

Letter to the Editor

SINGULARITIES IN WEDGE-SHAPED, ANTISYMMETRIC, COMPOSITE LAMINATES

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The partial differential equations which govern the behavior of unsymmetric laminates are coupled, in the sense that bending and in-plane deformations are not independent. This coupling, which is absent in the symmetric case, considerably complicates the analysis of singularities. General discussions of stress singularities in anisotropic media, a related field, can be found in Refs. [1], [2], [3].

The present investigation is concerned with singular fields in wedge-shaped, antisymmetric, composite laminates on whose edges are imposed typical boundary conditions. Expressions of the form

$$\begin{aligned} u_0 &= Ar^\lambda e^{s\theta} \\ v_0 &= Br^\lambda e^{s\theta} \\ w_0 &= Cr^{\lambda+1} e^{s\theta} \end{aligned} \quad (1)$$

are assumed for the dependent variables u_0 , v_0 and w_0 , which are the displacements of the midplane in the radial, tangential and transverse directions, respectively. In eqs. (1), A, B, C are constants, r and θ are polar coordinates and λ and s have to be determined. Satisfaction of the differential equations of equilibrium leads to a full 8th order polynomial in s , the

coefficients of which are functions of the unknown λ . Application of the boundary conditions leads to an eigenvalue problem involving an 8×8 matrix for the determination of λ . Since the roots of the polynomial in s cannot be written down explicitly, the eigenvalue problem is, in fact, nonlinear.

A graphite/epoxy laminate (T-300/5208) with 8 layers, each of thickness $1/4$ mm, is currently being investigated. Numerical results are being sought through an iterative procedure for various ply angles and boundary conditions. An initial guess is made for λ (the corresponding value for the symmetric cases will be used). Then the eigenvalue problem will be solved (using NAAS:EISPACK) to obtain a new value of λ and iteration will be performed until satisfactory accuracy is obtained.

REFERENCES

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