

THE UNIVERSITY OF MICHIGAN
COLLEGE OF ENGINEERING
Department of Engineering Mechanics

Technical Report

THE STABILITY OF PARALLEL FLOWS OF FLUIDS WITH MEMORIES

Dean T. Mook
W. P. Graebel

ORA Project 06505

under contract with:

DEPARTMENT OF THE NAVY
OFFICE OF NAVAL RESEARCH
CONTRACT NO. Nonr-1224(49), NR NO. 062-342
CHICAGO, ILLINOIS

administered through:

OFFICE OF RESEARCH ADMINISTRATION ANN ARBOR

September 1967

Distribution of this document is unlimited.

TABLE OF CONTENTS

	Page
LIST OF ILLUSTRATIONS	iv
ABSTRACT	v
I. INTRODUCTION	1
II. GOVERNING EQUATIONS	8
III. ASYMPTOTIC SOLUTIONS	20
A. The Determination of the Characteristic Equation for a Navier-Stokes Liquid	20
B. The Determination of the Characteristic Equation for a Viscoelastic Liquid of Type A' or B'	29
C. Solution of the Characteristics Equation	31
D. The Stability of a Viscoelastic Liquid of Type C'	36
BIBLIOGRAPHY	42
APPENDIX	44

LIST OF ILLUSTRATIONS

Table	Page
1. Velocity and Stresses for Primary Flow of Materials A', B', and C'	10

Figure	Page
1. Neutral stability curves for various values of λ .	37
2. Critical Reynolds number versus the parameter R_1 .	38
3. Neutral stability curves and curves of constant c for various values of λ .	39

**The Stability of Parallel Flows
of Fluids with Memories**

**D. T. Mook
Virginia Polytechnic Institute**

and

**W. P. Graebel
The University of Michigan**

Abstract

The equations governing the stability of plane parallel flows are developed for three models of fluids with memories. Asymptotic solutions valid for large Reynolds numbers are obtained and the effect of the memory are shown to be destabilizing. The approach to the problem allows evaluation of how fast a memory must fade to allow evaluation of the stresses in power series in the time interval. An alternate approach to inverting convected derivatives is also presented.

I. Introduction

A recent paper by Chan Man Fong and Walters (1965) considered the stability of parallel flows of two visco-elastic fluids with very short memories. The present work extends their analysis to such fluids with long but still fading memory, and also extends the analysis to a newer model which has been proposed by Goddard and Miller (1966). A discussion of the various convected derivatives is also presented in a manner which allows more ready physical interpretation as well as a quicker way of obtaining forms for convected integrals.

Quasi-linear models of visco-elastic fluids are usually written in the form

$$\underline{\underline{p}} + \sum_{n=1}^N \lambda_n L_t^{(n)} \underline{\underline{p}} = 2\mu_0 \left(\underline{\underline{d}} + \sum_{m=1}^M \tau_m L_t^{(m)} \underline{\underline{d}} \right) \quad (1)$$

where $\underline{\underline{p}}$ is related to the total stress $\underline{\underline{t}}$ by

$$\underline{\underline{t}} = -p \underline{\underline{I}} + \underline{\underline{p}}$$

$\underline{\underline{d}}$ is the rate of deformation tensor, and L_t is a time derivative operator satisfying the principle of material indifference. The constants λ_n and τ_m are related to the stress relaxation times and rate of deformation relaxation times, respectively. Several forms for these operators have been proposed in the past (See Oldroyd (1968) for a review); we present here briefly three of these definitions in a somewhat different manner which facilitates their physical interpretation.

If θ is a convected material reference frame, and $\underline{\underline{y}}^\alpha$ is a set of covariant base vectors defined by $\frac{\partial \underline{\underline{r}}}{\partial \theta^\alpha}$ ($\underline{\underline{r}}$ a position vector) so that they are tangent to the frame (see for example Sokolnikoff (1951),

Chapter 3), then a second order tensor $\underline{\underline{T}}$ can be written as

$$\underline{\underline{T}} = T^{\alpha\beta} \underline{\underline{\gamma}}_{\alpha} \underline{\underline{\gamma}}_{\beta}$$

or

$$\underline{\underline{T}} = T_{\alpha\beta} \underline{\underline{\gamma}}^{\alpha} \underline{\underline{\gamma}}^{\beta},$$

$\underline{\underline{\gamma}}^{\alpha}$ being the contravariant base vectors defined by

$$\underline{\underline{\gamma}}^{\alpha} = e^{\beta\alpha} \underline{\underline{\gamma}}_{\beta} \times \underline{\underline{\gamma}}_{\beta}.$$

The absence of a dot or cross between two vectors indicates the indefinite, or dyadic, product. Oldroyd (1950) proposed two separate definitions for L_t ; denoting time differentiation with material coordinates held fixed by D/Dt , they are

$$L_{tA} \underline{\underline{T}} \equiv \frac{DT_{\alpha\beta}}{Dt} \underline{\underline{\gamma}}^{\alpha} \underline{\underline{\gamma}}^{\beta} \equiv \frac{d_A T_{ij}}{dt} \underline{\underline{g}}^i \underline{\underline{g}}^j$$

for the model he called type A, and

$$L_{tB} \underline{\underline{T}} \equiv \frac{DT^{\alpha\beta}}{Dt} \underline{\underline{\gamma}}_{\alpha} \underline{\underline{\gamma}}_{\beta} \equiv \frac{d_B T^{ij}}{dt} \underline{\underline{g}}_i \underline{\underline{g}}_j$$

for the model he called type B. Latin indices and base vectors here refer to a space fixed reference frame. It is readily found by the normal tensor transformation laws that

$$\begin{aligned} \frac{d_A T_{ij}}{dt} &= \frac{DT_{ij}}{Dt} + (d^m_{.i} + \omega^m_{.i}) T_{mj} \\ &\quad + (d^n_{.j} + \omega^n_{.j}) T_{in}, \\ \frac{d_B T^{ij}}{dt} &= \frac{DT^{ij}}{Dt} - (d^i_{.m} + \omega^i_{.m}) T^{mj} \\ &\quad - (d^j_{.n} + \omega^j_{.n}) T^{in}, \end{aligned}$$

and that

$$\begin{aligned} \frac{DY_\alpha}{Dt} &= (d^\beta_{. \alpha} + \omega^\beta_{. \alpha}) \underline{Y}_\beta, \\ \frac{DY^\alpha}{Dt} &= - (d^\alpha_{. \beta} + \omega^\alpha_{. \beta}) \underline{Y}^\beta, \end{aligned}$$

where $\underline{\omega}$ is the vorticity tensor. Thus Oldroyd's definition of the convected rate is the material rate of those tensor components which an observer would measure with respect to a coordinate system both rotating and deforming with the material; for type A the components are measured with respect to the contravariant base vectors, while for type B the covariant base vectors are used, the base vectors in both cases being both stretched and rotated with the material.

The present form of writing the convected derivative allows ready inversion, for letting

$$W_{ij} = \frac{d_A T_{ij}}{dt},$$

then since

$$W_{\alpha\beta} = \frac{DT_{\alpha\beta}}{Dt},$$

and since the θ^{α} are constant in time for a material particle,

$$T_{\alpha\beta} = \int^t W_{\alpha\beta}(\theta, t') dt'$$

or

$$\begin{aligned} T_{ij} &= \frac{\partial \theta^{\alpha}}{\partial x^i} \frac{\partial \theta^{\beta}}{\partial x^j} \int^t \frac{\partial x'^m}{\partial \theta^{\alpha}} \frac{\partial x'^n}{\partial \theta^{\beta}} W_{mn}(x', t') dt' \\ &= \int^t \frac{\partial x'^m}{\partial x^i} \frac{\partial x'^n}{\partial x^j} W_{mn}(x', t') dt', \end{aligned}$$

and similarly for the type B derivative. These integrals were presented first by Oldroyd (1950); they were used by Walters (1962) in models designated as A' and B' by writing

$$p_{ij} = 2 \int_{-\infty}^t \psi(t-t') \frac{\partial x'^m}{\partial x^i} \frac{\partial x'^n}{\partial x^j} d_{mn}(x', t') dt' \quad (2)$$

and

$$p^{ij} = 2 \int_{-\infty}^t \psi(t-t') \frac{\partial x^i}{\partial x'^m} \frac{\partial x^j}{\partial x'^n} d^{mn}(x', t') dt', \quad (3)$$

respectively, where $\psi(t)$ is a material relaxation function. * When $\psi(t)$ consists of a combination of exponentials and Dirac delta functions,

* (it is frequently more convenient to work with N , the distribution of relaxation times, defined by $\psi(t) = \int_0^{\infty} N(p) \exp(-t/p) dp/p$.)

equations (2) and (3) are exactly equivalent to equation (1); otherwise they are generalizations of equation (1). (For example, when $N = M = 1$, equation (1) is obtained by taking $N(p) = \mu_0 \left[\frac{\tau_1}{\lambda_1} \delta(p) + \left(1 - \frac{\tau_1}{\lambda_1}\right) \delta(p - \lambda_1) \right]$.)

Jaumann (1911) proposed a different definition of L_t which we shall designate as L_{tC} . Introducing new base vectors $\underline{\Gamma}_\alpha$ by

$$\underline{\Gamma}_\alpha = S_{\alpha}^{\beta} \underline{Y}_\beta$$

where \underline{S} satisfies the equation

$$\frac{DS_{\alpha}^{\beta}}{Dt} = -S_{\alpha}^{\delta} d_{\delta}^{\beta}$$

and reduces to the identity matrix as an initial condition, then, denoting the inverse tensor with a minus unity superscript,

$$\underline{\underline{I}} = \tilde{T}^{\alpha\beta} \underline{\Gamma}_\alpha \underline{\Gamma}_\beta, \quad \tilde{T}^{\alpha\beta} = T^{\delta\epsilon} S_{\delta\alpha}^{-1} S_{\epsilon\beta}^{-1}$$

and

$$L_{tC} \underline{\underline{I}} \equiv \frac{D\tilde{T}^{\alpha\beta}}{Dt} \underline{\Gamma}_\alpha \underline{\Gamma}_\beta \equiv \frac{d_c T_{ij}}{dt} g_i g_j$$

as before it is readily found from the transformation laws that

$$\frac{d_c T_{ij}}{dt} = \frac{DT_{ij}}{Dt} + \omega_{.i}^m T_{mj} + \omega_{.j}^n T_{in}$$

and that

$$\frac{D \underline{\Gamma}_\alpha}{Dt} = \omega^{\beta s} S_\alpha^{\cdot s} \underline{y}_\beta.$$

Since by Ricci's theorem and the above $d_t g_{ij}/dt = 0$, raising and lowering of indices commutes with the operation of Jaumann differentiation.

The above results can be put in a simpler appearing form by introducing a further tensor \underline{R} , defined by

$$R_i^{\cdot \alpha} \equiv S_\rho^{-1 \cdot \alpha} \frac{\partial \theta^\rho}{\partial x_i};$$

then

$$\frac{D R_i^{\cdot \alpha}}{Dt} = \omega_{ij} R^{j\alpha} \tag{4}$$

and

$$R_i^{\cdot \alpha} \underline{\Gamma}_\alpha = \underline{g}_i, \text{ or } \underline{\Gamma}_\alpha = R_\alpha^{-1 \cdot i} \underline{g}_i$$

where \underline{R} is equal initially to the identity matrix. Thus \underline{S} is the tensor rotating the material base vectors \underline{y}_α into the material base vectors $\underline{\Gamma}_\alpha$, and \underline{R} is the tensor rotating the material base vectors $\underline{\Gamma}_\alpha$ into the fixed base vectors \underline{g}_i . As has been shown by Goddard and Miller (1965), \underline{R} corresponds to an orthogonal transformation, and hence its inverse and transpose are equivalent. Thus the Jaumann derivative is the material time rate of those tensor components which

an observer would measure with respect to the base vectors $\underline{\Gamma}_\alpha$ rotating locally with the same rate as the vorticity, i. e., moving with the principal axes of \underline{d} . The length of these base vectors changes also, but not directly with the material. Inversion of the derivative again follows readily from the definition; if now

$$Y_{ij} = \frac{dc T_{ij}}{dt},$$

then

$$\tilde{Y}_{\alpha\beta} = Y^{\delta\epsilon} S_{\delta\alpha}^{-1} S_{\epsilon\beta}^{-1} = \frac{D\tilde{T}_{\alpha\beta}}{Dt} = \frac{D}{Dt} (T^{\mu\nu} S_{\mu\alpha}^{-1} S_{\nu\beta}^{-1})$$

and

$$\tilde{T}_{\alpha\beta} = \int^t \tilde{Y}_{\alpha\beta}(\theta, t') dt',$$

$$T_{ij} = \frac{\partial\theta^\delta}{\partial x^i} \frac{\partial\theta^\epsilon}{\partial x^j} S_{\delta\alpha}^\alpha S_{\epsilon\beta}^\beta \int^t \frac{\partial\theta^\mu}{\partial x^m} \frac{\partial\theta^\nu}{\partial x^n} Y^{mn}(x', t') \cdot (S_{\mu\alpha}^{-1} S_{\nu\beta}^{-1})_{t'} dt'$$

$$= R_i'^\alpha R_j'^\beta \int^t (R_{m\alpha} R_{n\beta})_{t'} Y^{mn}(x', t') dt'$$

$$= \int^t R_i'^m R_j'^n Y_{mn}(x', t') dt',$$

the initial conditions on R_{ij} being imposed at time t' . A material of type C' could now be defined by

$$p_{ij} = 2 \int_{-\infty}^t \psi(t-t') R_i^m R_j^n d_{mn}(x', t') dt'. \quad (5)$$

We note that equation (5) is the constitutive equation presented by Goddard and Miller (1966), their integration being presented by other arguments. No simple relation has so far been found relating the various Oldroyd and Jaumann integrals.

II. Governing equations

The solution for steady flow between stationary parallel plates is next presented for materials of type A' , B' , and C' . The stability problem for parallel flows is then formulated for each by superimposing a wavy infinitesimal disturbance on the primary flow and then determining under what conditions this disturbance will grow.

Cartesian coordinates are used, the z -axis and the y -axis being chosen parallel and perpendicular to the plates, respectively. For the steady flow the velocity components are assumed in the form

$$U = V = 0, \quad W = W(y),$$

with $W = 0$ at $y = \pm h$. By inspection the motion is

$$x' = x, \quad y' = y, \quad \text{and} \quad z' = z - W(y)(t-t'),$$

primed coordinates denoting the position of a particle at time t' . The only non-zero component of the rate of deformation is

$$d_{23} = \frac{1}{2} DW,$$

where D represents differentiation with respect to y . The non-zero displacement gradients and rotation components are

$$\frac{\partial x'}{\partial x} = \frac{\partial y'}{\partial y} = \frac{\partial z'}{\partial z} = 1, \quad \frac{\partial z'}{\partial y} = -(t-t')DW,$$

$$R_{11} = 1, \quad R_{22} = R_{33} = \cos \left[\frac{1}{2}(t-t')DW \right],$$

$$\text{and } R_{23} = -R_{32} = -\sin \left[\frac{1}{2}(t-t')DW \right].$$

Substitution of these into equations (2), (3) and (5) and the equations of motion yields the non-zero partial stress components and velocity as shown in Table I, where

$$B_n = \int_0^\infty \tau^n N(\tau) d\tau = J_n^0, \quad (6)$$

and

$$J_n^m = \int_0^\infty \tau^n N(\tau) [1 + (\tau DW)^2]^{-m} d\tau. \quad (7)$$

Hence, fluids A', B', and C' all predict different normal stresses. Normal stress differences are consistent with the sudden expansion or contraction of the stream when some non-Navier-Stokes liquids suddenly emerge from a tube into the atmosphere (the Merrington effect, which would occur in A' and C') as well as with the differences in the shape of the free surface for such different liquids undergoing

	A'	B'	C'
P_{xx}	0	0	0
P_{yy}	$-2B_1(DW)^2$	0	$-2J_1^1(DW)^2$
P_{zz}	0	$2B_1(DW)^2$	$2J_1^1(DW)^2$
P_{yz}	$B_0 DW$	$B_0 DW$	$J_0^1 DW$
P	$\frac{\partial p}{\partial z} z - 2B_1(DW)^2$	$z \frac{\partial p}{\partial z}$	$\frac{\partial p}{\partial z} z - 2J_1^1(DW)^2$
W	$W_0 \left(1 - \frac{y^2}{2h^2}\right)$	$W_0 \left(\frac{1}{2} - \frac{y^2}{2h^2}\right)$	Satisfies $J_0^1 DW = \frac{\partial p}{\partial z} y$
$\frac{\partial p}{\partial z}$	$B_0 D^2 W$	$B_0 D^2 W$	$(J_0^1 - 2J_2^2 DW) D^2 W$

TABLE I
Velocity and Stresses for primary flow of materials A', B', and C'.

Couette flow (the Weissenberg effect, which would also occur in A' and C'). Only C' shows a variable effective viscosity.

In the development of the Orr-Sommerfeld equation (the stability equation for Navier-Stokes liquids), consideration is limited to a disturbance that corresponds to a velocity field which is both temporally and spatially (in \mathbf{z}) periodic. Subject to this limitation, Squire (1933) has shown that it is sufficient to consider a disturbance that corresponds to a two-dimensional velocity field. In the visco-elastic case, only such disturbances will be considered also, although no proof of the sufficiency of this exists for these fluids. (In fact, Listrov (1966) has shown that at least for a Stokesian fluid three dimensional disturbances are less stable than two dimensional disturbances.) Accordingly, the disturbance velocity components are taken in the form

$$u^* = 0, \quad v^* = v(y)E, \quad \text{and} \quad w^* = w(y)E,$$

where $E = \exp iA(z - Ct)$; A is the (real) wave number, and c the (complex) wave speed.

We first derive the stability equation for materials A' and B'. The total displacement is assumed to be the sum of the primary flow displacement and the displacement resulting from the disturbance, namely,

$$x' = x,$$

$$y' = y + f(y, z, t) - f(y', z', t'),$$

and

$$z' = z - [w(y) + k(y, z, t)](t - t') + g(y, z, t) - g(y', z', t').$$

Taking

$$f = \sum_{n=1}^{\infty} f_n(y) E^n$$

and similar expressions for g and k , since

$$\frac{Dx'}{Dt} = \frac{Dy'}{Dt} = \frac{Dz'}{Dt} = 0,$$

the solutions are readily obtained as

$$f_1 = -v/iA(W-C),$$

$$f_{n+1} = (f_n Dv - v Df_n)/iA(W-C),$$

$$g_1 = -(k_1 + \omega)/iA(W-C),$$

$$g_{n+1} = (-k_n + g_n Dv - v Dg_n)/iA(W-C),$$

$$k_1 = vDW/iA(W-C),$$

$$k_{n+1} = (k_n Dv - v Dk_n)/iA(W-C).$$

Expanding all quantities about values at time t , it follows that to the lowest order

$$y' = y - \frac{vE(1-F)}{iA(W-C)} \quad (8)$$

and

$$z' = z - (t-t')W + \frac{vDWEF(t-t')}{iA(W-C)} + \left[\frac{vDW}{A^2(W-C)^2} + \frac{Dv}{A^2(W-C)} \right] E(1-F), \quad (9)$$

where $F = \exp iA(C-W)(t-t')$.

If consideration is restricted to a short time interval, then equations (8) and (9) can be approximated by expanding F in a power series in $(t-t')$ and retaining only the first order terms. Hence,

$$y' = y - v(t-t')E$$

and

$$z' = z - \left(W - \frac{Dv}{iA} E \right) (t-t').$$

These are the expressions used by Walters (1962). The more general forms given by equations (8) and (9) will, however, be used here.

We note also that equations (8) and (9) are well-behaved at the critical point where W and C have the same value. Applying the limiting process $W \rightarrow C$ results in

and
$$\lim_{W \rightarrow C} (y' - y) = -vE(t-t')$$

$$\lim_{W \rightarrow C} [z' - z + W(t-t')] = \frac{Dv}{iA} E(t-t') - \frac{1}{2} v E D W (t-t')^2,$$

in agreement with the limit as t approaches t' to the order of the linear terms in $t-t'$.

The non-zero rate of deformation components at time t are

$$d_{yy} = -d_{zz} = E D v$$

and

$$d_{yz} = \frac{1}{2} \left[D W + (iA v - \frac{D^2 v}{iA}) E \right].$$

To the same approximation used for the displacements, the non-zero rate of deformation components at time t' are

$$d_{yy} = -d_{zz} = D v E F$$

and

$$d_{yz} = \frac{1}{2} \left[D W + (iA v - \frac{D^2 v}{iA}) E F - \frac{v D^2 W E (1-F)}{iA(W-C)} \right].$$

Making use of the continuity equation, the non-zero derivatives of the displacements can thus be written as

$$\frac{\partial y'}{\partial y} = 1 - \left(Dv - \frac{vDW}{W-C} \right) \frac{E(1-F)}{\iota A(W-C)},$$

$$\frac{\partial y'}{\partial z} = - \frac{vE(1-F)}{W-C},$$

$$\frac{\partial z'}{\partial z} = 1 + \frac{vDWE(t-t')}{W-C} + \left(Dv - \frac{vDW}{(W-C)} \right) (t-t'),$$

and

$$\begin{aligned} \frac{\partial z'}{\partial y} = & - (t-t')DW + (t-t') \frac{DWEF}{\iota A(W-C)} \left(Dv - \frac{vDW}{W-C} \right) + \frac{(t-t')E}{\iota A(W-C)} \left(vD^2W + DvDW - \frac{v(DW)^2}{W-C} \right) - \frac{E(1-F)}{A^2(W-C)} \left[D^2v + \frac{2v(DW)^2}{(W-C)^2} - \frac{2DvDW + vD^2W}{W-C} \right]. \end{aligned}$$

Combining these results with the constitutive equations for A' (equation (2)) leads to the disturbance stresses

$$\begin{aligned} \bar{P}_{zz} = & -2K_0^1 Dv - 2\iota ADWK_1^1 v, \\ \bar{P}_{yz} = & -\frac{K_0^1}{\iota A} D^2v + 2K_1^2 DW Dv \\ & + [\iota AK_0^1 - K_1^1 D^2W + 2\iota A(K_2^2 + K_2^1)] v, \end{aligned}$$

and

$$\begin{aligned} \bar{P}_{yy} = & \frac{2DW}{iA} (K_1^1 + K_1^2) D^2 v + 2[K_0^1 \\ & + 2(DW)^2 iA(W-C)(K_3^2 + K_3^3)] Dv \\ & + 2[DWD^2 W (2K_2^1 + K_2^2) \\ & - 2(DW)^3 iA(K_3^2 + K_3^3) - iADWK_1^2] v. \end{aligned}$$

The functions $K_n^m(y)$, are generalizations of Walters' constants and are defined by

$$K_n^m = \int_0^\infty \tau^n N(\tau) [1 + iA(W-C)\tau]^m d\tau. \quad (10)$$

Upon considering the equations of motion, one obtains, after subtracting the equations satisfied by the velocity components of the primary flow and eliminating the pressure by cross-differentiation,

$$\begin{aligned} \rho \left[(W-C)(D^2 - A^2) - D^2 W \right] v \\ = iAD(\bar{P}_{yy} - \bar{P}_{zz}) - (D^2 + A^2) \bar{P}_{yz}. \quad (11) \end{aligned}$$

In order to write equation (11) in terms of non-dimensional variables, the following dimensionless functions of y are introduced:

$$\begin{aligned} P_0 = K_0^1 / \rho W_0 h, \quad P_1 = K_1^1 / \rho h^2, \\ P_2 = K_2^2 W_0 / \rho h^3, \quad P_3 = K_3^3 W_0^3 / \rho h^5. \end{aligned}$$

Putting these and the previous expressions for \bar{P}_{yy} , \bar{P}_{zz} and \bar{P}_{yz} into equation (11) results in

$$\begin{aligned}
 & i[(U-c)(D^2-\alpha^2)v - vD^2U] \\
 & = \frac{P_0}{\alpha} (D^2-\alpha^2)^2 v - \alpha^2 P_2 (U \\
 & - c)(2DU)D^3v - \alpha(U-c)[3P_2 D^3U \\
 & - 2\alpha^2(U-c)P_3]D^2v - \alpha(U \\
 & - c)\{4\alpha^2 DU[(DU)^2 - D^2U(U \\
 & - c)]P_3 - (2\alpha^2 DU - D^3U)P_2\}Dv \\
 & + \{[4DU(\alpha^2 DU + D^3U) + \alpha^2(U-c)D^3U \\
 & + 3(D^2U)^2]\alpha P_2 - [\alpha^2(U-c)^2 \\
 & + 3(U-c)D^2U - 2(DU)^2]2\alpha^3(DU)^2 P_3 \\
 & + iD^4U P_1\}v, \quad (12)
 \end{aligned}$$

where α is the non-dimensional wave number Ah , c is the non-dimensional wave speed C/W_0 , U is now the non-dimensional primary flow velocity, W/W_0 , and D represents differentiation with respect to the non-dimensional y . For a Navier-Stokes liquid all the P_n are zero except P_0 , and equation (12) would then reduce to the Orr-Sommerfeld equation.

Walters (1962) has suggested that for some viscoelastic liquids, called "slightly viscoelastic," it is reasonable to expect the $\psi(\tau)$ to

vanish rapidly as τ increases. This is the justification he gave for using the approximate forms for equations (8) and (9). This is equivalent to replacing the upper limit in the expressions for the P_n by a finite limit (say T) which may even be quite small. Assuming $|iA(W-C)| \ll T^{-2}$ everywhere in the flow field and neglecting all B_n for $n \geq 2$, equation (12) reduces to

$$\begin{aligned} & i[(U-c)(D^2 - \alpha^2) - D^2U]v \\ & = [1 - i\alpha R_1(U-c)](\alpha R)^{-1}(D^2 - \alpha^2)^2 v \\ & + iR_1 R^{-1} v D^4 U \end{aligned}$$

where

$$R = \rho W_0 h / B_0$$

and

$$R_n = B_n (W_0)^n / B_0 h^n$$

which is the equation given by Walters.

Two remarks are appropriate. Equation (12) was derived without specifying the form of $W(y)$ and hence is not restricted to the primary flow here considered. Also, if equation (3) (the constitutive equation for material B') had been used in place of equation (2) (the constitutive equation for material A'), the resulting equation for v would have been exactly equation (12) - a perhaps surprising result, considering the primary flows involving the two constitutive equations give quite different normal stress results.

The stability equation for the C' material is developed in much the same way. For the perturbation velocities, the solution to equation (4) is found to be, to the order of the linearization,

$$R_{yy} = R_{zz} = \cos\left[\frac{1}{2}(t-t')DW\right] + EH \sin\left[\frac{1}{2}(t-t')DW\right],$$

$$R_{yz} = -R_{zy} = -\sin\left[\frac{1}{2}(t-t')DW\right] + EH \cos\left[\frac{1}{2}(t-t')DW\right],$$

where

$$H = \frac{(1-F)}{2iA(W-C)} \left\{ v \left(iA - \frac{D^2W}{iA(W-C)} \right) + \frac{D^2v}{iA} \right\} + \frac{(t-t')v D^2W}{2iA(W-C)}.$$

From equation (5), the disturbance stresses are

$$\bar{p}_{22} = -\bar{p}_{33} = v DW \left\{ -2iAL_1^1 + \left[iA - \frac{D^2W}{iA(W-C)} \right] (J_0^1 - L_0^1) + D^2W (J_1^2 + \frac{1}{2} J_1^1) \right\} \left[iA(W-C) \right]^{-1} + 2Dv \left[L_0^1 + iA(W-C)L_1^1 \right] + D^2v \frac{DW}{A^2(W-C)} (L_0^1 - J_0^1), \quad (13)$$

$$\bar{p}_{23} = v \left\{ iAL_0^1 + [-A^2(W-C) + D^2W]L_1^1 + \left[iA(W-C) \right]^{-1} \left[(L_0^1 - J_0^1)D^2W + \left(iA - \frac{D^2W}{iA(W-C)} \right) (J_1^1 - L_1^1) (DW)^2 + 2J_2^2 (DW)^2 D^2W \right] \right\}$$

$$\begin{aligned}
& + 2DWL_1^1 Dv + \frac{D^2 v}{cA} \left\{ - \left[L_0^1 \right. \right. \\
& \left. \left. + cA(W-C)L_1^1 \right] + \frac{(DW)^2}{cA(W-C)} (J_1^1 - L_1^1) \right\} \quad (14)
\end{aligned}$$

where

$$\begin{aligned}
L_n^m = \int_0^\infty \tau^n N(\tau) \left\{ \left[1 + cA(W-C)\tau \right]^2 \right. \\
\left. + (\tau DW)^2 \right\}^{-m} d\tau. \quad (15)
\end{aligned}$$

Because of the complexity of these stress terms we do not write out the stability equation here, but consider the appropriate approximation in the next section.

III. Asymptotic Solutions

An approximate solution to the stability equation (12) for the primary flow $U = 1 - y^2$ is next obtained. The determination of the characteristics equation for a Navier-Stokes liquid is first briefly discussed in Part A, the procedure used being that presented by Graebel (1966) with only slight modifications. The counterpart characteristics equation for viscoelastic liquids A' and B' is next presented in Part B, and the procedure used to actually solve this characteristic equation and to determine the points on the neutral stability curve is presented in Part C. Viscoelastic liquid C' is discussed in Part D.

A. The Determination of the Characteristic Equation for a Navier-Stokes Liquid

It is anticipated that both c and $1/\alpha R$ will be small for the case of interest, which suggests a solution in terms of matched asymptotic expansions. The flow region is first divided into an inner and an outer region. The inner region is a strip that includes the rigid bottom boundary at $y = -1$ and the "critical point" at $y = y_c$,

where $U(y_c) = c$. The outer region extends from the inner region to the centerline between the plates at $y = 0$ which, as a result of the symmetry of the geometry and equations, serves as the other boundary. It is assumed that the two regions overlap.

The procedure is to obtain a solution valid in the inner region and a solution valid in the outer region and then to merge the two solutions. The inner solution is made to satisfy the boundary conditions at $y = -1$ and the outer solution the conditions at $y = 0$. The merging then produces a characteristic equation which gives α as a function of R and c , the plot of α versus R for $c_1 = 0$ being the neutral stability curve.

In the inner region v is obtained by introducing the change in variable (coordinate stretching)

$$\eta = (y - y_c) / \mu = z / \mu$$

and by putting

$$v(y) = \chi(\mu, \eta) = \chi^{(0)}(\eta) + \mu \chi^{(1)}(\eta) + \dots;$$

where μ is a function of αR , expected to be small, but unknown at this point.

Substitution of the above into equation (11) with all of the R_n for $n \geq 1$ set equal to zero suggests that the proper choice for μ is $\mu = (\alpha R)^{-1/3}$, and equation (12) becomes

$$\begin{aligned} \frac{d^4 \chi^{(0)}}{d\eta^4} - i\eta DU_c \frac{d^2 \chi^{(0)}}{d\eta^2} + \mu \left[2i \chi^{(0)} \right. \\ \left. + i\eta^2 \frac{d^2 \chi^{(0)}}{d\eta^2} + \frac{d^4 \chi^{(1)}}{d\eta^4} \right. \\ \left. - i\eta DU_c \frac{d^2 \chi^{(1)}}{d\eta^2} \right] + \dots = 0. \end{aligned}$$

$\chi^{(0)}$ and $\chi^{(1)}$ are the solutions of

$$\frac{d^4 \chi^{(0)}}{d\eta^4} - i\eta DU_c \frac{d^2 \chi^{(0)}}{d\eta^2} = 0$$

and

$$\frac{d^4 \chi^{(1)}}{d\eta^4} - i\eta DU_c \frac{d^2 \chi^{(1)}}{d\eta^2} = i \left(\eta^2 \frac{d^2 \chi^{(0)}}{d\eta^2} + 2\chi^{(0)} \right).$$

Hence,

$$\chi_1^{(0)} = \eta,$$

$$\chi_2^{(0)} = 1,$$

$$\chi_3^{(0)} = \int_{-\infty}^{\eta} d\eta \int_{-\infty}^{\eta} d\eta h_1 [i\eta (DU_c)^{1/3}],$$

and

$$\chi_4^{(0)} = \int_{-\infty}^{\eta} d\eta \int_{-\infty}^{\eta} d\eta h_2 [i\eta (DU_c)^{1/3}],$$

where

$$h_1[x] = \frac{2}{3} x^{1/2} H_{1/3}^{(1)} \left[\frac{2}{3} x^{3/2} \right],$$

$$h_2[x] = \frac{2}{3} x^{1/2} H_{1/3}^{(2)} \left[\frac{2}{3} x^{3/2} \right];$$

$H^{(1)}$ and $H^{(2)}$ are Hankel functions of order one-third. Since h_2 increases exponentially with large positive η at a much faster rate than does the outer solution, $\chi_4^{(0)}$ cannot be merged with the outer solution and is therefore discarded. For $\chi^{(1)}$ we have

$$\chi_1^{(1)} = -\eta^2 / DU_c.$$

Graebel (1966) gives the solutions for $\chi_2^{(1)}$ and $\chi_3^{(1)}$ as

$$\chi_2^{(1)} = 2(i DU_c)^{-2/3} \int_0^\eta d\eta \int_0^\eta d\eta \int_0^\infty ds.$$

$$\cdot \exp\{-i[(i DU_c)^{1/3} \eta s + \frac{1}{3} s^3]\}$$

and

$$\chi_3^{(1)} = 2i \left[\frac{7}{10(DU_c)^2} \frac{d^3 \chi_3^{(0)}}{d\eta^3} - \frac{9}{10(DU_c)^2} \frac{d^2 \chi_3^{(0)}}{d\eta^2} + \frac{i\eta}{2DU_c} \chi_3^{(0)} \right] + \gamma \chi_2^{(1)},$$

where

$$\gamma = 0.67830 (DU_c)^{-1} - 0.39099 + i(0.39160 (DU_c)^{-1} - 0.67896).$$

For large η ,

$$\chi_2^{(1)} \sim \frac{-2\eta}{DU_c} \ln \eta + \frac{2i}{3\eta^2 (DU_c)^2},$$

where

$$-\frac{7}{6} \pi < \arg \eta < \frac{\pi}{6}.$$

In the outer region v is obtained by introducing the expansion

$$v(\mu, y) = \epsilon_0(\mu) v^{(0)}(y) + \epsilon_1(\mu) v^{(1)}(y) + \dots$$

Substituting this into equation (12) gives

$$(U - c)(D^2 - \alpha^2)v^{(0)} - v^{(0)}D^2U = 0,$$

and $v^{(0)} = v^{(n)}$ for all n such that $O(\epsilon_n(\mu)) > O(\mu^3)$. The Tollmien solutions (1936) are

$$v_1^{(0)}(y) = F_1(z) = \sum_{n=0}^{\infty} A_n z^{n+1},$$

and

$$v_2^{(0)}(y) = F_2(z) = F_1 \ln z + \sum_{n=0}^{\infty} B_n z^n,$$

where

$$A_0 = 1, \quad A_1 = -(DU_c)^{-1}, \quad A_2 = \frac{1}{6} \alpha^2,$$

$$A_{n+3} = \frac{(n+1)(n+4)A_{n+2} + \alpha^2 DU_c A_{n+1} - \alpha^2 A_n}{(n+3)(n+4)DU_c},$$

and

$$B_0 = -\frac{1}{2} DU_c, \quad B_1 = 0, \quad B_2 = \frac{2}{DU_c} - \frac{1}{4} \alpha^2 DU_c, \text{ etc.}$$

In the inner region Graebel used

$$v(y) \approx \chi^{(0)}(\eta) = C_1 \eta + C_2 + C_3 \chi_3^{(0)},$$

where the C_i are arbitrary constants. For the inner and outer solutions to merge for large positive η the proper choice is $1/\mu$ for ϵ_0 , $\ln \mu$ for ϵ_1 , and 1 for ϵ_2 . In the outer region v is given by

$$v(y) = \frac{1}{\mu} C_1 F_1 + \ln \mu \left(\frac{2C_2 F_1}{DU_c} \right) - \frac{2C_2}{DU_c} F_2 + A F_1 + O(\mu \ln \mu),$$

where A is a constant that cannot be determined until higher order terms are considered. $\chi_3^{(0)}$ decreases exponentially with large positive η and, hence, no terms are needed to merge with it.

The boundary conditions are

$$v = Dv = 0 \quad \text{at} \quad y = -1$$

and

$$Dv = D^3 v = 0 \quad \text{at} \quad y = 0$$

If v is an even function of y , or

$$v = D^2 v = 0 \quad \text{at} \quad y = 0$$

if v is an odd function of y .

The Orr-Sommerfeld equation allows separation into even and odd parts when U is an even function of y ; since v_0 in the outer region satisfies a second order equation

which is even in y , satisfaction of only one boundary condition at $y = 0$ is sufficient. Using symmetrical disturbances since they are less stable than asymmetrical disturbances, the boundary conditions give

$$\frac{1}{\mu} C_1 DF_1(z_0) - \frac{2C_2}{DU_c} DF_2(z_0) = 0,$$

$$C_1 \eta + C_2 + C_3 \chi_3^{(0)}(\eta_1) = 0,$$

and

$$C_1 + C_3 \frac{d\chi_3^{(0)}(\eta_1)}{d\eta} = 0$$

to the lowest non-trivial order in μ , where $\eta_1 = -(1 + y_c)/\mu = z_1/\mu$ and $z_0 = -y_c$. For the existence of a non-trivial solution,

$$\frac{1}{z_1} \left[\frac{DU_c DF_1(z_0)}{2DF_2(z_0)} + 1 \right] = \frac{\chi_3^{(0)}(\eta_1)}{\eta_1 \frac{d\chi_3^{(0)}(\eta_1)}{d\eta}}$$

which is the characteristic equation obtained by Graebel. The plot of α vs R for $c_i = 0$ obtained from this, however, does not contain the characteristic loop of a neutral stability curve, but gives only the lower branch of the curve. This shortcoming is explained by considering the magnitudes of the $\chi^{(1)}$ terms in the inner expansion, which in turn requires some knowledge of η_1 . With Lin's (1945) results ($\eta_1 \approx 2.5$, $\mu \approx 1/20$) as a guide it appears that $\chi^{(0)}$ does not adequately describe the solution near $y = -1$, for $\text{Im}(\mu \chi_2^{(1)}) \approx 0.3$ as compared with the $\text{Im} \chi_2^{(0)} = 0$. The real parts of $\mu \chi_2^{(1)}$ and

$\mu \chi_2^{(1)}$ are however small compared to $\chi_1^{(0)}$ and $\chi_2^{(0)}$ respectively, and $\mu \chi_3^{(1)}$ is small compared to $\chi_3^{(0)}$. Thus by including the $\chi^{(1)}$ terms the imaginary part of χ_2 is altered significantly, this additional term apparently being responsible for generating the loop in the stability curve. It seems then that a better choice for the solution in the inner region is

$$v(y) = C_1 \left(\eta - \frac{\mu}{DU_c} \eta^2 \right) + C_2 (1 + \mu \chi_2^{(1)}) + C_3 \chi_3^{(0)}. \quad (16)$$

For merging equation (16) with the outer solution, again the proper choice is $1/\mu$ for ϵ_0 , $\ln \mu$ for ϵ_1 , and 1 for ϵ_2 , as before. Then $v(y)$ in the outer region is given by

$$v(y) = \frac{C_1 F_1}{\mu} + \ln \mu \left(\frac{2C_2 F_1}{DU_c} \right) - \frac{2C_2}{DU_c} F_2 + O(\mu)$$

Substituting the appropriate boundary conditions leads to

$$C_1 \left(\eta_1 - \frac{\mu \eta_1^2}{DU_c} \right) + C_2 (1 + \mu \chi_2^{(1)}(\eta_1)) + C_3 \chi_3^{(0)}(\eta_1) = 0,$$

$$C_1 \left(1 - \frac{2\mu \eta_1}{DU_c} \right) + C_2 \mu \frac{d\chi_2^{(1)}(\eta_1)}{d\eta} + C_3 \frac{d\chi_3^{(0)}(\eta_1)}{d\eta} = 0,$$

$$\frac{1}{\mu} C_1 DF_1(z_0) - \frac{2C_2}{DU_c} DF_2(z_0) = 0,$$

$$P_0 = R^{-1} \left\{ 1 - i\alpha R_1 z DU_c - \left[\frac{1}{2} i\alpha R_1 D^2 U_c + \alpha^2 R_2 (DU_c)^2 \right] z^2 + \dots \right\},$$

$$P_1 = R^{-1} \left\{ R_1 - i\alpha R_2 z DU_c - \left[\frac{1}{2} i\alpha R_2 D^2 U_c + i\alpha^2 R_3 (DU_c)^2 \right] z^2 + \dots \right\},$$

$$P_2 = R^{-1} \left\{ R_2 - 2i\alpha R_3 z DU_c - \left[i\alpha R_3 D^2 U_c + 3\alpha R_4 (DU_c)^2 \right] z^2 + \dots \right\},$$

and

$$P_3 = R^{-1} \left\{ R_4 - 3i\alpha R_5 z DU_c - \left[\frac{3}{2} i\alpha R_5 D^2 U_c + 6\alpha R_6 (DU_c)^2 \right] z^2 + \dots \right\},$$

where $D^n U_c = D^n U$ evaluated at $y = y_c$. These are valid in the neighborhood of the critical point for any distribution function that vanishes as τ becomes large, specifically for $N(\tau)$ negligible when $\tau > T$, where $|iA(W-C)T| < 1$.

A change to a stretched variable and inner and outer expansions are introduced as in part A. When all these are substituted into equation (12), the stability equation in the inner region is now modified to

and thus to the characteristics equation

$$\frac{\chi_3^{(0)}(\eta_1)}{\eta_1 \frac{d\chi_3^{(0)}(\eta_1)}{d\eta}} = \frac{\frac{1}{2}DU_c \left[1 + \mu \frac{d\chi_2^{(1)}(\eta_1)}{d\eta}\right] DF_1(z_0) + z_1 \left(1 - \frac{z_1}{DU_c}\right) DF_2(z_0)}{z_1 \left[\left[1 - \frac{2\mu\eta_1}{DU_c}\right] DF_2(z_0) + \frac{1}{2}DU_c DF_1(z_0) \frac{d\chi_2^{(1)}(\eta_1)}{d\eta} \right]} \quad (17)$$

The plot of α versus R resulting from this does have a loop and the equation is essentially that used by Lin, although he elected to express the outer solution as an expansion in powers of α^2 . With present computers the expansion in terms of the coordinate rather than α^2 seems to be much simpler and more accurate.

B. The Determination of the Characteristic Equation for a Viscoelastic Liquid of Type A' or B'

For a viscoelastic liquid which does not depart too drastically from a Navier-Stokes liquid the solution of equation (11) can be carried out in a manner analogous to the solution presented in part A. Specifically, it is anticipated that in equation (12) both $|c|$ and $|P_n/\alpha|$ will be small. Hence, the flow region can be divided into an inner and an outer region as previously done.

In the inner region the P_n can be expressed as series in z by expanding the denominators and then integrating term by term. The results are

$$\begin{aligned} & \frac{d^4 \chi^{(0)}}{d\eta^4} - i\eta DU_c \frac{d^2 \chi^{(0)}}{d\eta^2} \\ & - \mu [i\alpha R_1 \eta DU_c \frac{d^4 \chi^{(0)}}{d\eta^4} \\ & - 2i\chi^{(0)} + i\eta^2 \frac{d^2 \chi^{(0)}}{d\eta^2} - \frac{d^4 \chi^{(1)}}{d\eta^4} \\ & + i\eta DU_c \frac{d^2 \chi^{(1)}}{d\eta^2}] + \dots = 0, \end{aligned}$$

and thus

$$(1 - i\epsilon \eta DU_c) \frac{d^4 \chi^{(0)}}{d\eta^4} - i\eta DU_c \frac{d^2 \chi^{(0)}}{d\eta^2} = 0, \quad (18)$$

where $\epsilon = \mu\alpha R_1$. The outer equation is unchanged. Since the term containing R_1 multiplies the fourth derivative, χ_1 and χ_2 remain as in part A. Taking ϵ to be small but still of larger order than μ , χ_3 is approximated by

$$\chi_3^{(0)}(\eta) \approx \phi_0(\eta) + \epsilon \phi_3(\eta), \quad (19)$$

where

$$\frac{d^4 \phi_0}{d\eta^4} - i\eta DU_c \frac{d^2 \phi_0}{d\eta^2} = 0$$

and

$$\frac{d^4 \phi_1}{d\eta^4} - i\eta DU_c \frac{d^2 \phi_1}{d\eta^2} = i\eta DU_c \frac{d^4 \phi_0}{d\eta^4}.$$

Then ϕ_0 is the same as $\chi_3^{(0)}$ in Part A, and

$$\phi_1 = \frac{1}{5} \left(\eta \frac{d^3 \phi_0}{d\eta^3} - 4 \frac{d^2 \phi_0}{d\eta^2} \right).$$

Thus for the solution in the inner region, χ_1 and χ_2 as given in Part A are used, and $\phi_0 + \epsilon \phi_1$ is used for χ_3 .

Because of the additional linearization introduced by equation (19), the present results are limited to small ϵ and serve mainly to indicate a trend. For larger values of ϵ the perturbation scheme used to solve equation (18) may not be adequate. In this case, one would have to resort to an exact solution of equation (18) as presented in the Appendix.

Since the outer solution remains unchanged and χ_3 is not directly involved in the merging, the characteristic equation retains the same form as equation (17), the only difference being χ_3 is now given by equation (19).

C. Solution of the Characteristics Equation

For calculation purposes, it is convenient to introduce the change in variable

$$\xi = \beta \eta$$

where

$$\beta = (DU_c)^{1/3}.$$

Then the left hand side of equation (17) becomes

$$\frac{\chi_3^{(0)}(\eta_1)}{\eta_1 \frac{d\chi_3^{(0)}(\eta_1)}{d\eta_1}} = \frac{\int_{-\infty}^{s_1} dS \int_{-\infty}^S dS h_1(iS) + \lambda \left[s_1 \frac{dh_1(iS_1)}{dS} - 4h_1(iS_1) \right]}{\int_{-\infty}^{s_1} \left\{ \int_{-\infty}^{s_1} dS h_2(iS) + \lambda \left[i s_1^2 h_2(iS_1) - 3 \frac{dh_2(iS_1)}{dS} \right] \right\}} \quad (20)$$

where

$$\lambda = \frac{1}{5} \epsilon \beta^2$$

$$= \frac{(\alpha W_0 D W_c)^{2/3} \int_0^\infty N(\tau) d\tau}{5 (\rho h^4)^{1/3} \left(\int_0^\infty N(\tau) d\tau \right)^{2/3}} \quad (21)$$

The functions h_1 and h_2 are discussed, and tables of h_1 and h_2 and their derivatives are given in Annals (1945). Putting $x = iS$, it follows that

$$\begin{aligned} \operatorname{Re} h_1(iS) = & \sum_{m=0}^{\infty} \left\{ \frac{(-1)^{m+1} B_{2m+1} S^3}{(200)^{2m+1}} \right. \\ & \left. + \frac{\sqrt{3}}{3} \left[\frac{2(-1)^m A_{2m+1} S^2}{(200)^{2m+1}} + \frac{(-1)^{m+1} B_{2m}}{(200)^{2m}} \right] S^{4m+1} \right\}, \end{aligned}$$

and

$$\begin{aligned} \Im_m h_1(i, s) = & \sum_{m=0}^{\infty} \left\{ \frac{(-1)^m s B_{2m}}{(200)^{2m}} \right. \\ & + \frac{\sqrt{3}}{3} \left[\frac{(-1)^{m+1} s^4 B_{2m+1}}{(200)^{2m+1}} \right. \\ & \left. \left. + \frac{2(-1)^{m+1} A_{2m}}{(200)^{2m}} \right] s^{6m} \right\} \end{aligned}$$

where

$$A_0 = 2^{1/3} / \Gamma(1/3), \quad A_m = 200 A_{m-1} / 3m(3m-1),$$

$$B_0 = 2^{1/3} / 3^{2/3} \Gamma(4/3), \quad B_m = 200 B_{m-1} / 3m(3m+1)$$

are also tabulated.

For the integration indicated on the left hand side of equation (20) the above series were integrated term by term for $s \leq 5$. For $s > 5$, the integration was continued numerically using Simpson's rule and the asymptotic expansion of h_1 until there were no noticeable changes in the value of the integral in the seventh significant figure. These asymptotic expansions are

$$\begin{aligned} \operatorname{Re} h_1(i, s) \sim & s s^{-1/4} e^{-\sqrt{2}/3 s^{3/2}} \cos\left(\frac{\sqrt{2}}{3} s^{3/2}\right) \\ & + \frac{13\pi}{24} \Big|, \end{aligned}$$

and

$$\operatorname{Im} h_1(iS) \sim -S S^{-1/4} e^{-\frac{\sqrt{3}}{3} S^{3/2}} \sin\left(\frac{\sqrt{3}}{3} S^{3/2}\right) + \frac{13\pi}{24}$$

where $S = (2\sqrt{3})^{2/3} / \sqrt{\pi}$ and $-\frac{2\pi}{6} < \arg S < \frac{5\pi}{6}$.
The results of the integration are

$$\int_{\infty}^S dS \operatorname{Re} h_1(iS) = 0.504361$$

$$+ \sum_{m=0}^{\infty} \left\{ \frac{(-1)^{m+1} S^3 B_{2m+1}}{(6m+5)(200)^{2m+1}} + \frac{\sqrt{3}}{3} \left[\frac{2(-1)^m A_{2m+1} S^2}{(6m+4)(200)^{2m+1}} + \frac{(-1)^{m+1} B_{2m}}{(6m+2)(200)^{2m}} \right] \right\} S^{6m+2} ,$$

$$\int_{\infty}^S dS \int_{\infty}^S dS \operatorname{Re} h_1(iS) = -0.390992$$

$$+ 0.504361 S$$

$$+ \sum_{m=0}^{\infty} \left\{ \frac{(-1)^{m+1} S^3 B_{2m+1}}{(6m+5)(6m+6)(200)^{2m+2}} + \frac{\sqrt{3}}{3} \left[\frac{2(-1)^m S^2 A_{2m+1}}{(6m+4)(6m+5)(200)^{2m+2}} + \frac{(-1)^{m+1} B_{2m}}{(6m+2)(6m+3)(200)^{2m}} \right] \right\} S^{6m+3} ,$$

$$+ \frac{\sqrt{3}}{3} \left[\frac{2(-1)^m S^2 A_{2m+1}}{(6m+4)(6m+5)(200)^{2m+2}} + \frac{(-1)^{m+1} B_{2m}}{(6m+2)(6m+3)(200)^{2m}} \right] \right\} S^{6m+3} ,$$

$$+ \frac{(-1)^{m+1} B_{2m}}{(6m+2)(6m+3)(200)^{2m}} \right\} S^{6m+3} ,$$

$$\begin{aligned}
\int_0^5 dS \operatorname{Im} h_1(iS) &= 0.873588 \\
&+ \sum_{m=0}^{\infty} \left\{ \frac{(-1)^m B_{2m}}{(6m+2)(200)^{2m}} \right. \\
&+ \frac{\sqrt{3}}{3} \left[\frac{(-1)^{m+1} S^4 B_{2m+1}}{(6m+5)(200)^{2m+1}} \right. \\
&\left. \left. + \frac{2(-1)^{m+1} A_{2m}}{(6m+1)(200)^{2m}} \right] \right\} S^{6m+1}
\end{aligned}$$

and

$$\begin{aligned}
\int_0^5 dS \int_0^5 dS \operatorname{Im} h_1(iS) &= -0.678640 \\
&+ 0.873588 S \\
&+ \sum_{m=0}^{\infty} \left\{ \frac{(-1)^m B_{2m}}{(6m+2)(6m+3)(200)^{2m}} \right. \\
&+ \frac{\sqrt{3}}{3} \left[\frac{(-1)^{m+1} S^4 B_{2m+1}}{(6m+5)(6m+6)(200)^{2m+1}} \right. \\
&\left. \left. + \frac{2(-1)^{m+1} A_{2m}}{(6m+1)(6m+2)(200)^{2m}} \right] \right\} S^{6m+2}
\end{aligned}$$

For given values of λ and c the solution of equation (20) was obtained by plotting the real part against the imaginary part for each side. The intersections of these two families of curves gave α and S_1 . R was computed from

$$R = \frac{DUc}{\alpha} \left(-\frac{1 + \frac{1}{2}c}{S_1} \right)^3.$$

The results are shown in Figure 1. The graph for $\lambda = 0$, which corresponds to a Navier-Stokes liquid, is seen to be in close agreement with Lin's results, and the preceding statements regarding the anticipated size of the various parameters are seemingly consistent with the final results. The results are qualitatively in agreement with those of Chan Man Fong and Walters, shown also in Figure 1. (The λ in their paper is defined as five times the value of the present one.) The quantitative disagreement of the two results is not understood; it is noted, however, that Chan Man Fong and Walters results for $\lambda = 0$, departing considerably from the results of Lin, do agree with the results of Stuart (1954). On this basis, it is believed that the present results are the more accurate ones.

D. The Stability of a Viscoelastic Liquid of Type C'

Introducing stretched coordinates again as in part B, near the critical point the stresses become

$$\bar{P}_{yy} = -\bar{P}_{zz} = \frac{2DWc}{iA\tilde{\mu}^2} J_{1c}^2 \frac{d^2v}{d\eta^2} + O\left(\frac{v}{\tilde{\mu}}\right),$$

$$\bar{P}_{yz} = \frac{1}{iA\tilde{\mu}^2} \left[J_{0c}^1 - 2(DWc)^2 J_{2c}^2 \right] \frac{d^2v}{d\eta^2}$$

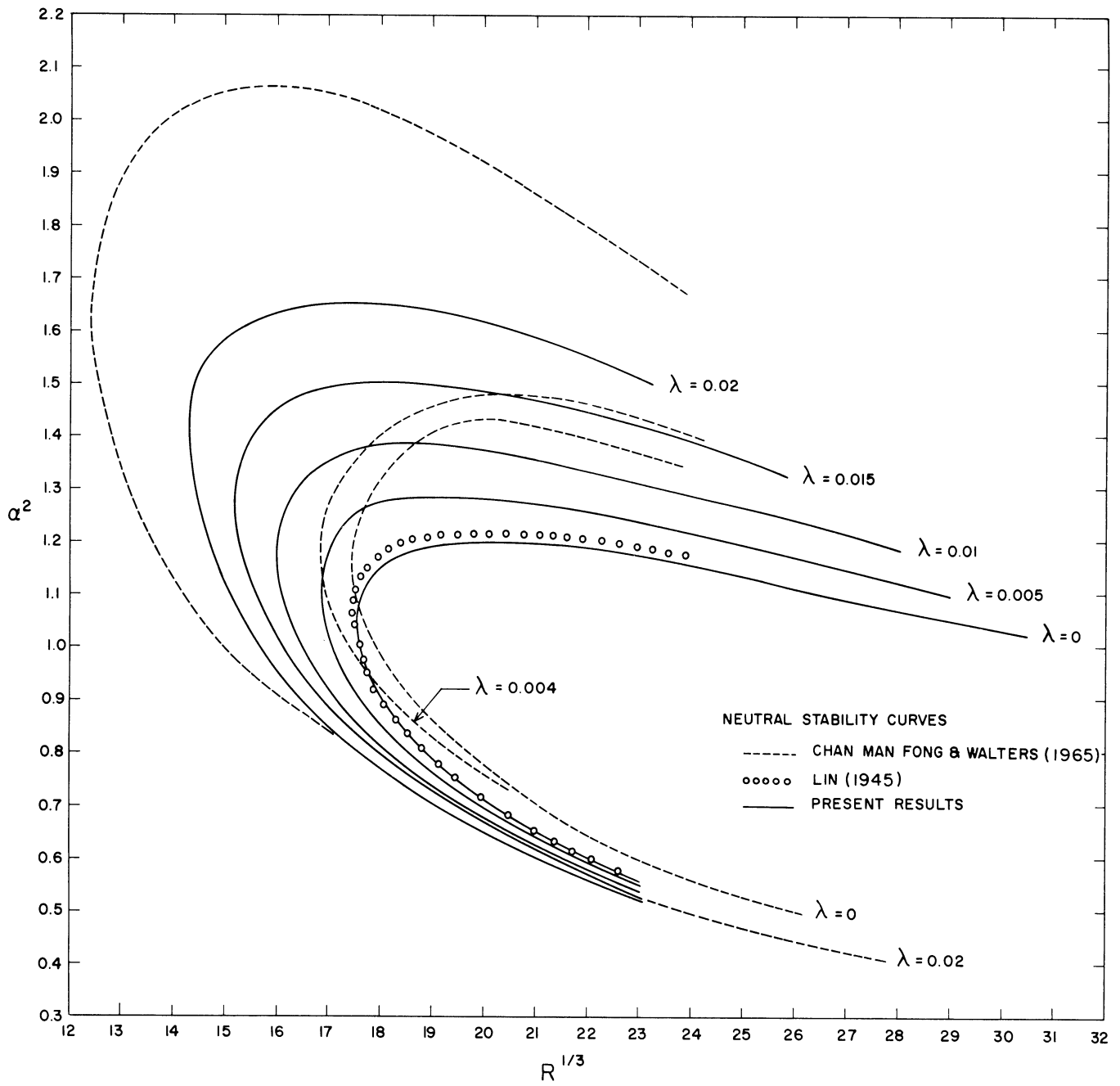


Figure 1. Neutral stability curves for various values of λ .

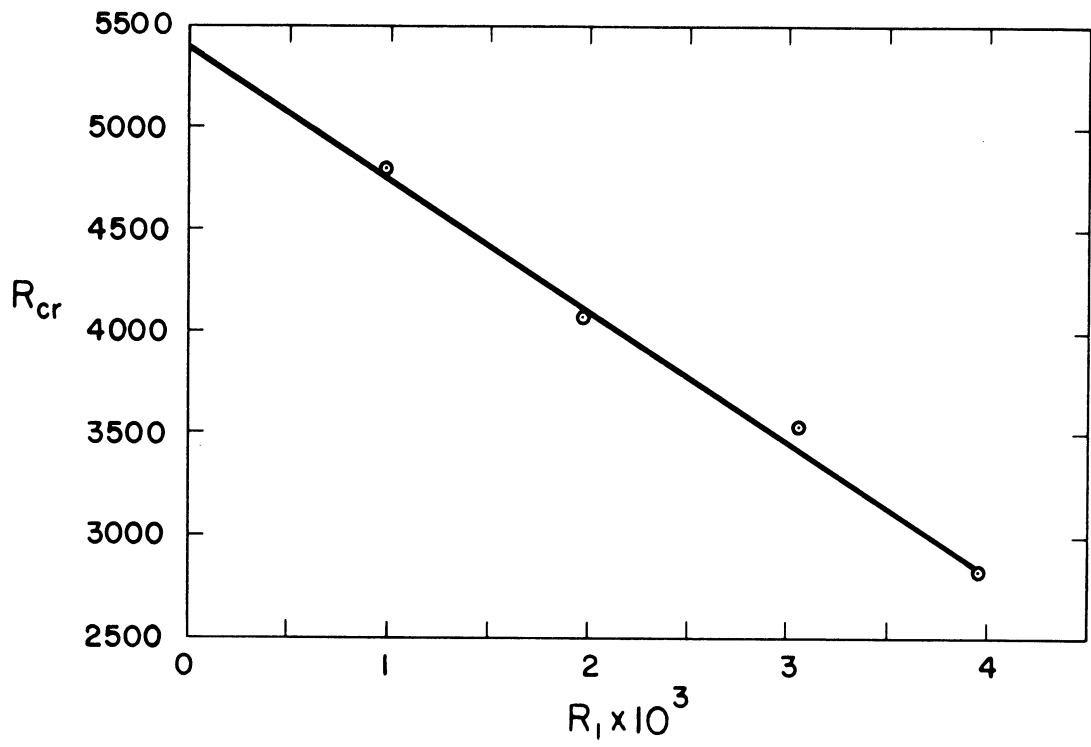


Figure 2. Critical Reynolds number versus the parameter R_1 .

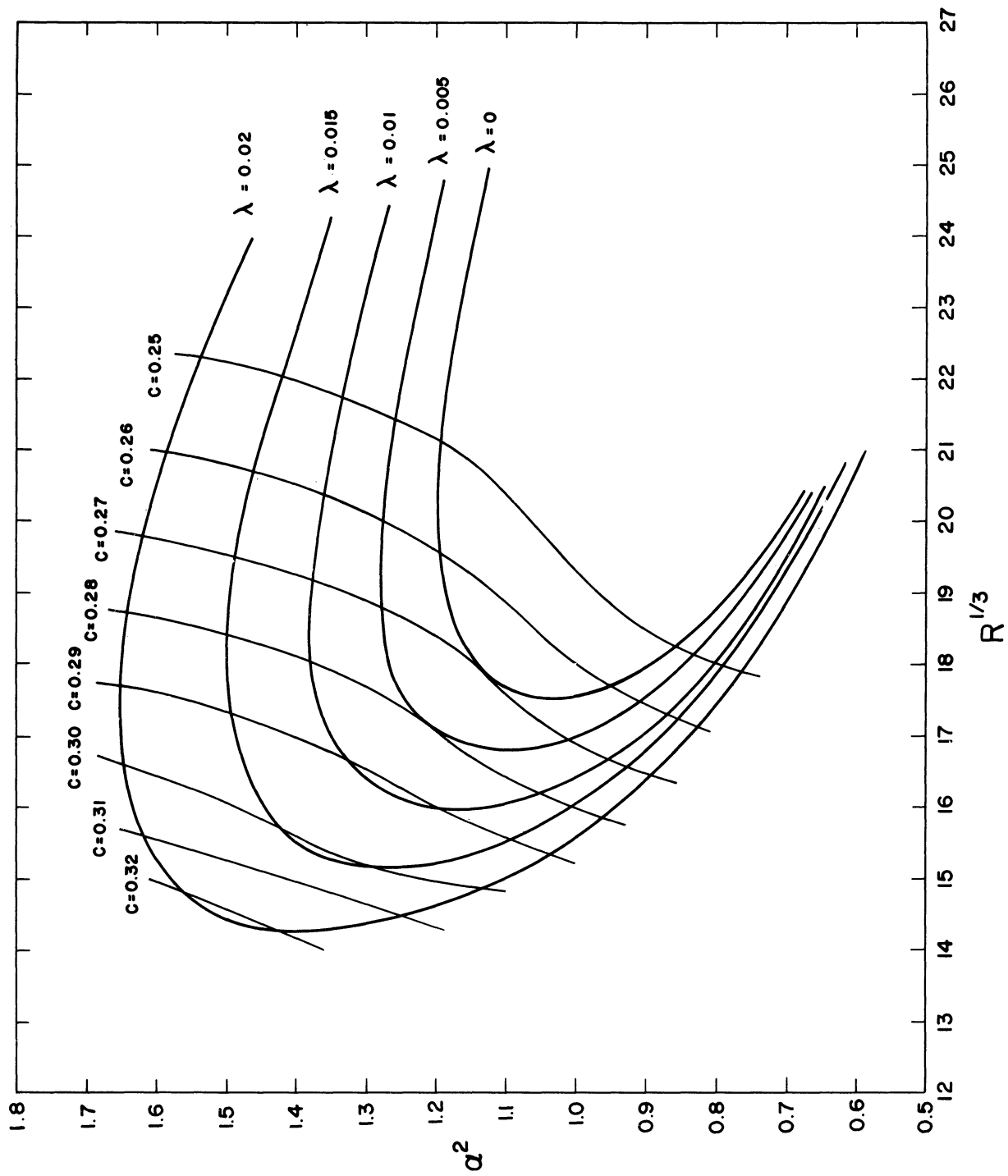


Figure 3. Neutral stability curves and curves of constant c for various values of λ .

$$\begin{aligned}
& -\frac{1}{iA\tilde{\mu}} \left\{ \eta \frac{d^2v}{d\eta^2} \left[6DW_c D^2W_c J_{2c}^2 \right. \right. \\
& - 8(DW_c)^3 D^2W_c J_{4c}^3 + iADW_c (2J_{1c}^2 \\
& - J_{1c}^1 + (DW_c)^2 (J_{3c}^2 - 4J_{3c}^3)) \left. \left. \right] \right. \\
& \left. + 2iADW_c J_{1c}^1 \frac{dv}{d\eta} \right\} + O(v)
\end{aligned}$$

where the subscript appended to the J_n^m 's indicates that they are evaluated at y_c ; the J_n^m 's and K_n^m 's have been assumed to be all of the same order of magnitude. The parameter $\tilde{\mu}$ is defined by

$$\tilde{\mu} = (\alpha \tilde{R})^{-1/2}, \text{ where } \tilde{R} = \rho h W_0 D^2W_c (\partial P / \partial z)^{-1} = \frac{\rho h y_c W_0 D^2W_c}{J_{1c}^1 DW_c}.$$

Making the distances and velocities dimensionless as before, and defining

$$\tilde{R}_1 = (\tilde{R} / \rho h^2) \left[J_{1c}^2 - 4 J_{3c}^3 \left(\frac{W_0 D U_c}{h} \right)^2 \right],$$

$$\tilde{R}_2 = (\tilde{R} / \rho h^2) \left[J_{1c}^1 + J_{1c}^2 + 8 J_{3c}^3 \left(\frac{W_0 D U_c}{h} \right)^2 \right],$$

substitution of these stresses into equation (11) along with use of the equation for the primary flow results in

$$\begin{aligned}
& \frac{d^4v}{d\eta^4} - i\eta D U_c \frac{d^2v}{d\eta^2} + \tilde{\mu} \left\{ -\frac{1}{2} i\eta^2 D^2 U_c \frac{d^2v}{d\eta^2} \right. \\
& - \eta \left(\frac{D^3 U_c}{D^2 U_c} + i\alpha D U_c \tilde{R}_1 \right) \frac{d^4v}{d\eta^4} \\
& \left. + 2 \left(-\frac{D^3 U_c}{D^2 U_c} + i\alpha D U_c \tilde{R}_2 \right) \frac{d^3v}{d\eta^3} \right. \\
& \left. + O(\tilde{\mu}^2) = 0.
\end{aligned}$$

Proceeding further as in case B, with $\epsilon_1 = \alpha \tilde{\mu} \tilde{R}_1$, $\epsilon_2 = \alpha \tilde{\mu} \tilde{R}_2$, we have

$$\begin{aligned} (1 - i\epsilon_1 \eta DU_c) \frac{d^4 \chi^{(0)}}{d\eta^4} + 2i\epsilon_2 DU_c \frac{d^3 \chi^{(0)}}{d\eta^3} \\ - i\eta DU_c \frac{d^2 \chi^{(0)}}{d\eta^2} = 0 \end{aligned} \quad (22)$$

as the governing equation in the inner region. It is seen then that if ϵ_1 and ϵ_2 are both small compared to unity, but larger than $\tilde{\mu}$, again $\chi_1^{(0)}$ and $\chi_2^{(0)}$ remain unchanged, but a first approximation to $\chi_3^{(0)}$ is now

$$\chi_3^{(0)} \approx (1 - \frac{1}{2} i\epsilon_2 \eta DU_c) \phi_0 + \epsilon_1 \phi_1$$

where ϕ_0 , ϕ_1 are as given in part B.

Away from the critical region, the inviscid equation will again hold, but the primary velocity profile is of course different from the parabolic one. (For the model typified by equation (1) with $M = N = 1$ and $L_t = L_{tC}$, for instance, DW satisfies a cubic equation where the coefficients are linear functions of y .) Comparing a fluid of type A' (or B') with type C', if N and $\partial p / \partial z$ are the same for both cases, the fluid of type C' will have a steeper velocity profile than the parabolic one. To carry out the details of the solution, it is necessary to specify N . When this is known, W can readily be found by numerical methods, and equation (17) then used to determine the neutral stability curve.

BIBLIOGRAPHY

1. The Annals of the Computation Laboratory of Harvard University, Volume II, Harvard University Press, Cambridge, Mass., 1945.
2. Chan Man Fong, C. F., and Walters, K., "The solution of flow problems in the case of materials with memory, Part II," Journal de Mechanique 4, 1965, 439-453.
3. Goddard, J. D., and Miller, C., "An inverse for the Jaumann derivative and some applications to the rheology of viscoelastic fluids," Rheologica Acta 5, 1966, 177-183.
4. Graebel, W. P., "On the determination of the characteristic equations for the stability of parallel flow," Journal of Fluid Mechanics 24, 1966, 497-508.
5. Jaumann, G., "Geschlossenes System physikalischer und chemischer Differenzialgesetze," Sitzber. Akad. Wiss. Wien (IIa) 120, 1911, 385-530.
6. Lin, C. C., "On the stability of two dimensional parallel flows," Quarterly of Applied Mathematics 3, 1945, 277-301.
7. Listrov, A. T., "Parallel flow stability of non-Newtonian media," Soviet Physics, Dokl. 10, 1966, 912-914.
8. Oldroyd, J. G., "On the formulation of rheological equations of state," Proceedings of the Royal Society (A) 200, 1950, 523-541.
9. Oldroyd, J. G., "Non-Newtonian effects in steady motion of some idealized elastico-viscous liquids," Proceedings of the Royal Society (A) 245, 1958, 278-297.
10. Sokolnikoff, I. S., Tensor Analysis, John Wiley and Sons, Inc., New York, 1951.
11. Squire, H. B., "On the stability for three-dimensional disturbances of viscous fluid flow between parallel walls," Proceedings of the Royal Society (A) 142, 1933, 621-628.

12. Stuart, J. T., Proceedings of the Royal Society (A) 221, 1954, 189-205.
13. Tollmien, W., "General instability criterion of laminar velocity distributions," Technical Memorandum 792, NACA, 1936.
14. Walters, K., "The solution of flow problems in the case of materials with memory, Part I," Journal de Mechanique 2, 1962, 479-486.

APPENDIX

An exact solution of equation (18) is possible and has, in fact, been given by Chan Man Fong and Walters (1965). A modified and more complete version of their results is presented here to show its use in the present method.

Equation (18) is of the form

$$(1 - i\lambda\eta) \frac{d^2\Phi}{d\eta^2} - i\eta \Phi = 0.$$

Writing

$$z = 2(i\lambda\eta - 1)\lambda^{-3/2}, \quad \delta = \frac{1}{2}\lambda^{-3/2},$$

and

$$\Phi = e^{-z/2} f(z),$$

the equation for f is the confluent hypergeometric form

$$z f'' - z f' - \delta f = 0$$

with solutions

$$U(z, \delta) = \frac{1}{\Gamma(\delta)} \int_0^{\infty e^{i\beta}} e^{-zv} v^{\delta-1} (v+1)^{-\delta-1} dv, \quad (23)$$

$$\left(-\pi < \beta < \pi; -\frac{\pi}{2} < \beta + \arg z < \frac{\pi}{2} \right),$$

and

$$V(z, \delta) = \frac{e^z \Gamma(\delta)}{2\pi i} \int_{-\infty, 0+}^{-\infty} e^{vz} v^{-\delta} (v+1)^\delta dv.$$

As z approaches $i\infty$, $U(z)$ approaches $z^{-\delta}$ and $V(z)$ approaches $z^\delta e^\delta$. This suggested that U must be the solution corresponding to χ_3 and V to χ_4 . To verify that this is indeed the case, replace v in equation (23) by $1/2(-1 + s\sqrt{\lambda})$. Then in the limit as λ approaches zero with $\arg z = -\pi$ and $\beta = 2\pi/3$,

$$\begin{aligned} & \lambda^{-1/2} \Gamma(\delta) (-1)^\delta e^{-z/2} U(z) \\ & \xrightarrow{\infty e^{i\beta}} -2 \int_{\infty} \exp[-i\eta s - \frac{1}{3}s^3] \left[1 \right. \\ & \quad \left. + \lambda(s^2 - \frac{1}{5}s^5) + \dots \right] ds, \end{aligned}$$

or, upon expressing the integral in terms of Hankel functions,

$$\begin{aligned} & \lambda^{-1/2} \Gamma(\delta) (-1)^\delta e^{-z/2} U(z) \\ & \rightarrow -2 \left[I - \lambda \left(\frac{d^2 I}{d\eta^2} + \frac{i}{5} \frac{d^5 I}{d\eta^5} \right) \right] \\ & = -2 \left[I + \frac{1}{5} \lambda \frac{d^2}{d\eta^2} \left(\eta \frac{dI}{d\eta} - 4I \right) \right], \end{aligned}$$

where

$$\begin{aligned} I & = \int_{\infty}^{\infty e^{2\pi i/3}} \exp \left[-i s \eta - \frac{1}{3} s^3 \right] ds \\ & = -\pi \sqrt{\eta/3} e^{-i\pi/12} H_{1/3}^{(1)} \left[\frac{2}{3} (i\eta)^{3/2} \right]. \end{aligned}$$

This is the desired result apart from a multiplicative constant. Thus the general form of $\chi_3^{(0)}$ for arbitrary λ is given by integrating

$$\frac{d^2 \chi_3^{(0)}}{d\eta^2} = \int_0^\infty e^{2\pi i/\lambda} \exp[-z(v + 1/2)] (v + 1)^{-s-1} v^{s-1} dv$$

and using this $\chi_3^{(0)}$ in the characteristics equation (17).

DISTRIBUTION LIST

(One copy unless noted otherwise)

Chief of Naval Research Department of the Navy Washington, D.C. 20360 Attn: Code 438 Code 463 Code 466		Chief, Bureau of Ships Department of the Navy Washington, D.C. 20360 Attn: Code 300 Code 305 Code 335 Code 341 Code 342A Code 345 Code 420 Code 421 Code 440 Code 442 Code 634A
Commanding Officer Office of Naval Research Branch Office 495 Summer Street Boston, Massachusetts 02110		Chief, Bureau of Naval Weapons Department of the Navy Washington, D.C. 20360 Attn: Code R Code R-12 Code RR Code RRRE Code RU Code RUTO
Commanding Officer Office of Naval Research Branch Office 219 S. Dearborn Street Chicago, Illinois 60604		
Commanding Officer Office of Naval Research Branch Office 207 W. 24th Street New York, New York 10011		
Commanding Officer Office of Naval Research Branch Office Fleet Post Office Box 39 New York, New York 09510	(25)	Commander Military Sea Transportation Service Department of the Navy Washington, D.C. 20360
Commanding Officer Office of Naval Research Branch Office 1030 E. Green Street Pasadena, California 91101		Special Projects Office Department of the Navy Washington, D.C. 20360 Attn: Code 001 Code 201
Commanding Officer Office of Naval Research Branch Office 1076 Mission Street San Francisco, California 94103		Chief, Bureau of Yards and Docks Department of the Navy Washington, D.C. 20360 Attn: Code D-202

Commanding Officer and Director
David Taylor Model Basin
Washington, D.C. 20007
Attn: Code 500
Code 513
Code 520
Code 521
Code 522
Code 580
Code 589

Superintendent
U. S. Naval Academy
Annapolis, Maryland 21402
Attn: Library

Commanding Officer and Director
U. S. Naval Civil Engineering
Laboratory
Port Hueneme, California 93041

Commanding Officer and Director
U. S. Navy Mine Defense Laboratory
Panama City, Florida 32402

Commander
Naval Ordnance Test Station
China Lake, California 93557

Code 4032
Code 5014
Code 753

Officer in Charge
Naval Ordnance Test Station
Pasadena Annex
3203 E. Foothill Boulevard
Pasadena, California 91107
Attn: Mr. J. W. Hoyt
Research Division
Code P508
Code P804
Code P807
Code P80962

Superintendent
U. S. Naval Postgraduate School
Monterey, California 93940
Attn: Library

Director (6)
Naval Research Laboratory
Washington, D.C. 20390
Attn: Code 2027

Commanding Officer
NROTC and Naval Administrative Unit
Massachusetts Institute of Tech-
nology
Cambridge, Massachusetts 02139

Commanding Officer
U. S. Naval Underwater Ordnance
Station
Newport, Rhode Island
Attn: Research Division

Commander
U. S. Naval Weapons Laboratory
Dahlgren, Virginia 22448
Attn: Tech. Library Div. (AAL)

Commander
Boston Naval Shipyard
Boston, Massachusetts 02129

Commander
Charleston Naval Shipyard
U. S. Naval Base
Charleston, South Carolina 29408

Commander
Long Beach Naval Shipyard
Long Beach, California 90802

Commander
Mare Island Naval Shipyard
Vallejo, California 94592

Defense Documentation Center (5)
Cameron Station
Alexandria, Virginia 22314

Commander
Norfolk Naval Shipyard
Portsmouth, Virginia 23709

Commander
Pearl Harbor
Navy No. 128, Fleet Post Office
San Francisco, California 96614

Commander
Philadelphia Naval Shipyard
Naval Base
Philadelphia, Pennsylvania 19112

Commander
Portsmouth Naval Shipyard
Portsmouth, New Hampshire 03804
Attn: Design Division

Commander
Puget Sound Naval Shipyard
Bremerton, Washington 98314

Commander
San Francisco Naval Shipyard
San Francisco, California 94135

Commander
Air Force Cambridge Research Lab.
L. G. Hanscom Field
Bedford, Massachusetts 01731
Attn: Research Library, CRMXL-R

Air Force Office of Scientific Research
Mechanics Division
Washington, D.C. 20333

Director
U. S. Army Engineering Research
and Development Laboratories
Fort Belvoir, Virginia
Attn: Technical Documents Center

Commanding Officer
U. S. Army Research Office—Durham
Box CM, Duke Station
Durham, North Carolina 27706

Commander
Hq., U. S. Transportation Research
and Development Command
Transportation Corps
Fort Eustis, Virginia

Commander
Hq., U. S. Transportation Research
and Development Command
Transportation Corps
Fort Eustis, Virginia
Attn: Marine Transport Division

Commandant
U. S. Coast Guard
1300 E Street, N.W.
Washington, D.C.

Superintendent
U. S. Merchant Marine Academy
Kings Point, L. I., New York
Attn: Department of Engineering

Director of Research
National Aeronautics and Space
Administration Headquarters
600 Independence Avenue, S.W.
Washington, D.C. 20546
Attn: Code RR

Director
Langley Research Center
National Aeronautics and Space
Administration
Langley Field, Virginia

Coordinator of Research
Maritime Administration
441 G Street, N.W.
Washington, D.C. 20235

Division of Ship Design
Maritime Administration
441 G Street, N.W.
Washington, D.C. 20235

Research and Development
Maritime Administration
441 G Street, N.W.
Washington, D.C. 20235

National Academy of Sciences
National Research Council
2101 Constitution Avenue, N.W.
Washington, D.C. 20418

Dr. G. B. Schubauer
Fluid Mechanics Section
National Bureau of Standards
Washington, D.C. 20360

Mr. E. S. Turner
National Research Council
Montreal Road
Ottawa 2, Canada

Director
Engineering Science Division
National Science Foundation
Washington, D.C.

Mr. C. A. Gongwer
Aerojet General Corporation
6352 N. Irwindale Avenue
Azusa, California

Mr. W. R. Wiberg, Chief
Marine Performance Staff
The Boeing Company
Aero-Space Division
P. O. Box 3707
Seattle 24, Washington

Hydrodynamics Laboratory
California Institute of Technology
Pasadena, California 91109

Dr. J. Laufer
California Institute of Technology
Pasadena, California 91109

Professor T. Y. Wu
California Institute of Technology
Pasadena, California 91109

Department of Engineering
University of California
Berkeley, California 94720

Professor P. Lieber
University of California
Berkeley, California 94720

Professor M. S. Uberoi
Department of Aeroanautical
Engineering
University of Colorado
Boulder, Colorado

Mr. R. H. Oversmith
Hydrodynamics Laboratory
Convair
San Diego 12, California

Professor A. B. Metzner
University of Delaware
Newark, Delaware

Engineering Societies Library
29 W. 39th Street
New York 18, New York

Mr. P. Eisenberg, President
Hydronautics, Inc.
Pindell School Road
Howard County
Laurel, Maryland

Professor L. Landweber
Iowa Institute of Hydraulic Research
State University of Iowa
Iowa City, Iowa

Dr. C. Elata
Hydraulics Laboratory
Israel Institute of Technology
Haifa, Israel

Professor S. Corrsin
The Johns Hopkins University
Baltimore 18, Maryland

Dr. R. H. Kraichnan
Peterborough, New Hampshire

Professor A. T. Ippen
Massachusetts Institute of Technology
Cambridge, Massachusetts 02139

Professor C. C. Lin
Massachusetts Institute of Technology
Cambridge, Massachusetts 02139

Dr. B. Sternlicht
Mechanical Technology, Inc.
968 Albany-Shaker Road
Latham, New York

Acting Director
St. Anthony Falls Hydraulic
Laboratory
University of Minnesota
Minneapolis, Minnesota 55414

Professor J. Ripken
St. Anthony Falls Hydraulic
Laboratory
University of Minnesota
Minneapolis, Minnesota 55414

Dr. E. R. Van Driest
Missile Development Division
North American Aviation, Inc.
Downey, California

Dr. T. R. Goodman
Oceanics, Inc.
Plainview, L.I., New York

Mr. L. M. White
U. S. Rubber Company
Research and Development Department
Wayne, New Jersey

Society of Naval Architects
and Marine Engineers
74 Trinity Place
New York 6, New York

Dr. J. P. Breslin
Stevens Institute of Technology
Davidson Laboratory
Hoboken, New Jersey

Transportation Technical Research
Institute
Investigation Office
Ship Research Institute
700 Shinkawa Mituka
Tokyo, Japan

Dr. A. Sacks
Vidya, Inc.
2626 Hanover Street
Palo Alto, California

Technical Library
Webb Institute of Naval
Architecture
Glen Cove, L.I., New York 11542

Mr. H. Crawford
Westco Research
Division of the Western Company
1171 Empire Central
Dallas 7, Texas

Professor David T. Pratt
Engineering Department
U. S. Naval Academy
Annapolis, Maryland 21402

Mr. Ralph Little
Naval Research Laboratory
Washington, D.C. 20390
Attn: Code 6170

Dr. C. R. Singleterry
Naval Research Laboratory
Washington, D.C. 20390
Attn: Code 6170

Dr. C. S. Wells, Jr.
LTV Research Center
Ling-Temco-Vought, Inc.
P. O. Box 5003
Dallas 22, Texas

Professor S. R. Keim
College of Engineering
University of California
Davis, California

Mr. W. E. Ferrin
Commercial Exploration Manager
Archer Daniels Midland Company
10701 Lyndale Avenue
South Minneapolis, Minnesota

Office of Research Administration
The University of Michigan
Ann Arbor, Michigan
Attn: File Room

Professor R. B. Couch
Department of Naval Architecture
and Marine Engineering
The University of Michigan
Ann Arbor, Michigan

Professor W. W. Willmarth
Department of Aerospace Engineering
The University of Michigan
Ann Arbor, Michigan

Dr. C. S. Yih
Department of Engineering Mech-
anics
The University of Michigan
Ann Arbor, Michigan

DOCUMENT CONTROL DATA - R&D

(Security classification of title, body of abstract and indexing annotation must be entered when the overall report is classified)

1. ORIGINATING ACTIVITY (Corporate author) The University of Michigan Department of Engineering Mechanics Ann Arbor, Michigan		2 a. REPORT SECURITY CLASSIFICATION Unclassified	
		2 b. GROUP	
3. REPORT TITLE The Stability of Parallel Flows of Fluids with Memories			
4. DESCRIPTIVE NOTES (Type of report and inclusive dates) Technical Report			
5. AUTHOR(S) (Last name, first name, initial) Mook, Dean T. Graebel, W. P., Project Director			
6. REPORT DATE September 1967		7 a. TOTAL NO. OF PAGES 51	7 b. NO. OF REFS --
8 a. CONTRACT OR GRANT NO. Nonr-1224(49)		9 a. ORIGINATOR'S REPORT NUMBER(S) 06505-3-T	
b. PROJECT NO. NR 062-342		9 b. OTHER REPORT NO(S) (Any other numbers that may be assigned this report)	
c.			
d.			
10. AVAILABILITY/LIMITATION NOTICES Distribution of this document is unlimited.			
11. SUPPLEMENTARY NOTES		12. SPONSORING MILITARY ACTIVITY Department of the Navy Office of Naval Research Washington, D.C.	
13. ABSTRACT The equations governing the stability of plane parallel flows are developed for three models of fluids with memories. Asymptotic solutions valid for large Reynolds numbers are obtained and the effect of the memory are shown to be destabilizing. The approach to the problem allows evaluation of how fast a memory must fade to allow evaluation of the stresses in power series in the time interval. An alternate approach to inverting convected derivatives is also presented.			

Security Classification

14.	KEY WORDS	LINK A		LINK B		LINK C	
		ROLE	WT	ROLE	WT	ROLE	WT

INSTRUCTIONS

1. ORIGINATING ACTIVITY: Enter the name and address of the contractor, subcontractor, grantee, Department of Defense activity or other organization (*corporate author*) issuing the report.

2a. REPORT SECURITY CLASSIFICATION: Enter the overall security classification of the report. Indicate whether "Restricted Data" is included. Marking is to be in accordance with appropriate security regulations.

2b. GROUP: Automatic downgrading is specified in DoD Directive 5200.10 and Armed Forces Industrial Manual. Enter the group number. Also, when applicable, show that optional markings have been used for Group 3 and Group 4 as authorized.

3. REPORT TITLE: Enter the complete report title in all capital letters. Titles in all cases should be unclassified. If a meaningful title cannot be selected without classification, show title classification in all capitals in parenthesis immediately following the title.

4. DESCRIPTIVE NOTES: If appropriate, enter the type of report, e.g., interim, progress, summary, annual, or final. Give the inclusive dates when a specific reporting period is covered.

5. AUTHOR(S): Enter the name(s) of author(s) as shown on or in the report. Enter last name, first name, middle initial. If military, show rank and branch of service. The name of the principal author is an absolute minimum requirement.

6. REPORT DATE: Enter the date of the report as day, month, year; or month, year. If more than one date appears on the report, use date of publication.

7a. TOTAL NUMBER OF PAGES: The total page count should follow normal pagination procedures, i.e., enter the number of pages containing information.

7b. NUMBER OF REFERENCES: Enter the total number of references cited in the report.

8a. CONTRACT OR GRANT NUMBER: If appropriate, enter the applicable number of the contract or grant under which the report was written.

8b, 8c, & 8d. PROJECT NUMBER: Enter the appropriate military department identification, such as project number, subproject number, system numbers, task number, etc.

9a. ORIGINATOR'S REPORT NUMBER(S): Enter the official report number by which the document will be identified and controlled by the originating activity. This number must be unique to this report.

9b. OTHER REPORT NUMBER(S): If the report has been assigned any other report numbers (*either by the originator or by the sponsor*), also enter this number(s).

10. AVAILABILITY/LIMITATION NOTICES: Enter any limitations on further dissemination of the report, other than those

imposed by security classification, using standard statements such as:

- (1) "Qualified requesters may obtain copies of this report from DDC."
- (2) "Foreign announcement and dissemination of this report by DDC is not authorized."
- (3) "U. S. Government agencies may obtain copies of this report directly from DDC. Other qualified DDC users shall request through _____."
- (4) "U. S. military agencies may obtain copies of this report directly from DDC. Other qualified users shall request through _____."
- (5) "All distribution of this report is controlled. Qualified DDC users shall request through _____."

If the report has been furnished to the Office of Technical Services, Department of Commerce, for sale to the public, indicate this fact and enter the price, if known.

11. SUPPLEMENTARY NOTES: Use for additional explanatory notes.

12. SPONSORING MILITARY ACTIVITY: Enter the name of the departmental project office or laboratory sponsoring (*paying for*) the research and development. Include address.

13. ABSTRACT: Enter an abstract giving a brief and factual summary of the document indicative of the report, even though it may also appear elsewhere in the body of the technical report. If additional space is required, a continuation sheet shall be attached.

It is highly desirable that the abstract of classified reports be unclassified. Each paragraph of the abstract shall end with an indication of the military security classification of the information in the paragraph, represented as (TS), (S), (C), or (U).

There is no limitation on the length of the abstract. However, the suggested length is from 150 to 225 words.

14. KEY WORDS: Key words are technically meaningful terms or short phrases that characterize a report and may be used as index entries for cataloging the report. Key words must be selected so that no security classification is required. Identifiers, such as equipment model designation, trade name, military project code name, geographic location, may be used as key words but will be followed by an indication of technical context. The assignment of links, rules, and weights is optional.

UNIVERSITY OF MICHIGAN



3 9015 03483 1324