# ANALYTICAL SOLUTIONS FOR THE INCOMPRESSIBLE VISCOUS FLOW WITHIN A RECTANGULAR DOMAIN

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Analytical solutions of the velocity, pressure and stream function are developed for the slow incompressible viscous fluid in a simple rectangular domain. All solutions can be applied to the boundary integral techniques as typical fundamental solutions.

#### 1. Introduction

The finite element method has been zealously applied to the slow incompressible viscous flow (simply called the Stokes flow). One of the major difficulties arises through divergent-free constraints due to incompressibility. We have several alternatives; the mixed formulations [1–3], stream function methods [2, 3] and reduced-integration penalty approximations [4–9]. In the second approach by the use of stream functions, we need some genuity to compute the pressure. In other finite element approximations, some smoothing procedures are necessary for the pressure [8, 9], although the higher order simplex interpolations [10] are always free from locking [11].

In the boundary element approach, on the other hand, the divergent-free conditions are realized quite easily [12]. However, another difficulty arises through incompressibility with regards to traction forces as shown in Appendix B. The usual boundary element approximations developed in linear elasticity [13] cannot be applied to the Stokes flow.

This paper is devoted to the analytical solutions for the steady Stokes flow within a simple rectangular domain, by which numerical techniques currently used can be examined directly. Not only the velocity and pressure but also the stream function are dealt with. Examples are then presented to which boundary element solutions in a least-square scheme are compared.

#### 2. Constitutive law

We consider a slow incompressible viscous fluid in the domain  $\Omega$  with isotropic homogeneous viscosity  $\mu$ . Let  $(f_x, f_y)$  be a body force placed at a source point  $x_s = (x_s, y_s)$ . Then the

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velocity u = (u, v) at the field point x = (x, y) satisfies the equilibrium equation of the form

$$\mu \nabla^2 u(\mathbf{x}) + (\mu + 1/\varepsilon) \partial [\operatorname{div} \mathbf{u}(\mathbf{x})] / \partial \mathbf{x} + f_{\mathbf{x}} \delta(\mathbf{x} - \mathbf{x}_s) = 0 ,$$

$$\mu \nabla^2 v(\mathbf{x}) + (\mu + 1/\varepsilon) \partial [\operatorname{div} \mathbf{u}(\mathbf{x})] / \partial \mathbf{y} + f_{\mathbf{y}} \delta(\mathbf{x} - \mathbf{x}_s) = 0 .$$
(1)

Here  $\nabla^2$  denotes the Laplacian, div designates the divergence,  $\delta$  is the Dirac distribution and  $\varepsilon$  is a parameter such that  $\varepsilon \to 0^+$ .

Let p be the pressure defined by

$$p(\mathbf{x}) = -[\operatorname{div} \mathbf{u}(\mathbf{x})]/\varepsilon. \tag{2}$$

Then (1) can be written as

$$\mu \nabla^2 u(\mathbf{x}) - (1 + \mu \varepsilon) \partial p(\mathbf{x}) / \partial x + f_x \delta(\mathbf{x} - \mathbf{x}_s) = 0,$$
  
$$\mu \nabla^2 v(\mathbf{x}) - (1 + \mu \varepsilon) \partial p(\mathbf{x}) / \partial y + f_y \delta(\mathbf{x} - \mathbf{x}_s) = 0.$$
(3)

In definition, the incompressible fluid is divergent-free such that

$$\operatorname{div} \boldsymbol{u}(\boldsymbol{x}) = 0 \,, \tag{4}$$

while the pressure should not be zero, i.e. div u(x) is  $O(\varepsilon)$ . At the limit of  $\varepsilon \to 0^+$ , (3) can simply be written as

$$\mu \nabla^2 u(\mathbf{x}) - \partial p(\mathbf{x}) / \partial x + f_x \delta(\mathbf{x} - \mathbf{x}_s) = 0, \qquad (5a)$$

$$\mu \nabla^2 v(\mathbf{x}) - \partial p(\mathbf{x})/\partial y + f_y \delta(\mathbf{x} - \mathbf{x}_s) = 0, \qquad (5b)$$

The boundary conditions shall be described for each problem.

We further define the stream function  $\psi$  by

$$\partial \psi / \partial y = u \,, \qquad \partial \psi / \partial x = -v \,. \tag{6}$$

REMARK 2.1. In the so-called penalty method,  $\varepsilon$  is regarded as a penalty parameter which is positive and very small. However, in this paper, the solutions are given at the limit of  $\varepsilon \to 0^+$  (i.e. independent of  $\varepsilon$ ).

REMARK 2.2. For  $\varepsilon > 0$ , (1) and (3) can be identified with the equilibrium equation in linear elasticity (plane strain), but not (5).

REMARK 2.3. If  $\Omega$  is the infinite domain, then velocities at infinity should be zero. The natural boundary condition at infinity requires multiple body forces satisfying the force balance conditions of the form

$$\sum_{i} f_x^i = 0 \tag{7a}$$

and/or

$$\sum_{i} f_{y}^{i} = 0. \tag{7b}$$

Here superscript i designates the body force number.

In this paper, for simplicity, we deal only with a point force  $(f_x, f_y)$ . If the force balance condition is needed in some direction, therefore, our solutions are inadequate. Notice that the linear superposition makes them adequate.

### 3. Whole-space problem

We first consider the whole-space  $\Omega$ . Here only the natural boundary condition at infinity is applied.

THEOREM 3.1. For the whole-space problem, the velocity and pressure at x can be expressed explicitly as

$$u = A[-f_x \ln R - r_y^0 (f_x r_y^0 - f_y r_x^0)/R^2],$$
  

$$v = A[-f_y \ln R - r_x^0 (f_y r_x^0 - f_x r_y^0)/R^2],$$
(8)

and

$$p = C(f_x r_x^0 + f_y r_y^0)/R^2. (9)$$

Here

$$r_x^0 = x - x_s$$
,  $r_y^0 = y - y_s$ ,  $R = [(r_x^0)^2 + (r_y^0)^2]^{1/2}$ , (10)

$$A = 1/4\pi\mu$$
,  $C = 1/2\pi$ . (11)

**PROOF.** In the case of  $x \neq x_s$ , it is easy to verify that the velocity and pressure by (8) and (9) satisfy the governing equations (5).

Let us introduce the polar coordinates  $(R, \theta)$  of Fig. 1 origined at  $x_s$  such that

$$\cos \theta = r_x^0 / R \,, \qquad \sin \theta = r_y^0 / R \,. \tag{12}$$

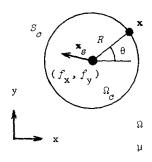


Fig. 1. Geometry of an infinitesimal sphere centered at  $x_s$ .

Then (8) and (9) can be written as

$$u = A\{-f_x(\ln R + \sin^2\theta) + f_y\sin\theta\cos\theta\},\$$
  

$$u = A\{-f_y(\ln R + \cos^2\theta) + f_x\sin\theta\cos\theta\}$$
(13)

and

$$p = C(f_x \cos \theta + f_y \sin \theta)/R. \tag{14}$$

Let  $\Omega_c$  be an infinitesimal closed circle with radius R around  $x_s$ , and let  $S_c$  be its surface. Then from (5a), we have

$$0 = \int_{\Omega_{c}} \left[ \mu \nabla^{2} u(\mathbf{x}) - \partial p(\mathbf{x}) / \partial x + f_{x} \delta(\mathbf{x} - \mathbf{x}_{s}) \right] d\Omega$$

$$= \int_{S_{c}} \left[ \mu \left\{ (\partial u / \partial x) \cos \theta + (\partial u / \partial y) \sin \theta \right\} - p \cos \theta \right] dS + f_{x}.$$
(15)

Here the divergence theorem is applied. Noting  $dS = R d\theta$ , (15) can be expressed as

$$f_x = \int_0^{2\pi} [p\cos\theta - \mu\{(\partial u/\partial x)\cos\theta + (\partial u/\partial y)\sin\theta\}] R d\theta = \pi f_x(C + 2\mu A).$$
 (16)

Evidently A and C by (11) realize (16), and hence the velocity and pressure by (8) and (9) satisfy (5a) even at  $x = x_s$ . In the same way, we can prove that (5b) is always adequate.

REMARK 3.2. The whole-space problem is not well posed for a single body force. The natural boundary condition at infinity requires the force balance condition (7a) and (7b). Therefore, solutions of (8) and (9) should be linearly superposed for the balanced body forces.

THEOREM 3.3. For the whole-space problem, the stream function by (6) can be given as

$$\psi = -A(f_x r_y^0 - f_y r_x^0) \ln R \,. \tag{17}$$

Here we neglect the integral constant.

**REMARK** 3.4. The incompressible viscous stream function of (17) is not harmonic, i.e.  $\nabla^2 \psi \neq 0$ .

**REMARK** 3.5. The potential function  $\phi$  (defined by  $\partial \phi/\partial x = u$ ,  $\partial \phi/\partial y = v$  and  $\phi \perp \psi$ ) does not exist in the Stokes flow.

#### 4. Half-space problem

We next consider the half-space  $\Omega$ . On the half-space boundary  $\Gamma_h^+$  of y = h, v should be zero (see Fig. 2). The natural boundary condition at infinity should be applied also.

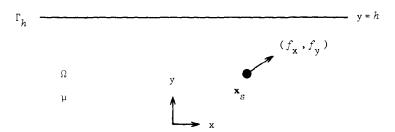


Fig. 2. Geometry of the half-space problem.

THEOREM 4.1. For the half-space problem, the velocity and pressure at x can be written explicitly as

$$u = A[-f_x \ln R - r_y^0 (f_x r_y^0 - f_y r_x^0)/R^2 - f_x \ln R_{0,1} - r_y^1 (f_x r_y^1 + f_y r_x^0)/R_{0,1}^2],$$

$$v = A[-f_y \ln R - r_x^0 (f_y r_x^0 - f_x r_y^0)/R^2 + f_y \ln R_{0,1} + r_x^0 (f_y r_x^0 + f_x r_y^1)/R_{0,1}^2]$$
(18)

and

$$p = C[(f_x r_x^0 + f_y r_y^0)/R^2 + (f_x r_x^0 - f_y r_y^1)/R_{0,1}^2]$$
(19)

where

$$y_s^1 = -y_s + 2h$$
,  $r_y^1 = y - y_s^1$ ,  $R_{0,1} = \{(r_x^0)^2 + (r_y^1)^2\}^{1/2}$ . (20)

**PROOF.** In whole-space, we place an image force  $(f_x, -f_y)$  at the image point  $x_s^1 = (x_s, y_s^1)$ . Then we can solve the half-space problem as the whole-space one with the image body force. On the half-space boundary  $\Gamma_h^+$ , v due to  $f_x$  at  $x_s$  is cancelled with v related to the image force  $f_x$  at  $x_s^1$ , while v caused by  $f_y$  at  $x_s$  is extinguished by v associated with v at v. Hence v = 0 on  $\Gamma_h^+$ .

**REMARK** 4.2. In the half-space problem, the force balance condition is required only in the x-direction, since  $f_y$  is balanced with its image force  $-f_y$ . Evidently  $f_y$  should not be placed on  $\Gamma_h^+$ .

**REMARK** 4.3. On  $\Gamma_h^+$  we have no tangential traction force, since  $\partial u/\partial y = \partial v/\partial x = 0$ .

THEOREM 4.4. For the half-space problem, the stream function can be written as

$$\psi = -A[(f_x r_y^0 - f_y r_x^0) \ln R + (f_x r_y^1 + f_y r_x^0) \ln R_{0,1}]. \tag{21}$$

Here we neglect the integral constant.

## 5. Infinite fissure problem

The domain we next consider is the infinite fissure with two walls  $\Gamma_h^+$  of y = h and  $\Gamma_h^-$  of y = -h (see Fig. 3). On the walls, v should be zero. The infinite fissure problem with the natural boundary condition at infinity can be solved by applying, successively, the image method.

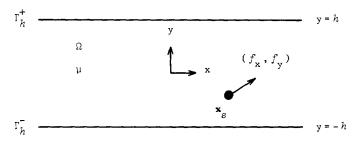


Fig. 3. Geometry of the infinite fissure problem.

THEOREM 5.1. For the infinite fissure problem, the velocity and pressure can be expressed as

$$u = A \sum_{n=-\infty}^{\infty} \left[ -f_x \ln R_{0,n} - r_y^n \{ f_x r_y^n - (-1)^n f_y r_x^0 \} / R_{0,n}^2 \right],$$

$$v = A \sum_{n=-\infty}^{\infty} \left[ -(-1)^n f_y \ln R_{0,n} - r_x^0 \{ (-1)^n f_y r_x^0 - f_x r_y^n \} / R_{0,n}^2 \right]$$
(22)

and

$$p = C \sum_{n=-\infty}^{\infty} \{f_x r_x^0 + (-1)^n f_y r_y^n\} / R_{0,n}^2.$$
 (23)

Here

$$y_s^n = (-1)^n y_s + 2nh$$
,  $r_y^n = y - y_s^n$ ,  $R_{0,n} = \{(r_x^0)^2 + (r_y^n)^2\}^{1/2}$ ,  $n = 0, \pm 1, \ldots, \pm \infty$ . (24)

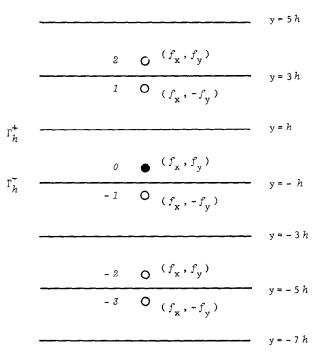


Fig. 4. Infinite series of image sources.

**PROOF.** Let  $(f_x, f_y)$  at  $x_s$  correspond to n = 0. With respect to  $\Gamma_h^+$ , its image force  $(f_x, -f_y)$  is placed at  $(x_s, y_s^1)$  associated with n = 1. Similarly with respect to  $\Gamma_h^-$ , we place the image force  $(f_x, -f_y)$  at  $(x_s, y_s^{-1})$  related to n = -1. These image procedures are successively applied such that the (n+1)st force is the image of the -nth force with respect to  $\Gamma_h^+$ , while the nth force yields the -(n+1)st one through  $\Gamma_h^-$  (see Fig. 4).

Our infinite fissure problem can thus be solved as the whole-space one with an infinite series of image forces  $(f_x, (-1)^n f_y)$  at  $(x_s, y_s^n)$ .

**REMARK** 5.2. In the infinite fissure problem, only the force balance condition (7a) is needed just like in the previous half-space problem.

THEOREM 5.3. For the infinite fissure problem, the stream function can be written as

$$\psi = -A \sum_{n=-\infty}^{\infty} \{ f_x r_y^n - (-1)^n f_y r_x^0 \} \ln R_{0,n} . \tag{25}$$

Here the integral constant is neglected.

## 6. Analytical solutions for the rectangle problem based on the image method

We now consider the rectangular domain  $\Omega$  with  $\Gamma_h^{\pm}$  of  $y=\pm h$  and  $\Gamma_g^{\pm}$  of  $x=\pm g$  (Fig. 5). On  $\Gamma_h^{\pm}$  v should be zero, while u should be zero on  $\Gamma_g^{\pm}$ .

THEOREM 6.1. The analytical solutions for the rectangle problem can be given as

$$u = A \sum_{m=-\infty}^{\infty} \sum_{n=-\infty}^{\infty} \left[ -(-1)^m f_x \ln R_{m,n} - r_y^n \{ (-1)^m f_x r_y^n - (-1)^n f_y r_x^m \} / R_{m,n}^2 \right],$$

$$v = A \sum_{m=-\infty}^{\infty} \sum_{n=-\infty}^{\infty} \left[ -(-1)^n f_y \ln R_{m,n} - r_x^m \{ (-1)^n f_y r_x^m - (-1)^m f_x r_y^n \} / R_{m,n}^2 \right]$$
(26)

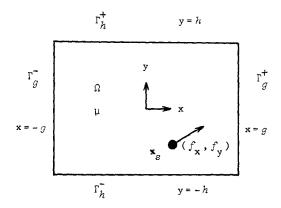


Fig. 5. Geometry of the rectangle problem.

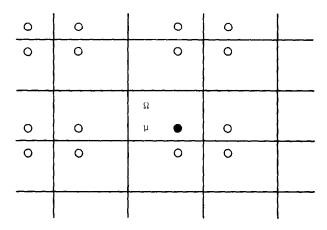


Fig. 6. Distributions of two-dimensional image sources.

and

$$p = C \sum_{m=-\infty}^{\infty} \sum_{n=-\infty}^{\infty} \left[ (-1)^m f_x r_x^m + (-1)^n f_y r_y^n \right] / R_{m,n}.$$
 (27)

Here

$$x_s^m = (-1)^m x_s + 2mg$$
,  $r_x^m = x - x_s^m$ ,  $R_{m,n} = \{(r_x^m)^2 + (r_y^n)^2\}^{1/2}$ ,   
  $m, n = 0, \pm 1, \dots, \pm \infty$ . (28)

**PROOF.** To the infinite series of forces in the preceding fissure problem, we apply the image procedure with respect to  $\Gamma_g^{\pm}$  (see Fig. 6). Our rectangle problem can then be solved as the whole-space one with  $((-1)^m f_x, (-1)^n f_y)$  at  $(x_s^m, y_s^n)$ .

**REMARK** 6.2. The incompressible rectangle problem is well posed even for a point force. It is evident, however, that  $f_x$  should not be placed on  $\Gamma_g^{\pm}$ , while  $f_y$  should not be on  $\Gamma_h^{\pm}$ . For velocity constraints see Fig. 7.

THEOREM 6.3. For the rectangle problem, the stream function is given as

$$\psi = -A \sum_{m=-\infty}^{\infty} \sum_{n=-\infty}^{\infty} \left[ (-1)^n f_x r_y^n - (-1)^m f_y r_x^m \right] \ln R_{m,n} . \tag{29}$$

Here the integral constant is neglected.

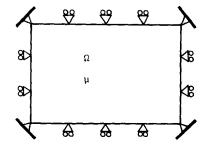


Fig. 7. Velocity constraints on the boundaries.

REMARK 6.4. The velocity and pressure are unbounded naturally at the source point  $x_s$ . The stream function, on the other hand, is bounded everywhere even at  $x = x_s$ .

Consider, for example,  $r_y^0 \ln R_{0,0}$ . Then putting  $r_x^0 = \alpha t$  and  $r_y^0 = \beta t$  (t > 0) at  $x = x_s$ , it extinguishes such that  $\beta t \ln[t\sqrt{\alpha^2 + \beta^2}] \to 0$  as  $t \to 0^+$ . Here  $\alpha$  and  $\beta$  denote appropriate constants.

## 7. Some formulae related to the potential problem

Analytical solutions for the steady Stokes flow within the rectangular domain are given in the form of infinite series with m and n, which are not so preferable in computations. In this section, we present some formulae to simplify our solutions.

The problem we consider is the electrical infinite fissure  $\Omega$  of unit specific conductivity between infinite spaces  $\Omega^+$  and  $\Omega^-$  of conductivity  $\infty$  (Fig. 8). Let only a point source of unit intensity be placed at  $x_s^m = (x_s^m, y_s)$  with  $\Omega$ . Then the potential  $\phi_m(x)$  at x in  $\Omega$  is governed by

$$\nabla^2 \phi_m(\mathbf{x}) + \delta(\mathbf{x} - \mathbf{x}_s^m) = 0. \tag{30}$$

The boundary conditions on  $\Gamma_h^{\pm}$  of  $y = \pm h$  are

$$\phi_m(\mathbf{x}) = 0 \,, \quad \mathbf{x} \in \Gamma_h^{\pm} \,. \tag{31}$$

LEMMA 7.1. For the electrical fissure problem, the potential at x in  $\Omega$  can be written as

$$4\pi\phi_m(x) = -\ln[F^-(x, y, x_s^m, y_s, h)/F^+(x, y, x_s^m, y_s, h)]$$
(32)

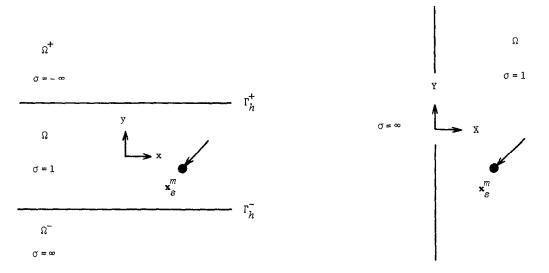


Fig. 8. Geometry of the electrical infinite fissure problem. An infinite fissure is embedded in otherwise perfectly conductive whole-space.

Fig. 9. Transformed space after the conformal mapping of (34).

where

$$F^{\pm}(x, y, x_s^m, y_s, h) = e^{\pi x/h} + e^{\pi x_s^m/h} \pm 2 e^{\pi (x + x_s^m)/2h} \cos \pi (y \pm y_s)/2h,$$
 (33)

PROOF. Under the conformal mapping of the form

$$\pi(x+iy) = 2h \ln(X+iY), \tag{34}$$

the infinite fissure of Fig. 8 can conformally be translated into the half-space of Fig. 9. Here i designates the imaginary unit. For further details, see [14, 15].

LEMMA 7.2. For the electrical fissure problem, the image procedure yields the solutions in another form of

$$2\pi\phi_m(x) = -\sum_{n=-\infty}^{\infty} (-1)^n \ln R_{m,n}.$$
 (35)

REMARK 7.3. The solution of (32) or (35) is adequate only after the linear superposition for balanced current sources just like in the preceding Stokes flow within the infinite fissure.

Lemmas 7.1 and 7.2 thus produce the following theorems, which are applicable to the steady Stokes flow.

THEOREM 7.4. Let us define the  $V_m$  function by

$$V_m = \frac{1}{2} \ln[F^-(x, y, x_s^m, y_s, h)/F^+(x, y, x_s^m, y_s, h)]. \tag{36}$$

Then we have

$$V_m = \sum_{n=-\infty}^{\infty} (-1)^n \ln R_{m,n} + D_m,$$
 (37)

where  $D_m$  designates an appropriate constant.

THEOREM 7.5. Let us define the  $U_n$  function by

$$U_n = \frac{1}{2} \ln[F^-(y, x, y_s^n, x_s, g)/F^+(y, x, y_s^n, x_s, g)].$$
(38)

Then we have

$$U_n = \sum_{m=-\infty}^{\infty} (-1)^m \ln R_{m,n} + D_n$$
 (39)

where  $D_n$  designates an appropriate constant.

THEOREM 7.6. The following relations hold:

$$\partial V_m/\partial y = \frac{\pi}{2h} e^{\pi(x+x\frac{m}{s})/2h} \left[ \frac{\sin \pi(y-y_s)/2h}{F^-(x,y,x^m_s,y_s,h)} + \frac{\sin \pi(y+y_s)/2h}{F^+(x,y,x^m_s,y_s,h)} \right]$$

$$= \sum_{n=-\infty}^{\infty} (-1)^n r_y^n / R_{m,n}^2, \qquad (40)$$

$$\partial U_n/\partial x = \frac{\pi}{2g} e^{\pi(y+y_s^n)/2g} \left[ \frac{\sin \pi(x-x_s)/2g}{F^-(y,x,y_s^n,x_s,g)} + \frac{\sin \pi(x+x_s)/2g}{F^+(y,x,y_s^n,x_s,g)} \right]$$

$$= \sum_{m=-\infty}^{\infty} (-1)^m r_x^m/R_{m,n}^2, \qquad (41)$$

$$\partial V_m/\partial x = \frac{\pi}{2h} \left[ \frac{e^{\pi x/h} - e^{\pi(x + x_s^m)/2h} \cos \pi(y - y_s)/2h}{F^-(x, y, x_s^m, y_s, h)} - \frac{e^{\pi x/h} + e^{\pi(x + x_s^m)/2h} \cos \pi(y + y_s)/2h}{F^+(x, y, x_s^m, y_s, h)} \right]$$

$$= r_x^m \sum_{n = -\infty}^{\infty} (-1)^n / R_{m,n}^2, \tag{42}$$

$$\partial U_n/\partial y = \frac{\pi}{2g} \left[ \frac{e^{\pi y/g} - e^{\pi (y+y_s^n)/2g} \cos \pi (x-x_s)/2g}{F^-(y,x,y_s^n,x_s,g)} - \frac{e^{\pi y/g} + e^{\pi (y+y_s^n)/2g} \cos \pi (x+x_s)/2g}{F^+(y,x,y_s^n,x_s,g)} \right]$$

$$= r_y^n \sum_{m=-\infty}^{\infty} (-1)^m/R_{m,n}^2. \tag{43}$$

#### 8. Simplified solutions for the rectangle problem

We can now simplify our solutions in Theorems 6.1 and 6.3 by using the preceding formulas. It is especially noted that the simplified solutions are given also in the form of infinite series, but only the one-directional expansion is needed.

THEOREM 8.1. For the rectangle problem, the incompressible viscous velocity and pressure can be written simply as

$$u = A \left[ -f_x \sum_{n=-\infty}^{\infty} \left\{ U_n + r_y^n (\partial U_n / \partial y) \right\} + f_y \sum_{m=-\infty}^{\infty} r_x^m (\partial V_m / \partial y) \right],$$

$$v = A \left[ -f_y \sum_{m=-\infty}^{\infty} \left\{ V_m + r_x^m (\partial V_m / \partial x) \right\} + f_x \sum_{m=-\infty}^{\infty} r_y^n (\partial U_n / \partial x) \right]$$
(44)

and

$$p = C \left[ f_x \sum_{n = -\infty}^{\infty} (\partial U_n / \partial x) + f_y \sum_{m = -\infty}^{\infty} (\partial V_m / \partial y) \right].$$
 (45)

THEOREM 8.2. For the rectangle problem, the incompressible viscous stream function can be written simply as

$$\psi = -A \left[ f_x \sum_{n=-\infty}^{\infty} r_y^n U_n - f_y \sum_{m=-\infty}^{\infty} r_x^m V_m \right]. \tag{46}$$

Here we neglect the integral constant.

We then deal with the stress-strain relation for the incompressible viscous fluid.

THEOREM 8.3. For the rectangle problem, stress components  $\sigma_x$ ,  $\sigma_y$  and  $\tau_{xy}$  are given by

$$\sigma_{x} = C \left[ -f_{x} \sum_{n=-\infty}^{\infty} \left\{ 2(\partial U_{n}/\partial x) + r_{y}^{n} (\partial^{2} U_{n}/\partial x \partial y) \right\} + f_{y} \sum_{m=-\infty}^{\infty} r_{x}^{m} (\partial^{2} V_{m}/\partial x \partial y) \right],$$

$$\sigma_{y} = C \left[ -f_{y} \sum_{m=-\infty}^{\infty} \left\{ 2(\partial V_{m}/\partial y) + r_{x}^{m} (\partial^{2} V_{m}/\partial x \partial y) \right\} + f_{x} \sum_{n=-\infty}^{\infty} r_{y}^{n} (\partial^{2} U_{n}/\partial x \partial y) \right],$$

$$\tau_{xy} = C \left[ f_{x} \sum_{n=-\infty}^{\infty} \left\{ r_{y}^{n} (\partial^{2} U_{n}/\partial x^{2}) - \partial U_{n}/\partial y \right\} + f_{y} \sum_{m=-\infty}^{\infty} \left\{ r_{x}^{m} (\partial^{2} V_{m}/\partial y^{2}) - \partial V_{m}/\partial x \right\} - \pi (f_{x} + f_{y}) \delta(x - x_{s}) \right].$$

$$(47)$$

**PROOF.** Strain-rate components  $\varepsilon_x$ ,  $\varepsilon_y$  and  $\gamma_{xy}$  are defined by

$$\varepsilon_{x} = \partial u/\partial x$$
,  $\varepsilon_{y} = \partial v/\partial y$ ,  $2\gamma_{xy} = \partial u/\partial y + \partial v/\partial x$ . (48)

Stress components can then be written as

$$\sigma_{\rm x} = 2\mu\varepsilon_{\rm x} - p \,, \qquad \sigma_{\rm y} = 2\mu\varepsilon_{\rm y} - p \,, \qquad \tau_{\rm xy} = 2\mu\gamma_{\rm xy} \,.$$
 (49)

Substitution of (44) and (45) into (48) and (49) thus completes the proof.

**REMARK** 8.4. In deriving  $\tau_{xy}$ , notice that

$$\nabla^2 V_0 - 2\pi \delta(\mathbf{x} - \mathbf{x}_s) = 0. \tag{50}$$

Evidently  $V_m$   $(m \neq 0)$  as well as  $U_n$   $(n \neq 0)$  are harmonic.

**REMARK** 8.5. Let the outward normal be expressed as  $(n_x, n_y)$  on the boundaries. Then the traction force  $(t_x, t_y)$  can be obtained as

$$t_{x} = n_{x}\sigma_{x} + n_{y}\tau_{xy}, \qquad t_{y} = n_{x}\tau_{xy} + n_{y}\sigma_{y}. \tag{51}$$

In our rectangle problem, no tangential forces exist.

## 9. Examinations of boundary element solutions

We can now examine currently used numerical techniques such as the finite element and boundary element methods. Examples we present are associated with a rectangle of g=4 and h=2 with a homogeneous viscosity of  $\mu=1/2\pi$  (Fig. 10). Body forces of  $(\frac{1}{2}\sqrt{3},\frac{1}{2})$  and  $(-\frac{1}{2}\sqrt{3},-\frac{1}{2})$  are applied at source points (2,1) and (2,-1), respectively. Our computations are made by using finite series of  $\sum_{-N}^{N}$  in place of infinite ones of  $\sum_{-\infty}^{\infty}$ . Here N is increased until the sufficient convergence is attained.

Table 1 shows the convergence of the velocity u, pressure p and stress  $\sigma_x$  at the origin (0,0). All values converge quite rapidly, while larger N is needed in computing the stream function as shown in Table 2. At nodes regularly placed on a grid without vacancy, computed results are listed in Tables 3 to 5.

We then apply the boundary element method (see Appendices A and B) associated with the whole-space fundamental solutions to this rectangle problem. The boundary surface is divided into 24 boundary elements of length 1, on which the piecewise linear approximations are made

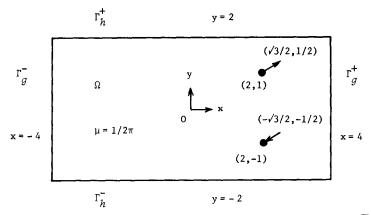


Fig. 10. Geometry of the example problem of g=4 and h=2. Body forces of intensity  $(\frac{1}{2}\sqrt{3},\frac{1}{2})$  and  $(-\frac{1}{2}\sqrt{3},-\frac{1}{2})$  are applied at the source points of (2,1) and (2,-1), respectively. The isotropic and uniform viscosity of  $\mu=1/2\pi$  is assumed.

Table 1 Numerical solutions at the origin (0,0) under the finite approximations. The infinite series of  $\sum_{-\infty}^{\infty}$  is approximated by the finite one of  $\sum_{-N}^{N}$ 

N	u (×10 <sup>-8</sup> )	$p \ (\times 10^{-8})$	$\sigma_x \ (\times 10^{-8})$
0	6775377	-1078335	3375079
1	6813287	-1080356	3394133
2	6813287	-1080356	3394133
3	6813287	-1080356	3394133

Table 2 Computed stream function values under the finite approximations at the origin (0,0)

N	$\psi$ (×10 <sup>-7</sup> )	N	$\psi$ (×10 <sup>-7</sup> )
0	-6803068	8	-7569628
1	-9199912	9	-7569866
2	-6936128	10	-7569811
3	-7762598	11	-7569823
4	-7517060	12	-7569820
5	-7583426	13	-7569821
6	-7566445	14	-7569821
7	-7570637	15	-7569821

Table 3 Solutions of the stream function and velocities in the rectangle problem

				x					$\psi \ (\times 10^{-4})$ $u \ (\times 10^{-5})$
4	3	2	1	0	-1	-2	-3	-4	$v (\times 10^{-5})$
(	0	0	0	0	0	0	0	0	
(	59692	53765	56988	49784	32502	18071	7850	0	2
(	0	0	0	0	0	0	0	0	
(	-6041	-8082	-6472	-4795	-2999	-1644	-711	0	
(	54000	$+\infty$	73226	43102	24885	13241	5658	0	1
-6346	-52989	+∞	12453	19018	16030	11172	7853	6735	
(	-8201	-13799	-11639	-7570	-4457	-2381	-1018	0	
(	-13534	587	15803	6813	2125	587	144	0	y 0
-8324	-78321	-16565	41479	36810	25528	16565	11314	9623	
	-4201	-8157	-8601	-5665	-3270	-1719	-729	0	
1	-54000	$-\infty$	-73226	-43102	-24885	-13241	-5658	0	-1
$-4490^{\circ}$	-36620	-∞	25752	28363	19383	12167	8137	6872	
(	0	0	0	0	0	0	0	0	
(	-32624	-54939	-88594	-63411	-36752	-19244	-8137	0	-2
(	0	0	0	0	0	0	0	0	

Table 4
Solutions of the pressure and strain components in the rectangle problem

				x					$p (\times 10^{-5})$ $\varepsilon_x (\times 10^{-5})$
4	3	2	1	0	-1	-2	-3	-4	$\gamma_{xy} (\times 10^{-5})$
14313	18681	13865	-5011	-4735	-2550	-1271	-687	-522	
-68524	-38460	25582	-6306	16000	16639	12131	8651	7449	2
0	0	0	0	0	0	0	0	0	
9407	14532	±∞	-12039	-4518	-2000	-936	-493	-372	
-50597	-62463	±∞	41718	22626	14458	9234	6287	5345	1
0	7961	±∞	-4500	9078	10239	6863	3223	0	
-2157	-5206	-12547	-4991	-1080	-225	-47	-10	-4	
14431	9515	-38495	7106	7269	2632	781	223	107	y 0
0	24924	-13197	39414	30824	19343	10948	4853	0	-
-9407	-14532	±∞	12039	4518	2000	936	493	372	
50597	62463	±∞	-41718	-22626	-14458	-9234	-6287	-5345	-1
0	-11122	$\pm \infty$	20241	23829	15504	8413	3616	0	
-10000	-8268	11228	14994	6895	3000	1365	707	530	
39662	19430	51408	-7905	-30538	-21904	-13694	-9097	-7663	-2
0	0	0	0	0	0	0	0	0	

Table 5
Solutions of the stress components in the rectangle problem.

				x					$\sigma_x \ (\times 10^{-5})$ $\sigma_y \ (\times 10^{-5})$
4	3	2	1	0	-1	-2	-3	-4	$\tau_{xy} \ (\times 10^{-5})$
-36125	-3092	-5722	3004	9827	7847	5133	3441	2893	
7499	-6439	-22008	7018	-358	-2746	-2590	-2067	-1849	2
0	0	0	0	0	0	0	0	0	
-25513	-34414	±∞	25318	11720	6602	3875	2494	2074	
6698	5351	±∞	-1240	-2684	-2602	-2004	-1508	-1329	1
0	2534	±∞	-1432	2890	3259	2185	1026	0	
6750	8235	293	7253	3394	1063	296	81	38	
-2437	2178	24800	2730	-1233	-613	-202	-61	-30	y 0
0	7934	-4201	12546	9812	6157	0 3485	1545	0	•
25513	34414	$\pm \infty$	-25318	-11720	-6620	-3875	-2494	-2074	
-6698	-5351	$\pm \infty$	1240	2684	2602	2004	1508	1329	-1
0	-3540	<u>+</u> ∞	6443	7585	4935	2678	1151	0	
22624	14453	5135	-17510	-16616	-9972	-5724	-3603	-2969	
-2625	2084	-27592	-12477	2825	3972	2994	2189	1909	-2
0	0	0	0	0	0	0	0	0	

for the tangential velocity with the piecewise constant normal traction forces. Under the tangential-free boundary conditions that we are concerned with, we must determine 20 velocities at nodes on smooth boundaries and 24 traction forces on the boundary elements. For the tangential components nodes on smooth boundaries are chosen as  $x_m$ , while centers of gravity of every boundary elements are selected as  $x_l$  for the normal components.

Computed velocities on the boundaries are plotted in Figs. 11 and 12 together with the exact profiles. We also show the velocity and strain components along the line of y = 0 in Figs. 13 and 14, respectively. Extremely accurate results are thus obtained.

Computed traction forces on  $\Gamma_h^+$  are drawn in Fig. 15, which are in good agreement with exact solutions. But the superconvergence like in the velocities cannot be attained. We then compute the pressure, and the results along the line of y=0 are plotted in Fig. 16. It is thus obvious that the accuracy of boundary element stress components is strongly influenced by that of the pressure.

## 10. Concluding remarks

We have developed analytical solutions of the steady Stokes flow in a rectangular domain as well as in an infinite fissure, which are applicable to the boundary integral techniques as typical fundamental solutions. Alternative expressions by the use of Fourier series have been already developed also, but they converge very slowly. It is emphasized that the quite rapid convergence inherent to our solutions is of great importance in the boundary element applications.

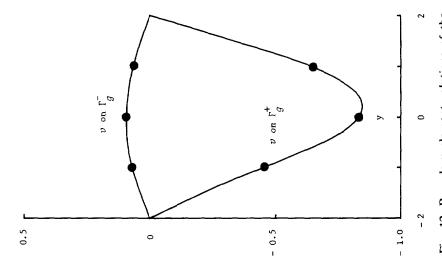


Fig. 12. Boundary element solutions of the tangential velocity on  $\Gamma_s^{\pm}$  with exact profiles.

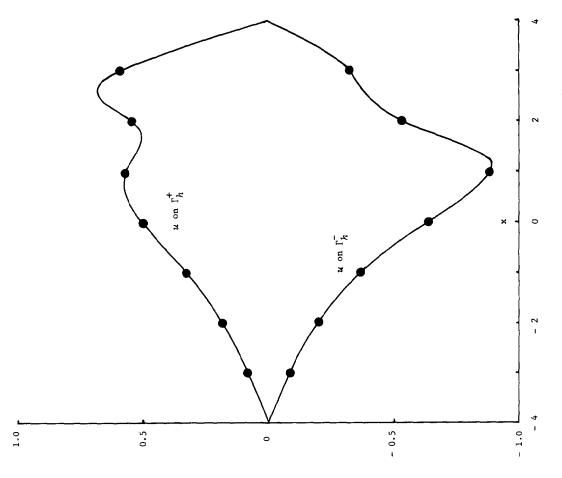


Fig. 11. Boundary element solutions of the tangential velocity on  $F_{\pi}^{\pm}$  with exact profiles.

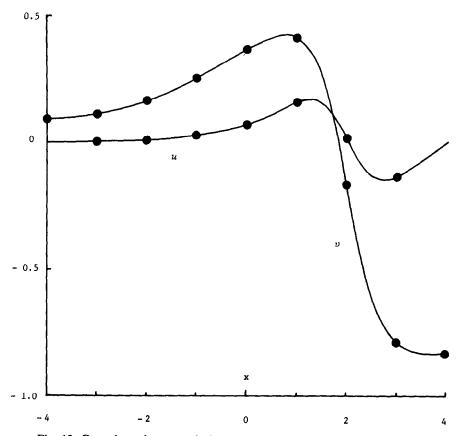


Fig. 13. Boundary element solutions of velocities along the line of y = 0.

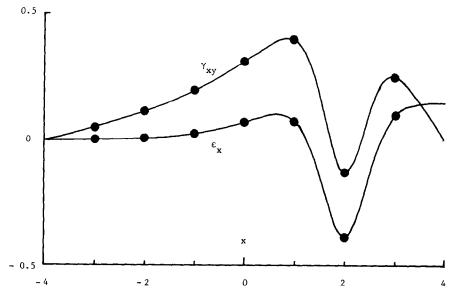


Fig. 14. Boundary element solutions of strain components along y = 0.

0.1

0

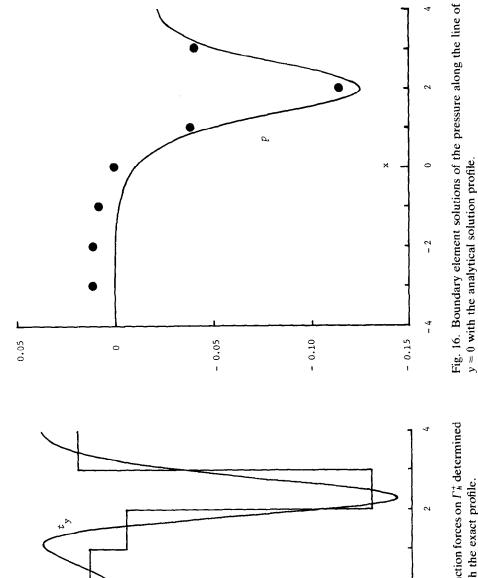


Fig. 15. Piecewise constant normal traction forces on  $\Gamma^+_h$  determined in our boundary element method with the exact profile.

0

- 0.3

- 0.2

- 0.1

×

## Appendix A. Boundary integral equations associated with the whole-space fundamental solutions

Referring to the whole-space solutions in (8) and (9), we introduce the fundamental solutions  $\hat{u} = (\hat{u}, \hat{v})$  and  $\hat{p}$  by

$$\hat{u}(\tilde{\mathbf{x}}, \mathbf{x}) = \hat{u}^1 \delta_{1i} + \hat{u}^2 \delta_{2i},$$

$$\hat{v}(\tilde{\mathbf{x}}, \mathbf{x}) = \hat{v}^1 \delta_{1i} + \hat{v}^2 \delta_{2i},$$

$$\hat{p}(\tilde{\mathbf{x}}, \mathbf{x}) = \hat{p}^1 \delta_{1i} + \hat{p}^2 \delta_{2i}.$$
(A.1)

Here  $\delta_{ij}$  is the Kronecker delta (i, j = 1, 2) and

$$4\pi\mu\hat{u}^{1} = -\ln r - (\tilde{y} - y)^{2}/r^{2},$$

$$4\pi\mu\hat{v}^{1} = (\tilde{x} - x)(\tilde{y} - y)/r^{2},$$

$$2\pi\hat{p}^{1} = (\tilde{x} - x)/r^{2},$$
(A.2)

$$4\pi\mu\hat{u}^{2} = (\tilde{x} - x)(\tilde{y} - y)/r^{2},$$

$$4\pi\mu\hat{v}^{2} = -\ln r - (\tilde{x} - x)^{2}/r^{2},$$

$$2\pi\hat{p}^{2} = (\tilde{y} - y)/r^{2}$$
(A.3)

where r denotes the distance between x and  $\tilde{x} = (\tilde{x}, \tilde{y})$  by

$$r = [(\tilde{x} - x)^2 + (\tilde{y} - y)^2]^{1/2}. \tag{A.4}$$

Then the governing equations for the fundamental solutions can be written as

$$\tilde{L}_{u}[\hat{\boldsymbol{u}}(\tilde{\boldsymbol{x}}, \boldsymbol{x})] + \lambda(\boldsymbol{x})\delta_{1i}\delta(\tilde{\boldsymbol{x}} - \boldsymbol{x}) = 0, 
\tilde{L}_{v}[\hat{\boldsymbol{u}}(\tilde{\boldsymbol{x}}, \boldsymbol{x})] + \lambda(\boldsymbol{x})\delta_{2i}\delta(\tilde{\boldsymbol{x}} - \boldsymbol{x}) = 0,$$
(A.5)

Here the differential operators  $\tilde{L}_u$  and  $\tilde{L}_v$  are given by using the Laplacian  $\tilde{\nabla}^2$  with respect to  $\tilde{x}$  such that

$$\tilde{L}_{u}[\hat{\boldsymbol{u}}(\tilde{\boldsymbol{x}}, \boldsymbol{x})] = \mu \tilde{\nabla}^{2} \hat{\boldsymbol{u}}(\tilde{\boldsymbol{x}}, \boldsymbol{x}) - \partial \hat{\boldsymbol{p}}(\tilde{\boldsymbol{x}}, \boldsymbol{x}) / \partial \tilde{\boldsymbol{x}}, 
\tilde{L}_{v}[\hat{\boldsymbol{u}}(\tilde{\boldsymbol{x}}, \boldsymbol{x})] = \mu \tilde{\nabla}^{2} \hat{\boldsymbol{v}}(\tilde{\boldsymbol{x}}, \boldsymbol{x}) - \partial \hat{\boldsymbol{p}}(\tilde{\boldsymbol{x}}, \boldsymbol{x}) / \partial \tilde{\boldsymbol{y}},$$
(A.6)

and  $\lambda(x)$  designates the positioning constant which can be geometrically defined [16].

Noting the governing equations (5) and (A.5), we adopt the weighted residual scheme of the form

$$0 = \int_{\Omega} \hat{u}(\tilde{\mathbf{x}}, \mathbf{x}) \{\tilde{L}_{u}[\mathbf{u}(\tilde{\mathbf{x}})] + f_{x}\delta(\tilde{\mathbf{x}} - \mathbf{x}_{s})\} d\tilde{\Omega} + \int_{\Omega} \hat{v}(\tilde{\mathbf{x}}, \mathbf{x}) \{\tilde{L}_{v}[\mathbf{u}(\tilde{\mathbf{x}})] + f_{y}\delta(\tilde{\mathbf{x}} - \mathbf{x}_{s})\} d\tilde{\Omega}$$

$$- \int_{\Omega} u(\tilde{\mathbf{x}}) \{\tilde{L}_{u}[\hat{\mathbf{u}}(\tilde{\mathbf{x}}, \mathbf{x})] + \lambda(\mathbf{x})\delta_{1i}\delta(\tilde{\mathbf{x}} - \mathbf{x})\} d\tilde{\Omega}$$

$$- \int_{\Omega} v(\tilde{\mathbf{x}}) \{\tilde{L}_{v}[\hat{\mathbf{u}}(\tilde{\mathbf{x}}, \mathbf{x})] + \lambda(\mathbf{x})\delta_{2i}\delta(\tilde{\mathbf{x}} - \mathbf{x})\} d\tilde{\Omega}.$$
(A.7)

Here the integrations and differentiations are taken with respect to  $\tilde{x}$ . Applying the Green's theorem, (A.7) can be written as

$$\lambda(\mathbf{x})\{\delta_{1i}u(\mathbf{x}) + \delta_{2i}v(\mathbf{x})\} = \hat{u}(\mathbf{x}_{s}, \mathbf{x})f_{x} + \hat{v}(\mathbf{x}_{s}, \mathbf{x})f_{y} + \int_{\Gamma} \{t_{x}(\tilde{\mathbf{x}})\hat{u}(\tilde{\mathbf{x}}, \mathbf{x}) + t_{y}(\tilde{\mathbf{x}})\hat{v}(\tilde{\mathbf{x}}, \mathbf{x})\}d\tilde{\Gamma}$$

$$-\int_{\Gamma} \{u(\tilde{\mathbf{x}})\hat{t}_{x}(\tilde{\mathbf{x}}, \mathbf{x}) + v(\tilde{\mathbf{x}})\hat{t}_{y}(\tilde{\mathbf{x}}, \mathbf{x})\}d\tilde{\Gamma}. \tag{A.8}$$

Here the integrations are taken over the boundary  $\Gamma$  with respect to  $\tilde{x}$ , and  $\hat{t}_x$  and  $\hat{t}_y$  are given by

$$\hat{t}_{x}(\tilde{\mathbf{x}}, \mathbf{x}) = -2(\tilde{\mathbf{x}} - \mathbf{x})\tilde{\mathbf{n}} \cdot (\tilde{\mathbf{x}} - \mathbf{x})\hat{p}(\tilde{\mathbf{x}}, \mathbf{x})/r^{2},$$

$$\hat{t}_{y}(\tilde{\mathbf{x}}, \mathbf{x}) = -2(\tilde{\mathbf{y}} - \mathbf{y})\tilde{\mathbf{n}} \cdot (\tilde{\mathbf{x}} - \mathbf{x})\hat{p}(\tilde{\mathbf{x}}, \mathbf{x})/r^{2}$$
(A.9)

where  $\tilde{n} = (\tilde{n}_x, \tilde{n}_y)$  is the outward normal at  $\tilde{x}$  on  $\Gamma$ .

Differentiation of (A.8) further gives

$$\hat{p}(\mathbf{x}) = -\hat{p}^{1}(\mathbf{x}_{s}, \mathbf{x})f_{x} - \hat{p}^{2}(\mathbf{x}_{s}, \mathbf{x})f_{y} - \int_{\Gamma} \{t_{x}(\tilde{\mathbf{x}})\hat{p}^{1}(\tilde{\mathbf{x}}, \mathbf{x}) + t_{y}(\tilde{\mathbf{x}})\hat{p}^{2}(\tilde{\mathbf{x}}, \mathbf{x})\}d\tilde{\Gamma}$$

$$+2\mu \int_{\Gamma} \{u(\tilde{\mathbf{x}})\tilde{\mathbf{n}} \cdot \nabla \hat{p}^{1}(\tilde{\mathbf{x}}, \mathbf{x}) + v(\tilde{\mathbf{x}})\tilde{\mathbf{n}} \cdot \nabla \hat{p}^{2}(\tilde{\mathbf{x}}, \tilde{\mathbf{x}})\}d\tilde{\Gamma}, \quad \mathbf{x} \in \Omega.$$
(A.10)

For further details, see [12]. Notice that  $\Gamma$  is composed of  $\Gamma_g^{\pm}$  and  $\Gamma_h^{\pm}$  in our rectangle problem.

## Appendix B. A boundary element scheme in least-squares

Since the fundamental solutions of (A.1) are divergent-free, we have

$$0 = \int_{\Omega} (\partial \hat{u}/\partial x + \partial \hat{v}/\partial y) d\tilde{\Omega}, \qquad (B.1)$$

which yields immediately

$$0 = \int_{\Gamma} (\tilde{n}_x \hat{u} + \tilde{n}_y \hat{v}) d\tilde{\Gamma}.$$
 (B.2)

It is then obvious in (A.8) that the constant traction forces  $t_x$  and  $t_y$  do not influence upon the velocities. In the boundary element approximations for Stokes flow, therefore, we should extinguish these degrees of freedom on traction forces.

On the smooth part of  $\Gamma$ , let  $u_t$  and  $t_n$  denote the tangential velocity and normal traction force, respectively. At the smooth point x on  $\Gamma$ , (A.8) can then be written for the tangential velocity as

$$\frac{1}{2}u_t(\mathbf{x}) = K_t[\mathbf{x}, u_t(\tilde{\mathbf{x}}), t_n(\tilde{\mathbf{x}})], \qquad (B.3a)$$

and for the normal velocity as

$$0 = K_n[\mathbf{x}, u_t(\tilde{\mathbf{x}}), t_n(\tilde{\mathbf{x}})]. \tag{B.3b}$$

We then express the traction force balance conditions as

$$C_{x}[\tilde{n}_{x}t_{n}(\tilde{x})] = f_{x} + \int_{\Gamma} t_{x}(\tilde{x})d\tilde{\Gamma} = 0 ,$$

$$C_{y}[\tilde{n}_{y}t_{n}(\tilde{x})] = f_{y} + \int_{\Gamma} t_{y}(\tilde{x})d\tilde{\Gamma} = 0 ,$$
(B.4)

Denoting the boundary element approximations by superscript h, we introduce a least-square functional of the form

$$\chi = \frac{1}{2} \sum_{m} \left\{ \frac{1}{2} u_{t}^{h}(\mathbf{x}_{m}) - K_{t}[\mathbf{x}_{m}, u_{t}^{h}(\tilde{\mathbf{x}}), t_{n}^{h}(\tilde{\mathbf{x}})] \right\}^{2} + \frac{1}{2} \sum_{l} \left\{ K_{n}[\mathbf{x}_{l}, u_{t}^{h}(\tilde{\mathbf{x}}), t_{n}^{h}(\tilde{\mathbf{x}})] \right\}^{2} + \alpha_{x} \left\{ C_{x}[\tilde{n}_{x} t_{n}^{h}(\tilde{\mathbf{x}})] \right\}^{2} + \alpha_{y} \left\{ C_{y}[\tilde{n}_{y} t_{n}^{h}(\tilde{\mathbf{x}})] \right\}^{2}.$$
(B.5)

Here  $\alpha_x$  and  $\alpha_y$  are Lagrange multipliers, and the smooth points  $x_m$  and  $x_l$  are sampled on  $\Gamma$  so that the sampled number is greater than or equal to that of the discretized variables to be determined plus two of Lagrange multipliers. Minimization of  $\chi$  by (B.5) then determines  $u_t^h$  and  $t_n^h$ .

It is emphasized that imposition of the constraints of (B.4) guarantees the uniqueness of our boundary element solutions.

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