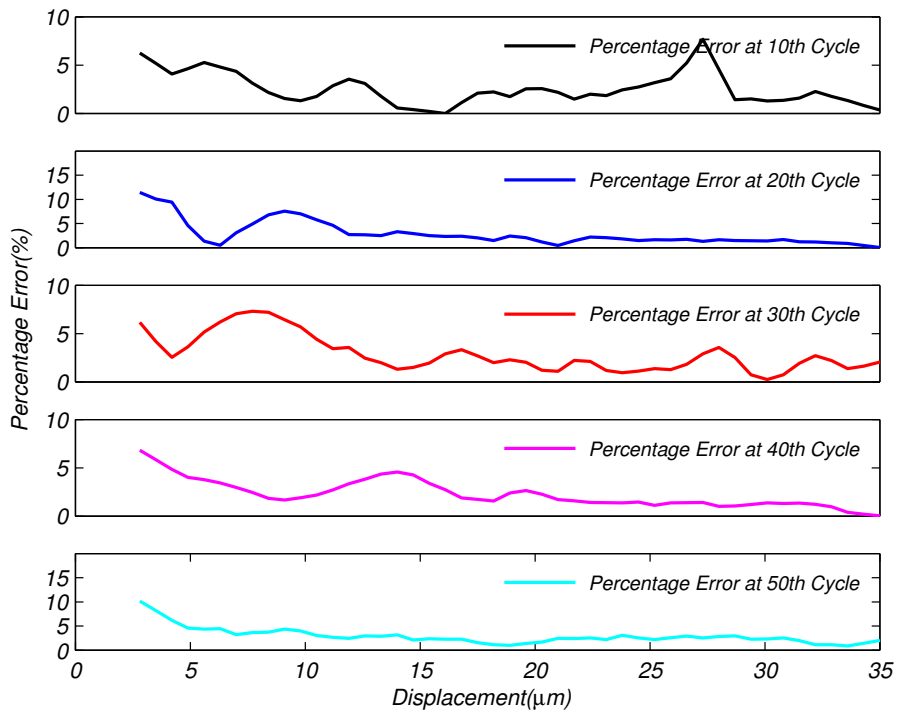


Figure 9: Repeatability test over 50 cycles with all points

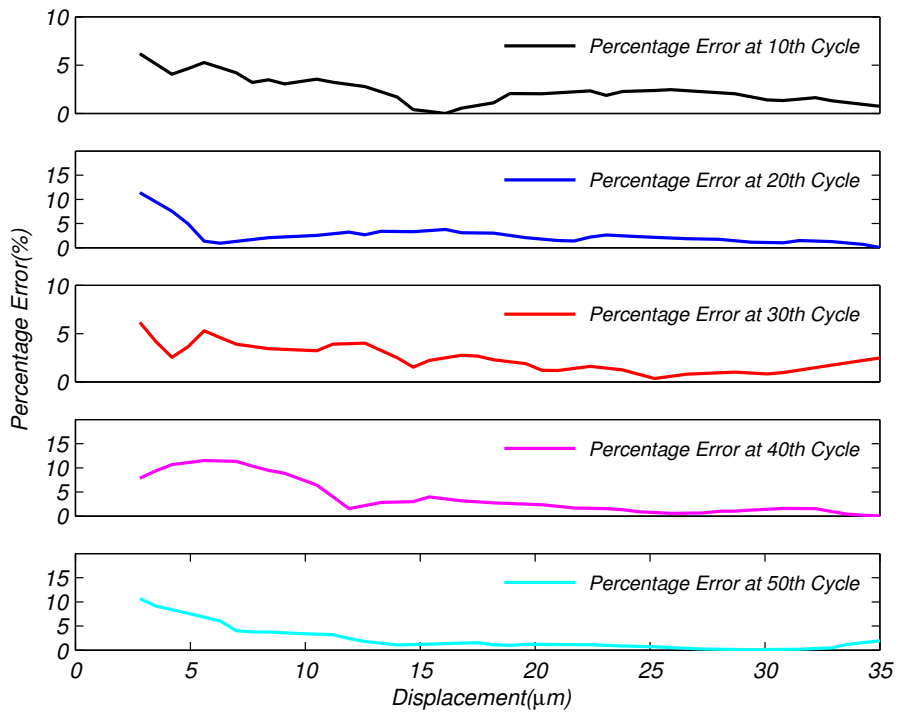


Figure 10: Repeatability test over 50 cycles with seizure points removed

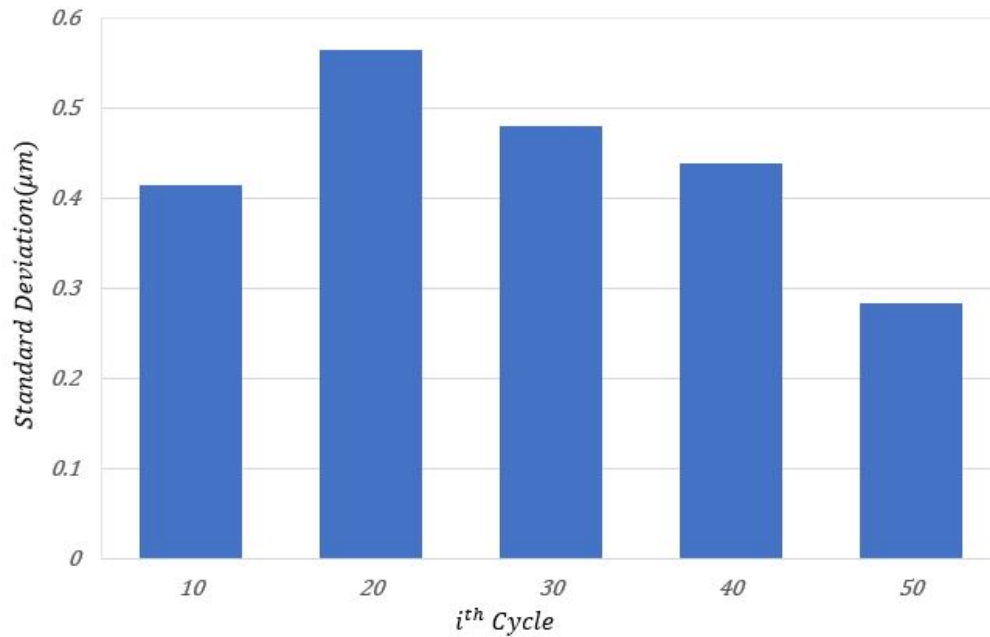


Figure 11: Repeatability Standard Deviation

### 357 3.5. Manufacturing Variability

358 As the SAS design requires very precise parts, near the limits of current  
359 3D printing capability, it was important to study the print-to-print variability  
360 of the device produced by the 3D printer. Four SASs printed were compared.  
361 These devices were printed on the same machine, with the same material in  
362 one print batch, but printed independently in the resin bath. Each SAS  
363 underwent 50 cycles as was completed in the above repeatability test and  
364 their variability was evaluated.

365 Figure 12 shows the standard deviation of the SAS measurement, com-  
366 pared across four sensors. The results show that three sensors are close in  
367 their standard deviation while one is significantly different. This may be  
368 attributed to the manufacturing tolerances which are  $\pm 0.05\text{mm}$ . The pin's  
369 nominal diameter is  $1.47\text{mm}$ , if a pin diameter is close to  $1.52\text{mm}$ , it could  
370 lead to a tight pin-hole connection with increasing internal friction which  
371 makes SAS more likely to experience sticking. If a pin diameter is close to  
372  $1.42\text{mm}$ , a loose pin-hole connection could appear and transfer movement  
373 less accurately. However, the overall variability between the four gauges was

374 small, even with the higher variability on gauge three. This indicate a good  
375 degree of repeatability exists with current additive manufacturing techniques,  
and that large production runs of such gauges would be feasible.

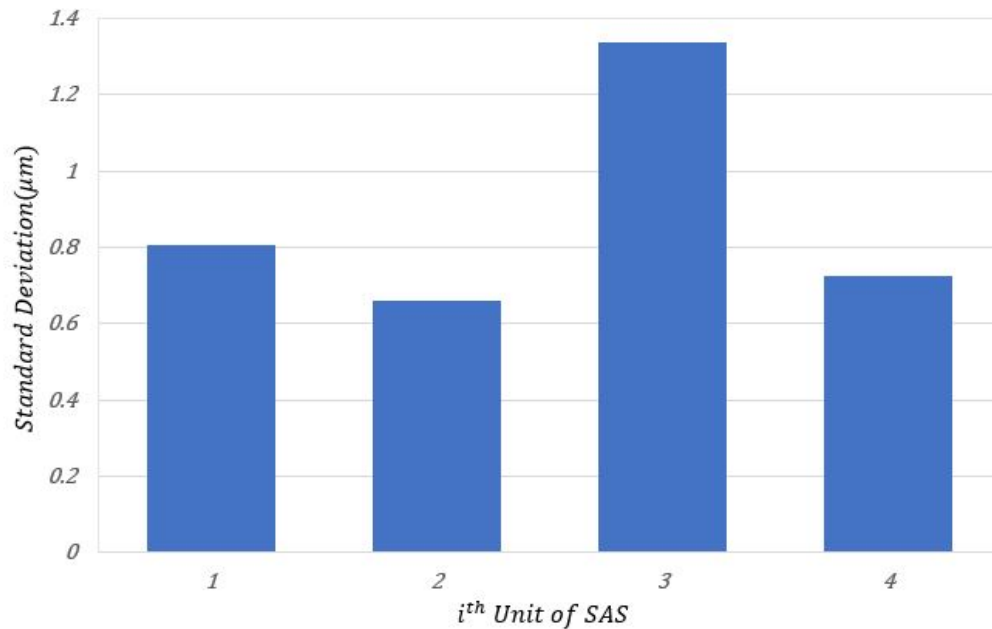


Figure 12: SAS displays small manufacturing variability gauge-to-gauge

376

### 377 3.6. Pull Tester Validation

378 The complete SAS assembly was tested on the pull tester in tensile stress  
379 to validate the calibration conducted on the piezo motor. SAS measurements  
380 were compared to data from two perpendicular strain gauges (for temperature  
381 compensation) wired in a 1/4 wheatstone bridge, as shown in Figure 13. A  
382 strong agreement between the measurements was observed. The average  
383 difference in measurement was 5%.

## 384 4. Weld Trial

385 Weld distortion during fabrication processes can result in significant mis-  
386 alignment of shipbuilding assemblies [15]. When upstream residual stresses  
387 and deformation cause structural components not to fit at assembly time,

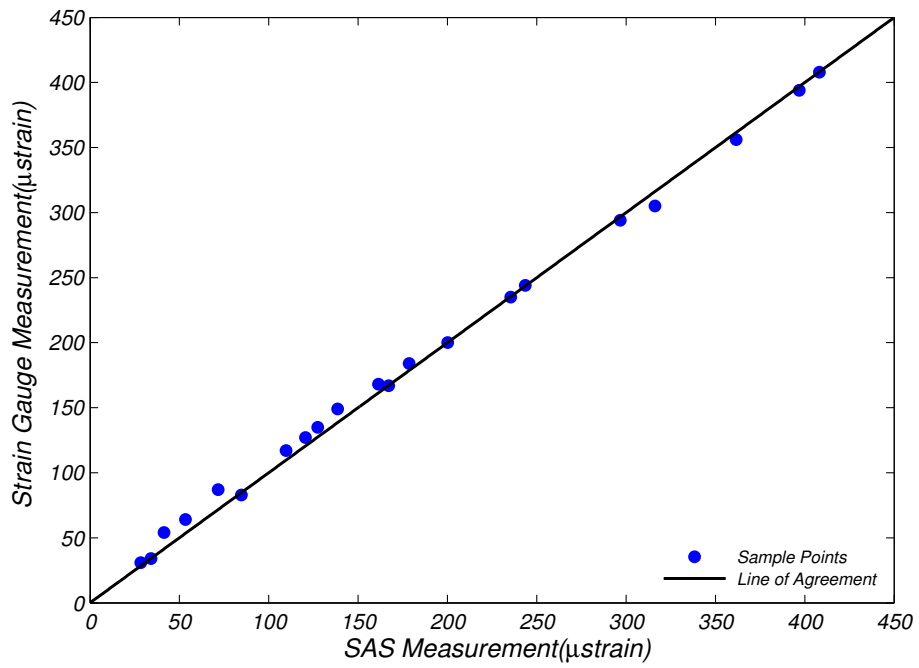


Figure 13: Comparison between SAS and conventional strain gauge on pull tester

388 mechanical force as well as thermal heating are used to re-align the struc-  
389 ture. This process is time consuming, can damage the structure if done too  
390 aggressively, and often damages machinery and equipment pre-outfitted to  
391 the structure. Past work has shown that such errors at assembly or grand  
392 block integration results in significant cost during shipbuilding [16]. An inex-  
393 pensive strain sensing technique such as SAS could give fabricators real-time  
394 visual feedback to the residual stress state in the structure. This information  
395 can translate to procedural changes and quality control checks that reduce  
396 the likelihood of misalignment and the resulting rework from weld deforma-  
397 tion. To explore the suitability for using SAS during welding for real-time  
398 strain measurements, a weld trial was conducted. Given the small size of  
399 laboratory specimen compared to a large ship or bridge structure, this work  
400 focused on transient thermal stresses from the welding process. Such stresses  
401 can reach similar magnitudes in small and large structures, where more com-  
402 plex residual stresses often need large structures with high degrees of restraint  
403 to develop.

404 Two 1/4 inch thick, 12 inch across square low carbon steel plates were butt  
405 tig welded together with a single pass. The SAS' sensor arm's was centered  
406 in the plate parallel to the weld. To help reduce the heat transferred to  
407 SAS during the weld process, given its proximity to the weld (6 inches), a  
408 copper bar was fixed beneath the plate and adjacent to the weld location.  
409 Additionally, a second piece of steel was clamped to the plate to form a  
410 barrier to prevent any weld spatter from striking the SAS and to block the  
411 light of the welding process so that the SAS movement could be captured  
412 by video camera. This barrier steel was not welded. Figure 14 shows the  
413 installation setup.

414 SAS reacted in real time during the welding process and cool down. Fig-  
415 ure 15 shows a time lapse of the movement while the weld bead was being  
416 laid and thereafter. During welding, the assembly compressed along the weld  
417 axis and SAS' measurement corresponded. SAS responded smoothly to the  
418 transient thermal stresses being generated, providing clear visual feedback to  
419 the welder and observers of the current state of the material. After the weld  
420 was completed, the camera was kept on while the material started to cool  
421 down. During cooling, the material's internal compression decreased, and the  
422 SAS measured the reduction of stress. SAS followed the entire strain change  
423 process and made it visibly observable. This trial demonstrated the practi-  
424 cality of a low-cost strain sensing system. Using only 3D-printed plastic parts  
425 and magnetic attachment, it was possible to make thermal welding strains

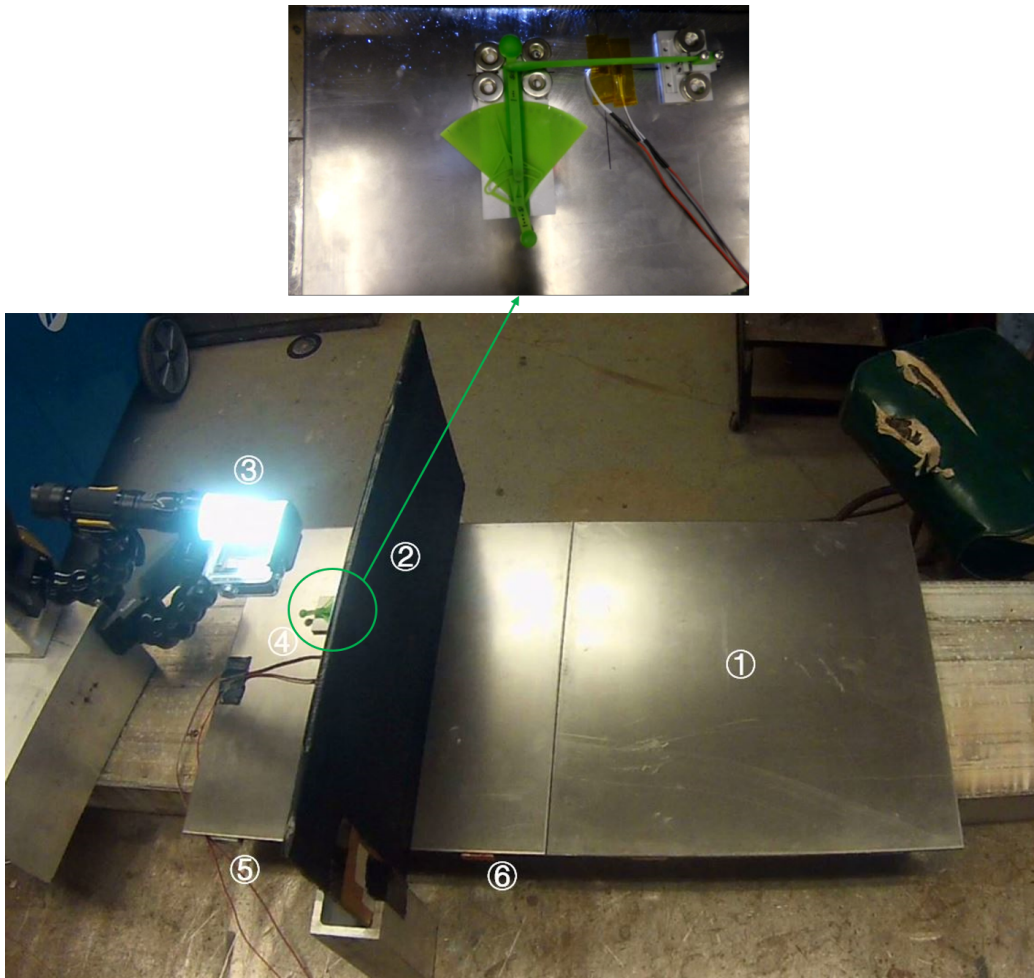


Figure 14: Welding preparation (① Steel Plate ② Protection Wall ③ Light and Camera ④ SAS ⑤ Aluminum Bar ⑥ Copper Bar)



426 visible to the welder and any inspectors examining the work piece. The ease  
 427 of installation and low cost nature of the gauge could allow structural strain  
 428 sensing in a variety of fabrication and incident response setting.

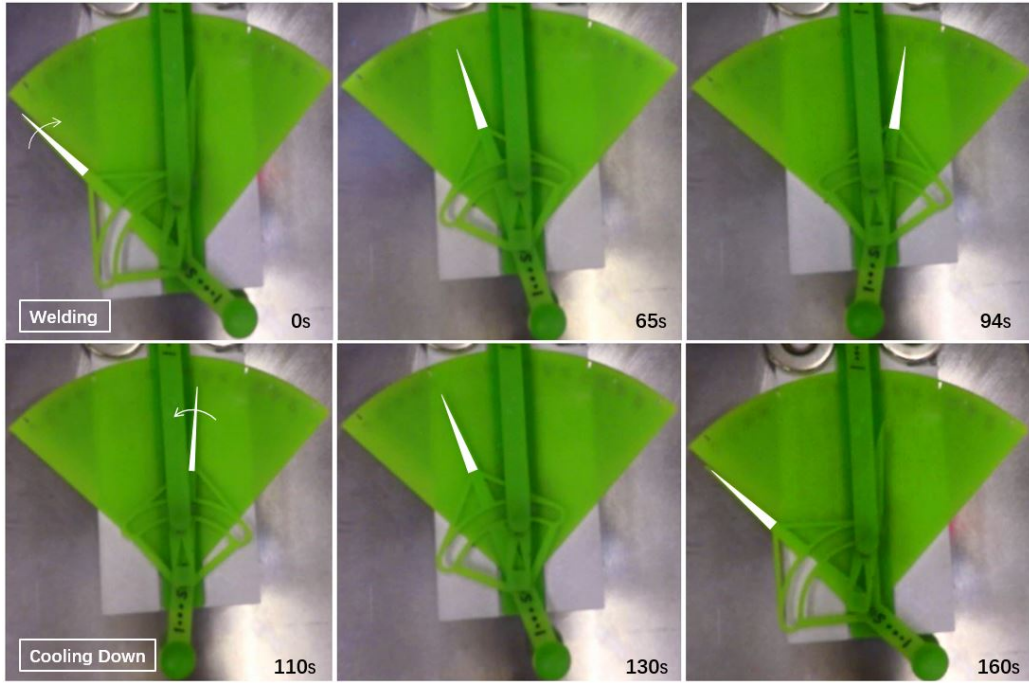


Figure 15: SAS movement during welding test (the first row indicates pointer was moving clockwise and steel was compressing during welding; the second row shows pointer was moving back counterclockwise which means steel was tensing during the cool down process.

429 **5. Conclusion**

430 A 3-D printable strain gauge was proposed as a new tool to allow *in situ*  
 431 strain monitoring when rapid installation time, low cost, and direct human  
 432 readability are important characteristics. Using a mechanical advantage ap-  
 433 proach, the gauge magnifies the change in length between two measuring  
 434 bases. This strain is then visually displayed on a large gauge with a needle.  
 435 The principle challenge in constructing such a gauge is minimizing the in-  
 436 ternal friction from the moving parts given the current accuracy of additive  
 437 manufacturing, especially for hole and slot construction.

438 A series of prototypes were tested to evaluate the accuracy, repeatability,  
439 and durability of the gauge. Overall, the gauge was shown to be sensitive and  
440 accurate, with a resolution on the order of a single microstrain. The primary  
441 problem encountered was temporary sticking of the gauge from internal fric-  
442 tion. This occurred in approximately 5% of the measuring points. There was  
443 some gauge-to-gauge variability in the consistency of the measurement, with  
444 three of the four tested gauges agreeing well, and one gauge with a larger  
445 variability. Over 50 strain cycles no significant degradation of the gauge was  
446 experienced, indeed, it appeared that the gauges may be improving after a  
447 “break-in” period, though this finding has not been validated by tear-down  
448 and inspection of the gauge geometry. Finally, the gauge was shown to be  
449 able to capture the build up and release of transient thermal stresses present  
450 during a TIG butt weld, displaying these stresses in real-time to the welder.

451 Overall, this initial work validated the concept of using additive manu-  
452 facturing to achieve low-cost strain sensing for short-term strain monitoring  
453 projects. While such a gauge in no way competes with conventional monitor-  
454 ing techniques for long-term time-history monitoring, for rapid applications  
455 during construction and incident response it is now practical to provide real-  
456 time strain feedback. Further print material-specific properties need to be  
457 examined, including the effect of temperature changes on the gauge, and the  
458 impact of humidity and water on the smoothness of the mechanism. Ad-  
459 ditionally, approaches to mechanically record the maximum value of strain  
460 experience would also allow the gauge to be left unattended for short periods  
461 of time.

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## 470 7. References

- 471 [1] Kenneth Lim, Leslie Wong, Wing Kong Chiu, and Jayantha Kodikara.  
472 Distributed fiber optic sensors for monitoring pressure and stiffness

- 473 changes in out-of-round pipes. *Structural Control and Health Monitoring*, 23(2):303–314, February 2016.  
474
- 475 [2] S. Laflamme, M. Kollosche, J. J. Connor, and G. Kofod. Soft capacitive  
476 sensor for structural health monitoring of large-scale systems. *Structural*  
477 *Control and Health Monitoring*, 19(1):70–81, February 2012.
- 478 [3] Ole J. D Kristensen and Forskningscenter Riso. *Fundamentals for remote*  
479 *structural health monitoring of wind turbine blades - a preproject, annex*  
480 *E: full-scale test of wind turbine blade, using sensors and NDT*. Riso  
481 National Laboratory, Roskilde, 2002.
- 482 [4] TingHua Yi, HongNan Li, and Ming Gu. Full-scale measurements of  
483 dynamic response of suspension bridge subjected to environmental loads  
484 using GPS technology. *Science China Technological Sciences*, 53(2):469–  
485 479, February 2010.
- 486 [5] M.B. Kane, C. Peckens, and J.P. Lynch. Design and selection of wire-  
487 less structural monitoring systems for civil infrastructures. In *Sensor*  
488 *Technologies for Civil Infrastructures*, pages 446–479. Elsevier, 2014.
- 489 [6] Haksoo Choi, Sukwon Choi, and Hojung Cha. Structural Health Moni-  
490 toring system based on strain gauge enabled wireless sensor nodes. pages  
491 211–214. IEEE, June 2008.
- 492 [7] J.J. McCullagh, T. Galchev, R.L. Peterson, R. Gordenker, Y. Zhang,  
493 J. Lynch, and K. Najafi. Long-term testing of a vibration harvesting  
494 system for the structural health monitoring of bridges. *Sensors and*  
495 *Actuators A: Physical*, 217:139–150, September 2014.
- 496 [8] Nephi R. Johnson, Jerome P. Lynch, and Matthew D. Collette. Re-  
497 sponse and fatigue assessment of high speed aluminium hulls using short-  
498 term wireless hull monitoring. *Structure and Infrastructure Engineering*,  
499 0(0):1–18, 2017.
- 500 [9] St350 Strain Transducer. [http://bditest.com/product/sensors/  
501 strain-sensors/bdi-strain-transducers/](http://bditest.com/product/sensors/strain-sensors/bdi-strain-transducers/). Accessed: 2017-11-13.
- 502 [10] David Bak. Rapid prototyping or rapid production? 3d printing  
503 processes move industry towards the latter. *Assembly Automation*,  
504 23(4):340–345, December 2003.

- 505 [11] Robert Bogue. 3d printing: the dawn of a new era in manufacturing?  
506 *Assembly Automation*, 33(4):307–311, September 2013.
- 507 [12] Yusuf Yagci, Steffen Jockusch, and Nicholas J. Turro. Photoinitiated  
508 Polymerization: Advances, Challenges, and Opportunities. *Macro-*  
509 *molecules*, 43(15):6245–6260, August 2010.
- 510 [13] Ferry P.W. Melchels, Jan Feijen, and Dirk W. Grijpma. A review on  
511 stereolithography and its applications in biomedical engineering. *Bio-*  
512 *materials*, 31(24):6121–6130, August 2010.
- 513 [14] Amit Joe Lopes, Eric MacDonald, and Ryan B. Wicker. Integrating  
514 stereolithography and direct print technologies for 3d structural elec-  
515 tronics fabrication. *Rapid Prototyping Journal*, 18(2):129–143, March  
516 2012.
- 517 [15] N.R. Mandal and P. Biswas. A review on development of weld induced  
518 distortion analysis. volume 2, pages 901–908, Lisbon, 2011. CRC press.
- 519 [16] T. D. Huang, Michael Harbison, Lee Kvidahl, David Niolet, John Walks,  
520 J. P. Christein, Mark Smitherman, Mark Phillippi, Pingsha Dong,  
521 Larry DeCan, Vince Caccese, Paul Blomquist, David Kihl, Rick Wong,  
522 Matthew Sinfield, Natale Nappi, James Gardner, Catherine Wong,  
523 Michael Bjornson, and Allen Manuel. Reduction of Overwelding and  
524 Distortion for Naval Surface Combatants. Part 2: Weld Sizing Effects  
525 on Shear and Fatigue Performance. *Journal of Ship Production and*  
526 *Design*, 32(1):21–36, February 2016.