# Fracture surface characterization of epoxy-based GFRP laminates

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Fractographic investigation has been carried out on glass fabric–epoxy composite laminates using scanning electron microscopy. Focusing on the flexural failure of lap shear specimens, some unique fracture features have been identified, and their likely origin suggested and explained. The influence of voids, present in the matrix, on the appearance of the fracture surface has been illustrated.

## 1. Introduction

Fractography as an important tool to analyse the sequential processes leading to failure has been widely used for characterizing failures in metallic materials. However, this branch of study for non-metallics in general, and polymers in particular, has not witnessed the same degree of progress; though investigations were initiated as early as a few decades ago [1, 2]. With the growing interest in this field, the literature on fractography and failure analysis of polymers and polymeric composites is rapidly picking up. Nevertheless, the quantum of work done seems to be insufficient for arriving at any detailed generalization of the micromechanisms responsible for development of particular fracture features which are often indicative of specific failure modes. While reasons for this situation could be traced to a variety of factors, two aspects, namely, the difficulties faced in characterizing the fracture features and the general lack of interest to correlate these features to the state of stress causing the failure are to be reckoned first.

A relatively recent effort by Purslow [3] deserves considerable attention and, of course, appreciation for the scientific approach and attempts on resolving the important aspect of ambiguity involved in the use of different terminologies for essentially the same features. He has also presented simple physical models with a view to illustrate logically the formation of typical fracture features. This system of nomenclature will be followed in the present paper, and these models referred to while discussing our own findings in the area of composite fractography.

In fibrous composites, most of the fracture features are registered in the matrix. The present work is principally directed to matrix fractography of glass fabric reinforced epoxy laminates.

Much of this work concentrates on identification and interpretation of principal fracture features found in lap shear specimens made to fail in three-point flexure. Attempts are made to pinpoint the exact failure mode and also to explain the origin of some unique fracture features hitherto not reported. New terminologies, first introduced in this paper, are in italics. The role of voids present in the matrix and their influence on the appearance of the fracture surface have been looked into with the help of typical illustrations taken from specimens failed in transverse tension, as well as in mixed mode. The stress situation responsible for failure has always been borne in mind and taken into account at the time of the fractographic analyses. Thus, the results are claimed to be valid universally, irrespective of the exact test procedures employed, as long as the active stress situation remains the same.

## 2. Materials and methods

E-glass fabric of 7 mil (  $\simeq 0.25400 \times 10^{-6} \,\mu\text{m}$ ), plain weave was used to reinforce the matrix material made of epoxy resin Araldite LY 556 cured at room temperature with the hardener type HY 951. Laminates were prepared by hand lay up followed by vacuum bagging procedure [4]. Lap shear specimens (Fig. 1a) were prepared by a single-stage lay up using proper spacers. At this juncture it should be noted that no adhesive jointing was done. The total length  $(L_t)$  of each specimen was 200 mm, the overlapped length  $(L_0)$  was 46 mm, thickness  $(t) \sim 1.35$  mm and width (w)25.5 mm. It was tested under flexural loading in a three-point bend set-up of span length (21) 80 mm, using a cross-head speed of  $0.05 \text{ mm sec}^{-1}$ . To prevent the sliding tendency of the specimen during the test, recourse to a small adhesive piecing to the sample (Fig. 1a) was made [4]. Fracture surfaces were studied in a Jeol model JSM 840A scanning electron microscope (SEM). Experiments were repeated a number of times and essentially similar fracture features could be observed in almost all cases. Unless otherwise mentioned, any fracture feature under discussion is understood to have been seen distributed on the fracture

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Figure 1 Illustrations of (a) geometry and loading configuration, (b) failure pattern, and (c) the state of stress ahead of the advancing failure front (O) of the lap shear specimen. XY is the plane of lamination.

surface of both the separating halves of the failed lap shear specimen.

In this paper the term "axial" has been used to indicate the direction of the long axis of the test specimen. Care was taken during specimen preparation to ensure that the warp fibres were always oriented along the axial direction.

#### 3. Results and discussion

When subjected to three-point flexural loading (Fig. 1a), apparently the lap shear specimen may be expected to fail in shear along the plane QQ'. Assuming simple bending, the state of stress operative there may be thought to be pure shear, because this plane is located exactly at the mid thickness position of the

overlapped section. That a shear field was active in effecting the interlaminar failure along QQ' was evident from the common occurrence of shear cusps [3] in the interfibre matrix regions of the fracture surface. (Whenever required, the overall fracture direction has been indicated on the illustrations by means of an arrowhead,  $\rightarrow$ , while the broken arrow,  $\rightarrow$ , and the half arrow,  $\rightleftharpoons$ , have been used to indicate the direction of local crack/fracture propagation and active shear stress respectively.) Although lacking sharpness at this high magnification, Fig. 2a is significant for documenting one such cusp formation just at the point of inception. But the myriad of rivers present at the resinrich portions of the fracture surface, bore testimony to the existence of a tensile (transverse) stress too, in action at the failure front. It is to be appreciated that an intermediate configuration of the specimen during the failure process can be represented schematically as shown in Fig. 1b. Evidently, the bending action as shown introduces an axial tensile stress,  $\sigma_r$ , along the crack face OS'. An exaggerated view (shown schematically in Fig. 1c) of the crack tip region is helpful in understanding the actual state of stress operative there. Although the bending action introduces only axial stress, in reality, because of the well known "notch effect" [5] operative at the crack tip, a transverse tensile stress ( $\sigma_z$ ) comes into play. It is this  $\sigma_z$ that has actually caused the development of tensile failure features, like rivers, on the fracture surface. Further, the lower face OS' bends more than the upper face OQ. This would tend to effect a strain difference between these two faces, resulting in the development of a shear stress,  $\tau_{xz}$ , active at the crack tip O. The occurrence of shear cusps in the interfibre matrix owes its origin to this  $\tau_{xz}$  only. The deduction, based on fractographic evidence, that lap shear specimens under flexure actually fail in the mixed mode, is thus upheld by a qualitative analysis of the state of stress active at the progressing failure front. Further, the failure scheme envisioned (Fig. 1b) strongly suggests a peel type of fracture [3]. And in fact, the occurrence of typical peel fracture features like tied zones were quite common on the fracture surface (Fig. 2b).

Now, that the flexural failure has been classified and confirmed as peel type, occurrence of well-developed



Figure 2 SEM fractographs from lap shear specimen failed in flexure illustrating (a) incipient cusp and (b) tied zones.

ridges on the interlaminar fracture surface is perhaps expected. Although Fig. 3a beautifully brings out the "rivery" (confluencing towards the ridge) and the "smooth" surfaces across the ridge (R-R), the micromechanism of ridge formation [3] is more clearly revealed in Fig. 3b. As is evident from the river direction, a crack front was progressing, under the action of the notch-induced transverse tension ( $\sigma_z$ ), from left to right along the general peel plane (OO'SS' in Fig. 1b). The feathers [3] on the face of the cusp (right) in Fig. 3b testify to another crack front moving along it (the cusp face) under the influence of the shear stress field  $(\tau_{xx})$  set up ahead of the former crack tip. The ridge (R-R) was formed (Fig. 3b) when the transverse tensile crack front advancing from the left met the preexisting fracture on the cusp face. However, Fig. 3b can claim originality for capturing the same facedirection scarps, e.g. C, located on a ridge. Because the ridge formation mechanism is established, their unique location proves beyond doubt the proposition [3]that same face-direction scarps originate due to "a combination of tensile and shear stresses".

Fig. 4a to c, taken from the interlaminar fracture surface of the same lap shear specimen, exhibit the different aspects of a unique failure feature hitherto perhaps not emphatically reported. It is essentially characterized by separation of two axial fibres due to failure of the resin band in between (Fig. 4a). It is also interesting to note how the separation is arrested. The term interfibre separation will henceforth be used to describe this particular phenomenon. Such a failure can be brought about only by the action of a tensile stress, active on the general peel plane (say OO'SS' in Fig. 1b) in a direction perpendicular to the specimen axis, i.e. along the y-direction. In order to explain the origin of this stress we have to consider the peeled lower half of the specimen (Fig. 1b) to constitute a cantilever fixed at the crack root and subjected to a bending moment, M, at the free end. Because the thickness of this cantilever beam was very small compared to its width ( $t/w \simeq 0.125$ ), the proper approach would be to analyse the system as a "wide" beam [6] subjected to bending. Consequently, this would mean bending under plane strain conditions which leads to development of a stress,  $\sigma_v$ , in the same plane of, but in

a direction perpendicular to the axial stress introduced by the bending action. The magnitude of this separating stress  $\sigma_v$  is  $v\sigma_x$ , where v is the Poisson's ratio of the beam material. Because v is normally very small,  $\sigma_{\rm r}$  has to be considerably large in order that the magnitude of  $\sigma_{v}$  becomes sufficient for effecting the separation. Appreciating that OO'SS' (Fig. 1b) is the plane of highest tension, it is evident that the maximum value of  $\sigma_x$  is encountered just at the crack tip O. The probability of any interfibre separation to take place is, therefore, highest at the tip O, which is also the point of maximum  $\sigma_v$ . The fact that evidence of interfibre separation was found only on the lower fracture surface, namely OS' in Fig. 1b, and not on the upper countersurface, namely OQ, supports the above contention. However, the concept of development of a  $\sigma_v$  alone cannot account for the origin of the *faceted* (sawtooth) morphology of the failed resin band encountered at some locations of interfibre separation (Fig. 4b). In addition, it is to be appreciated that the beam is actually subjected to a system of biaxial tensile stress comprising  $\sigma_x$  and  $\sigma_y$ , which gives rise to a system of shear stress,  $\tau$ , having a maximum magnitude at an angle of  $45^{\circ}$  (Fig. 5) and acting on a plane containing  $\sigma_x$  and  $\sigma_y$ . When the magnitude of this stress,  $\tau$ , exceeds the shear strength, the material may actually fail in shear (as opposed to the cusp formation phenomenon where the actual failure is caused by the resolved tensile component of the existing shear field [3]), giving rise to the faceted appearance of the cleaved faces. Faceted morphology was generally noticed when the interfibre resin band had a rather long, defect-free stretch; and such cases were comparatively rare. Normally the resin band contained many voids [4] which acted as stress raisers and precipitated a tensile failure under  $\sigma_v$ , rather than under shear. Fig. 4c shows two mating facet faces, e.g. PQ and RS, at a higher magnification. It is to be noted that both of these faces are approximately at an angle of 45° to the axial direction – a fact that lends credence to the proposed mechanism.

Fig. 6 displays another unique interlaminar fractograph encountered in the lap shear specimen. The terminology *in plane cross cracking* is being introduced here to describe this particular aspect of failure,



Figure 3 Fractographs showing (a) a ridge and (b) a ridge with same face-direction scarps found in lap shear failure.







as no previous report on similar features is available. The unique feature of this fractograph is, evidently, the fibre(s) piercing through the matrix; although the cross crack lying on the general peel plane but grossly normal to the axial direction (evident from fibre imprint), also deserves consideration. Noting that evidence of in plane cross cracking could be seen only on the lower fracture surface (OO'SS' in Fig. 1b), it is understandable (Fig. 7a) that the propagation direction of the cross crack  $(\mathbf{RR'})$  is commensurate with the operative axial stress  $\sigma_x$  introduced due to the bending action. Further, it is to be noted that because  $\sigma_x$  is maximum just at the tip (O) of the primary peel fracture front, this secondary cross crack is naturally expected to originate very close to the point O (a fact not highlighted in Fig. 7a for the sake of clarity and simplicity). However, in order to understand the origin and significance of the penetrating fibres, it is necessary to invoke the concept of "fibre bridging" [7]. The idea is, during interlaminar failures it is not guaranteed that the advancing crack front will com-

Figure 4 Photographs of (a) interfibre separation and (b) the typical appearance of faceted morphology as well as (c) two mating facet faces (PQ and RS).

pletely separate all the fibres on one side from those on the other. These unseparated fibres, rooted at both the faces run across or, in other words, bridge the crack opening behind the crack tip. Essentially this is the phenomenon of fibre bridging, as illustrated schematically in Fig. 7b. The base of the bridging fibre (P in Fig. 7b) lies on a plane that is subjected to an axial stress (Fig. 7c) resulting from the bending action, while the fibre itself is subjected to some tension,  $T_a$ , because of the bridging action. The stress situation is made further acute by the stress raising effect of the corners (C and C') at the root (Fig. 7c). As a consequence, the fibre-matrix interface at that location may give way, leading to debonding. This is how the secondary failure is precipitated, which subsequently propagates as the cross crack under the action of the axial stress,  $\sigma_x$ . It is very clear from Fig. 6 that the mating faces of the cross crack tend to converge away from the fibre root. This is a confirmatory evidence to justify the argument that such cracks originate at the fibre roots. Further, little resin attachment to fibre 1 near the root conforms to the envisioned debonding phenomena there. However, considerable resin attachment to fibre 2 at a similar location is apparently incommensurate. In fact, it is the region A–B that had actually suffered from debonding during the inception of the cross crack. Subsequently, the segment B-C was pulled out as the tension,  $T_{a}$ , continued to build up with the primary peel fracture front continuing to progress ahead of the bridging fibres and, thus, increasing the separation between the primary crack faces. It would be appropriate to close this topic on in-plane cross



Figure 5 Schematic illustration of the mechanics of formation of faceted morphology during interfibre separation.



Figure 6 SEM fractograph illustrating in-plane cross cracking in lap shear failure.



Figure 7 Schematic illustrations of (a) in-plane cross crack (RR') formation during flexural failure of a lap shear specimen, (b) fibre bridging [7], and (c) debonding at the root (P of Fig. 7b) of a bridging fibre. XY is the plane of lamination.

cracking with the comment that it is believed [7] that the phenomenon of fibre bridging "causes an elevation in the measured interlaminar fracture energy" and we do actually see now that the in-plane cross cracking can constitute a potential energy absorbing mechanism that may be responsible for this enhancement.

The next fractograph (Fig. 8) is from a specimen failed in transverse tension. In view of the overall



Figure 8 Scanning electron micrographs showing wavy pattern.

direction of progress of the interlaminar failure front, the wavy nature of Fig. 8 suggests that the failure marks (e.g. Q) on the resin bands (elevations) located in between two fibres exhibit a distinct "phase lag" with respect to the failure marks (e.g. P) visible on the fibre imprints (depressions). The implication is, adhesive failure fronts [8] tended to advance faster than the neighbouring cohesive failure fronts; and the relative weakness of the adhesive bond between the fibre and the resin might have been the reason for it.

The presence of a three-dimensional defect in the form of a void inside a stressed solid is known to introduce a stress raising effect around it. The stress is maximum at the edge of the discontinuity, its intensity falling off as one moves away from the edge. The plethora of failure features concentrated in the vicinity of a void is perhaps the reflection of this effect projected on to fractographic evidence. In Fig. 9a, taken from a specimen failed in transverse tension, the concentration of failure features, mainly rivers directed outwardly, testifies to local crack fronts in the neighbourhood progressing away from the void. Note that the void is large and deep enough to reveal the impression of fibres belonging to the lamina beneath the fracture plane. There are, however, two principal features to be observed in this fractograph, namely, the curved lines (e.g. PQR) toward the left and the finger-like pattern [3] just below the void, diverging away from it. A magnified view (Fig. 9b) of the latter clearly reveals how the fracture progressed locally, with a curvilinear crack front indicated by MN. It is also interesting to note how some of these finger-like features develop into a river form before terminating at the regions of fibre-matrix interface. Fig. 9c is noteworthy for capturing these features originating just at the edge of this void. It is known that crack fronts always try to reach a free surface via the shortest route. This micrograph remarkably brings it out by demonstrating how crack fronts close enough to the edge of the void (finding it energetically favourable) took a turn and terminated into that void, from which it had originated. To behave in a similar fashion, crack fronts located (and also directed) away from the void edge would be required to suffer a rather severe bend, which was obviously not favoured; and





Figure 9 SEM pictures from a specimen failed in transverse tension illustrating (a) typical appearance of the fracture surface in the neighbourhood of a void, (b) well-developed finger pattern progressing with a curvilinear front, and (c) origin of fracture features at the void edge and termination of some of the local crack fronts into the same void.

they progressed undeviated. The other important feature of Fig. 9a, namely, the curved lines (e.g. PQR) is to be identified as the parabolic feature. It has been suggested [3] that "where an inhomogeneity lies immediately ahead of the advancing crack and in the same plane, a secondary failure may initiate locally at the discontinuity before the main crack reaches it, causing a scarp to form in the shape of a parabola". The opposite direction of rivers present across a segment PQ (Fig. 10a) provides strong fractographic evidence for substantiating the above mechanism, perhaps for the first time. Fig. 10b concentrates on the lower portions of these parabolic lines. It is particularly interesting to find sudden and segmental discontinuities (e.g. A-B, C-D, etc.) in the otherwise continuous fibre imprints, just at the back of the parabolic lines. Also, the region of the discontinuity (e.g. ABCD) is characterized by a sloping (and grossly planar) surface. At the first instance, an explanation for this discontinuity tends to take shape in terms of broken fibres present during fabrication. However, a critical review rules out this possibility because the

Figure 10 Appearance of (a) parabolic features, and discontinuities in the fibre imprints (e.g. A–B, C–D, etc.) at (b) lower and (c) higher magnifications.











Figure 11 Schematic illustrations of (a) the track of the advancing cracks (J-J' and L-L') that merged to form (b) a discontinuity in the fibre imprints on the fracture surface.

occurrence of successive piecewise breakages (e.g. C-D and C'-D') along the same fibre length, and removal of the broken pieces without disturbing the linear arrangement of the remaining length at all, could be considered anything but realistic. A much more reasonable explanation is presented below with the help of a schematic illustration (Fig. 11) of the situation envisioned to be responsible for the origin of this feature. Fig. 11a represents the longitudinal section (not to scale) of the laminate taken through the centre of the fibres. It is postulated that the approaching crack fronts, responsible for parabola formation, were advancing along the fibre-matrix interface (e.g. J-J' and L-L'). At the point J', however, the crack approaching from the left ceased to continue as adhesive in nature and gently diverted into the matrix (probably because of a localized resin weakness there). Had the crack from the right maintained its usual course, a long overlapping of the two crack fronts would have resulted, bringing in unnecessary and excessive surface creation. This crack also, therefore, was made to divert, in fact more sharply into the resin in order to meet the other crack front at a point K located below the interface. An isometric view of the fractured bottom half of Fig. 11a is shown schematically in Fig. 11b. This view closely simulates the appearance of the actual fracture surface in the region of the discontinuity in fibre imprints, as magnified in Fig. 10c.

Referring to the lap shear specimens failed in flexure once again, Fig. 12 can claim uniqueness for documenting three important features condensed in a single fractograph. Needless to say, all these features owe their origin to the presence of a moderately large central void. The parabolic segments (PP) are quite evident. It is more important, however, to concentrate on the ribbons (R) which result from an overlapping of non-planar crack fronts approaching each other [3]. Because the mechanism of parabola formation has already been established, the occurrence of ribbons on the parabolic segments leaves no doubt about the mechanism of ribbon formation too. The last, but the most remarkable feature consists of the incipient



Figure 12 SEM picture revealing the formation of ribbons (R) and stress-induced voids in the vicinity of a large pre-existing void.

stress-induced voids of very small size, clustered around the top left corner. Other than stress concentration, a three-dimensional discontinuity in a stressed body also introduces a state of triaxial stress in its vicinity. In fact, stress-induced voids can form only under the action of hydrostatic tensile stress capable of effecting volume dilatation. Viewed in this light, these nascent voids in Fig. 12 bear micrographic testimony to the development of triaxial stress in the neighbourhood of a pre-existing three-dimensional defect.

It is known that at the microscopic level local crack fronts may progress along different planes and in different directions. A transverse tensile fractograph (Fig. 13a), illustrates the formation of a *cliff* (X–Y) at the edge of the void, when cut by two such non-planar crack fronts, namely 1 and 2. At higher magnification it is revealed (Fig. 13b) that X–Y is actually not a single cliff, but includes a small *ledge* (X'–Y') in the middle. The deep notch MN in Fig. 13a, with an overhanging slice of material testifies to a rather long overlap between the two non-planar crack fronts 3 and 4 progressing approximately in opposite directions.



Figure 13 Photographs illustrating (a) cliff (X-Y) formation, and (b) the ledge (X'Y') made visible at high magnification.

However, the phenomena of failure feature concentration around a void, as observed in transverse tensile and mixed mode (lap shear) failure, were notably absent in failures of interlaminar shear mode.

## 4. Conclusions

Efforts were directed to perform a meaningful analysis of fracture surfaces in polymeric composite laminates. It was intended to generalize the findings of fractographic analysis by way of relating them to the stress state prevailing on the fracture surface. The following conclusions were drawn.

1. Mixed mode failure was observed in lap shear specimen subjected to three-point flexural loading.

2. New fractographic evidence of mechanisms proposed earlier to account for certain features is found in this work.

3. Unique fracture features, such as in-plane cross cracking etc., have been identified in the failed samples. By analysing these features, particularly with reference to the state of stress imposed by the testing procedure, compatible micromechanisms and a possible sequence of formation of these features have been diagramatically illustrated.

4. Voids/three-dimensional discontinuities present in the matrix material were seen to influence significantly the nature and distribution of fracture features in all the examined failure modes.

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