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ON THE EROSION RATE - DROPLET SIZE RELATION FOR
AN EXPERIMENTAL STAND WITH A ROTATING SAMPLE

by

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ABSTRACT

There is still little known about the droplet size-erosion rate patterns for wet steam turbine stages. It has been already proved that the structure of the droplet stream may influence very much some of the impact parameters governing the erosion of the turbine blading (ref. 1). In order to shed more light on the problem, a series of experiments was initiated at IFFM.* In the test stand of IFFM the structure of the droplet stream may be readily controlled. The aim of this report is to present the algorithm relating the structure of the droplet stream in the mentioned stand with some impact parameters. Also an attempt is made toward preliminary evaluation of the experimental results presented in (ref. 3).

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1. NOMENCLATURE

$$A' = \sqrt{\frac{0.85 \mu \rho}{r_*^3 \rho_*^2 c_1}}$$

c_1 air velocity, m/s

c_* droplet velocity, m/s

D mean diameter of the rotor, m

D_s mean diameter of the sample, m

$E \equiv \frac{y_0}{D_s/2}$ coefficient of droplet separation

E_r rationalized erosion rate

F see Fig. 3, m

G constant, m^3/s

$f_n \equiv \frac{1}{\Delta r_*} \frac{n(r_*, c_1)}{N}$ number distribution function, l/m

k factor of proportionality, see Fig. 4, mg/m

L see Fig. 3, m

$n(r_*, c_1)$ number of the droplets of a given size r_* per unit area

N number of all droplets per unit area

N_a normalized erosion resistance

$N_{1,2,3}$ constants

r_* droplet radius, m

u circumferential velocity, m/s

U_a water flux over the sample surface, $m^3/m^2 s$

UA see (18) and (B7)

\bar{U}_a mean value of U_a , $m^3/m^2 s$

- \overline{UA} see (19) and (B9)
- U_{eM} maximum instantaneous value of the volumetric material loss per unit area and unit time m^3/m^2s
- U_{EM} see (23) and (B11)
- $\overline{U_{eM}}$ mean value of U_{eM} , m^3/ms
- $\overline{U_{EM}}$ see (24) and (B12)
- w_* droplets velocity in the relative frame of reference, m/s
- w_{*N} normal component of w_* , m/s
- $\overline{w_{*N}}_{AR}, \overline{w_{*N}}_{AR}, \overline{w_{*N}}_M, \overline{w_{*N}}_M$ mean values of w_{*N} , m/s
- w_1 air velocity in the relative frame of reference, m/s
- α, γ see Fig. 2, $^\circ$
- Δm loss of material eroded, mg
- ΔM_x amount of water supplied onto the trailing edge of the flat plate per its length ΔR_1 , kg/s
- ΔR_1 the length element of the trailing edge of the flat plate, m
- $\Delta(\Delta m)/\Delta\tau$ the maximum slope of the curve $\Delta m = f(\tau)$, $mg/min.$
- κ parameter, see (B1)
- μ viscosity of air, kg/ms
- φ, φ_n see Fig. 3, $^\circ$
- φ_m see Fig. 8, $^\circ$
- ρ density of air, kg/m^3
- ρ_* density of water, kg/m^3
- τ time, minutes

2. INTRODUCTION

There are some characteristic features of the kinematics of a droplet motion in the axial gap of a steam turbine blading.

The neighborhood of the leading edge of the rotor blade is exposed to the impact of droplets whose radius is a function of the location of the blade surface element. The further away from the leading edge of the blade the surface element is located, the smaller are the maximum and mean value of the droplet radius. Also the angle of attack, which is a function of the inclination of the blade surface element and the droplet size, changes.

Another characteristic feature of the droplet impact intensity is its strong dependence upon the structure of the droplet stream. This had been proved in (ref. 1), where the correlation between droplet stream structure and the impact parameters was considered.

In the experimental investigation of the material removal-time patterns for different material, these characteristic features of the droplet stream are usually not taken into account:

1. The droplets, no matter what their size, collide with the sample of the material under the same angle of attack. Usually this angle is 90 degrees.
2. The structure of the droplet stream is usually relatively homogeneous. In general, little is known about this structure and often not even the droplet stream is used in tests, but rather a liquid jet.

It appears, however, that both the angle of attack and the droplet structure have substantial influence upon erosion rate patterns. As far as the influence of the angle of attack is concerned, some information is already available. One comes roughly to the conclusion that the normal component of the impact velocity governs the erosion.* Much less is known about the influence of the droplet size. Only recently (ref. 2) an attempt was made to draw some preliminary conclusions from the meager experimental data; it is there assumed that the drop size effect can be represented by a factor of the form

$$w_{*N}^2 r_* = \text{const}$$

where the constant represents a critical or threshold combination of velocity and droplet size, such that, for $w_{*N}^2 r_* \leq \text{constant}$ no significant erosion occurs.

In order to shed more light on the droplet size effect, among others, a test rig with rotating sample was built in the Institute of Fluid Flow Machinery of the Polish Academy of Sciences. The sample intersects once per rotation the droplet stream generated in the aerodynamic wake of a flat plate. The particular feature of the stand is that the droplet stream structure may readily be controlled by changing the air velocity in the test section, the amount of water per unit time and unit width of the plate, the length of the plate, the shape of the trailing edge of the plate, etc. Mr. B. Weigle designed the stand; he and Mr. H. Severin are in charge of the

* See extensive reference data in (ref. 2).

experiments. The outline of the stand is shown in Fig. 1. More details are available elsewhere, (ref. 3).

The aim of this report is to present the algorithm relating the structure of the droplet stream with mean values of some selected impact parameters. The structure of the droplet stream is defined by the droplet size distribution function. This function is assumed to be known, and to depend upon the droplet size and air velocity. Particular attention is paid to the mean value of the amount of water impinging upon the unit area of the sample per unit time, the mean value of the product of this amount of water and $(w_{*N}/2550)^5$, and the mean value of the impact velocity. A preliminary evaluation of the experimental results presented in (ref. 3) is made.

3. FORMULATION

Before we formulate the conditions of the droplet impact with the rotating sample of the IFFM experimental stand, let us consider the kinematics of the individual droplet.

In some prior authors' papers, it has been indicated that the droplet motion in the aerodynamic wake may be described with sufficient accuracy by the equation:

$$\frac{c_*}{c_1} = 0.8 \left\{ 1 - \frac{1}{[1 + A'z + \sqrt{A'^2 z^2 + 2A'z}]^2} \right\} \quad (1)$$

c_* is the droplet velocity in the absolute reference frame of coordinates. The relationship between this velocity, the gas velocity, and droplet size is shown in Fig. 2. The group of parameters is relevant to the IFFM stand. The droplet velocity w_* in the relative reference frame of coordinates:

$$\vec{w}_* = \vec{c}_* - \vec{u} \quad (2)$$

and the droplet path in this reference frame are also shown.

The conditions of the droplet impact may be easily calculated under the assumption that the droplet path shown in Fig. 2 does not change its shape in the neighborhood of the sample. This assumption has been evaluated in Appendix B in more detail.

Let us now consider first the amount of water which hits the unit area of the sample per unit time. This parameter of the droplet impact is worth considering because it has been already shown that the erosion rate is proportional to it. Let us assume that the structure of the droplet stream is

defined by the droplet size distribution function

$$f_n(r_*, c_1) \equiv \frac{1}{\Delta r_*} \frac{n(r_*, c_1)}{N} \quad (3)$$

and that this function in the point where the sample intersects the droplet stream is given by the equation

$$f_n(r_*, c_1) = N_1 r_*^{N_2} e^{-N_3 r_*}, \quad N_1, N_2, N_3 = f(c_1) \quad (4)$$

Then the number of droplets of a particular size r_* impinging upon the surface element $\Delta R_1 \cdot \frac{D_s}{2} \cdot d\varphi$ of the sample per one crossing of the droplet stream is equal to

$$n(r_*, c_1) L \cdot F = n(r_*, c_1) L \cdot \frac{D_s}{2} d\varphi \sin \varphi \frac{1}{\cos \alpha(r_*, c_1)} \quad (5)$$

The volume of the water carried by these droplets is equal to

$$\frac{4}{3} \pi r_*^3 \cdot n(r_*, c_1) L \frac{D_s}{2} d\varphi \sin \varphi \frac{1}{\cos \alpha(r_*, c_1)} \quad (6)$$

since

$$\cos \alpha(r_*, c_1) = \frac{u \sin \gamma}{w_*(r_*, c_1)} \quad (7)$$

and the frequency with which the sample crosses the droplet stream per unit time is $u / \pi D$. Hence, the volumetric flux relevant to this volume of water is equal to

$$\delta U_a(r_*, c_1, \varphi) = \frac{4}{3} \pi r_*^3 \cdot LN f_n(r_*, c_1) \Delta r_* \frac{w_*(r_*, c_1) \sin \varphi}{u \sin \gamma} \cdot \frac{u}{\pi D} \cdot \frac{1}{\Delta R_1} \quad (8)$$

The product LN may be eliminated by means of the continuity equation

for the liquid phase formulated for the cross-section 0-0 of the droplet

stream. It has the form:

$$\Delta M_* = \sum_{r_*=0}^{\infty} L c_*(r_*, c_1) n(r_*) \cdot \frac{4}{3} \Pi r_*^3 \cdot \rho_*$$

$$= \sum_{r_*=0}^{\infty} L c_*(r_*, c_1) f_n(r_*, c_1) N \Delta r_* \frac{4}{3} \Pi r_*^3 \rho_* \quad (9)$$

The summation, which may be replaced by integration is extended over all the droplet radii in the stream. Instead of Eq. (9), one may write:

$$\Delta M_* = \frac{4}{3} \Pi \rho_* LN \int_0^{\infty} r_*^3 f_n(r_*, c_1) c_*(r_*, c_1) dr_* \quad (10)$$

Hence

$$LN = \frac{\Delta M_*}{\frac{4}{3} \Pi \rho_* \int_0^{\infty} r_*^3 f_n(r_*, c_1) c_*(r_*, c_1) dr_*} \quad (11)$$

For the particular droplet size the angle $\alpha(r_*, c_1)$ may be expressed by the equation:

$$\alpha = \arctan \alpha \quad (12)$$

where

$$\tan \alpha = \text{sign} \left[\sin \left(\frac{c_*(r_*, c_1) - u \cos \gamma}{w_*(r_*, c_1)} \right) \right] \cdot \frac{w_*(r_*, c_1)}{u \cdot \sin \gamma} \sqrt{1 - \frac{u^2 \sin^2 \gamma}{w_*(r_*, c_1)^2}} \quad (13)$$

The sign-function takes care of the appropriate sign of term $\tan \alpha$.

The second part of the R-H-S of the Eq. (13) may be easily deduced from the geometric relationship shown in Fig. 2. It must be remembered that

for each droplet size r_* , we obtain a different α or, in other words,

$\alpha = \alpha(r_*, c_1)$. Hence, in order to validate the Eq. (3) for all droplet

sizes, the $\sin \varphi$ function in it must be replaced by function $F(\varphi)$, such

that (see fig. 3):

$$\begin{aligned}
 F(\varphi) &= 0 && \text{for } \varphi < \varphi_n(r_*, c_1) - \frac{\pi}{2} \\
 F(\varphi) &= \overline{\sin} \left[\varphi - \varphi_n(r_*, c_1) + \frac{\pi}{2} \right] && \text{for } \varphi_n(r_*, c_1) - \frac{\pi}{2} \leq \varphi \leq \varphi_n(r_*, c_1) + \frac{\pi}{2} \\
 F(\varphi) &= 0 && \text{for } \varphi > \varphi_n(r_*, c_1) + \frac{\pi}{2} \quad (1)
 \end{aligned}$$

where:

$$\varphi_n(r_{*i}, c_1) = \varphi_n(r_{*i-1}, c_1) + \alpha(r_{*i-1}, c_1) - \alpha(r_{*i}, c_1) \quad (15)$$

$$i = 1, 2, 3, \dots,$$

also,

$$\varphi_n(r_{*0}, c_1) = \frac{\pi}{2} \text{ and } \alpha(r_{*0}, c_1) = \alpha(r_{*1}, c_1) \quad (16)$$

Rearranging (8) by means of (11) and (14) results in

$$\delta U_a(r_*, c_1, \varphi) = \frac{1}{\rho_*} \cdot \frac{\Delta M_*}{\Delta R_1} \frac{r_*^3 f_n(r_*, c_1)}{\int_0^\infty r_*^3 f_n(r_*, c_1) c_*(r_*, c_1) dr_*} \cdot \frac{w_*(r_*, c_1)}{\pi D_s \sin \gamma} \cdot \overline{\sin} \left[\varphi - \varphi_n(r_*, c_1) + \frac{\pi}{2} \right] \Delta r_* \quad (17)$$

One establishes the volumetric water flux $U_a(c_1, \varphi)$ for a surface element located by φ extending the integration of the value

$\delta U_a(r_*, c_*, \varphi)$ over the whole range of droplet size. Hence one finally

obtains:

$$U_a(c_1, \varphi) = \frac{1}{\rho_*} \frac{\Delta M_*}{\Delta R_1} \frac{1}{\Pi D_s \sin \gamma} \int_0^\infty \frac{r_*^3 f_n(r_*, c_1) w_*(r_*, c_1)}{\int_0^\infty r_*^3 f_n(r_*, c_1) c_*(r_*, c_1) dr_*} \cdot \overline{\sin \left[\varphi - \varphi_n(r_*, c_1) + \frac{\Pi}{2} \right]} dr_* = \frac{1}{\rho_*} \frac{\Delta M_*}{\Delta R_1} \frac{1}{\Pi D_s \sin \gamma} UA(c_1, \varphi) \quad (18)$$

In order to eliminate the dependence of φ one may average $U_a(c_1, \varphi)$ over the circumference of the sample. Hence:

$$\begin{aligned} \bar{U}_a(c_1) &= \frac{1}{\varphi_n(r_{*max}, c_1) + \frac{\Pi}{2} \int_0^\infty r_*^3 f_n(r_*, c_1) c_*(r_*, c_1) dr_*} \cdot \frac{1}{\rho_*} \frac{\Delta M_*}{\Delta R_1} \frac{1}{\Pi D_s \sin \gamma} \int_0^\infty r_*^3 f_n(r_*, c_1) w_*(r_*, c_1) \overline{\sin \left[\varphi - \varphi_n(r_*, c_1) + \frac{\Pi}{2} \right]} dr_* d\varphi \\ &= \frac{1}{\rho_*} \frac{\Delta M_*}{\Delta R_1} \frac{1}{\Pi D_s \sin \gamma} \bar{U}A(c_1) \end{aligned} \quad (19)$$

The erosion rate is obviously not only the function of the flux of water impinging upon the surface element of the rotating sample. It is also a function of the impact velocity. This relationship may be expressed as an empirical equation in the form (ref. 4):

$$\frac{\Delta m}{\tau} = k \frac{U_a}{N_a} \left(\frac{w_* N}{2550} \right)^5 \exp(-0.25 \Delta m / \Delta m_T) \quad (20)$$

For a flat sample the coefficient $k = F \cdot \rho$. Other symbols of Eq. (20) are explained on Fig. 4. The maximum instantaneous erosion rate

$\frac{\Delta(\Delta m)}{\Delta \tau}$ is according to (ref. 4) and Eq. (20) equal to:

$$\frac{\Delta(\Delta m)}{\Delta \tau} = k \frac{U_a}{N_a} \left(\frac{w_{*N}}{2550} \right)^5 \equiv U_{em} \quad (21)$$

Here U_{em} represents the maximum instantaneous value of erosion rate.

These empirical relationships have been established for the flat material sample attacked by a relatively homogeneous (in size and direction) droplet stream. Application of these formulae for the case of cylindrical, rotating samples demands assuming that the erosion damage caused by the droplet groups of a particular size may be superimposed. That assumption has not yet been proved and may be accepted here merely as a first approximation. Under this assumption, the theoretical evaluation of the maximum instantaneous value of erosion rate may be based on the value:

$$\delta U_{em}(r_*, c_1, \varphi) = \delta U_e(r_*, c_1, \varphi) \cdot \left(\frac{w_{*N}(r_*, c_1, \varphi)}{2550} \right)^5 \quad (22)$$

its integral extended over all droplet size:

$$\begin{aligned} U_{em}(c_1, \varphi) &= \int_0^{\infty} \delta U_{em} dr_* = \frac{1}{\rho_*} \frac{\Delta M_*}{\Delta R_1} \frac{1}{\Pi D_s \sin \gamma} \int_0^{\infty} \frac{r_*^3 f_n(r_*, c_1) w_*(r_*, c_1)}{\int_0^{\infty} r_*^3 f_n(r_*, c_1) c_*(r_*, c_1) dr_*} \cdot \\ &\quad \cdot \left[\frac{w_{*N}(r_*, c_1, \varphi)}{2550} \right]^5 \frac{1}{\sin [\varphi - \varphi_n(r_*, c_1) + \frac{\Pi}{2}]} dr_* \\ &= \frac{1}{\rho_*} \frac{\Delta M_*}{\Delta R_1} \frac{1}{\Pi D_s \sin \gamma} \cdot UEM(c_1, \varphi) \end{aligned} \quad (23)$$

and the mean value:

$$\begin{aligned}
 \bar{U}_{eM}(c_1) &= \frac{1}{\varphi_n(r_{*max}, c_1) + \frac{\pi}{2}} \int_0^{\varphi_n(r_{*max}, c_1) + \frac{\pi}{2}} U_{eM}(c_1, \varphi) d\varphi = \\
 &= \frac{1}{\varphi_n(r_{*max}, c_1) + \frac{\pi}{2} \int_0^\infty r_*^3 f_n(r_*, c_1) c_*(r_*, c_1) dr_*} \cdot \frac{1}{\rho_*} \frac{\Delta M_*}{\Delta R_1} \frac{1}{\pi D_s \sin \gamma} \int_0^{\varphi_n(r_{*max}, c_1) + \frac{\pi}{2}} \\
 &\int_0^\infty r_*^3 f_n(r_*, c_1) w_*(r_*, c_1) \left[\frac{w_{*N}(r_*, c_1, \varphi)}{2550} \right] \sin \left[\varphi - \varphi_n(r_*, c_1) + \frac{\pi}{2} \right] dr_* d\varphi = \frac{1}{\rho} \frac{\Delta M_*}{\Delta R_1} \frac{1}{\pi D_s \sin \gamma} \bar{U}_{eM}(c_1) \quad (24)
 \end{aligned}$$

The relevant programs of numerical calculations of these impact parameters are presented in Appendix A and B. In Appendix B, in addition, the equations for local and mean value of the impact velocity are given.

4. SOME RESULTS OF NUMERICAL CALCULATION AND SOME EXPERIMENTAL RESULTS

The theoretical analysis presented in the previous section and in Appendix B had been triggered by interesting experimental results of B. Weigle and H. Severin (ref. 3). They had found that for constant mass flux of water, G_w , supplied per unit width per unit time onto the plate, for constant circumferential velocity, u , of the sample but changing gas velocity c_1 in the range of

$$c_1 = 60 \frac{m}{s} \text{ through } 75, 92, 153 \frac{m}{s},$$

the erosion rate changes substantially. A fragment of the experimental results in the form of the relation $\Delta m = f(\tau)$ is shown in Fig. 5. There is also shown the droplet size distribution functions relevant to the gas velocity c_1 . It is apparent that with the increase of gas velocity c_1 the dispersion of the droplets increases, and the representative mean value of the droplet size

$$r_{*m} = \int_0^{\infty} r_*^3 f_n(r_*, c_1) dr_* \quad (25)$$

decreases, as does the erosion rate represented by the ratio $\Delta(\Delta m)/\Delta\tau$ defined in Fig. 4.

This significant change in erosion rate may be a result from several different effects. One of these may be the change in amount of water impacting the sample per unit area and unit time. As a matter of fact, Heymann's empirical relation (Eq. 20) says that the maximum instantaneous

value of erosion rate $\Delta(\Delta m)/\Delta \tau \equiv U_{em}$ is proportional to the volumetric water flux U_a over the eroded surface of the sample. The other reason for the change of experimental value of $\Delta(\Delta m)/\Delta \tau$ may be the change in the term $U_a \left(\frac{w^*N}{2550}\right)^5$. It is also proportional to $\Delta(\Delta m)/\Delta \tau$

Numerical calculation presented in this section have been performed for the following set of parameters relevant to the stand of IFFM:

$$c_1 = 60, 75, 92, 153 \text{ m/s}$$

$$u = 200 \text{ m/s}$$

$$\gamma = 80^\circ$$

$$D_s = 7 \cdot 10^{-3} \text{ m}$$

$$\Delta r_* = 10 \cdot 10^{-6} \text{ m}$$

$$\mu = 1.85 \cdot 10^{-5} \text{ kg/ms}$$

$$\rho = 1.205 \text{ kg/ms}$$

$$\rho_* = 1000 \text{ kg/ms}$$

$$z = 0.3 \text{ m}$$

In Fig. 6 the results of numerical calculations of \overline{UA} and \overline{UEM} are presented. Both \overline{UA} and \overline{UEM} are proportional to $\overline{U_a}$ and $\overline{U_{em}}$ respectively (see Eq. 19, 24). \overline{UA} and \overline{UEM} decrease, as does the experimental value $\Delta(\Delta m)/\Delta \tau$ when c_1 increases. The trend of change of both theoretical and experimental values is the same. However, it may be readily shown that

$$\left(\frac{\Delta(\Delta m)}{\Delta \tau}\right)_{c_1} / \left(\frac{\Delta(\Delta m)}{\Delta \tau}\right)_{c_1=75} = 75 \text{ m/s}$$

changes much faster than do $\overline{UA}_{C_1} / \overline{UA}_{C_1=75}$ and $\overline{UEM}_{C_1} / \overline{UEM}_{C_1=75}$.

For $c_1 = 153 \text{ m/s}$ there is already a difference in order of magnitude between the theoretical and experimental values. This difference clearly may not be explained as the consequence of averaging, or the assumption concerning the superposition of erosion caused by the droplet groups of particular size.

In searching for a reason for this discrepancy, attention must be paid to the problem of the correlation between the droplet stream structure and the erosion rate pattern. The program of experiments of B. Weigle and M. Severin is particularly suitable for these investigations because their stand makes it possible to change the droplet stream structure easily for almost unchanged mean impact velocity (Fig. 6).

Little is known, however, about the droplet size-erosion rate relationship, and only in exceptional experiments was the droplet size investigated. However, it has been indicated (ref. 2) that the droplet size is probably related to the maximum instantaneous value of rationalized erosion rate, through the influence on its threshold value. The rationalized erosion rate has been defined in (ref. 2) as follows:

$$E_r = \frac{\text{Volume of material lost per unit area per time}}{\text{Volume of liquid impinged per unit area per unit time}}$$

In the nomenclature of this report:

Volume of material lost per unit area per unit time =

$$= \Delta(\Delta m) \frac{l}{\rho_s} / \Delta \tau \cdot L$$

and

Volume of liquid impinged per unit area per unit time =

$$= \bar{U}_a = \frac{l}{\rho_*} \frac{\Delta M_*}{\Delta R_l} \frac{l}{\pi D_s \sin \gamma} \bar{U} \bar{A}$$

Hence

$$E_r(c_1) \sim \frac{\Delta(\Delta m) / \Delta \tau}{\bar{U} \bar{A}}$$

In Ref. 2 Heymann suggests the droplet size-erosion rate dependence in the form of the following possible relations:

$$E_r = f_1 \left[w_{*N} \left(1 - \frac{G}{w_{*N}^2 r_*} \right) \right] \quad \text{or} \quad E_r = f_2 \left[w_{*N} - \sqrt{\frac{G}{r_*}} \right] \quad (26)$$

For both of these relations, which strictly speaking, are not yet fully confirmed by experiment, the terms $\frac{G}{w_{*N}^2 r_*}$ and $\sqrt{G/r_*}$ relate to the threshold impact velocity w_{*Nc} such that for $w_{*N} \leq w_{*Nc}$ no erosion results. G may be considered as a material constant. Now, the experimental data of (ref. 3) for an aluminum sample, partially quoted in Fig. 5, may be used to determine this constant G , and thus the relationship between w_{*Nc} and r_* . It may be done by means of the plot $E_r = f(r_*)$ because in the experiments mentioned, the mean value of the impact velocity w_{*N} was almost constant. In fact, for $u = 200$ m/s the mean impact velocity changes only between

$$\bar{w}_{*N} = 119; 117.1; 115.5 \text{ and } 113.5 \text{ m/s}$$

for

$$c_1 = 60; 75; 92 \text{ and } 153 \text{ m/s respectively.}$$

Thus, from the appropriate extrapolation of the function $E_r = f(r_*)$ (see Fig. 7), for a given impact velocity $\bar{w}_{*N} = \sim 116$ m/s, results the threshold droplet size $r_{*c} = 44 \cdot 10^{-6}$ m. Hence, for the material considered

$$G = d_{*c} \cdot \bar{w}_{*N}^2 = 1,185 \frac{\text{m}^3}{\text{s}^2}$$

The relation $w_{*Nc} = f(r_*)$ for an aluminum sample used in (ref. 3) is shown in the Fig. 6. The results of the Busch and Hoff experiments (ref. 2) may be here recollected. They obtained for the rain-eroded aluminum the threshold velocity $w_{*Nc} \approx 33$ m/s. Hence, the corresponding droplet size d_* would be the order $1 \cdot 10^{-3}$ m which appears qualitatively reasonable for the rain droplet size.

It has to be pointed out here that in this report only a small part of the results of (ref. 3) has been used. In addition, in assessment of r_{*c} and G , the results for only one velocity w_{*N} have been used. More extensive experimental data are needed to shed more light on the problem. As far as the author is informed, the program of relevant experiments is being continued in IMMF.

5. CONCLUSIONS

1. The experimental investigation of B. Weigle and H. Severin, published in (ref. 3), indicated substantial dependence of the derosion damage-time patterns upon the droplet size.

2. These experimental results of IFFM deserve certainly some more consideration. To make it possible, a theoretical model of droplet impact for the experimental stand of IFFM has been presented in this report.

Attention has been particularly paid to the calculations of the mean value of the volumetric water flux U_a over the sample surface. Also the mean value of the product of water flux, U_a , and $\left(\frac{w_{*N}}{2550}\right)^5$ as well as the mean impact velocity w_{*N} have been calculated.

3. Comparison of experimental and theoretical data indicated that the experimentally established rationalized erosion rate, E_r , changes faster with the gas velocity c_1 than theoretically calculated water flux \bar{U}_a and the mean value \bar{U}_{eM} of the product $U_a \left(\frac{w_{*N}}{2550} \right)^5$.

4. It is likely that variation in droplet size causes this discrepancy. The threshold droplet size of order $r_* = \sim 44 \cdot 10^{-6}$ m for mean impact velocity $\bar{w}_{*N} = \sim 116$ m/s for soft aluminum has been established under the assumption (ref. 2) that the droplet size effect can be represented by a factor of the form:

$$w_{*N}^2 r_* = \text{const} = 1/G$$

5. The model presented of the droplet impact indicated that the experimental stand of IFFM is particularly suitable for experiments designed to shed more light on the problem of droplet size effect:

5.1 It provides the possibility of the exceptional change of the mean droplet size between $r_* = \sim 50 \cdot 10^{-6}$ m up to about $r_* = \sim 300 \cdot 10^{-6}$ m.

5.2 The mean value of the impact velocity may be controlled independently.

5.3 The droplet stream structure seems to approach the stream structure in the steam turbine.

6. Further investigation of the relationship between the erosion resistance and the droplet size are of importance because in (ref. 1) it has been already shown that the impact parameters are also seriously influenced by the droplet stream structure. Thus the control of the droplet stream

structure may offer a powerful method for the protection of steam turbine blading.

6. ACKNOWLEDGMENTS

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7. APPENDICES

APPENDIX A

The program of numerical calculations has been formulated in Algol for the Burroughs B 5500 computer. The text of the program is quoted below. It makes it possible to calculate:

$$\varphi_n(r_*, c_1), \alpha(r_*, c_1), r_{*m}(c_1), \varphi_n(r_{*max}, c_1),$$

$$UA(c_1, \varphi), UEM(c_1, \varphi), \overline{UA}(c_1), \overline{UEM}(c_1)$$

according to the equation (15), (12), (25), (15), (18), (23), (19), and (24), respectively.

The input data have to be sequenced according to the sequence numbers: 900 and 1000 where:

$$c_1 \equiv C_1, u \equiv U, \gamma \equiv GAMA, \Delta r_* \equiv DR, \mu \equiv MI, \rho \equiv RO,$$

$$\rho_* \equiv RO_1, z \equiv Z, N_1 \equiv N11, N_2 \equiv N22, N_3 \equiv N33$$

The output data are being printed in the sequence resulting from sequence numbers: 2350, 2500, 5050, and 5800 where:

$$r_* \equiv R[K], \varphi_n(r_*, c_1) \equiv FINDGR[K], \alpha(r_*, c_1) \equiv ALFADGR[K], r_{*m} \equiv J[51],$$

$$\varphi_n(r_{*max}, c_1) + \frac{\Pi}{2} \equiv FIMAXDGR, \Delta\varphi \equiv \frac{\varphi_n(r_{*max}, c_1) + \Pi/2}{N_{max} - 1} \equiv DELTAFIDGR,$$

$$N, \varphi(N) \equiv FIDGR[N], UA(c_1, \varphi) \equiv UA[N], UEM(c_1, \varphi) \equiv UEM[N],$$

$$\overline{UA}(c_1) \equiv UASR[N_{max}], \overline{UEM}(c_1) \equiv UEMSR[N_{max}].$$

```
100 BEGIN
200 COMMENT BADANIE UA,UEM DLA WIRUJACEJ PROBKI W FUNKCJI C1 (WAR II);
300 REAL C1,U,GAMA,N11,N22,N33,DR,MI,RO,RO1,Z,CALKA1,CALKA2,
350 FIMAX,DELTAFI,FIMAXDGR,DELTAFIDGR,CALKA3,CALKA4,
360 CALKA5,CALKA6;
400 INTEGER K,N;
500 LABEL ET1;
600 ARRAY R, FN, A1, CKR, H, I, G, J, L, WKR, M, O, UA, P, UEM,
650 TGALFA, ALFA, FIN, ALFADGR, FINDGR, WKRNC[1:51],
660 A3, A4, UASR, UEMSR, FI, FIDGR[1:21];
700 FILE JAK33 REMOTE(2,9);
800 FORMAT FORMAT1(5E14.3);
850 FORMAT FORMAT2(X1);
900 ET1: READ(JAK33,/,C1,U,GAMA,DR,MI,RO,RO1,Z);
1000 READ(JAK33,/,N11,N22,N33);
1010 CALKA1:=0;
1020 CALKA2:=0;
1100 K:=2;
1200 R[K-1]:=0;
1205 G[K-1]:=0;
1210 H[K-1]:=0;
1220 I[K-1]:=0;
1230 J[K-1]:=0;
1300 FOR K:=2 STEP 1 UNTIL 51 DO
1400 BEGIN
1500 R[K]:=R[K-1]+DR;
1600 FN[K]:=N11\R[K]*N22\EXP(-N33\R[K]);
1700 A1[K]:=SQRT((.85\MI\RO)/(R[K]*3\RO1*2\C1));
1800 CKR[K]:=0.8\CIN(1-1/(1+A1[K]\Z+
1900 SQRT(A1[K]*2\Z*2+2\A1[K]\Z))*2);
1910 WKR[K]:=SQRT(6*2*(SIN(GAMA))*2+(CKR[K]-UNCOS(GAMA))*2);
2000 G[K]:=R[K]*3\FN[K];
2050 H[K]:=R[K]*3\CKR[K]\FN[K];
2055 O[K]:=R[K]*3\WKR[K]\FN[K];
2100 CALKA1:=(H[K-1]+H[K])/2\DR;
2200 I[K]:=I[K-1]+CALKA1;
2300 CALKA2:=(G[K-1]+G[K])/2\DR;
2310 J[K]:=J[K-1]+CALKA2;
2320 TGALFA[K]:=SIGN(SIN((CKR[K]-UNCOS(GAMA))/WKR[K]))\
2323 WKR[K]/(UNSIN(GAMA))\SQRT(1-U*2*(SIN(GAMA))*2/
2326 WKR[K]*2);
2330 ALFA[K]:=ARCTAN(TGALFA[K]);
2333 IF K EQL 2
2336 THEN FIN[K]:=1.570796
2339 ELSE FIN[K]:=FIN[K-1]+ALFA[K-1]-ALFA[K];
2340 ALFADGR[K]:=ALFA[K]\360/6.283185;
2345 FINDGR[K]:=FIN[K]\360/6.283185;
2350 WRITE(JAK33,FORMAT1,R[K],FINDGR[K],ALFADGR[K]);
2360 END;
2370 FIMAX:=FIN[51]+1.570796;
2380 DELTAFI:=FIMAX/20;
2385 DELTAFIDGR:=DELTAFI\360/6.283185;
2390 J[51]:=J[51]*.3333;
2400 FIMAXDGR:=FIMAX\360/6.283185;
2450 WRITE(JAK33,FORMAT2);
2500 WRITE(JAK33,FORMAT1,J[51],FIMAXDGR,DELTAFIDGR);
2600 FOR N:=2 STEP 1 UNTIL 21 DO
2700 BEGIN
```



```
2701     FIC1J:=0;
2702     A3C1J:=0;
2704     UASRC1J:=0;
2706     A4C1J:=0;
2708     UEMSR1J:=0;
2800     FIENJ:=FIEN-1J+DELTAFI;
2850     FIDGRNJ:=FIENJ\360/6.283185;
2900     FOR K:=2 STEP 1 UNTIL 51 DO
3000         BEGIN
3100             WKRNKJ:=WKR[KJ]\SIN(FIENJ-ALFAC2J+ALFACKJ);
3200             M1J:=0;
3210             UAC1J:=0;
3250             P1J:=0;
3260             UEM1J:=0;
3300             IF FIENJ LSS (FIN[KJ]-1.570796)
3400                 THEN M[KJ]:=0
3500                 ELSE IF FIENJ GTR (FIN[KJ]+1.570796)
3600                     THEN M[KJ]:=0
3700                     ELSE M[KJ]:=1/I[51J]\O[KJ]\SIN(FIENJ-FIN[KJ]+
3800                         1.570796);
3900             CALKA3:=(M[KJ]+M[K-1J])/2\DR;
4000             UACKJ:=UACK-1J+CALKA3;
4100             IF FIENJ LSS (FIN[KJ]-1.570796)
4200                 THEN P[KJ]:=0
4300                 ELSE IF FIENJ GTR (FIN[KJ]+1.570796)
4400                     THEN P[KJ]:=0
4500                     ELSE P[KJ]:=1/I[51J]\O[KJ]\(WKRNKJ/2550)*5\
4600                         SIN(FIENJ-FIN[KJ]+1.570796);
4700             CALKA4:=(P[KJ]+P[K-1J])/2\DR;
4800             UEM[KJ]:=UEM[K-1J]+CALKA4;
5000         END;
5050         WRITE(JAK33,FORMAT1,N,FIDGRNJ,UAC[51J],UEMC[51J]);
5100     A3[NJ]:=1/FIMAX\UAC[51J];
5200     A4[NJ]:=1/FIMAX\UEM[51J];
5300     CALKA5:=(A3[NJ]+A3[N-1J])/2\DELTAFI;
5400     UASR[NJ]:=UASR[N-1J]+CALKA5;
5500     CALKA6:=(A4[NJ]+A4[N-1J])/2\DELTAFI;
5600     UEMSR[NJ]:=UEMSR[N-1J]+CALKA6;
5700 END;
5800 WRITE (JAK33,FORMAT1,UASR[21J],UEMSR[21J]);
5900 GO TO ET1;
6000 END.
```

APPENDIX B

In section 2 one has accepted the simplified assumption that the presence of the round-shaped sample does not change the droplet paths in the sample's neighborhood. The validity of this assumption has to be checked. Unfortunately, the solution of the equation of the droplet path for the case under consideration does not exist. There do exist, however, such solutions for the case where the droplet velocity and its direction far from the circular obstacle is equal to the gas velocity (ref. 5, 6). Let us observe, though, that in the case of a rotating sample:

1. For small droplets, where the presence of the sample may influence substantially the droplet path, the differences between the gas and the droplet velocities far from the sample are small.
2. For large droplets, these differences may be large, but the presence of the sample does not influence the droplet path.

One might expect then, that for certain convenient arrangements of parameters, one could apply an already existing solution for the case under consideration. Let us check whether this is not possible in our case.

To this end, let us establish with the aid of (ref. 6) the values of

$$E \equiv \frac{y_{om}}{D_s/2} \quad \text{and} \quad \zeta_m \quad (\text{Fig. 8}).$$

Both are functions of the parameters:

$$\kappa = \frac{2 \rho_* r_*^2 w_1}{9 \mu \cdot D_s/2} \quad (\text{B1})$$

and

$$\varepsilon = \frac{18 \cdot \rho^2 \cdot D_s/2 \cdot w_1}{\mu \rho_*} \quad (\text{B2})$$

For the parameters relevant to the stand of IFFM and $w_1 = 200$ m/s, we obtain $\varepsilon = \sim 1000$ and:

r_*	10^{-6} m	5	10	20	40	60	80	100	1000
κ	---	17.2	69.0	286	915	2540	4520	6900	69000
E	---	0.830	0.920	9.977	0.990	1.0	1.0	1.0	1.0
φ_m	rad	1.450	1.515	1.540	1.560	1.564	1.566	1.568	1.5

It appears that in the case under consideration, only for droplets smaller than about $r_* = \sim 50 \cdot 10^{-6}$ m, is the droplet path slightly modified in the neighborhood of the sample. For these small droplets, the application of the assumption that the droplet and gas velocity far from the obstacle does not differ significantly, seem to be fully justified.

One may apply the same reasoning as in Section 3. Then

$$\delta U_a = \frac{.1}{\rho_*} \frac{\Delta M_*}{\Delta R_1} \frac{1}{\Pi D_s \sin \gamma} \frac{\int_0^{\infty} r_*^3 f_n(r_*, c_1) c_*(r_*, c_1) dr_*}{\int_0^{\infty} r_*^3 f_n(r_*, c_1) w_*(r_*, c_1) \frac{\Delta y(r_*, c_1)}{\Delta \varphi \cdot D_s/2} \Delta r_*} \quad (B3)$$

The new elements of the nomenclature are explained in Fig. 8. Now, according to (ref. 6) one has:

$$\frac{y_o}{D_s/2} = \frac{y_{om}}{D_s/2} \left| \frac{1}{\sin \left[\frac{\Pi}{2\varphi_m} (\varphi - \varphi_n) \right]} \right| \quad (34)$$

Here the function $\overline{\sin}$ is defined such that

$$\begin{aligned} \left| \overline{\sin} \left[\frac{\Pi}{2\varphi_m} (\varphi - \varphi_n) \right] \right| &= 0 && \text{for } \varphi < \varphi_n - \varphi_m, \\ \left| \overline{\sin} \left[\frac{\Pi}{2\varphi_m} (\varphi - \varphi_n) \right] \right| &= \left| \sin \left[\frac{\Pi}{2\varphi_m} (\varphi - \varphi_n) \right] \right| && \text{for } \varphi_n - \varphi_m \leq \varphi \leq \varphi_n + \varphi_m, \\ \left| \overline{\sin} \left[\frac{\Pi}{2\varphi_m} (\varphi - \varphi_n) \right] \right| &= 0 && \text{for } \varphi > \varphi_n + \varphi_m \end{aligned} \quad (\text{B5})$$

Hence,

$$\frac{\Delta y(r_*, c_1)}{\Delta \varphi \cdot D_s/2} = \frac{1}{D_s/2} \frac{dy_o}{d\varphi} = \frac{y_{om}}{D_s/2} \frac{\Pi}{2\varphi_m} \overline{\cos} \left[\frac{\Pi}{2\varphi_m} (\varphi - \varphi_n) \right] \quad (\text{B6})$$

The parameters of the droplet path: $\frac{y_{om}}{D_s/2} \equiv E$, and φ_m and φ_n are functions of r_* and c_1 . From (B3), (B6), and the integration of

$\delta U_a(r_*, c_1, \varphi)$ extended over $0 \leq r_* \leq \infty$ results:

$$\begin{aligned} U_a(c_1, \varphi) &= \frac{1}{\rho_*} \frac{\Delta M_*}{\Delta R_1} \frac{1}{\Pi D_s \sin \gamma} \cdot \int_0^\infty \frac{r_*^3 f_n(r_*, c_1) w_*(r_*, c_1)}{\int_0^\infty r_*^3 f_n(r_*, c_1) c_*(r_*, c_1) dr_*} \cdot \frac{y_{om}}{D_s/2} \cdot \frac{\Pi}{2\varphi_m} \cdot \\ &\quad \overline{\cos} \left[\frac{\Pi}{2\varphi_m} (\varphi - \varphi_n) \right] dr_* = \frac{1}{\rho_*} \frac{\Delta M_*}{\Delta R_1} \frac{1}{\Pi D_s \sin \gamma} \cdot UA(c_1, \varphi) \end{aligned}$$

(B7)

Here again the $\overline{\cos}$ function is such that

$$\begin{aligned} \overline{\cos \left[\frac{\Pi}{2\varphi_m} (\varphi - \varphi_n) \right]} &= 0 && \text{for } \varphi < \varphi_n - \varphi_m \\ \overline{\cos \left[\frac{\Pi}{2\varphi_m} (\varphi - \varphi_n) \right]} &= \cos \frac{\Pi}{2\varphi_m} (\varphi - \varphi_n) && \text{for } \varphi_n - \varphi_m \leq \varphi \leq \varphi_n + \varphi_m \\ \overline{\cos \left[\frac{\Pi}{2\varphi_m} (\varphi - \varphi_n) \right]} &= 0 && \text{for } \varphi > \varphi_n + \varphi_m \end{aligned} \quad (\text{B8})$$

The mean value $\overline{U}_a(c_1)$ of the function $U_a(c_1, \varphi)$ calculated for $0 \leq \varphi \leq (\varphi_n + \varphi_m)_{r_{*max}, c_1}$ may be obtained from:

$$\begin{aligned} \overline{U}_a(c_1) &= \frac{1}{(\varphi_n + \varphi_m)_{r_{*max}, c_1}} \frac{\frac{1}{\rho_*} \frac{\Delta M_*}{\Delta R_1} \frac{1}{\Pi D_s \sin \gamma}}{\int_0^\infty r_*^3 f_n(r_*, c_1) c_*(r_*, c_1) dr_*} \cdot \int_0^{(\varphi_n + \varphi_m)_{r_{*max}, c_1}} \overline{\cos \left[\frac{\Pi}{2\varphi_m} (\varphi - \varphi_n) \right]} d\varphi \\ &= \frac{1}{(\varphi_n + \varphi_m)_{r_{*max}, c_1}} \frac{\frac{1}{\rho_*} \frac{\Delta M_*}{\Delta R_1} \frac{1}{\Pi D_s \sin \gamma}}{\int_0^\infty r_*^3 f_n(r_*, c_1) c_*(r_*, c_1) dr_*} \cdot \overline{UA}(c_1) \end{aligned} \quad (\text{B9})$$

In order to calculate U_{eM} and \bar{U}_{eM} one has to take into account that:

$$\frac{w_{*N}(r_*, c_1)}{w_*(r_*, c_1)} = \frac{dy_o}{d\varphi} = \frac{y_{om}}{D_s/2} \frac{\Pi}{2\varphi_m} \overline{\cos \left[\frac{\Pi}{2\varphi_m} (\varphi - \varphi_n) \right]}$$

(B10)

This relation results from the continuity equation. Hence,

$$U_{eM}(c_1, \varphi) = \frac{1}{\rho_*} \frac{\Delta M_*}{\Delta R_1} \frac{1}{\Pi D_s \sin \gamma} \cdot \frac{\int_0^\infty r_*^3 f_n(r_*, c_1) w_*(r_*, c_1) \frac{y_{om}}{D_s/2} \frac{\Pi}{2\varphi_m}}{\int_0^\infty r_*^3 f_n(r_*, c_1) c_*(r_*, c_1) dr_*} \cdot \overline{\cos \left[\frac{\Pi}{2\varphi_m} (\varphi - \varphi_n) \right]} \left(\frac{w_{*N}}{2550} \right)^5 dr_* = \frac{1}{\rho_*} \frac{\Delta M_*}{\Delta R_1} \frac{1}{\Pi D_s \sin \gamma} \cdot U_{EM}(c_1, \varphi)$$

(B11)

and:

$$\bar{U}_{eM}(c_1) = \frac{1}{(\varphi_n + \varphi_m) r_{*max, c_1}} \cdot \frac{\frac{1}{\rho_*} \frac{\Delta M_*}{\Delta R_1} \frac{1}{\Pi D_s \sin \gamma} \int_0^{(\varphi_n + \varphi_m) r_{*max, c_1}} \int_0^\infty r_*^3 f_n(r_*, c_1) c_*(r_*, c_1) dr_*}{\int_0^\infty r_*^3 f_n(r_*, c_1) w_*(r_*, c_1) \left(\frac{w_{*N}}{2550} \right)^5 \frac{y_{om}}{D_s/2} \frac{\Pi}{2\varphi_m} \overline{\cos \left[\frac{\Pi}{2\varphi_m} (\varphi - \varphi_n) \right]} dr_* d\varphi}$$

$$= \frac{1}{(\varphi_n + \varphi_m) r_{*max, c_1}} \cdot \frac{\frac{1}{\rho_*} \frac{\Delta M_*}{\Delta R_1} \frac{1}{\Pi D_s \sin \gamma}}{\int_0^\infty r_*^3 f_n(r_*, c_1) c_*(r_*, c_1) dr_*} \cdot \bar{U}_{EM}(c_1) \quad (B12)$$

The program of the numerical calculations is quoted below. It provides also the possibility to calculate the mean values of the impact velocity.

The following mean values have been introduced:

$$\overline{w_{*N}}_{AR}(c_1, \varphi) = \frac{\int_0^{\infty} w_{*N}(r_*, c_1, \varphi) f_n(r_*, c_1) dr_*}{\int_0^{\infty} f_n(r_*, c_1) dr_*} \quad (B9)$$

$$\overline{w_{*N}}_M(c_1, \varphi) = \frac{\int_0^{\infty} w_{*N}(r_*, c_1, \varphi) r_*^3 f_n(r_*, c_1) dr_*}{\int_0^{\infty} r_*^3 f_n(r_*, c_1) dr_*} \quad (B10)$$

and

$$\overline{w_{*N}}_{AR}(c_1) = \frac{1}{(\varphi_n + \varphi_m) r_{*max}} \int_0^{(\varphi_n + \varphi_m) r_{*max}} \overline{w_{*N}}_{AR}(c_1, \varphi) d\varphi \quad (B11)$$

$$\overline{w_{*N}}_M(c_1) = \frac{1}{(\varphi_n + \varphi_m) r_{*max}} \int_0^{(\varphi_n + \varphi_m) r_{*max}} \overline{w_{*N}}_M(c_1, \varphi) d\varphi \quad (B12)$$

The numerical calculations were performed for the same group of parameters as in Section 4. The results are shown also in that section.

The input data have to be sequenced according to the sequence numbers:

900, 1000, 1004, where:

$$c_1 \equiv C1, u \equiv U, \gamma \equiv GAMA, D_s/2 \equiv RP, \text{ the amount of terms in the } (r_*)\text{-array} \equiv Z1,$$

$$(\varphi)\text{-array} \equiv Z2, (\kappa, E, \varphi_m)\text{-array} \equiv Z3, \text{ respectively, } \Delta r_* \equiv DR, \mu \equiv MI, \rho \equiv RO,$$

$$\rho_* \equiv RO1, z \equiv Z, N_1 \equiv N11, N_2 \equiv N22, N_3 \equiv N33, \text{ array: } (\kappa \equiv KB1[S], E \equiv E1[S],$$

$$\varphi_m \equiv F1M1[S]) \text{ for } 1 \leq S \leq Z3,$$

The array (κ, E, φ_m) called JAK1 may be stored either on tape or in the memory of the computer. It is being put into the program in accordance with the sequence number 1004.

The output data are being printed in the sequence resulting from sequence numbers: 2350, 2500, 5050, and 5800 where:

$$r_* \equiv R[K], \varphi_n(r_*, c_1) \equiv F1NDGR[K], \alpha(r_*, c_1) \equiv ALFADGR[K], \kappa(r_*, c_1) \equiv KB[K],$$

$$E(r_*, c_1) \equiv E[K], \varphi_m(r_*, c_1) \equiv F1M[K], r_{*m} \equiv J[Z1], (\varphi_n + \varphi_m)_{r_{*max}} \equiv F1MAXDGR,$$

$$\Delta\varphi = \frac{(\varphi_n + \varphi_m)_{r_{*max}}}{Z2-1} \equiv DELTAF1DGR, N, \varphi[N] \equiv F1DGR[N], UA(c_1, \varphi) \equiv UA[Z1],$$

$$UEM(c_1, \varphi) \equiv UEM[Z1], \bar{w}_{*N}(c_1, \varphi) \equiv WKRNRAR[Z1], \bar{w}_{*N}(c_1, \varphi) \equiv WKRNMAS[Z1],$$

AR M

$$\overline{UA}(c_1) \equiv UASR[Z2], \overline{UEM}(c_1) \equiv UEMS[R[Z2]], \bar{w}_{*N}(c_1) \equiv WKRNRASR[Z2], \bar{w}_{*N}(c_1) \equiv$$

AR M

WKRNMASR[Z2].


```
100 BEGIN
200 COMMENT BADANI: UA,DEM DLA WIRUJACEJ PROBK I W FUNKCJI C1 (WARIII);
300 REAL C1,U,GAMA,N11,N22,N33,DR,MI,RO,RO1,Z,CALKA1,CALKA2,
350 FIMAX,DELTAFI,FIMAXDGR,DELTAFIGR,CALKA3,CALKA4,
360 CALKA5,CALKA6,CALKA7,CALKA8,CALKA9,CALKA10,RP;
400 INTEGER K,N,S,T,Z1,Z2,Z3;
500 LABEL ET1,ET2,ET3,ET4,ET5;
600 ARRAY R, FN, A1, CKR, H, I, G, J, L, WKR, M, O, UA, P, UEM,
650 TGALFA, ALFA, FIN, ALFADGR, FINDER, WKRN,
655 Q1, WKRNR, Q2, WKRNMAS, KB, FIM, EC[1:300],
660 A3, A4, UASR, UEMSR, FI, FIDGR, A5, A6, WKRNRSR, WKRNMASRC[1:200],
670 KB1, E1, FIM1[1:100];
700 FILE JAK33 REMOTE(2,9);
710 FILE JAK1 DISK SERIAL(2,10,30);
800 FORMAT FORMAT1(6E11.3);
850 FORMAT FORMAT2(X1);
900 ET1: READ(JAK33,/,C1,U,GAMA,RP,Z1,Z2,Z3,DR,MI,RO,RO1,Z);
1000 READ(JAK33,/,N11,N22,N33);
1001 FOR S:=1 STEP 1 UNTIL Z3 DO
1002 BEGIN
1004 READ(JAK1,/,KB1[S],E1[S],FIM1[S]);
1005 END;
1010 CALKA1:=0;
1020 CALKA2:=0;
1100 K:=2;
1200 R[K-1]:=0;
1205 G[K-1]:=0;
1210 H[K-1]:=0;
1220 I[K-1]:=0;
1230 J[K-1]:=0;
1300 FOR K:=2 STEP 1 UNTIL Z1 DO
1400 BEGIN
1500 R[K]:=R[K-1]+DR;
1600 FN[K]:=N11\NR[K]*N22\EXP(-N33\NR[K]);
1700 A1[K]:=SQRT((.85\MINFO)/(R[K]*3\RO1*2\C1));
1800 CKR[K]:=+.8\C1*(1-1/(1+A1[K]\Z+
1900 SQRT(A1[K]*2\Z*2+2\A1[K]\Z))*2);
1910 WKR[K]:=SQRT(U*2*(SIN(GAMA))*2+(CKR[K]-UNCOS(GAMA))*2);
1915 KB[K]:=(2\1000\NR[K]*2\WKR[K])/(9\MINRP);
1920 T:=0;
1925 ET2: T:=T+1;
1930 IF KB[K] GTR KB1[T] THEN GO TO ET3;
1935 ET3: IF KB[K] LSS KB1[T+1] THEN
1940 E[K]:=(E1[T+1]-E1[T])/(KB1[T+1]-KB1[T])\
1945 (KB[K]-KB1[T])+E1[T]
1960 ELSE GO TO ET2;
1965 T:=0;
1970 ET4: T:=T+1;
1975 IF KB[K] GTR KB1[T] THEN GO TO ET5;
1980 ET5: IF KB[K] LSS KB1[T+1] THEN
1985 FIM[K]:=(FIM1[T+1]-FIM1[T])/(KB1[T+1]-KB1[T])\
1990 (KB[K]-KB1[T])+FIM1[T]
1995 ELSE GO TO ET4;
2000 G[K]:=R[K]*3\FN[K];
2050 H[K]:=R[K]*3\CKR[K]\FN[K];
2055 O[K]:=R[K]*3\WKR[K]\FN[K];
```

```

2100 CALKA1:=(CHK-1)+HEK)/2\DR;
2200 I[K]:=I[K-1]+CALKA1;
2300 CALKA2:=(CEK-1)+ECK)/2\DR;
2310 J[K]:=J[K-1]+CALKA2;
2320 TGA[FACK]:=SIGN(SIN((CKREK)-UNCOS(GAMA))/WKR[K])\
2323 WKR[K]/(UN\SIN(GAMA))\SQRT(1-U*2\((SIN(GAMA))*2/
2326 WKR[K]*2));
2330 ALFACK:=ARCTAN(TGA[FACK]);
2333 IF K EQL 2
2336 THEN FIN[K]:=1.570796
2339 ELSE FIN[K]:=FIN[K-1]+ALFACK-1]-ALFACK];
2340 ALFADGR[K]:=ALFACK\360/6.283185;
2345 FINDGR[K]:=FIN[K]\360/6.283185;
2350 WRITE(JAK33,FORMAT1,K[K],FINDGR[K],ALFADGR[K],
2355 KB[K],E[K],FIN[K]);
2360 END;
2370 FIMAX:=FIN[Z1]+FIM[Z1];
2380 DELTAFI:=FIMAX/(Z2-1);
2385 DELTAFIDGR:=DELTAFI\360/6.283185;
2390 JU[Z1]:=JU[Z1]*.3333;
2400 FIMAXDGR:=FIMAX\360/6.283185;
2450 WRITE (JAK33,FORMAT2);
2500 WRITE(JAK33,FORMAT1,JU[Z1],FIMAXDGR,DELTAFIDGR);
2600 FOR N:=2 STEP 1 UNTIL Z2 DO
2700 BEGIN
2701 FI[1]:=0;
2702 A3[1]:=0;
2704 UASR[1]:=0;
2706 A4[1]:=0;
2708 UEMSR[1]:=0;
2710 A5[1]:=0;
2712 WKRNAPSR[1]:=0;
2714 A6[1]:=0;
2716 WKRNMASSR[1]:=0;
2800 FI[N]:=FI[N-1]+DELTAFI;
2850 FIDGR[N]:=FI[N]\360/6.283185;
2900 FOR K:=2 STEP 1 UNTIL Z1 DO
3000 BEGIN
3100 IF FI[N] LSS (FIN[K]-FIM[K])
3110 THEN WKR[K]:=0
3120 ELSE IF FI[N] GTR (FIN[K]+FIM[K])
3130 THEN WKR[K]:=0
3140 ELSE WKR[K]:=WKR[K]\E[K]\1.570796/FIM[K]
3150 \COS(1.570796/FIM[K]\(FI[N]-
3155 FIN[K]));
3200 MU[1]:=0;
3210 UA[1]:=0;
3250 P[1]:=0;
3260 UEM[1]:=0;
3265 Q1[1]:=0;
3270 WKRNAR[1]:=0;
3275 Q2[1]:=0;
3280 WKRNMASS[1]:=0;
3300 IF FI[N] LSS (FIN[K]-FIM[K])
3400 THEN ME[K]:=0
3500 ELSE IF FI[N] GTR (FIN[K]+FIM[K])

```

```

3600         THEN M[K]:=0
3700         ELSE M[K]:=1/|C[Z1]|N(O[K])\F[K]\1.570796/FIM[K]
3800             \COS(1.570796/FIM[K]\(F1[N]-FIM[K]));
3900     CALKA3:=(M[K]+M[K-1])/2\DR;
4000     UACK:=UACK-1]+CALKA3;
4100     IF F1[N] LSS (FIM[K]-FIM[K])
4200         THEN P[K]:=0
4300         ELSE IF F1[N] GTR (FIM[K]+FIM[K])
4400             THEN P[K]:=0
4500             ELSE P[K]:=1/|C[Z1]|N(O[K])\E[K]\1.570796/FIM[K]
4550                 \COS(1.570796/FIM[K]\(F1[N]-FIM[K]))
4600                 \((WKR[N[K]]/2550)*5;
4700     CALKA4:=(P[K]+P[K-1])/2\DR;
4800     UEM[K]:=UEM[K-1]+CALKA4;
4810     Q1[K]:=WKR[N[K]]\F[K];
4820     CALKA7:=(Q1[K]+Q1[K-1])/2\DR;
4830     WKRNAR[K]:=WKRNAR[K-1]+CALKA7;
4840     Q2[K]:=WKR[N[K]]\R[K]*3\F[K];
4850     CALKA8:=(Q2[K]+Q2[K-1])/2\DR/J[C[Z1]]*3;
4960     WKRNMA[K]:=WKRNMA[K-1]+CALKA8;
5000     FND;
5050     WRITE(CJAK33,FORMAT1,N,FIDGR[N],UAC[Z1],UEM[Z1],
5055         WKRNAR[Z1],WKRNMA[Z1]);
5100     A3[N]:=1/FIMAX\UAC[Z1];
5200     A4[N]:=1/FIMAX\UEM[Z1];
5300     CALKA5:=(A3[N]+A3[N-1])/2\DELTA FI;
5400     UASR[N]:=UASR[N-1]+CALKA5;
5500     CALKA6:=(A4[N]+A4[N-1])/2\DELTA FI;
5600     UEMSR[N]:=UEMSR[N-1]+CALKA6;
5610     A5[N]:=1/FIMAX\WKRNAR[Z1];
5620     A6[N]:=1/FIMAX\WKRNMA[Z1];
5630     CALKA9:=(A5[N]+A5[N-1])/2\DELTA FI;
5640     WKRNARSR[N]:=WKRNARSR[N-1]+CALKA9;
5650     CALKA10:=(A6[N]+A6[N-1])/2\DELTA FI;
5660     WKRNMASSR[N]:=WKRNMASSR[N-1]+CALKA10;
5700     END;
5800     WRITE (CJAK33,FORMAT1,UASR[Z2],UEMSR[Z2],WKRNARSR[Z2],
5850         WKRNMASSR[Z2]);
5900     GO TO ET1;
6000     END.

```

P JAK1

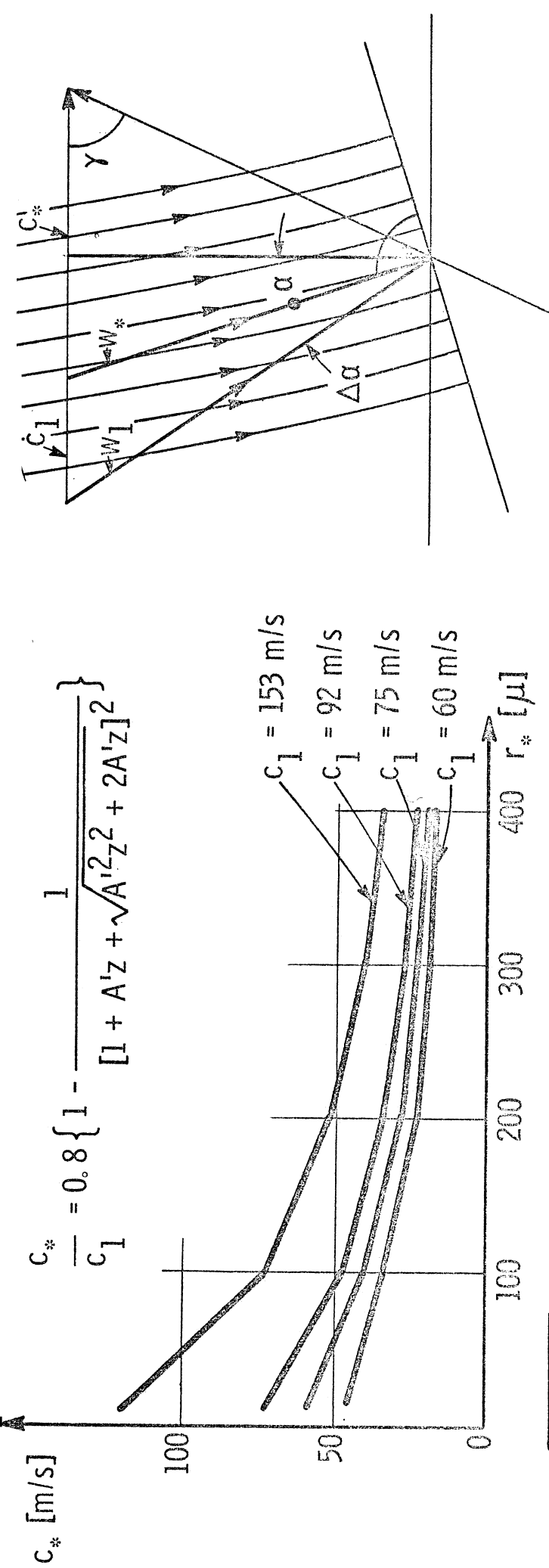
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200	17.1,	0.830,	1.45,
300	68.4,	0.920,	1.515,
400	274,	0.977,	1.540,
500	1095,	0.990,	1.560,
600	2460,	1.0,	1.564,
700	4380,	1.0,	1.566,
800	6840,	1.0,	1.566,
900	684000,	1.0,	1.570,

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9. LIST OF FIGURES

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$$\frac{c_*}{c_1} = 0.8 \left\{ 1 - \frac{1}{[1 + A'z + \sqrt{A'^2 z^2 + 2A'z}]^2} \right\}$$

$$A' = \sqrt{\frac{0.85 \mu \rho}{3 r_*^2 \rho_* c_1}}$$

$$u = 200 \text{ m/s}$$

$$\gamma = 80^\circ$$

$$D_s = 7 \cdot 10^{-3} \text{ m}$$

$$\mu = 1.85 \cdot 10^{-5} \text{ kg/ms}$$

$$\rho = 1.205 \text{ kg/m}^3$$

$$\rho_* = 1000 \text{ kg/m}^3$$

$$z = 0.3 \text{ m}$$

c_1	m/s	60	75	92	153
r^*	μ	10	400	10	400
w_1	m/s	198.5	199.0	200.5	220.0
w_*	m/s	197.5	198.5	197.5	197.0
α	$^\circ$	+ 3°42'	+ 7°06'	- 4°39'	+ 10°55'
$\Delta\alpha$	$^\circ$	3°57'	2°00'	13°45'	6°45'
		13°53'	2°00'	13°45'	20°18'
					214.5
					197.0
					+ 23°32'
					- 0°30'
					7°48'
					31°50'

Fig. 2

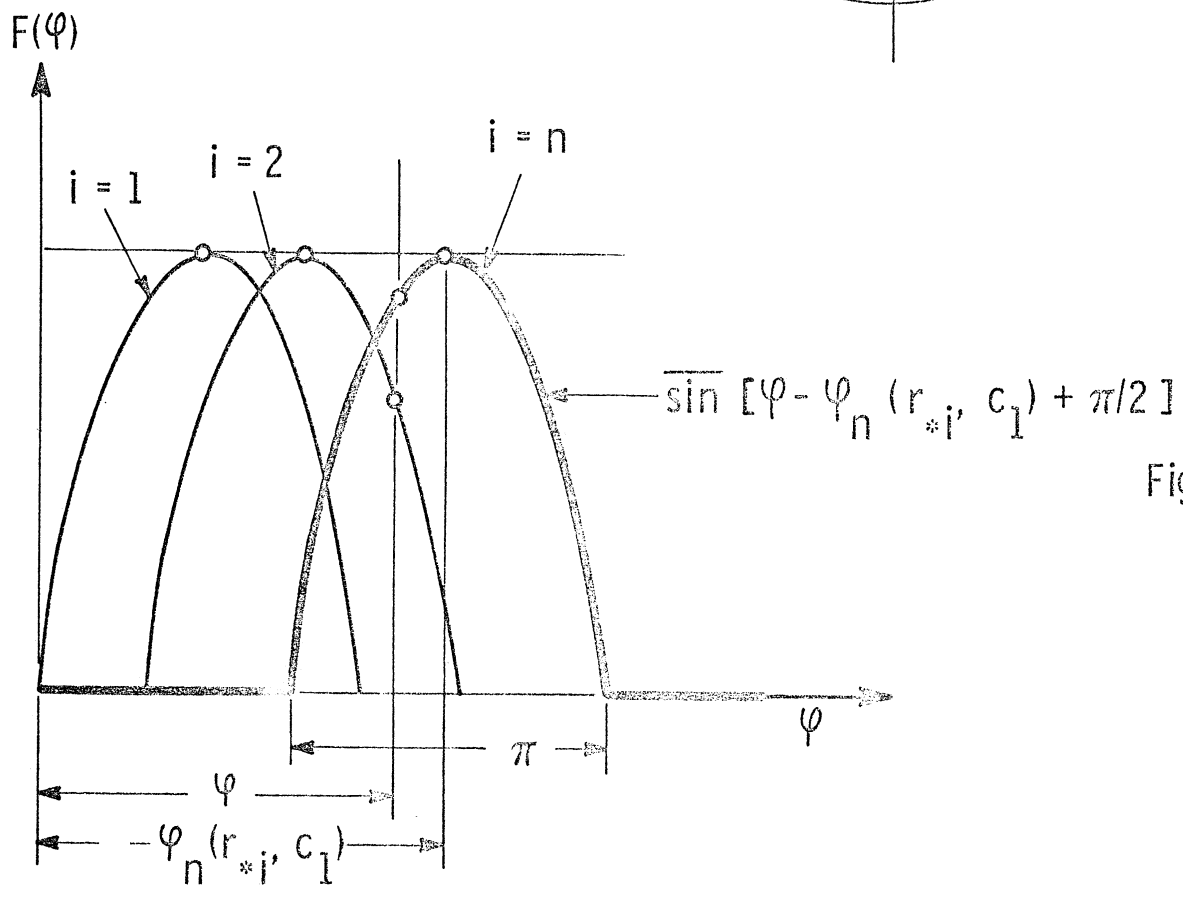
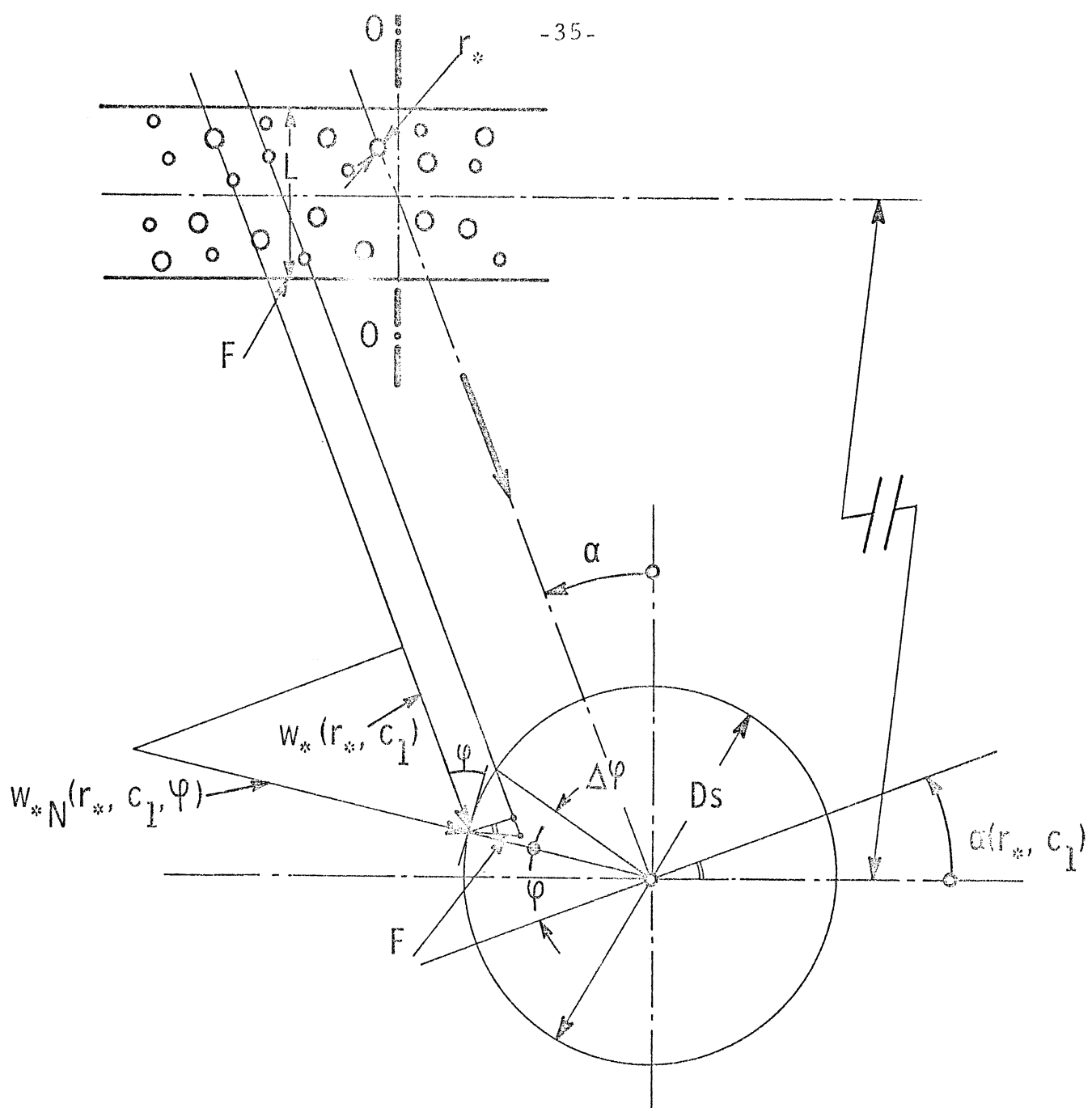
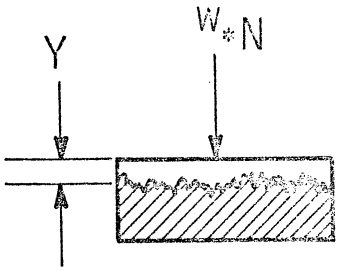
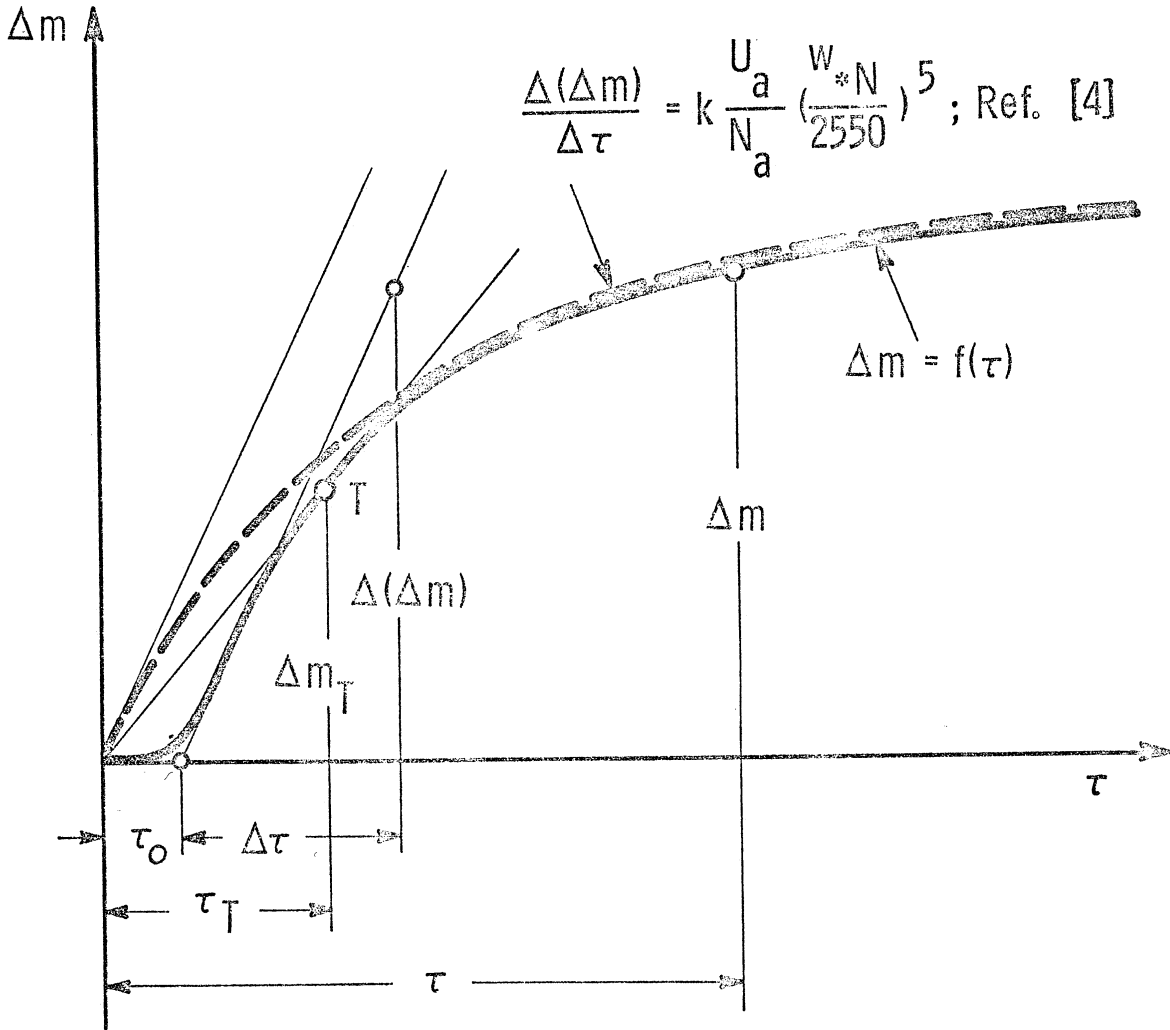


Fig. 3

$$\Delta m = F\rho Y \approx kY$$



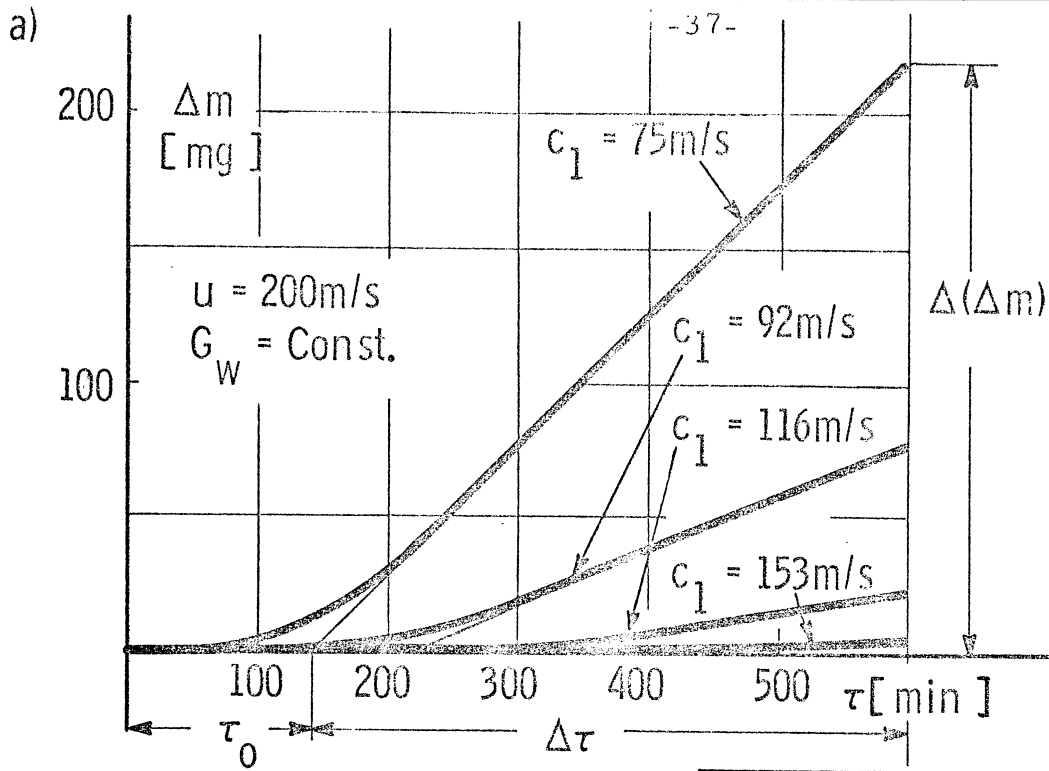
$$\frac{\Delta m}{\tau} = k \frac{U_a}{N_a} \left(\frac{W*N}{2550} \right)^5 \exp \left(-0.25 \frac{\Delta m}{\Delta m_T} \right),$$



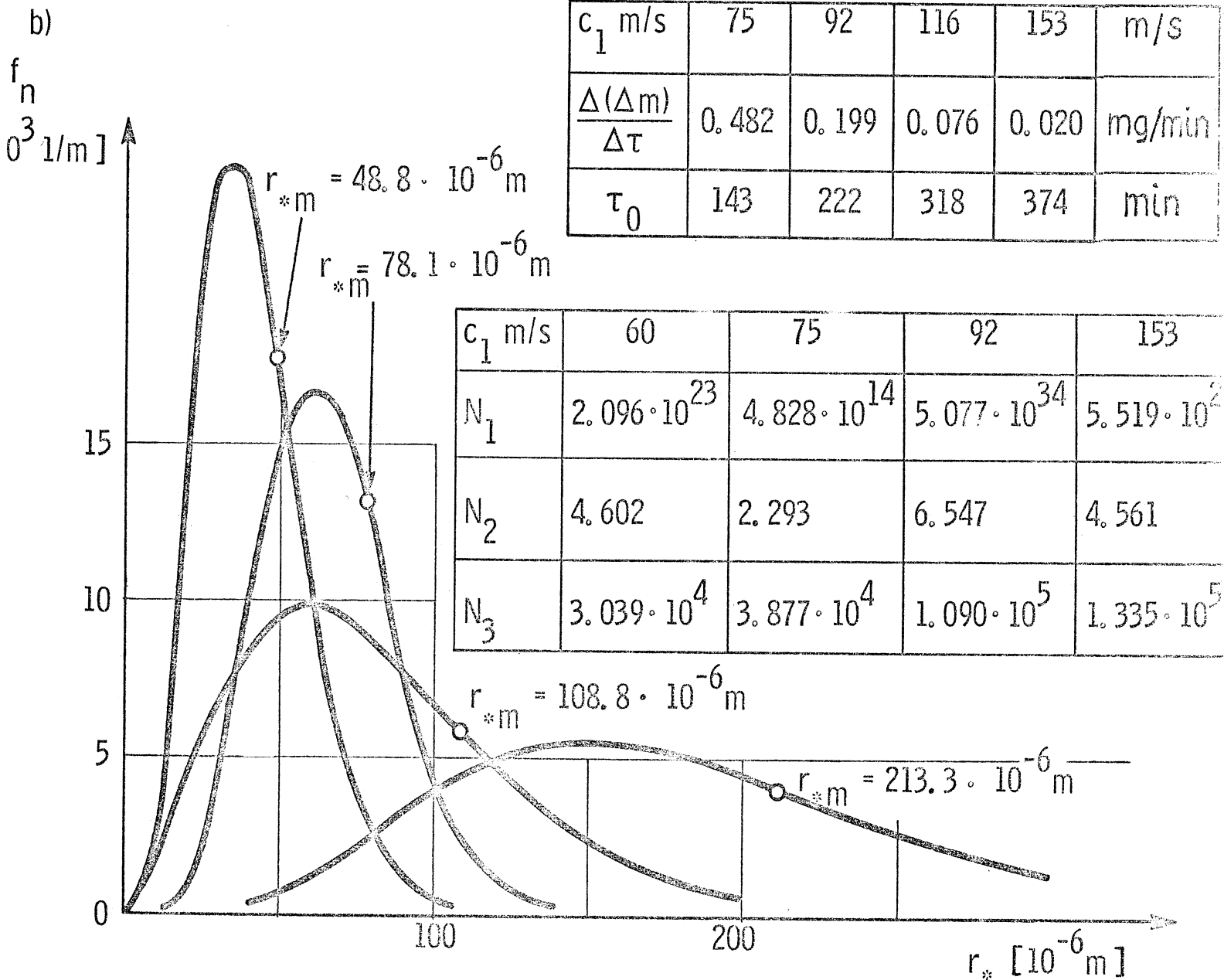
$$\frac{\Delta(\Delta m)}{\Delta \tau} = k \frac{U_a}{N_a} \left(\frac{W*N}{2550} \right)^5 ; \text{Ref. [4]}$$

Fig. 4

Fig. 5



c_1 m/s	75	92	116	153	m/s
$\frac{\Delta(\Delta m)}{\Delta\tau}$	0.482	0.199	0.076	0.020	mg/min
τ_0	143	222	318	374	min



c_1 m/s	60	75	92	153
N_1	$2.096 \cdot 10^{23}$	$4.828 \cdot 10^{14}$	$5.077 \cdot 10^{34}$	$5.519 \cdot 10^2$
N_2	4.602	2.293	6.547	4.561
N_3	$3.039 \cdot 10^4$	$3.877 \cdot 10^4$	$1.090 \cdot 10^5$	$1.335 \cdot 10^5$

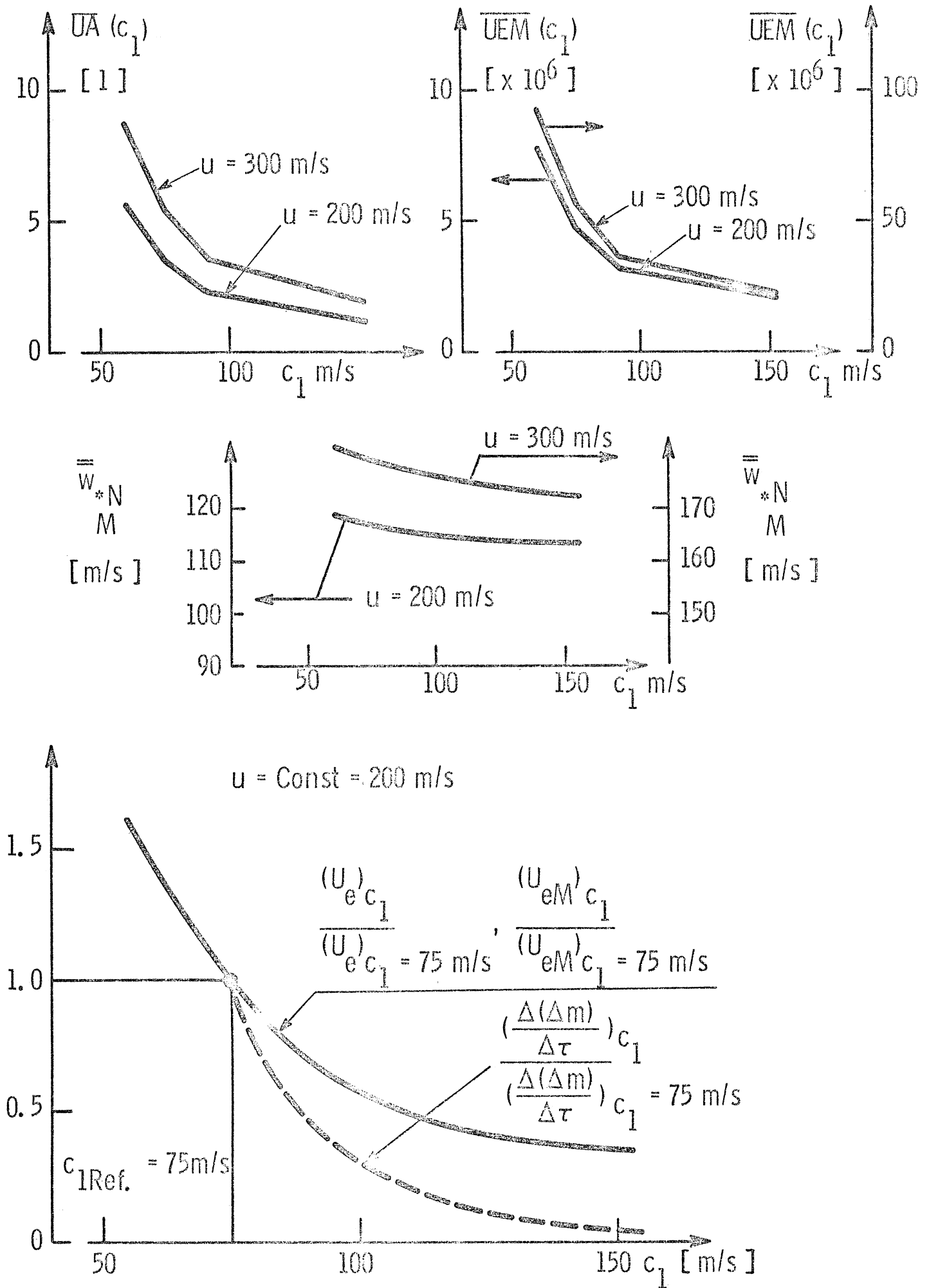
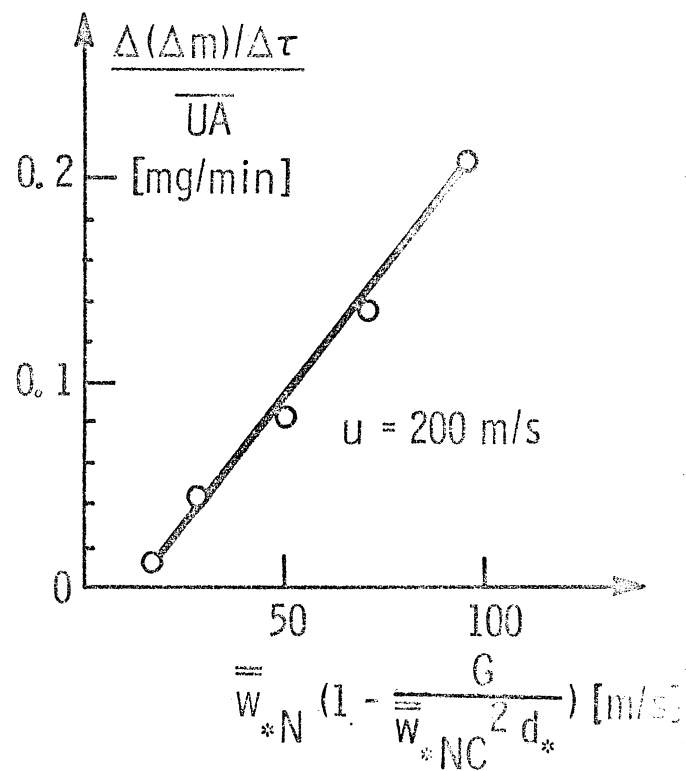
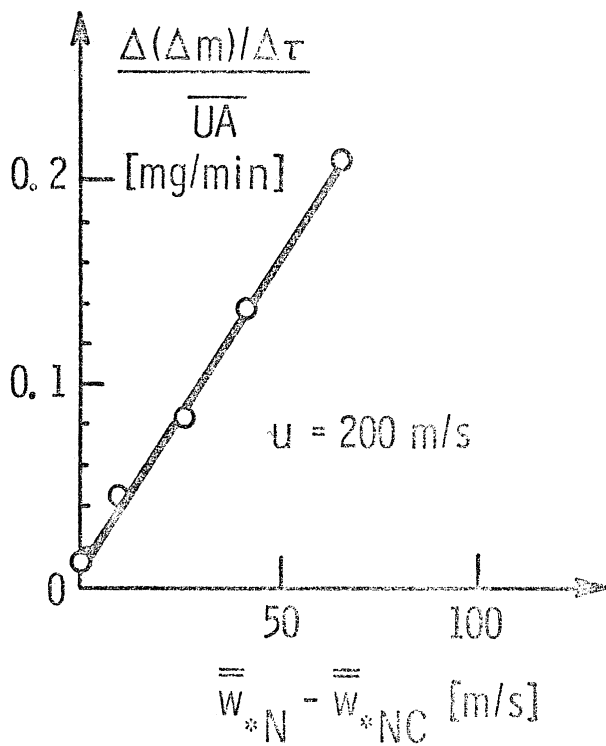
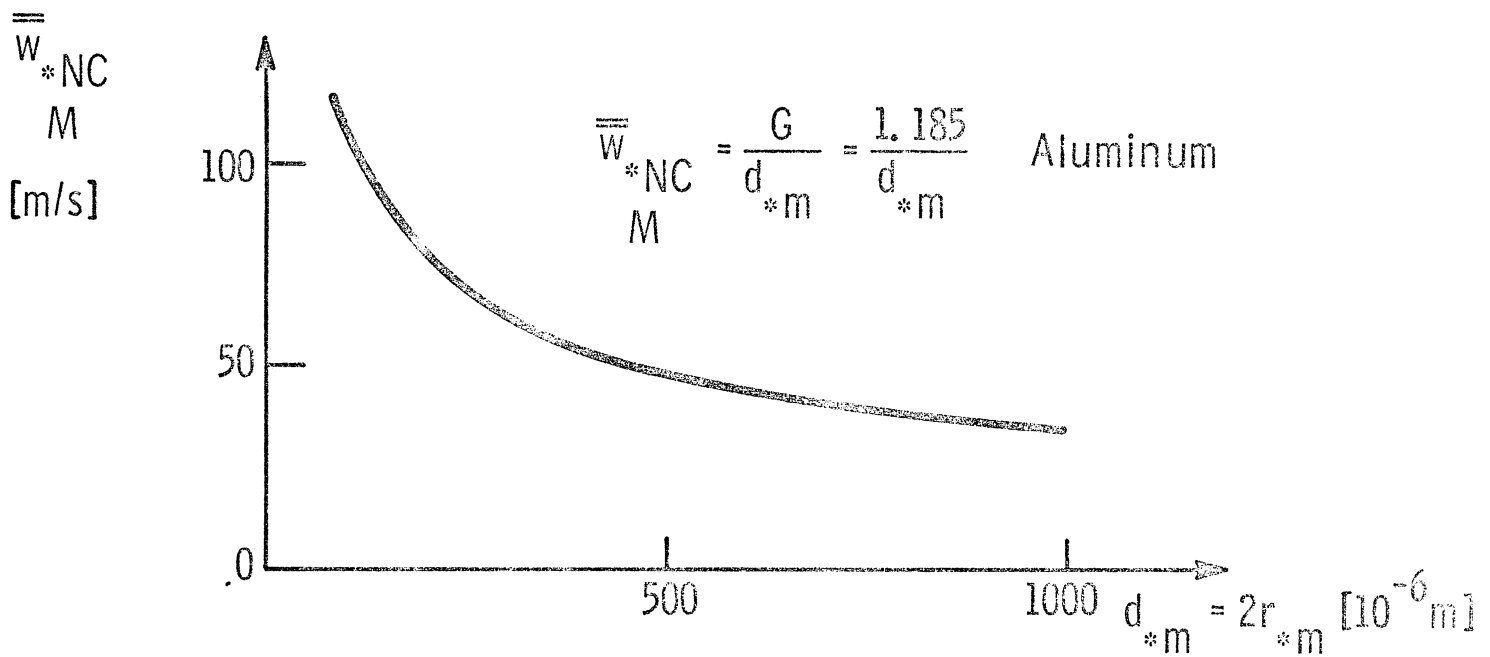
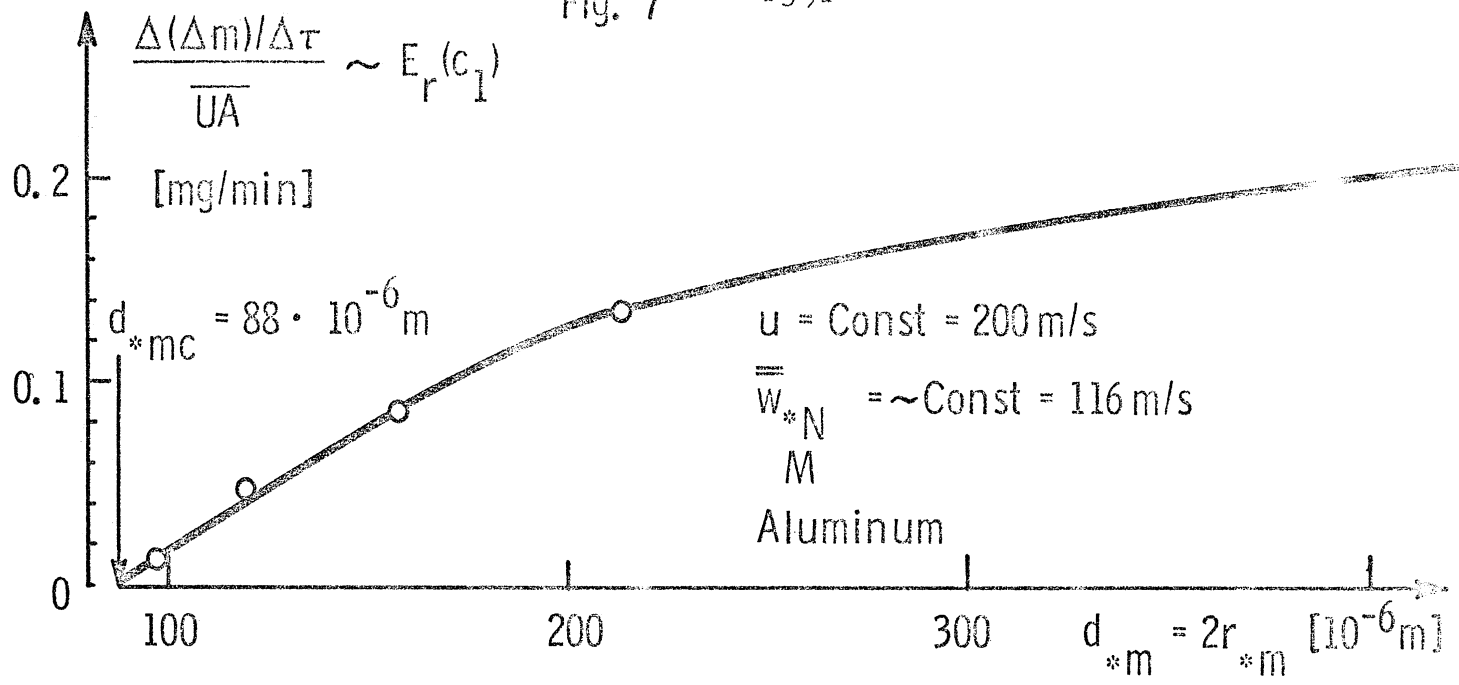


Fig. 6



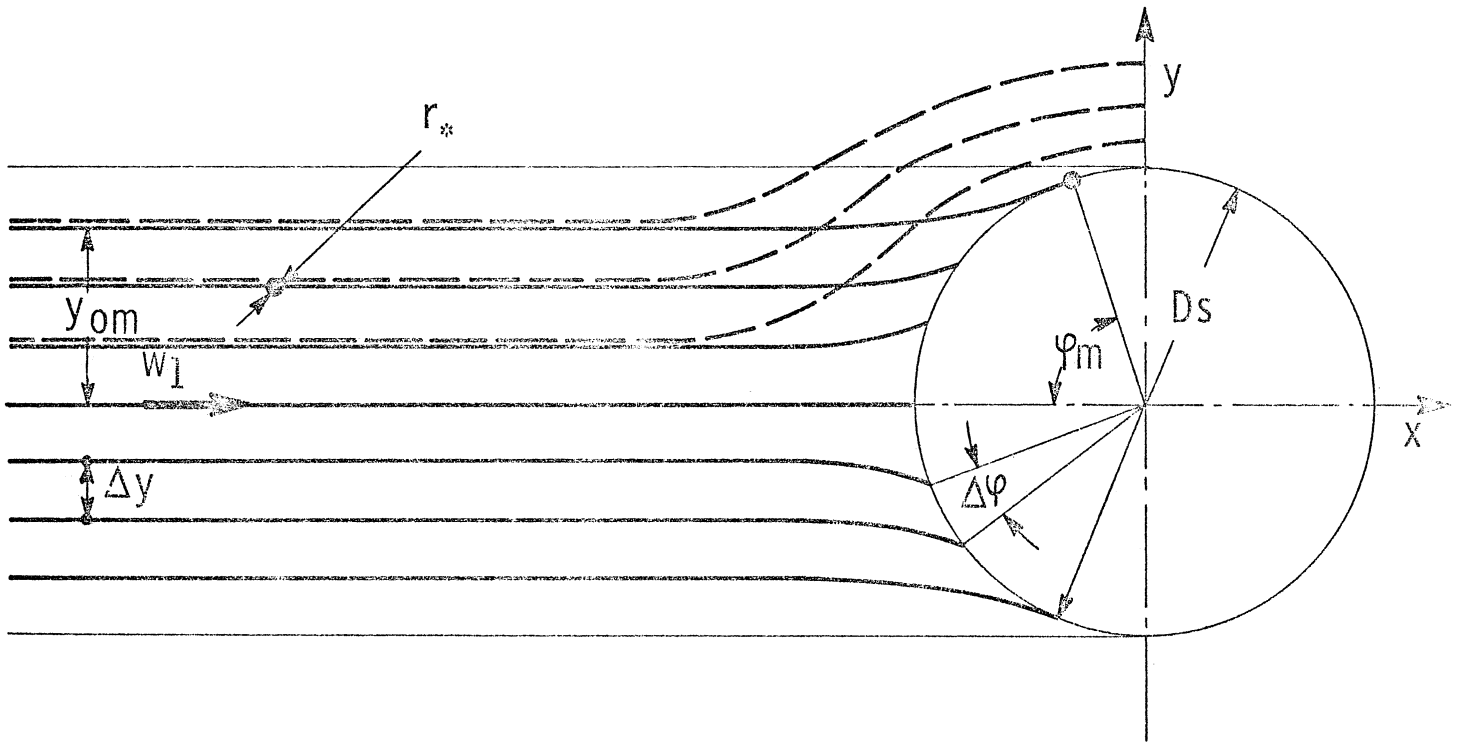


Fig. 8