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Department of Electrical Engineering
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Final Report

STUDIES TOWARD THE DEVELOPMENT OF A D-REGION PROBE

J. C. Pearl

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NOMENCLATURE

Alphabetic

A	surface area
A_i	see Eqn. (5.4-10)
a	probe radius ($= \frac{a'}{\lambda}$)
B_i	see Eqn. (5.4-10)
D	coefficient of diffusion
e	electronic charge
f	function representing r_0 (Eqn. (5.3-5)); function used in sheath cutoff (Eqn. (6.4-24))
g	function used in sheath cutoff (Eqn. (6.4-29))
H	parameter for finite difference calculations (Eqn. (5.2-8))
h	function used in sheath cutoff (Eqn. (6.4-30))
I	current
J	current density (positive species)
K	mobility constant ($= \frac{eD}{kT}$)
k	index for solutions of cubic equation (Eqn. (6.4-14)); Boltzmann constant
L	current density (negative species)
l	displacement
n	number of incremented steps in a calculation; number of particles in a Debye cube

NOMENCLATURE (Continued)

Alphabetic

P	arbitrary field point
p	parameter used in flow line computation (Eqn. (5.2-12))
Q	charge; point on central axis of torus (Fig. 11)
q	charge
R	distance from coordinate origin
$R_1, R_2, R(P)$	distances identified in Fig. 9.
r	radial cylindrical coordinate; minor radius of torus (Eqn. (5.4-18))
S	point in charge density field (Fig. 11)
s	distance normal to flow lines
T	point on torus (Fig. 11); temperature
t	time (= $\frac{t'V'}{\lambda}$); variable used in sheath cutoff (Eqn. (6.4-27))
V	neutral gas flow velocity (= $\frac{\lambda V}{D}$)
v	composite velocity term (Eqn. (5.2-2))
x	variable used in sheath cutoff (Eqn. (6.4-23))
y	see Eqn. (6.4-6)
z	axial cylindrical coordinate
α_i	dimensionless ratio (Eqn. (5.4-13))

NOMENCLATURE (Continued)

Alphabetic

β	dimensionless parameter (Eqn. (6.2-5))
γ	parameter used in sheath cutoff (Eqn. (6.4-27))
Δ	parameter in difference formula (Fig. 7)
δ	indicator of finite increment
ϵ	dielectric constant; error due to approximations
ζ	negative charge density ($= \frac{\zeta'}{\zeta'_{\infty}}$)
η	parameter used in sheath cutoff (Eqn. (6.4-33))
κ	complete elliptic integral of the first kind
λ	Debye length ($= \sqrt{\frac{\epsilon k T}{e \sigma'_{\infty}}}$)
μ	see Eqn. (6.4-12)
ρ	charge density (Eqn. (5.4-9))
Σ', Σ''	see Eqn. (5.4-26)
σ	positive charge density ($= \frac{\sigma'}{\sigma'_{\infty}}$)
τ	volume
Φ	electrical potential ($= \frac{e\Phi'}{kT}$)
ϕ	azimuthal cylindrical coordinate

Subscripts

i	counting or identifying index
n	evaluation after n increments; summation index

NOMENCLATURE (Concluded)

Subscripts

o	identifies flow line (see Section 5.3, p. 16). evaluation for sphere in torus routine (Eqn. (5.4-24))
p	condition at probe
r	radial component
z	axial component
∞	condition at infinity

Superscripts and Miscellaneous

'	quantity before being nondimensionalized; image (Eqn. (5.4-5)); evaluation for flow line tangent to probe (Eqn. 6.4-3))
"	evaluation for flow line target to probe (see p.30)
$\hat{}$	unit vector
∇	gradient operator ($= \lambda \hat{\nabla}'$)
a	absolute value of a
[a]	greatest integer in a.

1. INTRODUCTION

Since 1946, when University of Michigan investigators first proposed such experiments, Langmuir probes have been widely used to study the E and F regions of the ionosphere. The theory of operation of Langmuir probes, for these regions, where the mean free path of the particles is large, is well understood and therefore their volt-ampere characteristics can be accurately interpreted in terms of ambient charged particle parameters.

The growing interest in the physical properties of the lower parts of the ionosphere has provided impetus to study the feasibility of the use of such current collecting probes for the D region. Recently a number of Langmuir probe measurements have been made at these low altitudes. It is, however, difficult to interpret the data from such measurements, since to date no relevant theoretical work has been published dealing with the current collection characteristics of such probes in regions of high neutral gas density.

This report deals with the development of a probe for D region charged particle density measurements and the work done to enable the interpretation of the current measured by such a probe in terms of the ambient density. A test model of the probe was constructed and placed in the wind tunnel of the Aberdeen Proving Grounds. The results of these tests are also discussed.

2. SELECTION OF THE PROBE CONFIGURATION

A probe configuration enabling measurements to be made in the E as well as the D region was the aim of this study, since the sounding rockets which are likely to carry these experiments will pass through both of these regions. Because there is a great deal of confidence in E region Langmuir probe measurements, these results could also be used to provide some degree of calibration for the interpretation of the D region data.

First the question of the probe position received a great deal of consideration. The velocity of the vehicle is expected to be supersonic, resulting in the formation of a bow shock wave. As even the neutral gas flow well behind the shock is not understood it is considered imperative that the probe be located on the tip of the nose cone or on a boom of sufficient length to place it outside the shock cone of the vehicle. In either case the probe itself will create its own shock disturbance. A nose cone probe was chosen since the technical problems involved in its construction are considerably less than those for a boom supported probe. Because the tip of the nose cone is preferred for a number of other experiments it is felt that if the results from the nose cone probe experiments are encouraging work should be initiated towards the design of boom mounted probes.

Having established the nose tip as the preferred position of the probe, a configuration using a hemispherical tip was chosen. The main advantages of such a probe over a conical one, are:

- (a) the collector area projected in the direction perpendicular to the velocity vector is not very sensitive to small changes in angle of attack as would be the case for example with the usual conical nose tip;
- (b) the current collection theory for spherical collector has been worked out for the long mean free path operation to be encountered in the E region.

The principal disadvantage of the spherical over the sharp conical arrangement is the increased aerodynamic drag, but this increase is not expected to be significant for most applications. For both D and E region application it is considered important to minimize the effects of the distortion in the sheath about the probe, caused by the rest of the nose cone. All these considerations led to the choice of probe configuration indicated in Fig. 1.

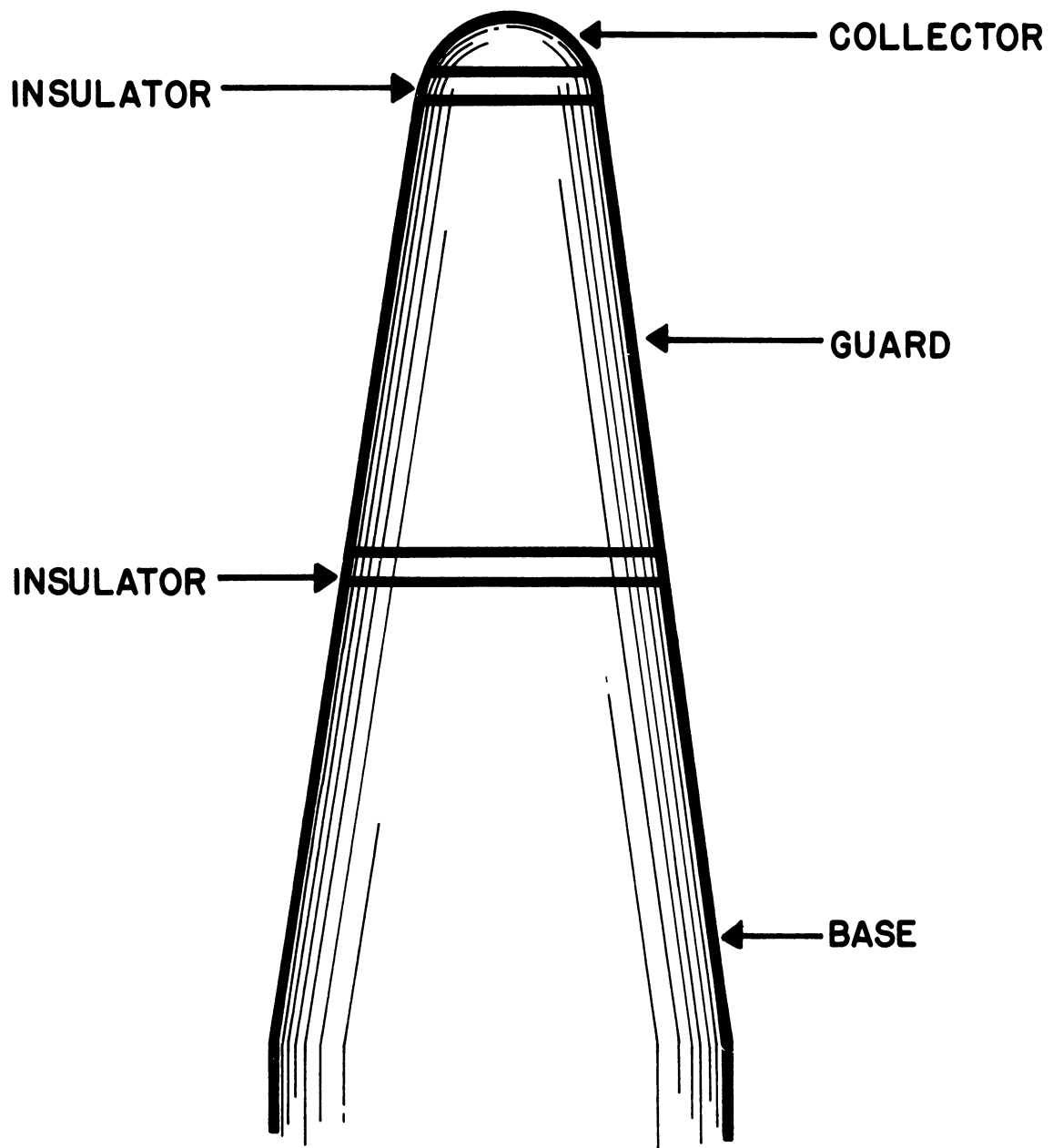


Fig. 1. Configuration of probe assembly.

3. WIND TUNNEL TESTS

A decision was made at the beginning of the contract period to observe the behavior of the probe in the wind tunnel at the Aberdeen Proving Grounds. The relatively high operating pressure of this wind tunnel (total pressure 100 torr) clearly does not permit the simulation of conditions typical of the D region but still tests were carried out in the hope of obtaining certain information about the behavior of the probe (e.g., saturation, etc.). A radioactive source was employed to achieve partial ionization in the tunnel, but since there was no way of knowing the degree of ionization, the percentage of electrons and negative ions and what the recombination rates were along the flow, no real interpretation of the experimental results was possible.

A photograph of the probe constructed for these tests shown in Fig. 2. The electrometer amplifier used in this test fixture was capable of measuring currents from 10^{-13} to 10^{-7} amperes in seven linear ranges. A remote controlled voltage stepping circuit allowed the application of eleven different voltages from -15 to +15 volts to the collector and guard.

Current versus voltage readings were taken for a number of Mach numbers and source strength values. Figure 3 shows the current versus voltage characteristics for $M=3.02$. These data indicate a strong relation between the current collected and the radioactive source strength. These tests gave no indication of current saturation, so the test setup was changed to permit the application of potentials up to 200V. The results of such a test are shown in Fig. 4; even here there is no clear indication of saturation. The current detector and the voltage source for these runs had to be situated outside the tunnel and connected to the probe via shielded cable and therefore considerable noise voltage may have been present. This noise however is not believed to have been of sufficient magnitude to mask the real results.

Numerous Schlieren photographs were taken during the test (Fig. 5). These show that the shock separation as expected, was very small. It was possible to vary the angle of attack in the wind tunnel from 0° to 15° . Changes in the angle of attack over this range resulted in no detectable change in the current collected. However, since these tests were also run with the electronics outside the tunnel the results have to be treated with caution.

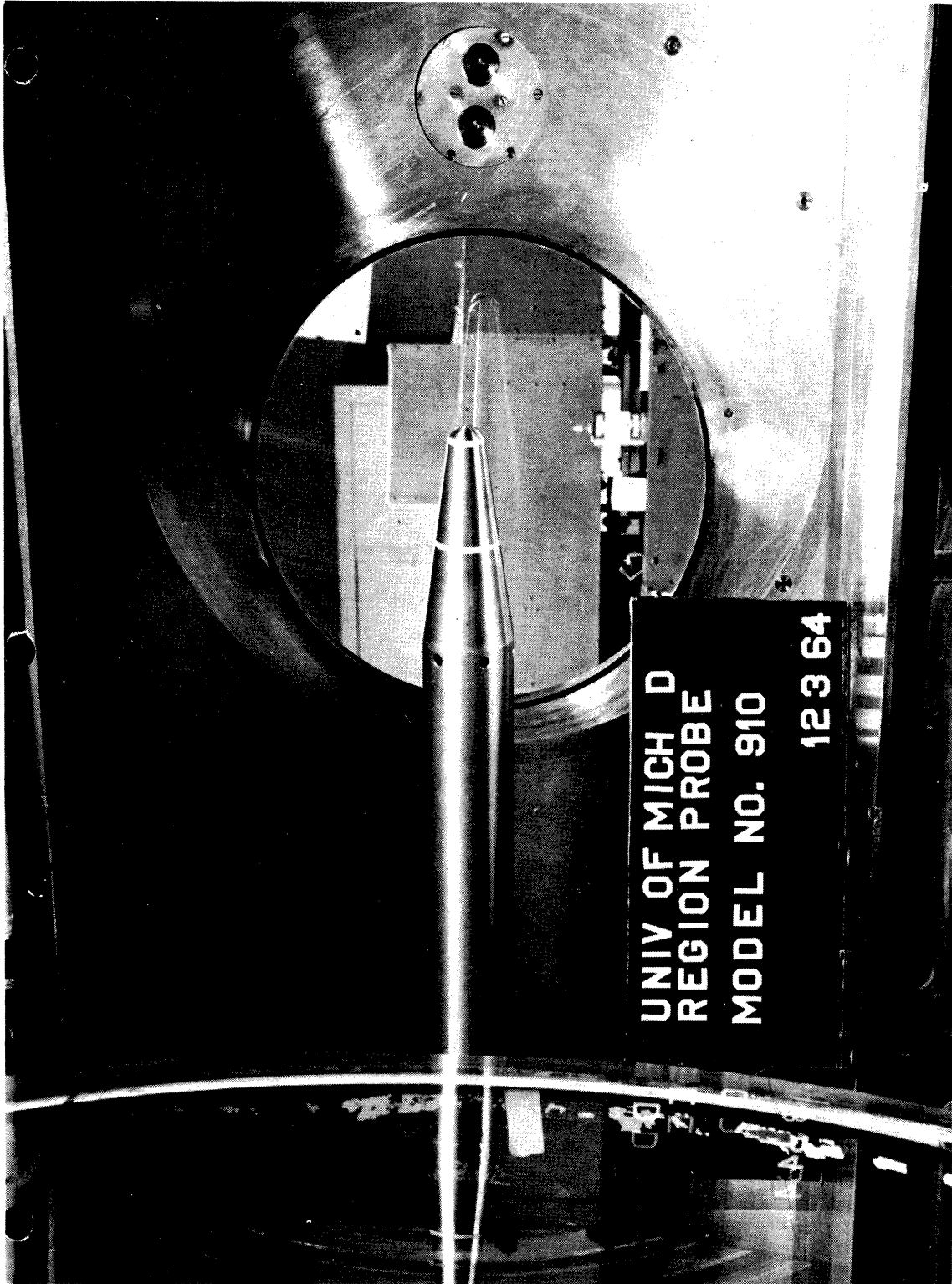


Fig. 2. Probe installed for wind tunnel test.

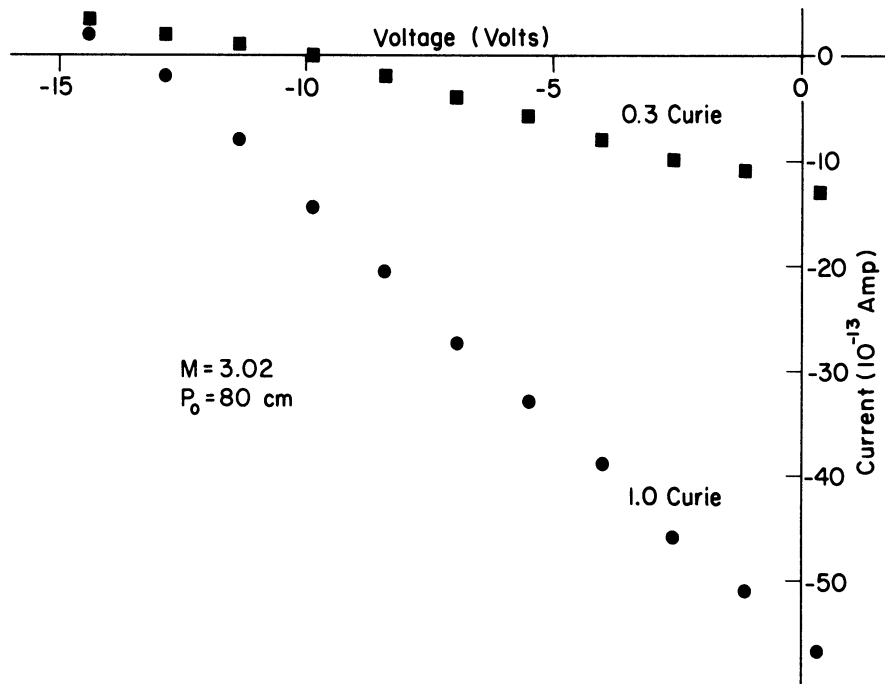


Fig. 3. Measured voltage-current characteristic at $M=3.02$ for low range of applied voltages.

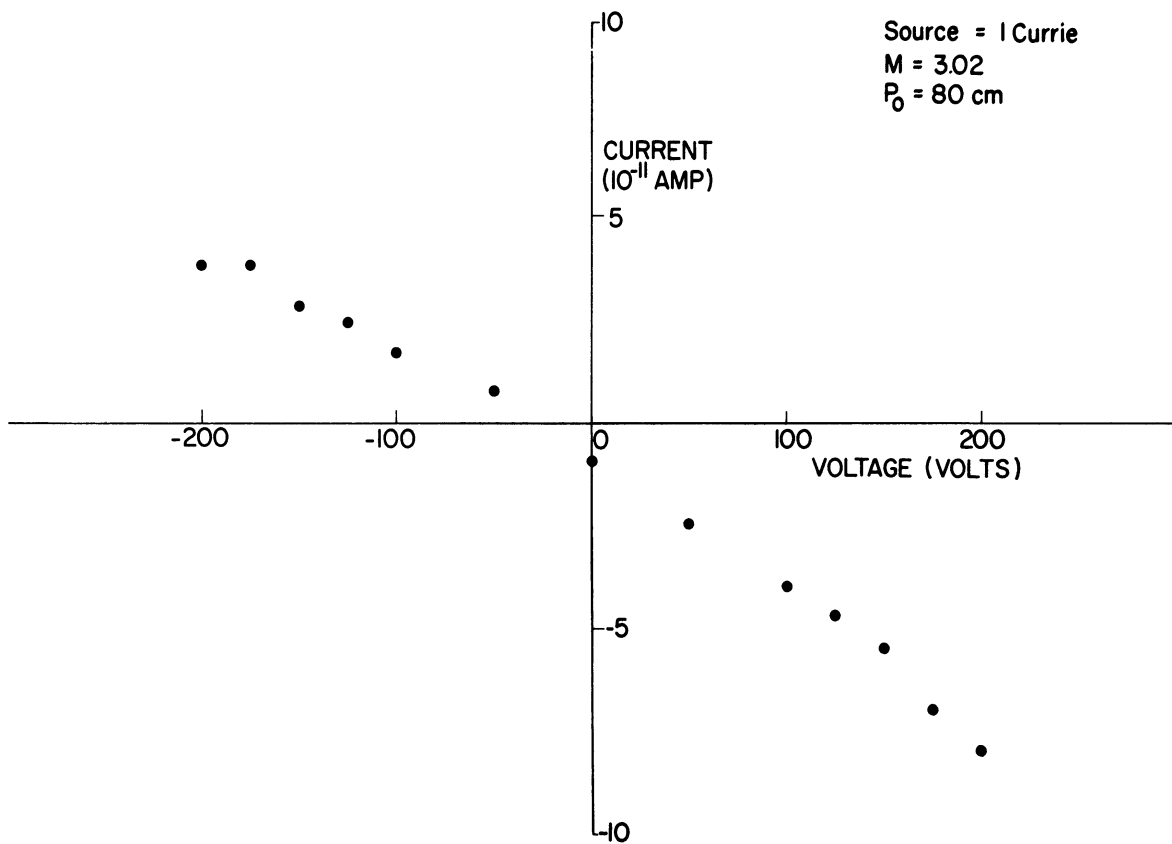


Fig. 4. Measured voltage-current characteristic at $M=3.02$ for high range of applied voltages.

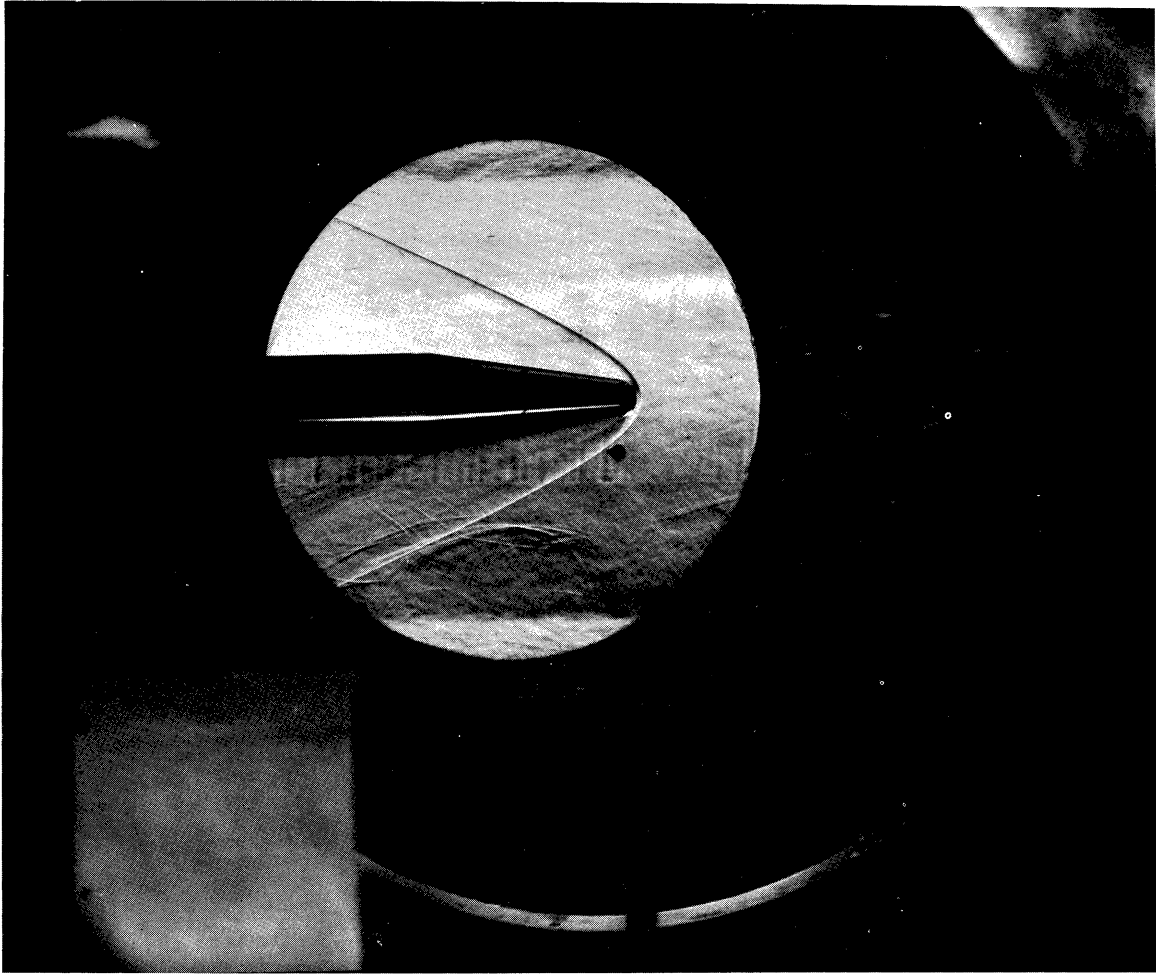


Fig. 5. Schlieren photograph of probe at $M=3.02$.

4. MATHEMATICAL INVESTIGATION OF THE CURRENT COLLECTING

CHARACTERISTICS OF A MOVING PROBE

4.1 THE PHYSICAL PROBLEM

To formulate a theoretical treatment for the collection characteristics of the probe discussed in Section 3, we consider a rounded electrode moving with supersonic velocity through a dense neutral gas containing a low density plasma. The principal features of the physical environment near such a body are indicated in Fig. 6.

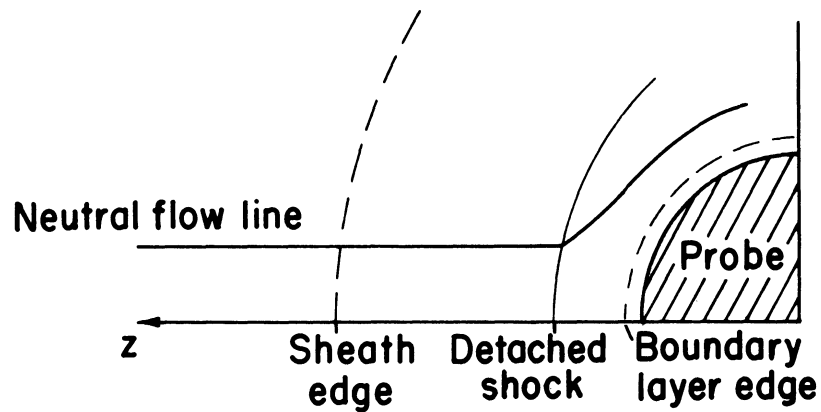


Fig. 6. Representation of the physical environment in front of a supersonic probe.

Here the flow of the nonionized component of the gas is uniform outside the shock, changes to a region of compressible flow behind this, and passes over into the boundary layer near the probe surface.

Significant electrical effects on the ionized components of the flow are restricted to a region around the probe designated the sheath. The scale of this region of electrical influence is determined mainly by the temperatures and densities of the charged particle components and by the potential of the probe relative to the plasma; for sufficiently low densities and/or high probe potentials, this region may extend well beyond the shock. Consequently, within the sheath, strong ionic currents may be created, resulting in a net current to the probe.

Our concern in the following work is to relate the charged particle densities in the undisturbed plasma to the current collected by the probe as a function of the probe voltage.

4.2 MATHEMATICAL FORMULATION

Since we wish to apply our results to data from a probe in the D-region of the ionosphere, the following assumptions regarding the gas are considered justified:

- a. The degree of ionization of the gas is small so that the flow of the neutral particle component is unaffected by the ionic currents;
- b. The neutral particle density is high enough so that ion migration is governed by mobility and diffusion;
- c. The shock is nonionizing and the flow is frozen;
- d. The entire system is in a steady state;
- e. The Einstein relation between mobility and diffusion constants is valid;
- f. effects of magnetic fields are negligible;
- g. Each ionic species and the neutral gas may be considered to have a well defined temperature.

The following two further assumptions are made to simplify calculations, though neither is essential in principle to the development:

- a. Only positive and negative ions of equal masses are present in the plasma;
- b. The plasma components are in thermal equilibrium with the neutral component.

Under these assumptions we may consider the system governed by the equations of charge conservation for each species and by the Poisson equation, written respectively:

$$\begin{aligned}
 \nabla' \cdot (\bar{V}' \sigma' - K \sigma' \nabla' \Phi' - D \nabla' \sigma') &= 0 \\
 \nabla' \cdot (\bar{V}' \zeta' + K \zeta' \nabla' \Phi' - D \nabla' \zeta') &= 0 \\
 \nabla' \cdot \nabla' \Phi' &= \frac{\zeta' - \sigma'}{\epsilon} .
 \end{aligned}
 \tag{4.2-1}$$

The boundary conditions for the system are:

on the probe surface^{2,5}

$$\sigma' = \zeta' = 0, \quad \Phi' = \Phi'_p; \quad (4.2-2)$$

at great distances from the probe:

$$\sigma' = \sigma'_\infty, \quad \zeta' = \zeta'_\infty, \quad \Phi' = 0.$$

Written in terms of the appropriate dimensionless variables, the constants K , D and ϵ can be absorbed (see nomenclature) and the system of differential equations may be written:

$$\begin{aligned} \nabla \cdot (\bar{\nabla} \sigma - \sigma \nabla \Phi - \nabla \sigma) &= 0 \\ \nabla \cdot (\bar{\nabla} \zeta + \zeta \nabla \Phi - \nabla \zeta) &= 0 \\ \nabla \cdot \nabla \Phi &= \zeta - \sigma, \end{aligned} \quad (4.2-3)$$

while the boundary conditions become:

on the probe surface

$$\sigma = \zeta = 0, \quad \Phi = \Phi_p, \quad (4.2-4)$$

at large distances from the probe

$$\sigma_\infty = \zeta_\infty = 1, \quad \Phi = 0$$

Throughout the remainder of this report all quantities will be considered in dimensionless form.

4.3 DISCUSSION OF PROBLEM

The problem as formulated is complicated in particular by the three flow regimes of the neutral gas which must be considered. Up to the present time, theoretical work has been done only on certain aspects of the problem as a whole.

Talbot¹ considers a small planar probe at the stagnation point of a blunt body in a low density supersonic flow. He assumes that the sheath is entirely within the boundary layer and that its thickness is much less than an ionic mean free path.

Chung² and Pollin³ treat a similar problem, but consider ion motion in the sheath to be governed by mobility and diffusion. Again, the sheath is

taken as imbedded in the boundary layer. In addition, Pollin considers ionization caused by a strong shock.

At the other extreme, Cohen,⁴ and Su and Lam⁵ have given the continuum treatment for a stationary sphere.

A more general treatment for an arbitrary low velocity probe in a continuum plasma has been given by Lam.⁶ Considering only incompressible flow of the neutral gas, a complete solution to the problem is provided, making no assumptions about the range of the electrical effects of the probe.

Unfortunately the various simplifying assumptions made in the above works render their results inextensible to the problem here considered. Indeed, an exact analytical solution to the entire problem seems out of the question. For this reason, this report describes work done toward its solution by a numerical method.

5. THE "FLOW LINE" METHOD

5.1 OUTLINE OF METHOD

The object of this method is to determine the flow of the plasma components by a series of approximations to the charge and potential distributions about the probe. Starting with the potential field of the probe (taken as a spherical conductor) in free space, flow lines of small elements of the plasma are obtained. Considering the resulting space charge distribution, a new potential field is determined which is in turn used to find new flow lines. The process is to be continued until the variations in succeeding approximations are considered sufficiently small. The current to the probe can then be obtained by calculating the current in the flux tubes containing all flow lines which terminate on the probe surface.

To determine the desired flow lines we apply the divergence Eqns. (4.2-3), to current flux tubes. These equations may be interpreted as expressing the constancy of current in any given tube. From these, we take the current densities for positive and negative species:

$$\begin{aligned}\bar{J} &= (\bar{V}-\nabla\Phi)\sigma-\nabla\sigma \\ -\bar{L} &= (\bar{V}+\nabla\Phi)\zeta-\nabla\zeta.\end{aligned}\tag{5.1-1}$$

Here the velocity field \bar{V} of the neutral gas is considered known; the flow ahead of the shock is uniform, while that between probe and shock can be determined numerically,^{7,8,9} and entered in the computer as a table.

In the supersonic region of the flow the convective contribution in Eqns. (5.1-1) will dominate the others. Lam⁶ has shown, by his "outer" solution, that in such a region the charge densities are uniform, while the potential is harmonic. Thus, we can expect the diffusive terms in Eqns. (5.1-1) to be smaller than the mobility terms except near the probe. Consequently, as a first approximation, we can consider large diffusive effects to be confined to the region behind the shock where the neutral flow is most complex.

This suggest that the program be constructed in two phases:

- I. The determination of flow lines in a region of uniform neutral flow, neglecting diffusive effects entirely;
- II. Extension of the above to include the flow region behind the shock and density gradient terms over the entire flow field.

This breakdown has several advantages:

- a. The phase I program alone should provide good estimates of the actual flow lines in the region considered. Further because the shock standoff distance is small a reasonable estimate of expected probe current can then be obtained for sufficiently large probe potentials simply by extending the flow lines for the attracted ion species down to the probe surface and neglecting the retarded species altogether.
- b. By excluding diffusive contributions ahead of the shock analytical solutions for the initial flow lines can be obtained. Comparing these with the corresponding results of the phase I program, the error in the computed flow lines can be estimated;
- c. To extend the phase I program to cover the entire flow field (phase II), two essentially similar interpolation routines must be introduced (see Section 8.1). Upon their insertion into the phase I program the general flow line program is completed.

This report deals only with the phase I program.

5.2 DETERMINATION OF FLOW LINES---PHASE I

In the following we treat flow lines of positive particles only; the development for negative species parallels this exactly with appropriate adjustments of signs.

Neglecting the diffusive term in the expression for the positive ion current density in Eqn. (5.1-1) we have:

$$\bar{J} = (\bar{V} \cdot \nabla \Phi) \sigma. \quad (5.2-1)$$

From this, the charges can be considered to move as the result of a composite velocity term:

$$\bar{v} = \bar{V} \cdot \nabla \Phi. \quad (5.2-2)$$

Integrating this over a small but finite time increment δt we have:

$$\delta \bar{l} = (\bar{V} \cdot \nabla \Phi) \delta t + O(\delta t^2) \quad (5.2-3)$$

where $\delta \bar{l}$ is the change in position of the charge element during time δt .

Taking a cylindrical coordinate system (r, ϕ, z) with origin at the center of the probe and z -axis directed into the flow, the neutral flow vector and potential gradient become respectively:

$$\bar{V} = -V \hat{z} \quad (5.2-4)$$

and

$$\nabla\Phi = \frac{\partial\Phi}{\partial r} \hat{r} + \frac{\partial\Phi}{\partial z} \hat{z}, \quad (5.2-5)$$

where \hat{r} and \hat{z} are unit vectors in the r and z directions respectively.

Knowing the potential field (see Section 5.4) the gradients in Eqn. (5.2-5) are found in the program by using a difference method.* Desiring the gradients at some point P, and taking a difference parameter Δ (see Fig. 7) we have:

$$\begin{aligned} \left. \frac{\partial\Phi}{\partial r} \right|_P &= \frac{\Phi_4 - \Phi_2}{2\Delta} + O(\Delta^2) \\ \left. \frac{\partial\Phi}{\partial z} \right|_P &= \frac{\Phi_1 - \Phi_3}{2\Delta} + O(\Delta^2) \end{aligned} \quad (5.2-6)$$

where Φ_i represents the potential at a point i at a distance Δ along lines parallel to the coordinate axes.

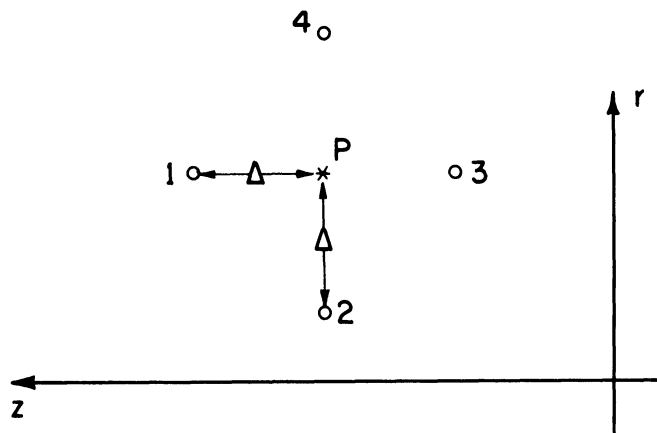


Fig. 7. The difference scheme for potential gradient calculations.

The quantity Δ is chosen to be consistent with the point spacings in the trajectory field. Starting points for the trajectories (taken along a line of constant z far ahead of the probe) are spaced so that:

$$\delta r = V \delta t \quad (5.2-7)$$

making the r and z spacings of the trajectory points of the same order. Δ is

*The only variation in this procedure occurs in the determination of the initial trajectory field. In this case an analytical expression for Φ is assumed (see part 6).

then written as:

$$\Delta = H V \delta t \quad (5.2-8)$$

where H is an input constant, usually taken as 0.5. Thus, Δ is of order δt .

Consequently, using Eqns. (5.2-6), Eqn. (5.2-3) gives us:

$$\delta \bar{l} = \left[-\frac{\Phi_4 - \Phi_2}{2\Delta} \hat{r} - \left(V + \frac{\Phi_1 - \Phi_3}{2\Delta} \right) \hat{z} \right] \delta t, \quad (5.2-9)$$

which is accurate to terms in δt^2 . Indeed, since the error incurred through the approximation (5.2-6) will contribute only a term in δt^3 to the above, the second order error term arises only from Eqn. (5.2-3), and has the form:

$$\bar{\epsilon} = (\bar{v} \cdot \nabla) \bar{v} \frac{\delta t^2}{2}. \quad (5.2-10)$$

To obtain the flow lines, Eqn. (5.2-9) is applied at a given point to determine the next. Reapplying the equation at the new point another is found, and the process is continued until the flow line is terminated. If one considers the flow line to be generated over a fixed time interval $t_n = n\delta t$, while letting n increase (by taking successively finer and finer steps) then the accumulated error behaves as

$$\bar{\epsilon}_n = \frac{1}{n} (\bar{v} \cdot \nabla) \bar{v} \frac{t_n^2}{2}, \quad (5.2-11)$$

so that the total error over the given time interval decreases inversely as the number of steps in the interval.

Because of limited storage facilities in the computer, the region of study must be restricted to a finite volume of space about the probe. Consequently, a region is considered which is bounded in the z direction by two planes normal to the z axis, one ahead of and one behind the probe, and radially by imposing a certain maximum on the initial radial coordinates of the flow lines. The positions of these planes and the value of the maximum radius are obtained below.

The initial points for the flow lines can be chosen by considering the flow along the axis. Here $r \equiv 0$, and the initial z coordinate may be determined by requiring that the mobility term $(\nabla\Phi)$ resulting from the starting potential be a certain fraction p of the free stream velocity V:

$$p = \frac{|\nabla\Phi|}{V}. \quad (5.2-12)$$

The initial r coordinate for the flow line farthest from the axis may then be taken as, say, twice the value of z so determined, to allow for the greater range of influence of the potential field normal to the flow. Since the terminal value of z is fixed by the parameters of the problem, as explained below, the value of p is chosen to make the total number of points on the flow lines compatible with the available storage space in the computer.

The terminal z coordinate of the flow line, z_{\min} , is determined by other means. In using the first approximation to the potential, it is found that the flow lines of the repelled charge species are deflected so as never to close in again behind the probe. Since this does not occur with the flow lines of the attracted species, the result is an infinite sheath trailing behind the probe. This is clearly nonphysical, and its elimination provides the criterion used for terminating the flow lines.

On physical grounds the probe must cause a disturbance in the plasma charge distribution which is commensurate with its own charge. Thus, the charge imbalance in the sheath must be equal to the charge residing on the sphere:

$$Q_{\text{sphere}} = - Q_{\text{sheath}} \quad (5.2-13)$$

Satisfying this condition requires that the charge "tail" behind the probe be cut off at a distance which can be calculated (see Section 6.4). This also restores charge neutrality in the plasma at infinity, as required by the boundary conditions.

5.3 DETERMINATION OF CHARGE DENSITIES

Knowing the flow lines, the variations in charge density along them can be obtained.

Since the flow far ahead of the probe is uniform each flow line in that region is parallel to and is characterized by its distance from the z axis. Let this characteristic distance be r_0 .

Recalling the symmetry of the problem, each flow line is recognized as a section of a surface of revolution about the axis. Two such surfaces whose characteristic distances (radii) differ by an infinitesimal amount, dr_0 , form a flux tube. By Gauss' Theorem and the first of Eqns. (5.1-1) we may write for such a tube:

$$\int_{\tau} \nabla \cdot \bar{J} \, d\tau = \int_A \bar{J} \cdot d\bar{A} = 0. \quad (5.3-1)$$

Here we consider a volume τ obtained by twice slicing the flux tube transversely. The only portions of the surface of this volume (denoted A) which contribute to the above integral are these transverse surfaces. For convenience we take these surfaces as being everywhere orthogonal to the trajectories. Considering one such surface infinitely far upstream of the probe, and the other a finite distance from it, we may write Eqn. (5.3-1):

$$J_{\infty} dA_{\infty} = J dA = I = \text{constant}, \quad (5.3-2)$$

where I is the current in the flux tube and the scalar product is neglected because J and dA are parallel.

By symmetry and using Eqns. (5.2-1) and (5.2-2), Eqn. (5.3-2) may be written:

$$I = \sigma_{\infty} V 2\pi r_0 dr_0 = \sigma v 2\pi r ds, \quad (5.3-3)$$

where ds is the distance between the two "flow line surfaces" at the point (r,z) (see Fig. 8). Since v is known, only ds needs to be determined in order to find the charge densities.

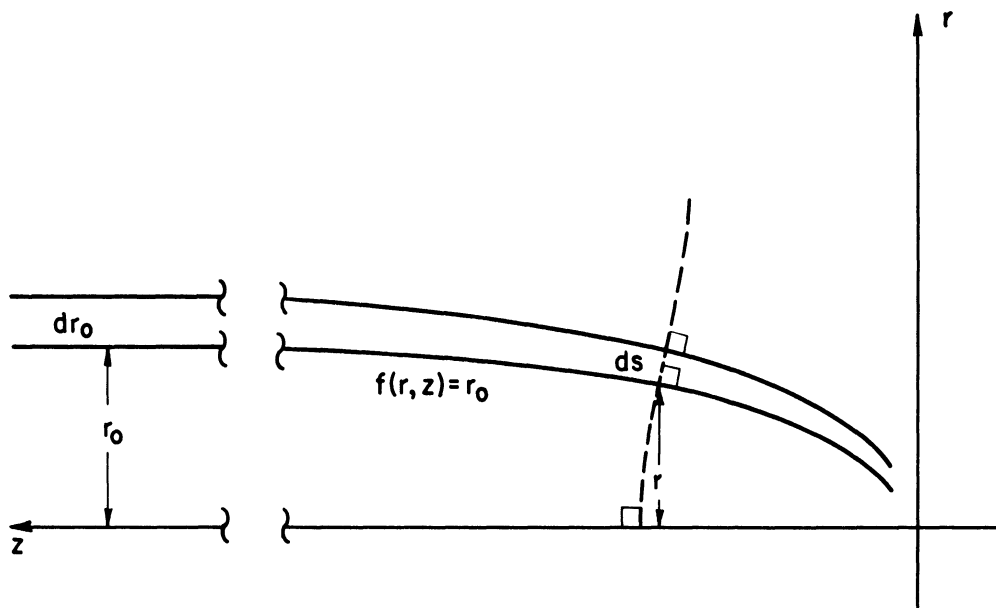


Fig. 8. Parametrization of flow lines.

The equation of a given flow line may be written as:

$$z = z(r, r_0), \quad (5.3-4)$$

or, solving for r_0 in terms of r and z :

$$r_0 = f(r, z). \quad (5.3-5)$$

Considering f as a function of its two independent variables and taking its gradient, we have:

$$|\nabla f| = \frac{df}{ds} \quad (5.3-6)$$

where ds is the magnitude of an elemental displacement taken normal to the curves of constant f . From Eqn. (5.3-5) $df \equiv dr_0$, and since we have taken our slicing surfaces normal to the flow lines, the ds of Eqn. (5.3-6) is seen to be the quantity we seek:

$$ds = \frac{dr_0}{|\nabla f|} \quad (5.3-7)$$

Thus, to determine the charge densities from Eqn. (5.3-3) it is sufficient to know the equations of the flow lines and the local flow velocity v .

In the phase I program the densities are not needed explicitly, but are used implicitly in calculating the potential.

5.4 CALCULATION OF POTENTIAL

In order to make use of Eqn. (5.2-9), it is necessary to know the potential distribution in the space surrounding the probe. This field will be governed by the Poisson Equation.

$$\nabla^2 \Phi = \zeta - \sigma \quad (5.4-1)$$

and the boundary conditions.

$$\begin{aligned} \Phi &= \Phi_p && \text{at the probe} \\ \Phi &\rightarrow 0 && \text{at large distances from the probe} \end{aligned} \quad (5.4-2)$$

Assuming the charge densities known, the potential at any point P may be expressed in terms of the probe potential Φ_p and integrals over the space charge distribution:

$$\Phi(P) = \Phi_p \frac{a}{R(P)} - \frac{1}{4\pi} \int \frac{dq}{R_1} + \frac{1}{4\pi} \int \frac{dq'}{R_2} \quad (5.4-3)$$

where

$$dq = (\sigma - \zeta) d\tau \quad (5.4-4)$$

$$dq' = \frac{a}{R} dq, \quad (5.4-5)$$

and a , R , R_1 , R_2 , $R(P)$ are the nondimensionalized distances shown in Fig. 9.

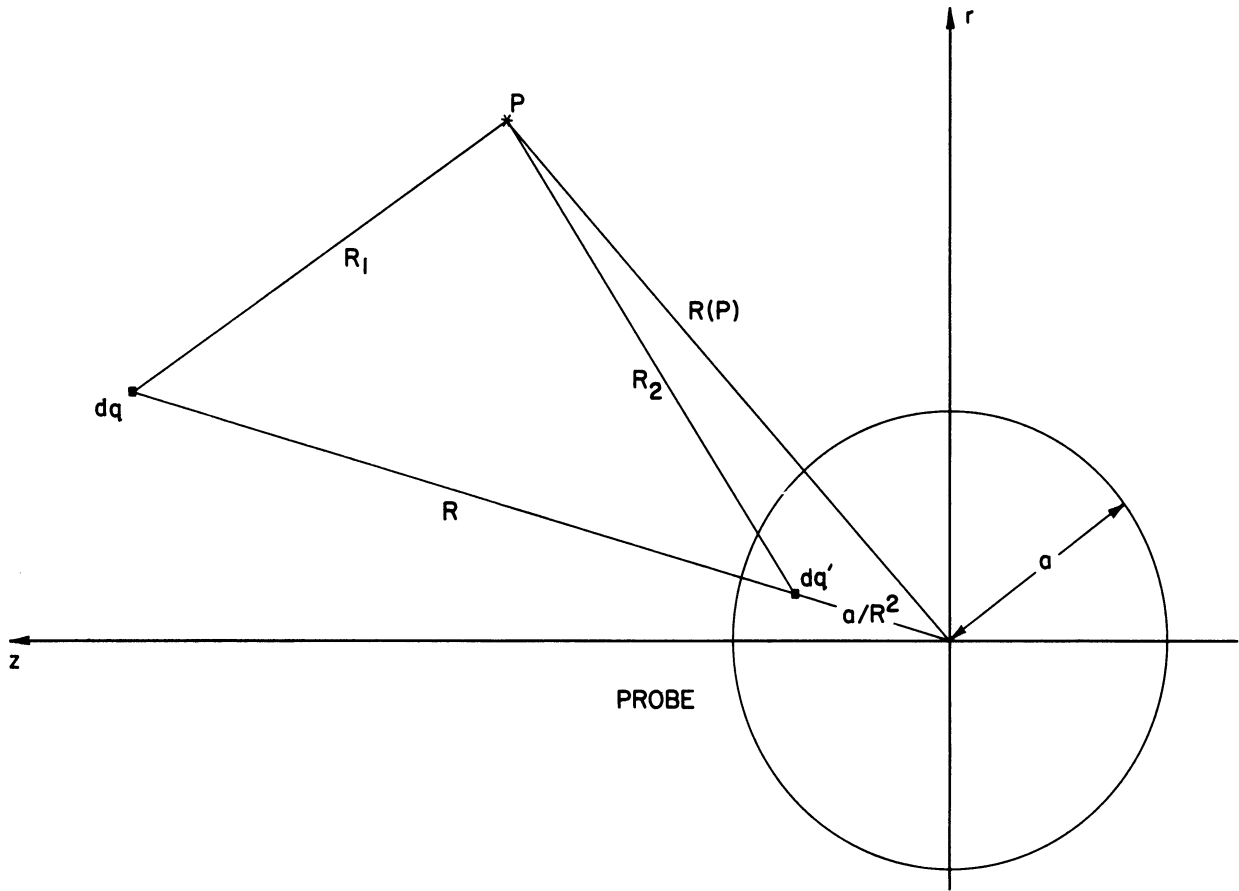


Fig. 9. Variables used in potential integration.

The terms in Eqn. (5.4-3) may be identified as the potentials at P due to the probe, the space charge distribution, and the surface charge induced on the sphere respectively.

In order to use this equation in the program for the calculation of potential, the integrals must be reduced to sums over the points composing the flow lines.

We consider the integrals over the two charge densities in Eqn. (5.4-3) separately. For either species, the flow lines and the corresponding family of "slicing" surfaces we have chosen form an orthogonal net over the r, z plane. The differential volume element may then be conveniently expressed as:

$$d\tau = r \, dl \, ds \, d\phi, \quad (5.4-6)$$

where l and s measure distances along the flow lines and the orthogonals, respectively, and r is the cylindrical radial coordinate of the volume element. Further, the elemental distance along a flow line may be expressed as:

$$dl = v \, dt. \quad (5.4-7)$$

Combining Eqns. (5.3-3), (5.4-6), and (5.4-7) we obtain:

$$\begin{aligned} \sigma d\tau &= \alpha_{\infty} \frac{V r_0 dr_0}{v r ds} \cdot r v dt ds d\phi \\ &= \alpha_{\infty} V dt r_0 dr_0 d\phi, \end{aligned} \quad (5.4-8)$$

so that the integrals of interest take the form:

$$\int \rho \frac{d\tau}{R_i} = V \rho_{\infty} \int \frac{r_0 dt dr_0 d\phi}{R_i} \quad (5.4-9)$$

where integration extends along a flow line (dt), over all flow lines (dr_0) and around the symmetry axis ($d\phi$). Here ρ and R_i are to be taken respectively as the appropriate charge density and the distance between charge element and field point.

Using the flow lines obtained previously (Section 5.2), Eqn. (5.4-9) is well suited for numerical evaluation. The r_0 of each flow line is known, as is the spacing between flow lines δr_0 . The step size in the flow lines is determined by the input value of δt . Thus, the integrations over r_0 and t in Eqn. (5.4-9) are easily transformed to sums over the points of the flow lines. Only the angular integration need be more thoroughly investigated.

Consider a charge element dq and its image charge in the sphere, dq' , as shown in Fig. 10. Let the potential be desired at the point $P(r, z, 0)$, let

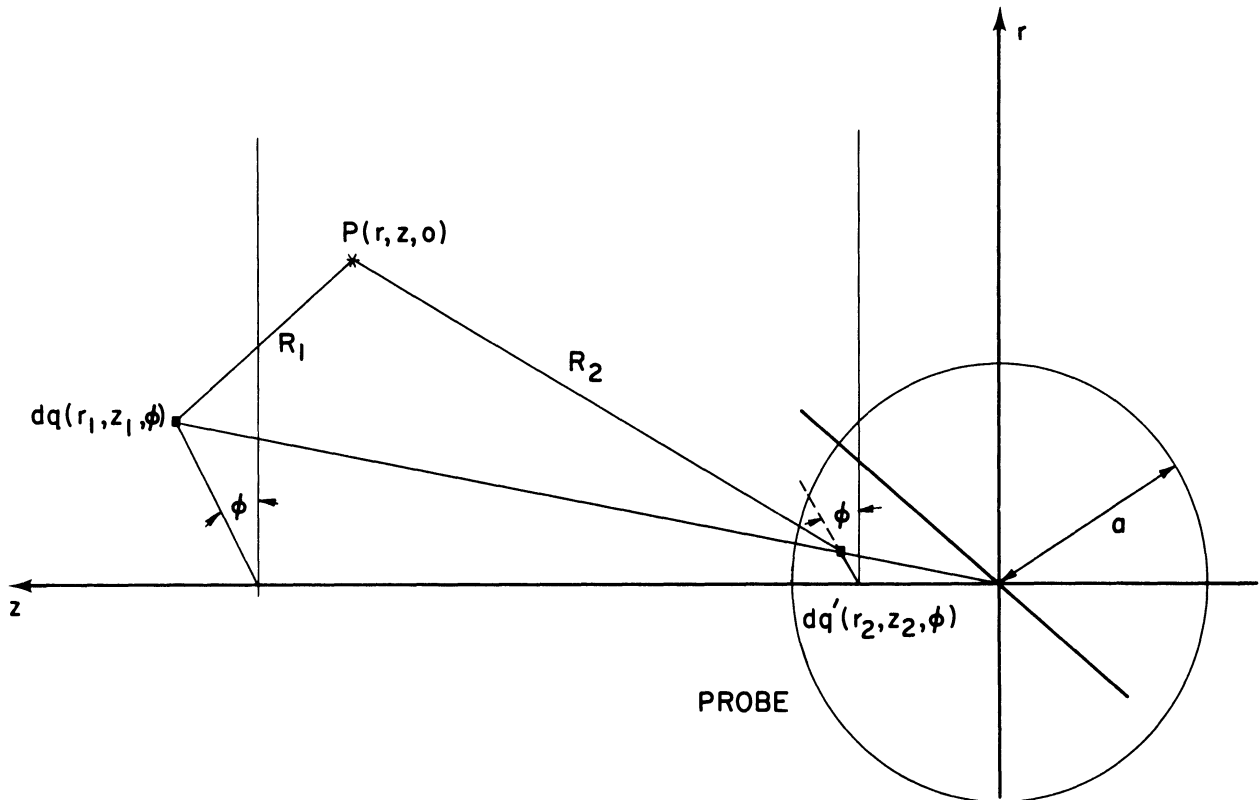


Fig. 10. Variables relating charge with its image charge.

dq be at (r_1, z_1, ϕ) , and let dq' be at (r_2, z_2, ϕ) . It is easily established from the figure that:

$$\begin{aligned} R_i^2 &= (z-z_i)^2 + r^2 + r_i^2 - 2rr_i \cos\phi \\ &= A_i - B_i \cos\phi \quad i = 1,2 \end{aligned} \quad (5.4-10)$$

where A_i and B_i have the obvious definitions. From the position of the image charge (see Fig. 9) it is known that:

$$\begin{aligned} r_2 &= \frac{a^2}{r_1^2 + z_1^2} r_1 \\ z_2 &= \frac{a^2}{r_1^2 + z_1^2} z_1. \end{aligned} \quad (5.4-11)$$

Using relations (5.4-10) and (5.4-11), all integrands in integrals of the form (5.4-9) can be expressed in terms of the coordinates of dq and the point P , i.e., in terms of the flow line point and the field point.

Using the notation of Eqn. (5.4-10) the azimuthal integral in Eqn. (5.4-9) takes the form

$$\int \frac{d\phi}{R_i} = \frac{1}{\sqrt{A_i}} \int_0^{2\pi} \frac{d\phi}{\sqrt{1 - \alpha_i \cos\phi}}. \quad (5.4-12)$$

where

$$\alpha_i = \frac{B_i}{A_i} \quad 0 \leq \alpha_i \leq 1. \quad (5.4-13)$$

The integral on the right side of Eqn. (5.4-12), may be expressed in terms of the complete elliptic integral of the first kind:

$$\begin{aligned} \int_0^{2\pi} \frac{d\phi}{\sqrt{1 - \alpha_i \cos\phi}} &= \frac{4}{\sqrt{1 + \alpha_i}} \int_0^{\pi/2} \frac{d\zeta}{\sqrt{1 - \frac{2\alpha_i}{1 + \alpha_i} \sin^2\zeta}} \\ &= \frac{4}{\sqrt{1 + \alpha_i}} \kappa \left(\frac{2\alpha_i}{1 + \alpha_i} \right) \end{aligned} \quad (5.4-14)$$

Thus, Eqn. (5.4-12) may be written as:

$$\int \frac{d\phi}{R_i} = \frac{4}{\sqrt{A_i+B_i}} \kappa \left(\frac{2\alpha_i}{1+\alpha_i} \right) \quad (5.4-15)$$

In the program this elliptic integral is evaluated by the series:

$$\kappa \left(\frac{2\alpha}{1+\alpha} \right) = \sum_{n=0}^4 a_n \left(\frac{1-\alpha}{1+\alpha} \right)^n - \ln \left(\frac{1-\alpha}{1+\alpha} \right) \sum_{n=0}^4 b_n \left(\frac{1-\alpha}{1+\alpha} \right)^n, \quad (5.4-16)$$

where¹⁰

$a_0 = 1.386 \quad 294 \quad 361 \quad 12$	$b_0 = .5$
$a_1 = .096 \quad 663 \quad 442 \quad 59$	$b_1 = .124 \quad 985 \quad 935 \quad 97$
$a_2 = .035 \quad 900 \quad 923 \quad 83$	$b_2 = .068 \quad 802 \quad 485 \quad 76$
$a_3 = .037 \quad 425 \quad 637 \quad 13$	$b_3 = .033 \quad 283 \quad 553 \quad 46$
$a_4 = .014 \quad 511 \quad 962 \quad 12$	$b_4 = .004 \quad 417 \quad 870 \quad 12$

(5.4-17)

Evaluating this series in the IBM 7090 computer using roughly 9 significant figures, the results are considered accurate to about one part in 10^7 .

It is noted that the above integral diverges as the logarithm of $(1-\alpha_i)^{-1}$ when $\alpha_i \rightarrow 1$, i.e., as the field point approaches a flow line point. Since this divergence is nonphysical, such situations must be treated by a special method.

In evaluating the potential integrals we obtain Eqn. (5.4-12) by considering the charge associated with a given flow line point to be concentrated on a circle concentric with the symmetry axis, rather than distributed in a toroid of finite volume. Since the potential of such a ring diverges as one approaches it, the misbehavior of our integral is explained. Clearly then, our method breaks down when the field point P enters the torus associated with the flow line point (Fig. 11). This provides a criterion for changing the procedure used for finding the potential due to the toroid in question.

Turning again to Fig. 11, the toroid associated with the point Q will generally have a roughly parallelogram-like cross-section with an area of order $(v\delta t)^2$. Since the exact shape of this figure cannot be determined without greatly increased complexity, it is approximated in the program by a circle about the point Q having area $(v\delta t)^2$. The radius of this circle is then given by:

$$r = \frac{v\delta t}{\sqrt{\pi}} \quad (5.4.18)$$

where the value of $v\delta t$ is taken as the separation of the trajectory point Q and the preceding point S. If a point P falls within distance r of Q, the potential at that point is found in the following manner.

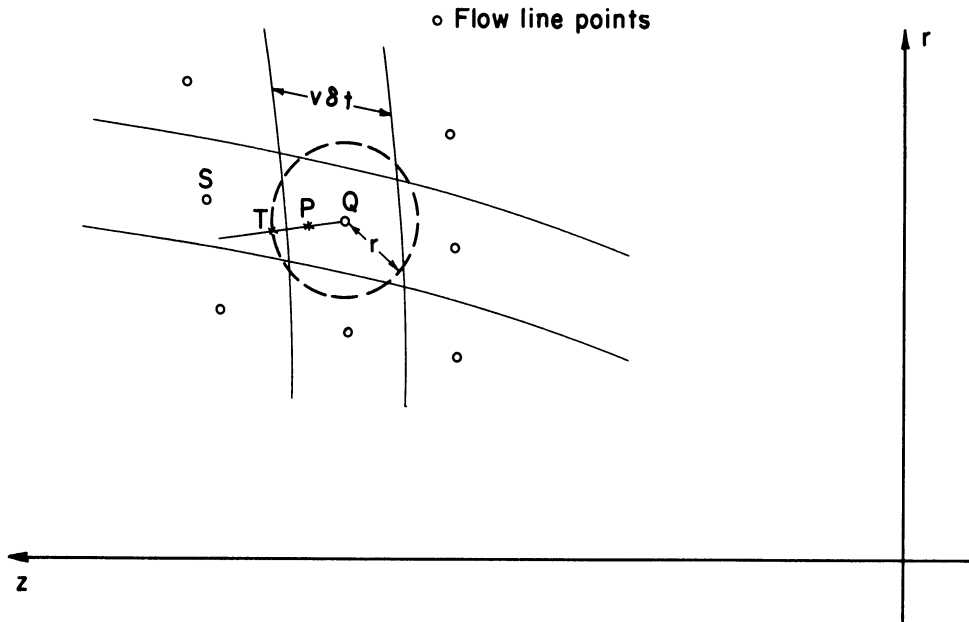


Fig. 11. Section of charge toroid.

The line PQ is constructed and extended through P to intersect the above circle at T. The potential at T is found in the usual manner, and the potential at Q is found as explained below. Then the potential at P is taken as that obtained by linear interpolation between the potentials at Q and T.

To obtain the potential at a point Q on the central circle of the torus, the torus is first approximated by a series of spheres centered on this circle. The potential due to this new charge distribution is taken as an approximation to that of the original toroidal distribution.

The torus is assumed to have major radius R , and minor radius r , given by Eqn. (5.4-18) (see Fig. 12). Measuring from the point Q the axis is cut into segments having length

$$R\delta\phi = \frac{2\pi R}{\left[\frac{\pi R}{r} \right]} \quad (5.4-19)$$

where $[\pi R/r]$ signifies the greatest integer in $\pi R/r$. At the end of each segment determined as above, a sphere of radius r is constructed. Each such sphere is considered to have charge given by:

$$\delta q = \frac{q}{\left[\frac{\pi R}{r} \right]} \quad (5.4-20)$$

where q is the total charge of the complete torus. The flow lines have been generated in such a way that this charge is identically equal to the charge in a torus at the top of the flow line considered, i.e.,

$$q = 2\pi r_0 \delta r_0 V \delta t. \quad (5.4-21)$$

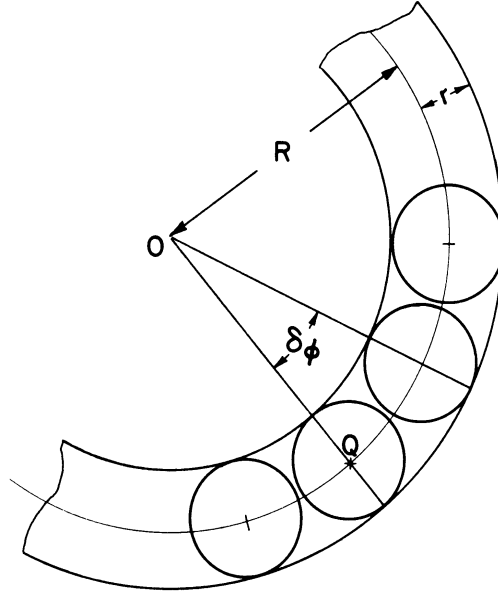


Fig. 12. Construction for integration when field point lies within charge toroid.

Calling

$$\phi_n = n \delta\phi, \quad n = 0, 1, \dots, \left[\pi \frac{R}{r} \right] - 1 \quad (5.4-22)$$

the potential at Q due to this distribution is

$$\Phi = \Phi_0 + \frac{1}{4\pi} \sum_{n=1}^{\left[\pi \frac{R}{r} \right] - 1} \frac{\delta q}{\sqrt{2R^2 - 2R^2 \cos \phi_n}}, \quad (5.4-23)$$

where

$$\Phi_0 = \frac{3 r_0 \delta r_0 V \delta t}{4r \left[\pi \frac{R}{r} \right]} \quad (5.4-24)$$

represents the potential at Q due to the sphere of charge δq centered at Q. Thus the total potential at Q due to the torus is:

$$\Phi = \frac{r_0 V \delta r_0 \delta t}{2\sqrt{2} \left[\pi \frac{R}{r}\right]} \left\{ \frac{3}{r\sqrt{2}} + \frac{1}{R} \sum_{n=1}^{\left[\pi \frac{R}{r}\right]-1} \frac{1}{\sqrt{1-\cos\phi_n}} \right\}. \quad (5.4-25)$$

The above relations now allow us to calculate the potential at any point P due to the charge distribution given by the flow lines. The potential expression (5.4-4) becomes:

$$\begin{aligned} \Phi(P) = & \Phi_P \frac{a}{R(P)} + \frac{V \delta t}{\pi} \left(\sum'_{\zeta} - \sum'_{\sigma} \right) r_0 \delta r_0 \left\{ \frac{\kappa \left(\frac{2\alpha_1}{1+\alpha_1} \right)}{\sqrt{A_1+B_1}} - \frac{a}{R} \frac{\kappa \left(\frac{2\alpha_2}{1+\alpha_2} \right)}{\sqrt{A_2+B_2}} \right\} \\ & + \frac{V \delta t}{2\sqrt{2}} \left(\sum''_{\zeta} - \sum''_{\sigma} \right) \frac{r_0 \delta r_0}{\left[\pi \frac{R}{r}\right]} \left\{ \frac{3}{r\sqrt{2}} + \frac{1}{R} \sum_{n=1}^{\left[\pi \frac{R}{r}\right]-1} \frac{1}{\sqrt{1-\cos\phi_n}} \right\} \end{aligned} \quad (5.4-26)$$

where all symbols are as previously indicated, and \sum'_{σ} , \sum'_{ζ} are sums over

all flow line points of the σ and ζ species respectively except points which fall into the case treated by Eqn. (5.4-25), which are accounted for by the

sums \sum''_{σ} and \sum''_{ζ} . The values of r_0 and δr_0 which are used in the program

are computed from the r coordinates at the tops of the flow line considered and of that with the next larger value of r ; r_0 is taken as the average of these values, and δr_0 as the difference. These calculations are performed for the farthest trajectory from the axis by considering another to be located δr_0 farther out, where δr_0 is taken as the last calculated value of that quantity.

6. INITIAL APPROXIMATIONS USED IN PROGRAMMING THE FLOW LINE METHOD

6.1 INTRODUCTION

In order to start the iterative procedure of the flow line method it is necessary to assume an initial potential field or charge distribution. We assume here an initial potential field: that due to a sphere of potential Φ_p in free space. This is not only reasonable physically but it also enables the flow lines and charge densities of the first iteration to be derived analytically, giving an insight into the operation of the procedure as well as an estimate of the errors involved due to finite step size.

6.2 INITIAL FLOW LINES

In a cylindrical coordinate system the initial potential is given by

$$\Phi = \Phi_p \frac{a}{\sqrt{r^2+z^2}}, \quad r^2+z^2 \geq a^2 \quad (6.2-1)$$

The paths followed by the charge elements will be described by a function $z = z(r, r_0)$ which, for given r_0 , will be everywhere tangent to the local velocity vector \bar{v} , as given by Eqn. (5.2-2). This function is determined by integrating the differential equation:

$$\frac{dr}{dz} = \frac{v_r}{v_z} \quad (6.2-2)$$

where v_r and v_z are the r and z components of the vector \bar{v} . Using Eqns. (5.2-2), (5.2-4), (5.2-5), and (6.2-1), we find these velocity components to be:

$$v_r = \frac{\Phi_p a r}{(r^2+z^2)^{3/2}} \quad (6.2-3)$$
$$v_z = -V + \frac{\Phi_p a z}{(r^2+z^2)^{3/2}} .$$

Substituting these relations in Eqn. (6.2-2), the resulting differential equation is easily integrated. Requiring that $r \rightarrow r_0$ as $z \rightarrow +\infty$ the constant of integration is evaluated, and the flow line equation is found to be:

$$r_0^2 - r^2 = 2\beta \left(\frac{z}{\sqrt{r^2+z^2}} - 1 \right), \quad (6.2-4)$$

where:

$$\beta = \frac{\Phi_p a}{V} . \quad (6.2-5)$$

The equation for flow lines of the negative species is identical to the above except that the sign of the right side is changed.

Taking β positive (positive probe) the flow lines of positive particles are deflected away from the probe, while those of negative particles are attracted, which results in a region where one species is excluded, i.e., a sheath. It can be shown from the equations of the flow lines that this region starts at a distance ahead of the origin equal to $\sqrt{|\beta|}$, that it extends behind the probe an infinite distance, and that it is always of finite width, having a radius of $2\sqrt{|\beta|}$ at an infinite distance behind the probe. For the ionic species which is attracted to the probe, this radius also corresponds to the r_0 of the flow line farthest from the axis which is "collected." In fact, it is easily shown that the flow lines of the positive and negative species are merely the mirror images of one another across the plane $z=0$ (see Figs. 13 and 14).

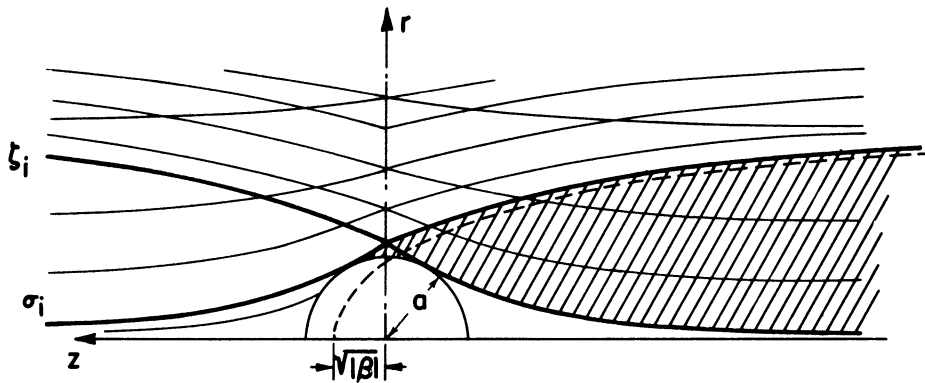


Fig. 13. Flow past sphere, ($\sqrt{|\beta|} < a$).

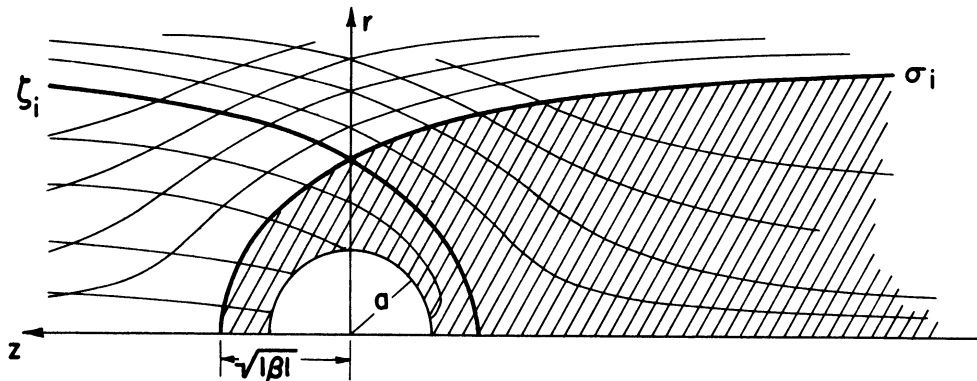


Fig. 14. Flow past sphere, ($\sqrt{|\beta|} > a$).

6.3 CHARGE DISTRIBUTION RESULTING FROM STARTING FLOW LINES

From Section 5.3 we know that the charge density distribution is given by Eqns. (5.3-3), (5.3-5), and (5.3-7). Combining these relations and using the fact that $\alpha_\infty = 1$ we obtain

$$\sigma = \frac{V}{v} \frac{|\nabla r_0^2|}{2r}, \quad (6.3-1)$$

since by Eqn. (5.3-5), $f \equiv r_0$.

Solving Eqn. (6.2-4) for r_0^2 and using Eqns. (6.2-3) we find that

$$\frac{\partial r_0^2}{\partial r} = -2r \frac{v_z}{V} \quad (6.3-2)$$

$$\frac{\partial r_0^2}{\partial z} = 2r \frac{v_r}{V}$$

so that Eqn. (6.3-1) reduces to:

$$\sigma \equiv 1. \quad (6.3-3)$$

A similar relation holds for ζ , so that the charge distribution is everywhere uniform except in the region where one species of particles is excluded. The net density is therefore zero everywhere outside the sheath, and unity inside, where it has the sign opposite to that of the sphere potential.

6.4 TERMINATION OF FLOW LINES

As mentioned in Section 5.2, the nonphysical nature of the infinite trailing sheath provides a criterion for terminating the flow lines. Since both charge densities are uniform everywhere outside the sheath, terminating the flow lines amounts to cutting off the sheath and restoring neutrality to the flow behind it, which also satisfies the boundary condition of charge neutrality at infinity.

Our requirement is that the net charge in the sheath shall be equal in magnitude to the charge on the sphere due to its natural capacitance:

$$Q_{\text{sphere}} = -Q_{\text{sheath}} \quad (6.4-1)$$

Considering the spherical probe to be initially in free space, the non-dimensionalized charge on it due to its own capacitance is easily found to be

$$Q_{\text{sphere}} = 4\pi a n_\infty \Phi_p \quad (6.4-2)$$

Here the charge is normalized with respect to the electronic charge and n_{∞} is the number of charged particles of either species in a cube at infinity having side equal to a Debye length, λ .

To determine the sheath charge for use in Eqn. (6.4-1) we must first find the boundary of the sheath.

Figure 13 indicates some of the flow lines of the first approximation in the vicinity of the probe when $|\beta| < a^2$. We note that the probe intercepts flow lines of both species. Assuming the probe to have a positive potential, the curves labeled σ_1 and ζ_1 represent the flow lines of positive and negative charges respectively which just graze the probe surface. All flow lines of both species having characteristic radii r_0 less than those of the corresponding flow line above are collected; all having values of r_0 greater than those of the above are not. Thus the sheath is the region (shaded) between the line σ_1 , and the line ζ_1 or the probe, whichever is appropriate. In this region there is negative charge only.

To determine the outer edge of the sheath we must find the flow line σ_1 , which has a point of tangency with the probe. Let this line have characteristic radius r'_0 . At the required point of tangency, call it (r', z') , the following conditions are satisfied:

$$r'^2 + z'^2 = a^2, \quad (6.4-3)$$

and:

$$r'_0{}^2 - r'^2 = 2\beta \left(\frac{z'}{\sqrt{r'^2 + z'^2}} - 1 \right). \quad (6.4-4)$$

In addition, the slope of the flow line is equal to that of the sphere here, so that the quantity $\frac{d}{dr} z^2$ is equal for both also. Finding z^2 from Eqns. (6.4-3) and (6.4-4), calculating these derivatives and equating the results we obtain:

$$(4\beta^2 - y^2)^2 = y^4 - 4\beta^2 y^2 + 8\beta^2 r' z y, \quad (6.4-5)$$

where:

$$y = r'_0{}^2 - r'^2 + 2\beta. \quad (6.4-6)$$

Solving the system of Eqns. (6.4-3) through (6.4-6) we obtain

$$r' = \sqrt{a^2 - \frac{\beta^2}{a^2}}$$

$$z' = \frac{\beta}{a} \quad (6.4-7)$$

$$r'_0 = a - \frac{\beta}{a}.$$

The outer boundary of the sheath is thus determined.

The inner boundary of the sheath consists of two parts. Because of the symmetry of the flow lines of the different species about the plane $z=0$, the line ζ_1 of Fig. 13 is tangent to the sphere at the point $(r', -z')$. Thus, the inner boundary of the sheath is the probe surface between the points (r', z') and $(r', -z')$, and is the flow line ζ_1 for all values of $z < -z'$.

The value of r_0 for this flow line, call it r_0'' , can be found rather simply in terms of r_0' . We consider the flow line equation for negative ions:

$$r_0''^2 - r^2 = -2\beta \left(\frac{z}{\sqrt{r^2+z^2}} - 1 \right). \quad (6.4-8)$$

Letting z go to minus infinity and denoting the value of r in this limit by r_∞ we have:

$$r_0''^2 - r_\infty^2 = 4\beta. \quad (6.4-9)$$

Because of the previously mentioned symmetry of the flow lines we know that for the ζ_1 trajectory $r_\infty = r_0'$, so that Eqn. (6.4-9) gives us:

$$r_0''^2 = 4\beta + r_0'^2. \quad (6.4-10)$$

Thus, the boundaries of the charge distribution in the sheath are known.

With this information we can now turn to the calculation of the sheath charge as required in Eqn. (6.4-1). Considering all flow lines terminated at some plane $z = z_{\min}$, the sheath charge may be expressed in terms of integrals:

$$Q_{\text{sheath}} = -\pi n \left\{ \begin{array}{l} \int_{z_{\min}}^{z'} dz \int_{a^2-z^2}^{r_2^2(z)} dr^2 \quad |\beta| < a^2; z' > z_{\min} > -a \\ \int_{-z'}^{z'} dz \int_{a^2-z^2}^{r_2^2(z)} dr^2 + \int_{z_{\min}}^{-z'} dz \int_{r_1^2(z)}^{r_2^2(z)} dr^2 \quad |\beta| < a^2; z_{\min} < -a \end{array} \right. \quad (6.4-11)$$

where n represents the number of ions per Debye cube in the sheath, and $r_1^2(z)$ and $r_2^2(z)$ are obtained from the equations for ζ_1 and σ_1 respectively, as shown below.

Letting

$$\mu = r_0'^2 + 2\beta \quad (6.4-12)$$

and expanding either of the equations for σ_i or ζ_i we obtain a single cubic equation in r^2 :

$$(r^2)^3 + (z^2 - 2\mu)(r^2)^2 + \mu(\mu - 2z^2)(r^2) + z^2(\mu^2 - 4\beta^2) = 0. \quad (6.4-13)$$

For values of the parameters in this equation which are of interest to us ($|\beta| > 0$; $r_0^2, z^2 > 0$) there are three real solutions for r^2 :

$$r^2(k, z) = \frac{2(r_0^2 + z^2 + 2\beta)}{3} \left[1 + \cos \left[\frac{2k\pi + \cos^{-1} \left(\frac{54\beta^2 z^2}{(r_0^2 + z^2 + 2\beta)^3} - 1 \right)}{3} \right] \right] - z^2$$

$$k = 0, 1, 2 \quad (6.4-14)$$

Considering appropriate limits of these solutions we find that:

$$r_1^2 = r^2(0, z)$$

$$r_2^2 = r^2(2, z) \quad (6.4-15)$$

while the solution $r^2(1, z)$ is physically impossible.

In the event that $|\beta| \geq a^2$, Fig. 13 is altered (see Fig. 14). In this case all of the σ flow lines miss the probe entirely. In addition, there is no region behind the probe which is free of charge, so that the range of integration for the sheath charge integral extends from the axis to σ_1 except between $a < z < -a$ where it is bounded on the inside by the sphere.

In this case flow line σ_1 has $r'_0 = 0$, so that μ in Eqn. (6.4-12) reduces to 2β . Equation (6.4-13) then becomes

$$(r^2) [(r^2)^2 + (z^2 - 4\beta)(r^2) + 4\beta(\beta - z^2)] = 0, \quad (6.4-16)$$

and the solutions of interest are:

$$r_1^2(z) = 0$$

$$r_2^2(z) = 2\beta - \frac{z^2}{2} - \frac{z}{2} \sqrt{8\beta + z^2}. \quad (6.4-17)$$

Using these results, the sheath charge integral now takes the forms

$$Q_{\text{sheath}} = -\pi n \left\{ \begin{array}{l} \int_{z_{\min}}^{\sqrt{|\beta|}} dz \int_0^{r_2^2(z)} dr r^2 \quad |\beta| \geq a^2; \quad \sqrt{|\beta|} \geq z_{\min} > a \\ \int_{z_{\min}}^{\sqrt{|\beta|}} dz \int_0^{r_2^2(z)} dr r^2 - \int_{z_{\min}}^a dz \int_0^{a^2-z^2} dr r^2 \quad |\beta| \geq a^2; \quad a > z_{\min} > -a \\ \int_{z_{\min}}^{\sqrt{|\beta|}} dz \int_0^{r_2^2(z)} dr r^2 - \int_{-a}^a dz \int_0^{a^2-z^2} dr r^2 \quad |\beta| \geq a^2; \quad z_{\min} < -a \end{array} \right. \quad (6.4-18)$$

where $r_2^2(z)$ is given by Eqn. (6.4-17).

Thus, depending on the relative magnitudes of a^2 and β , the appropriate form of Eqns. (6.4-11) or (6.4-18) is used with Eqn. (6.4-2) in Eqn. (6.4-1). This yields an equation for z_{\min} which involves only the parameters of the problem; when solved this provides the stopping point for the flow lines.

In the case where $|\beta| < a^2$ the evaluation of the charge integral is extremely difficult because of the complicated nature of the limits r_1^2 and r_2^2 . However, if $|\beta| \ll a^2$ an approximation may be made which reduces the difficulty considerably and the integral may be evaluated. When $|\beta| > a^2$ the sheath charge integral can be evaluated directly. We now consider these cases separately.

If $|\beta| \ll a^2$ ($a \gg \frac{|\Phi_p|}{v}$) the two flow lines determining the sheath will be tangent to the probe very near $(r, z) = (a, 0)$. In the limit of $\Phi_p = 0$ these flow lines will satisfy this condition exactly and will be everywhere parallel to the z -axis. We consider cases near this limit, taking $|\beta|$ to be small.

Demanding that the two flow lines of interest pass through the point $(a, 0)$, the flow line equations give us the relations

$$r_0^2 - a^2 = \pm 2\beta \quad (6.4-19)$$

where the upper sign is valid for flow lines of positive species, the lower for negative species. Substituting these results in the flow line equations we obtain:

$$a^2 - r^2 = \frac{\pm 2\beta z}{\sqrt{r^2 + z^2}} \quad (6.4-20)$$

In the limit we are considering r will not deviate appreciably from its value at $z = 0$, so we set $r = a$ in the radical of Eqn. (6.4-20) to obtain:

$$r^2 = a^2 \mp \frac{2\beta z}{\sqrt{a^2+z^2}} \quad (6.4-21)$$

This gives a good approximation to r^2 for all z , since near $z = 0$, r is very nearly equal to a , and for large negative z the importance of the r^2 in the radical is diminished so that the error introduced by the substitution $r = a$ is very small.

Using Eqn. (6.4-21), the sheath charge integrals (6.4-11) reduce to:

$$Q_{\text{sheath}} \approx -\pi n \int_{z_{\min}}^0 dz \int_{a^2 + \frac{2\beta z}{\sqrt{a^2+z^2}}}^{a^2 - \frac{2\beta z}{\sqrt{a^2+z^2}}} dr^2 \quad z_{\min} \leq 0$$

$$= -4\pi n \beta a \{ \sqrt{1+x^2} - 1 \} \quad (6.4-22)$$

where:

$$x = \frac{z_{\min}}{a} \quad (6.4-23)$$

Substituting Eqns. (6.4-21), (6.4-22), and (6.4-23) into Eqn. (6.4-1) and rearranging terms we have:

$$\frac{\Phi_p}{\beta} = \sqrt{1+x^2} - 1 = f(x) \quad x \leq 0 \quad (6.4-24)$$

which enables us to solve for z_{\min} . We note that the asymptotic form of $f(x)$ for large negative x , call it $f_{\infty}(x)$, is

$$f_{\infty}(x) = -x-1 \quad x \rightarrow -\infty \quad (6.4-25)$$

When $|\beta| > a^2$, we use Eqns. (6.4-18) for the sheath charge. These can be immediately integrated to give

$$Q_{\text{sheath}} = -\frac{\pi n \beta^{3/2}}{6} [t^3 + (8+t^2)^{3/2} - 12t - 16]$$

$$+ \frac{\pi n \beta^{3/2}}{3} \left\{ \begin{array}{ll} 0 & \underline{1} \geq \gamma^2 \quad 1 > t > \gamma \\ t^3 - 3\gamma^2 t + 2\gamma^3 & \underline{1} \geq \gamma^2 \quad \gamma > t > -\gamma \\ 4\gamma^3 & \underline{1} \geq \gamma^2 \quad t < -\gamma \end{array} \right. \quad (6.4-26)$$

where

$$t = \frac{z_{\min}}{\sqrt{|\beta|}}, \quad \gamma = \frac{a}{\sqrt{|\beta|}}. \quad (6.4-27)$$

Substituting Eqns. (6.4-27) and (6.4-26) into Eqns. (6.4-1) and (6.4-2), and rearranging terms we have

$$24\gamma \frac{\Phi_p}{\beta} = g(t) - \begin{cases} 0 & \underline{1} \geq \gamma^2 & 1 > t > \gamma \\ 2h(\gamma, t) & \underline{1} \geq \gamma^2 & \gamma > t > -\gamma \\ 8\gamma^3 & \underline{1} \geq \gamma^2 & t < -\gamma \end{cases} \quad (6.4-28)$$

where

$$g(t) = t^3 + (8+t^2)^{3/2} - 12t - 16 \quad (6.4-29)$$

$$h(\gamma, t) = t^3 - 3\gamma^2 t + 2\gamma^3. \quad (6.4-30)$$

The asymptotic form of $g(t)$ is

$$g_\infty(t) = -24t - 16 \quad t \rightarrow -\infty. \quad (6.4-31)$$

Having obtained solutions of Eqn. (6.4-1) for $|\beta| \geq a^2$ and $|\beta| \ll a^2$ it is of interest to see whether the results can be adapted to the intermediate case of $|\beta| \lesssim a^2$. Considering the relations for x , t and γ (Eqns. (6.4-23) and (6.4-27)) we have

$$t = \gamma x. \quad (6.4-32)$$

Defining:

$$\eta = \frac{\Phi_p}{\beta}, \quad (6.4-33)$$

Eqn. (6.4-24) can be solved for t :

$$\begin{aligned} t &= \gamma f^{-1}(\eta) \\ &= -\gamma \sqrt{(\eta+1)^2 - 1}. \end{aligned} \quad (6.4-34)$$

Turning to Eqn. (6.4-28) we consider values of γ beyond the stated range, i.e., $\gamma > 1$. From the nature of $h(\gamma, t)$ we know that

$$24\eta\gamma \leq g(t) \leq 24\eta\gamma + 8\gamma^3, \quad (6.4-35)$$

and hence, by the monotone property of $g(t)$, that:

$$g^{-1}(24\eta\gamma) \geq t \geq g^{-1}(24\eta\gamma+8\gamma^3). \quad (6.4-36)$$

Using Eqn. (6.4-34) and the inequalities (6.4-36), curves of t vs γ are given in Figs. 15, 16, and 17 for three values of η . The curves have been extended beyond their ranges of definition in γ for purposes of comparison. These indicate that for values of $\eta \geq 10$ the percentage variation between the values of t determined from f and g is quite small in the intermediate range of $1 \leq \gamma \leq 3$. Thus, extension of the use of $g(t)$ to values of up to $\gamma = 1.2$ (see Fig. 15) in the form

$$g(t) = 24\eta\gamma+8\gamma^3 \quad 1 \leq \gamma \leq 1.2 \quad (6.4-37)$$

and the use of $f(x)$ for values of $\gamma > 1.2$ can be considered to give a good approximation for determining z_{\min} in the intermediate range of γ .

The restriction that $\eta \geq 10$ is not a serious one in practical cases of interest, since it will be satisfied by most rocket mounted probes, e.g., considering a probe at an altitude of 70 km with $V' \geq 100$ m/sec, $\Phi'_p = 10$, $a' = 1.3$ cm, $T' = 300^\circ\text{K}$, and $n'_\infty = 1.3 \times 10^4/\text{cm}^3$ of singly ionized O_2 , we find that $\eta \geq 60$.

In order to obtain fast approximate solutions of the z_{\min} equations, graphs of the functions $f(x)$, $g(t)$, and $g(t) - 2h(\gamma, t)$ are given in Figs. 18 through 22. To obtain a graphical solution for z_{\min} the value of $|\beta|$ is calculated from the given data of the problem and compared with a^2 . If $|\beta| \ll a^2$, the parameter on the left side of Eqn. (6.4-24) is computed and the appropriate value of x determined from the graph of $f(x)$ in Fig. 18, or from the asymptotic relation for $f(x)$. Then z_{\min} is found from Eqn. (6.4-23).

If $|\beta| \geq a^2$, the parameter γ is evaluated from Eqn. (6.4-27). This determines a unique "branch" of the $g(t)-2h(\gamma, t)$ curve of Fig. 22. Computing the parameter on the left side of Eqn. (6.4-28), this figure should be consulted to find the appropriate value of t , interpolating between the curves for different values of γ if necessary. In the event that the parameter $24\gamma \frac{\Phi'_p}{\beta}$ is so large that the $g(t)-2h(\gamma, t)$ curves are not extended sufficiently far along the negative t axis to allow solution, $8\gamma^3$ should be added to it, and the value of t read directly from the $g(t)$ curve (Figs. 19, 20, and 21) or obtained from the asymptotic relation (6.4-31).

When the values of a and β are such that γ falls between the limiting cases discussed above, the curves are used in accordance with the results discussed in the paragraph following Eqn. (6.4-36).

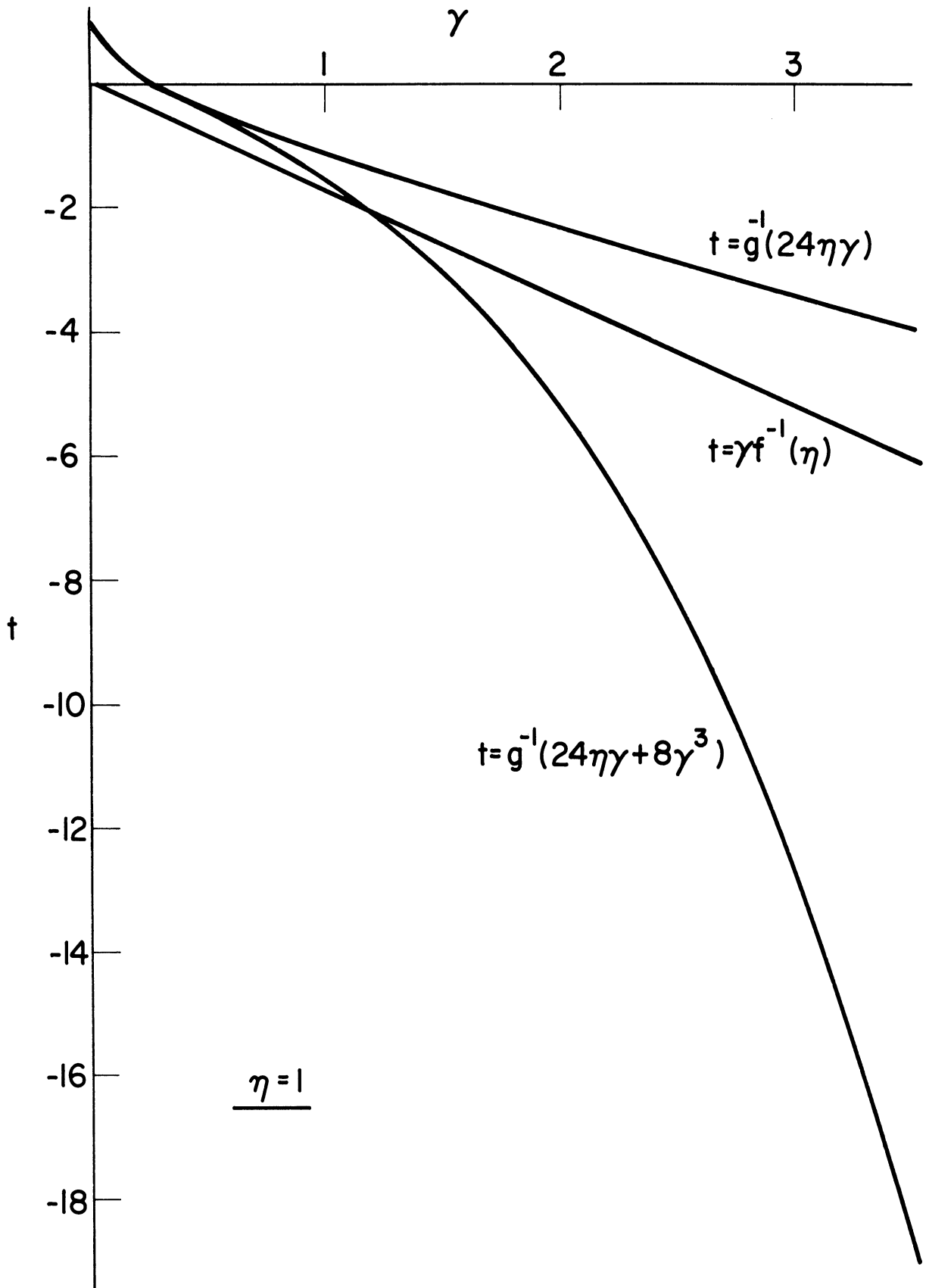


Fig. 15. Curve of t vs γ , ($\eta=1$).

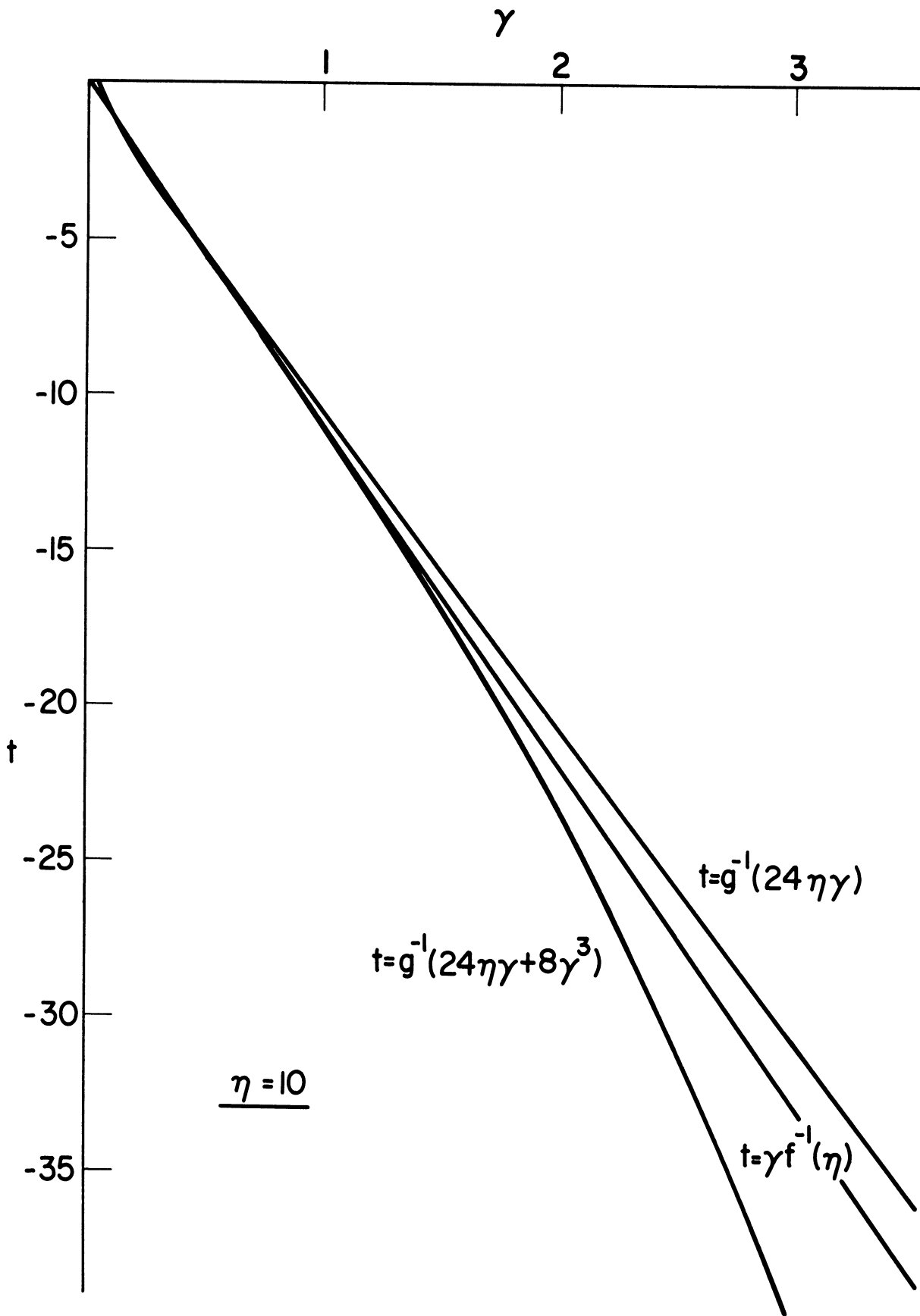


Fig. 16. Curve of t vs γ , ($\eta=10$).

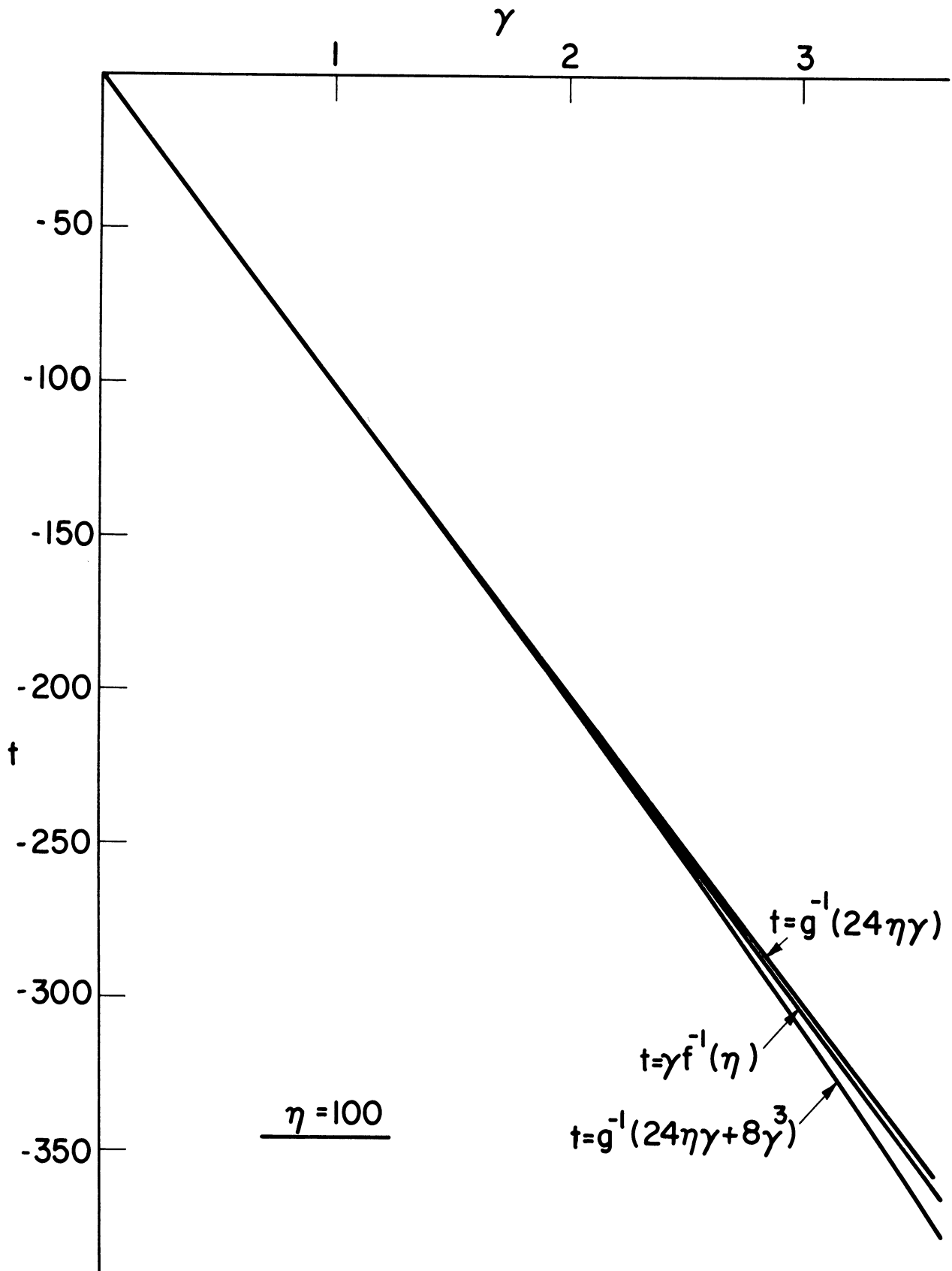


Fig. 17. Curve of t vs γ , ($\eta=100$).

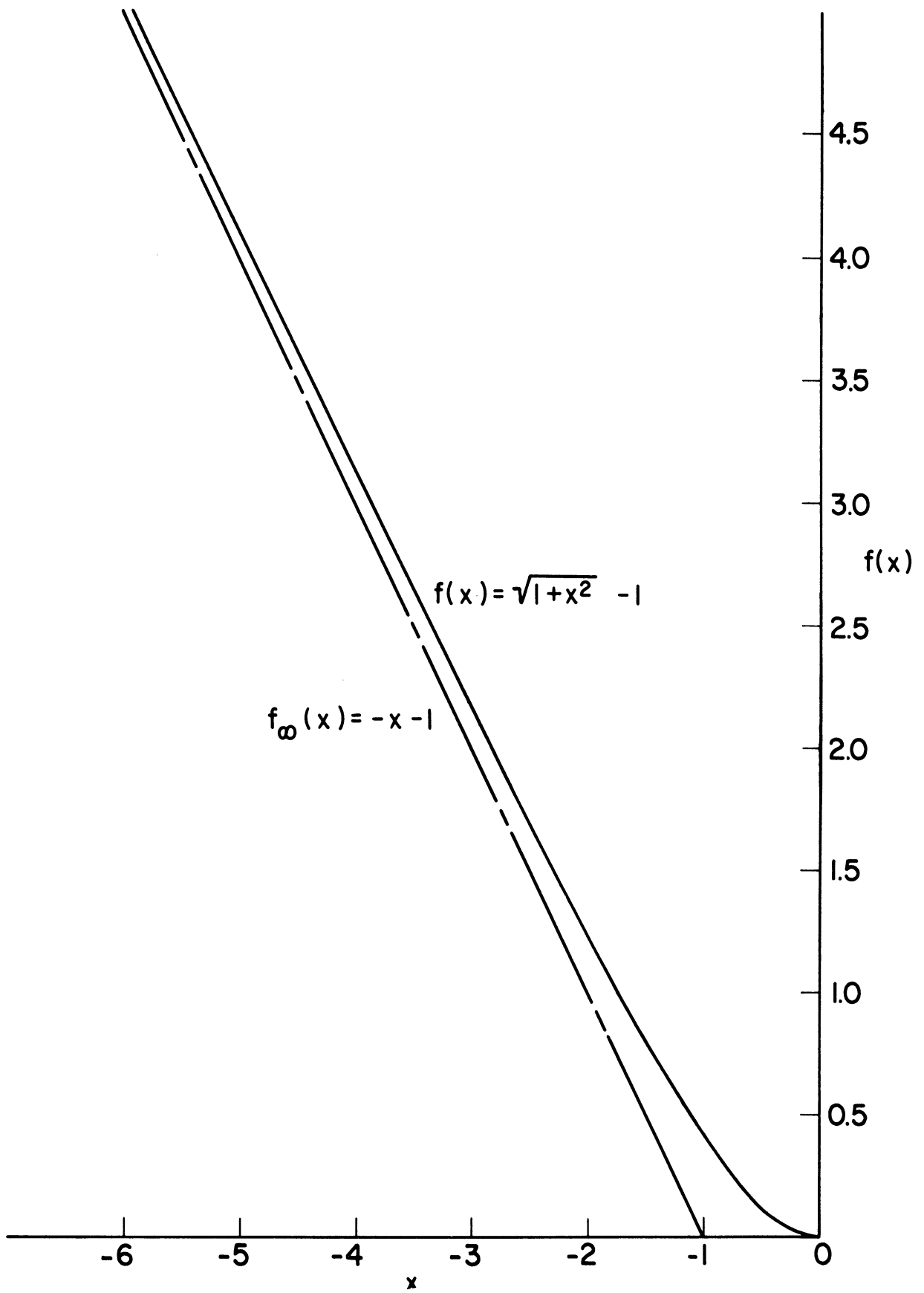


Fig. 18. Curve of $f(x)$ vs x .

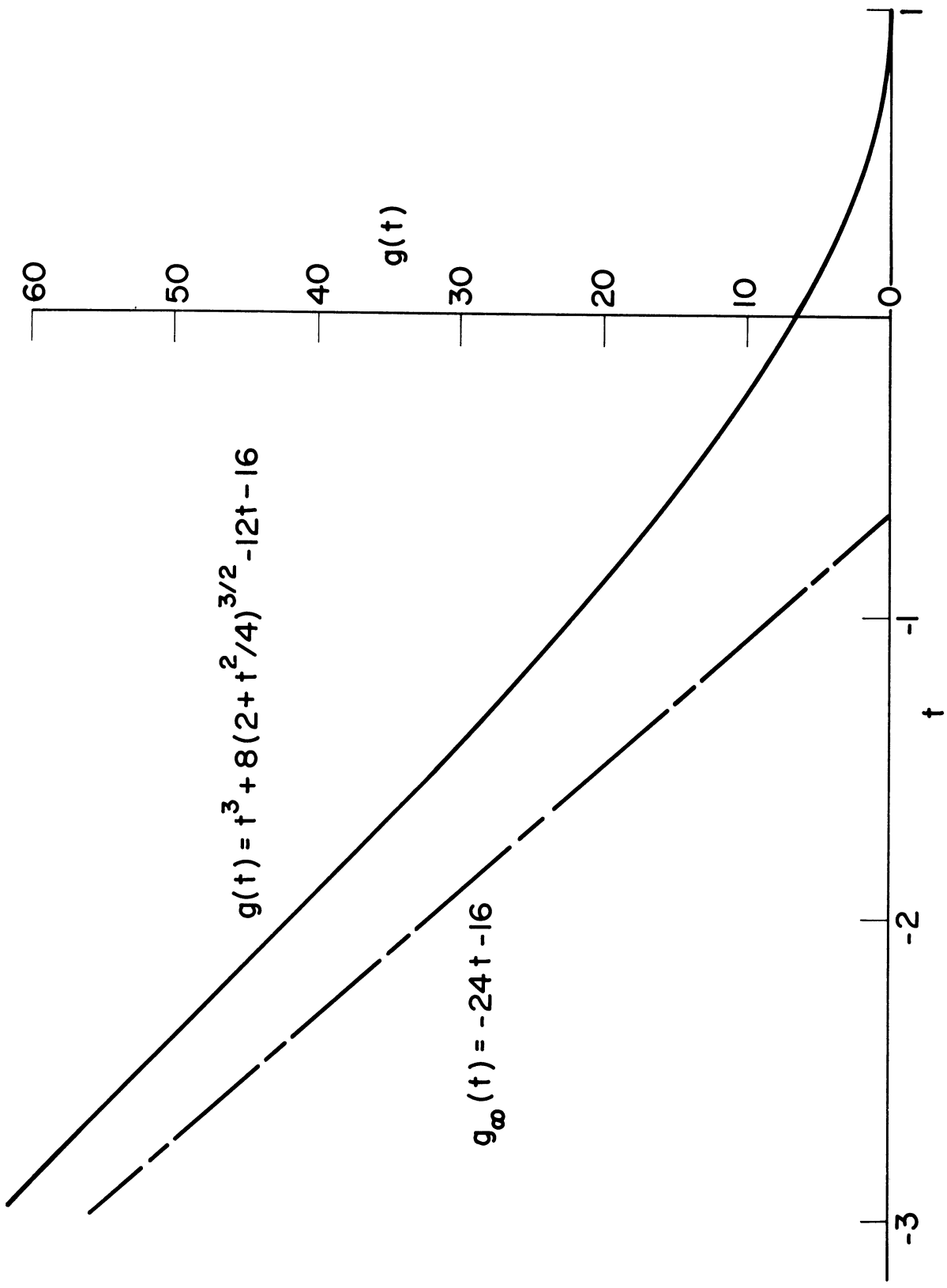


Fig. 19. Curve of $g(t)$ vs $-t$, ($\gamma=0$).

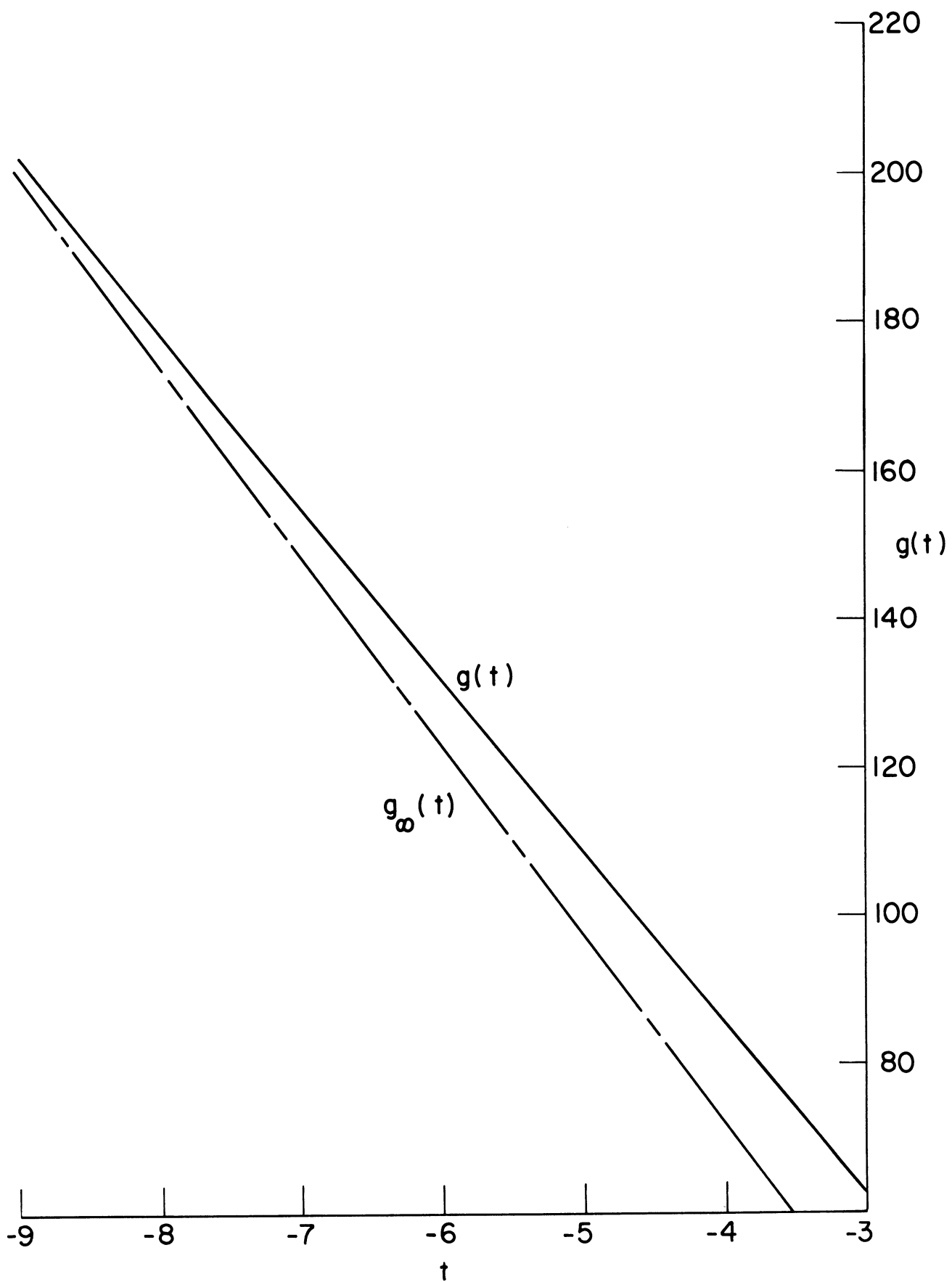


Fig. 20. Curve of $g(t)$ vs $-t$, ($\gamma=0$).

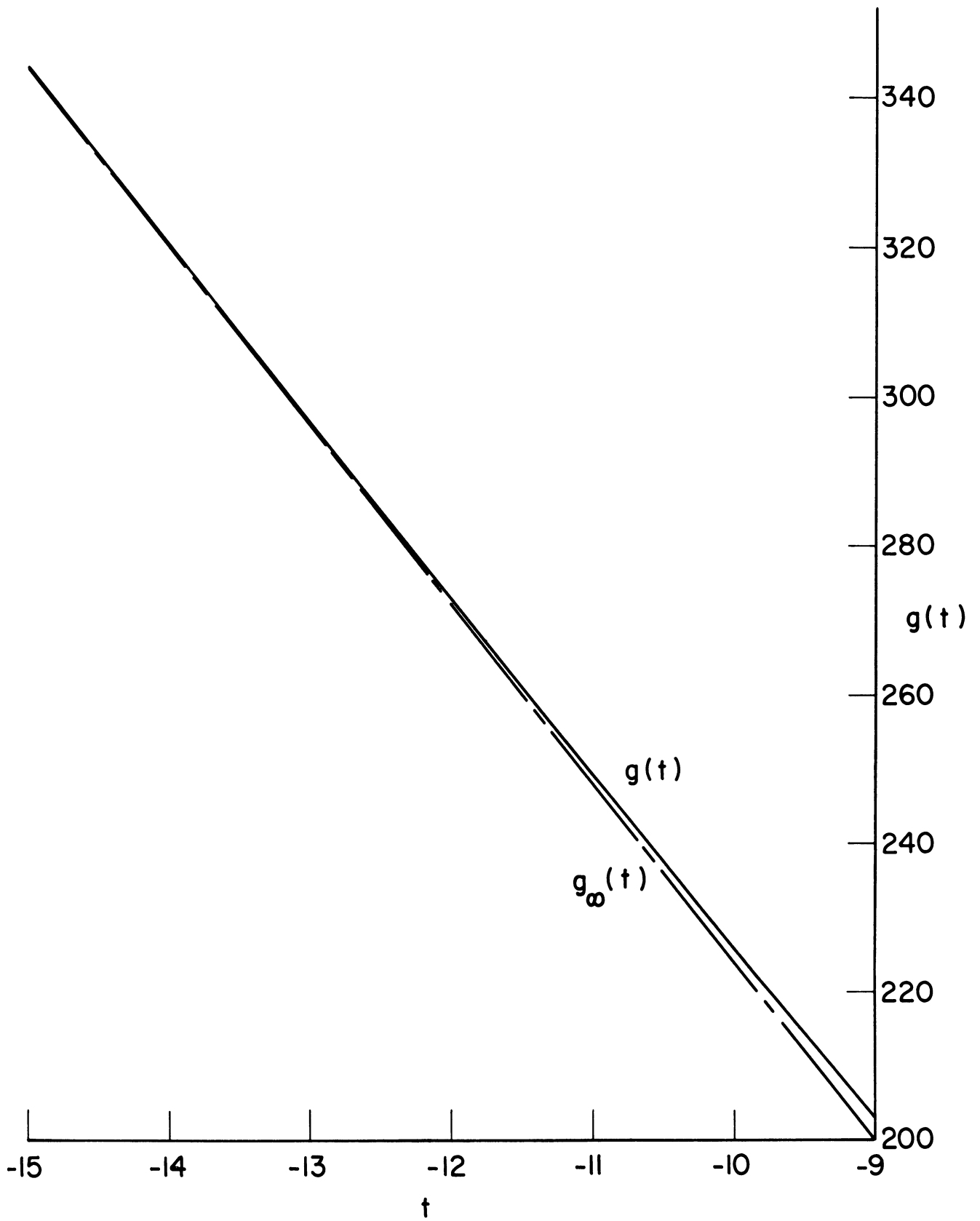


Fig. 21. Curve of $g(t)$ vs $-t$, ($\gamma=0$).

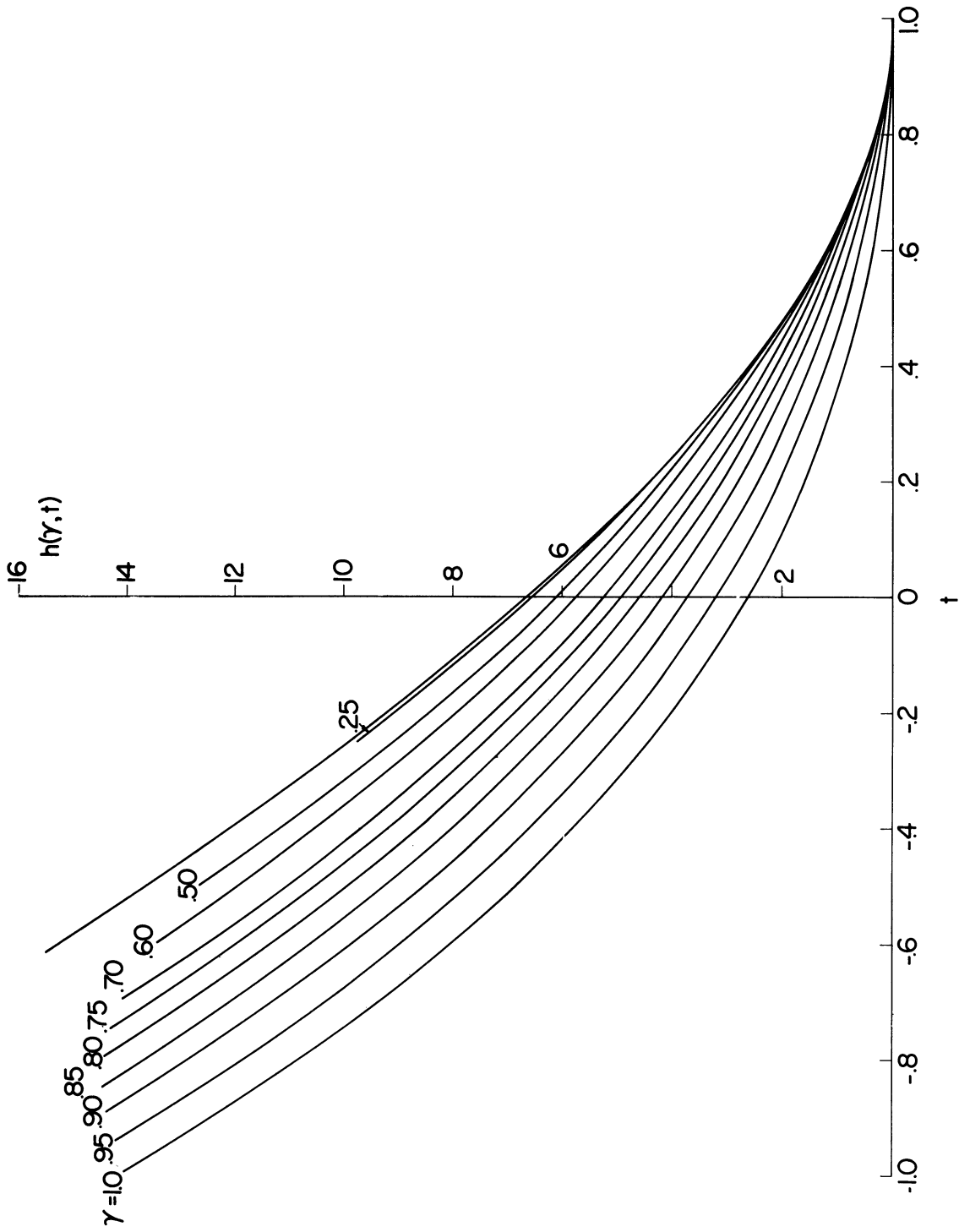


Fig. 22. Curve of $h(\gamma, t)$ vs $-t$, ($\gamma \neq 0$).

7. COMPUTER PROGRAM AND NUMERICAL RESULTS

7.1 COMPUTER PROGRAM

The input parameters for the program are:

1. V = probe velocity (dimensionless)
2. $\text{PHI} = \Phi_p$ = probe potential (dimensionless)
3. $\text{DELTAT} = \delta t$ (dimensionless)
4. $A = a$ = probe radius
5. $\text{ZMIN} = z_{\min}$ (dimensionless)
6. H = spacing parameter for derivative calculations.
7. NUM = number of starting points to be used in generating flow lines.
8. NMAX = maximum number of points per flow line, set by computer storage restrictions.
9. NITER = number of iterations to be made after computing initial flow lines.
10. PSW = intermediate output switch.
11. (RI, ZI) = starting points for flow lines (NUM pairs will be read).

From the starting points (RI, ZI) the program first computes an initial approximation to the flow lines from the neutral flow velocity and probe potential. These flow lines are stored in two pairs of arrays: RSA , ZSA and RZA , ZZA . The first letter of each array name indicates SIGMA (positive charge) or ZETA (negative charge).

Next a new approximation to the flow lines is made by using the same starting points and taking into account the charge distribution from the previous approximation. These new approximations are stored in arrays RSB , ZSB , RZB , and ZZB . If the number of new approximations is less than NITER , the "B" arrays are shifted into the "A" arrays and the next iteration is calculated and stored in the "B" arrays.

During calculation of all approximations to the flow lines, a flow line is terminated at the last point before any of the following occur:

1. The calculated increment in Z is positive for the flow line along the vertical axis.
2. R is negative.
3. Z is less than ZMIN.
4. The flow line intersects the probe.
5. More than NMAX points are calculated.

After each approximation to the flow lines is completed, it is written on the output tape. Because each iteration uses many minutes of computer time, it is sometimes desirable to print out each new flow line point as it is generated. Assigning a non-zero value to PSW accomplishes this, and allows the program to be stopped at any time without loss of output.

For details of the method see the flow diagram and program listing in the Appendix. Subroutines which are not expanded in the flow diagram or the program listing are part of The University of Michigan Computing Center Executive System.

7.2 PRESENTATION OF SAMPLE RUN

At the time of writing this report, results of programming the procedure previously discussed are not satisfactory. The output of a run using the program discussed in Section 4.1 is presented in Figs. 23 through 26; the dots indicate the computed points. Note also that the r and z axes are scaled differently.

Here the values of the parameters used are $V=15$, $\Phi_p=+10$, $a=3$, $\delta t = \frac{1}{6}$, $H = \frac{1}{2}$. It is found in this case that $|\beta| < a^2$, so the sheath cutoff distance is determined using Fig. 14; this gives $z_{\min} = -17.5$. The starting coordinates for the flow lines are arbitrarily chosen to be $z=+30$, $r=0,2,4,\dots,12$.

Figures 23 and 24 show the flow lines as computed using only the free space potential due to the probe. Figures 25 and 26 show the first iteration, using the fields of the previous figures in addition to the potential of the probe. Total running time on an IBM 7090 computer for the two iterations was 15 min. The results shown in Figs. 25 and 26 have several salient features:

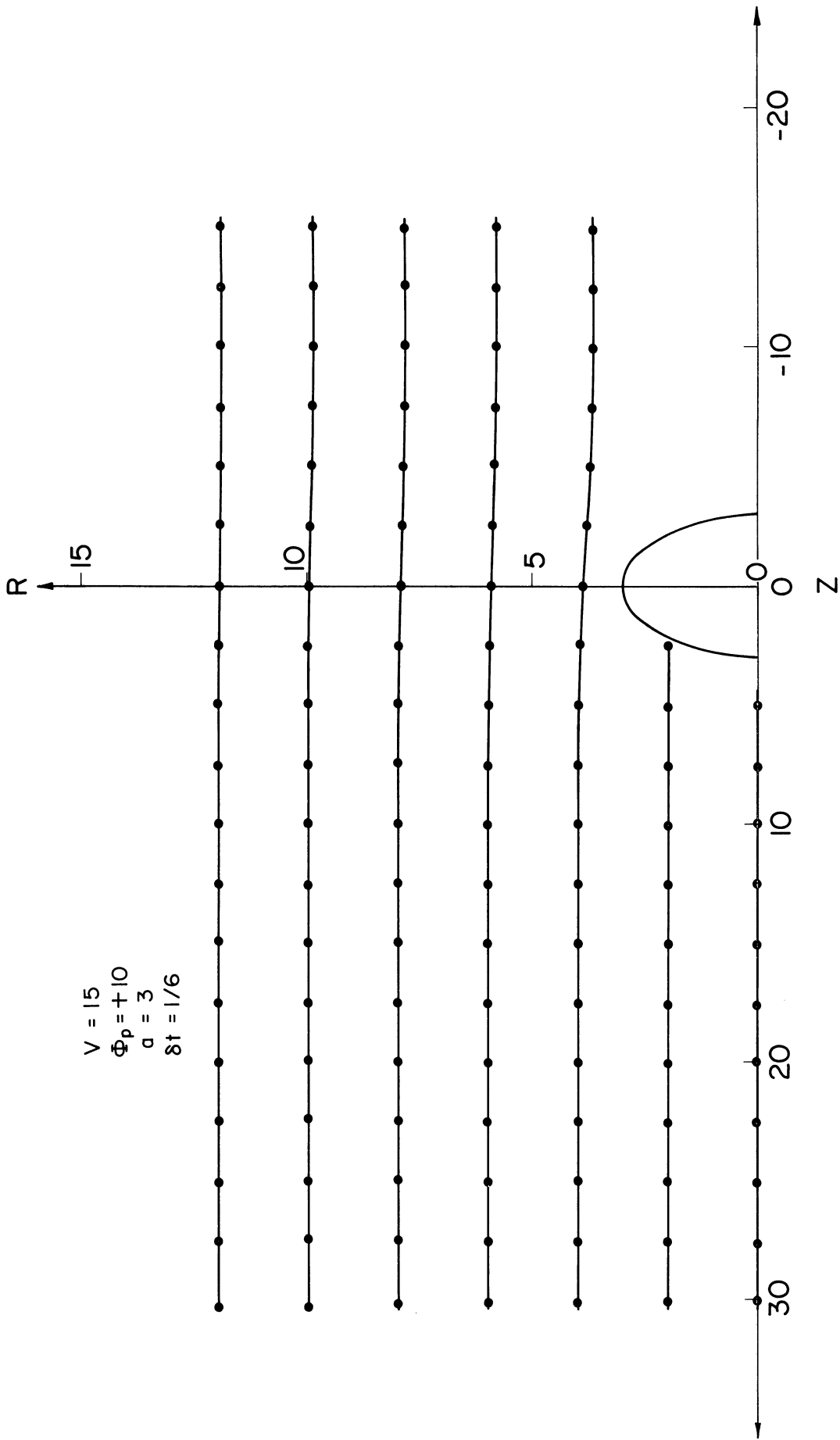


Fig. 23. σ flow lines (initial approximation).

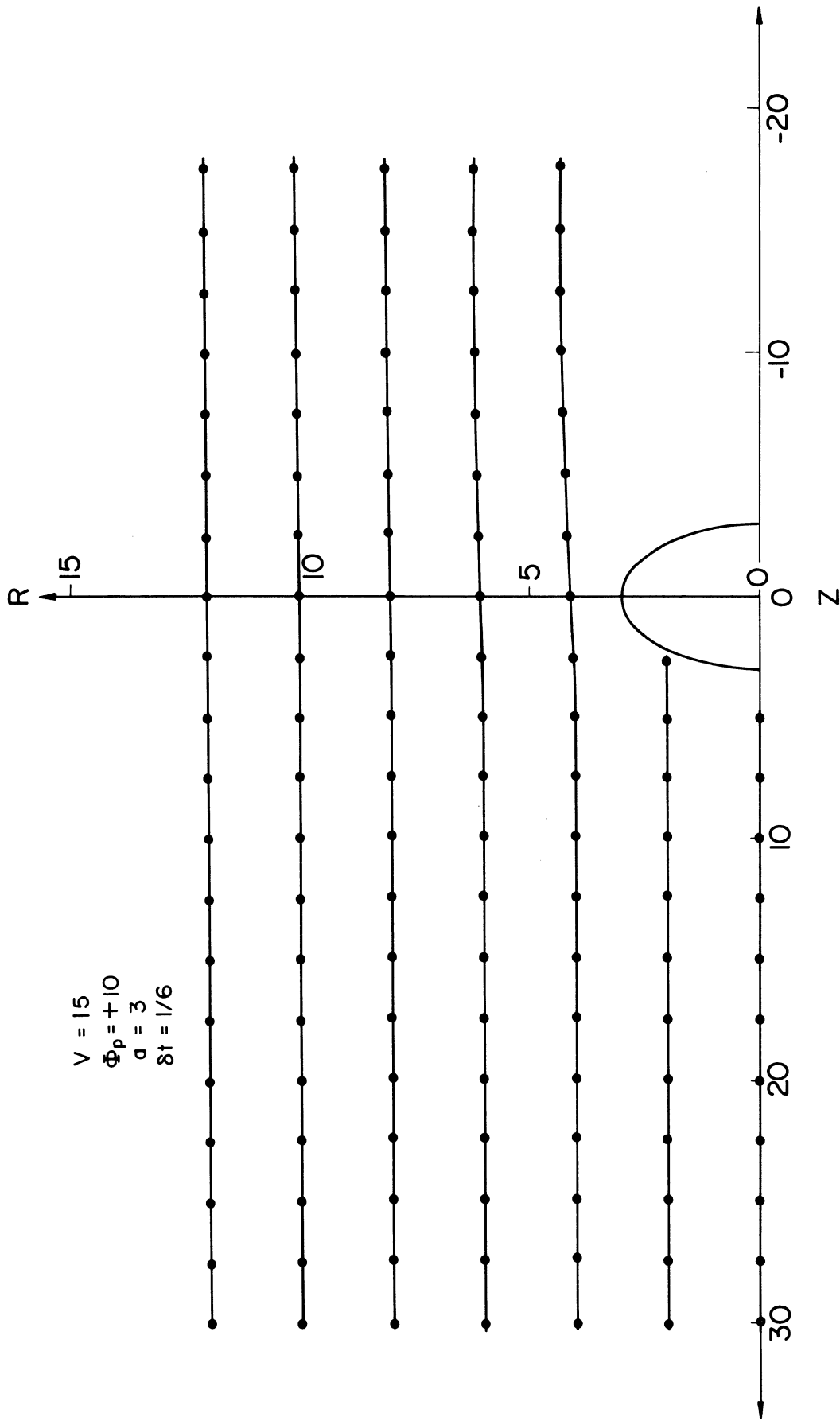


Fig. 24. ζ flow lines (initial approximation).

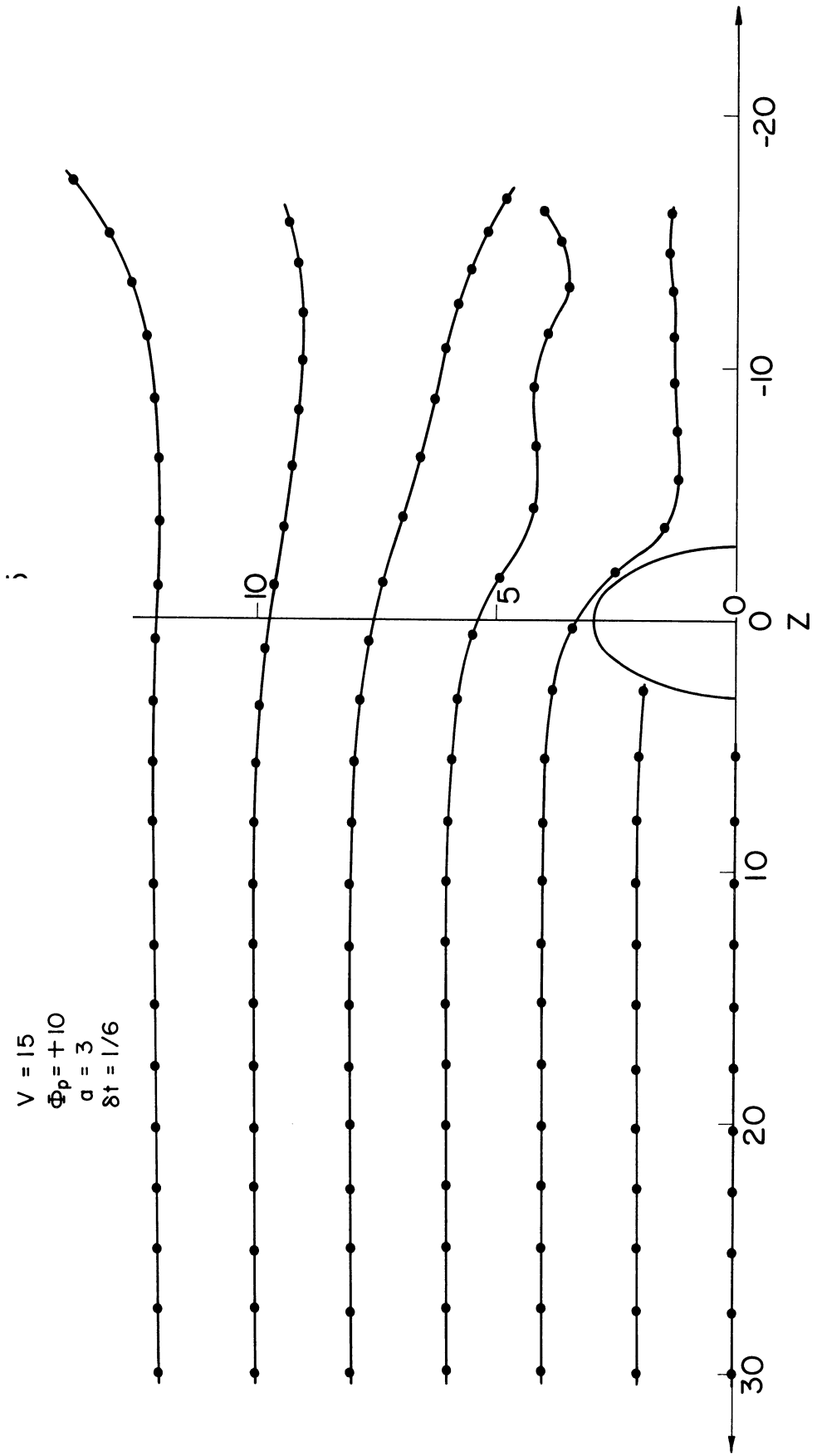


Fig. 25. σ flow lines (first iteration).

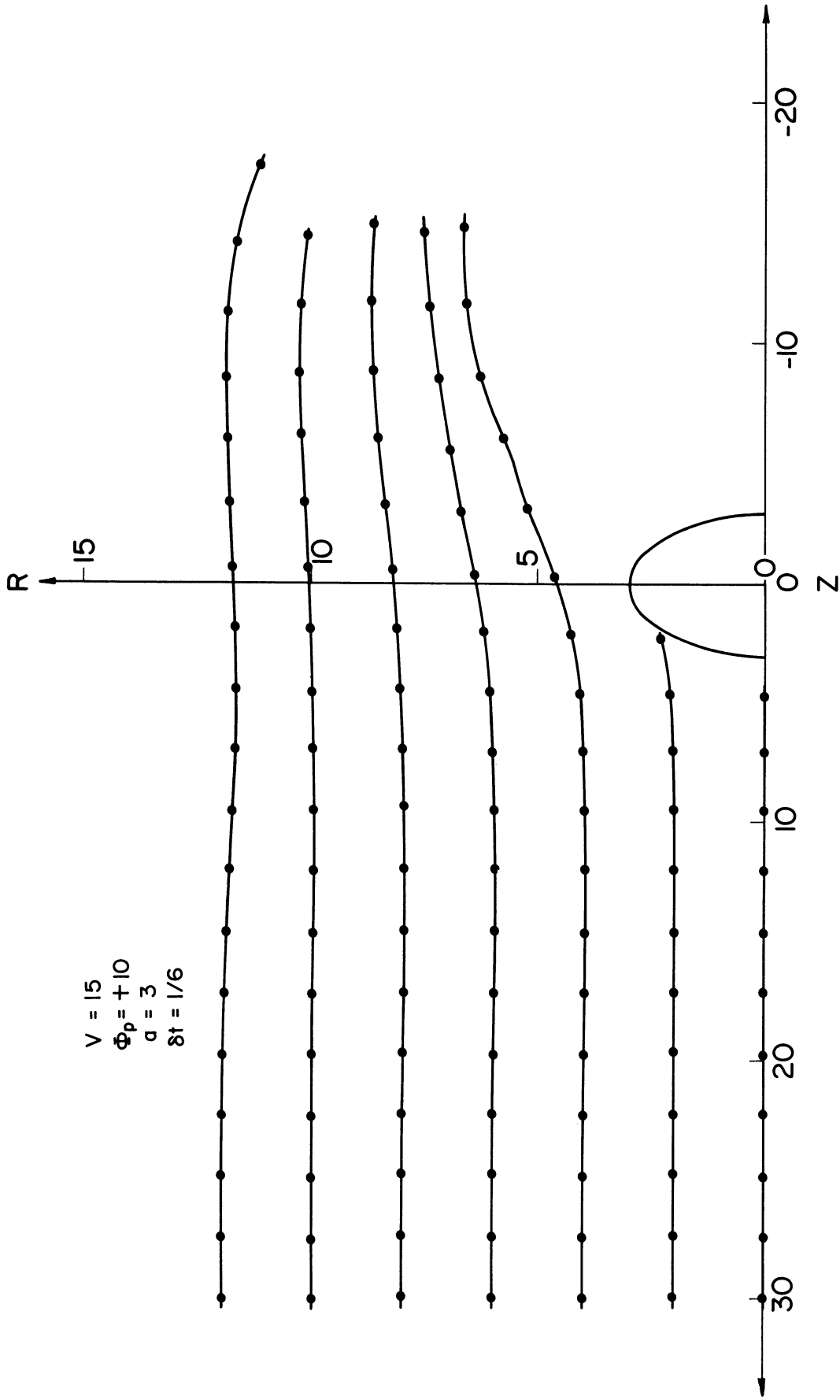


Fig. 26. ζ flow lines (first iteration).

1. The sigma flow line starting with $r=6$ has several irregularities near the lower end;
2. the magnitude of the overall interaction is markedly increased, as exemplified by the highly distorted flow lines;
3. near the probe the positive (negative) species are attracted (repelled) in the second iteration; and
4. the strength of the interaction seems to increase with distance from the probe, as indicated by the increasing distortion of the flow lines which begin with successively larger values of r .

At present, these effects, all of which appear to be non-physical have not been satisfactorily explained. Until these are resolved, further iterations do not appear warranted.

8. EXTENSIONS AND CONCLUSIONS

8.1 EXTENSION TO PHASE II

The extension of the phase I program to include nonuniform flow and diffusion can be achieved by considering these effects independently.

In principle the complications necessary in the program because of the change in the neutral flow regime at and behind the shock are of a relatively simple nature. In passing through the shock both the flow velocity and the density change discontinuously. Thus, the first necessary alteration to the phase I program is the determination of the intercepts of the flow lines and the shock. This involves checking the points along a flow line to locate the first point falling behind the shock, and then by say, linear interpolation, finding the point of intersection P of the flow line and the shock front (Fig. 27). The point so located should then be used as an "initial" point for further determination of the flow line behind the shock.

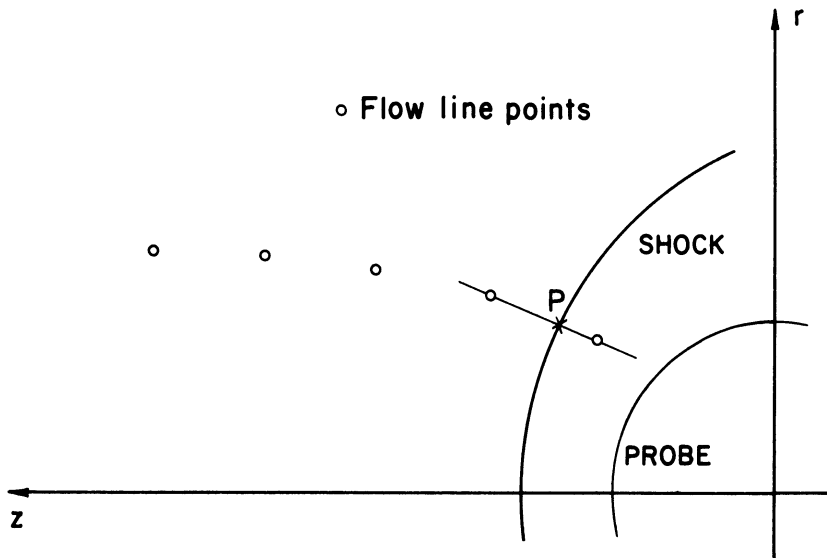


Fig. 27. Flow lines crossing shock.

The addition of an interpolation routine is also necessary in order to determine the neutral flow velocity at flow line points behind the shock. This will be required because such points cannot generally be expected to coincide with any of the points at which the neutral velocity is available, since the velocity field is entered in the computer as a table. Such an interpolation routine might be constructed as follows.

Consider a flow line point P at which the neutral stream velocity is desired (see Fig. 28). Let the three points nearest to P at which the velocity

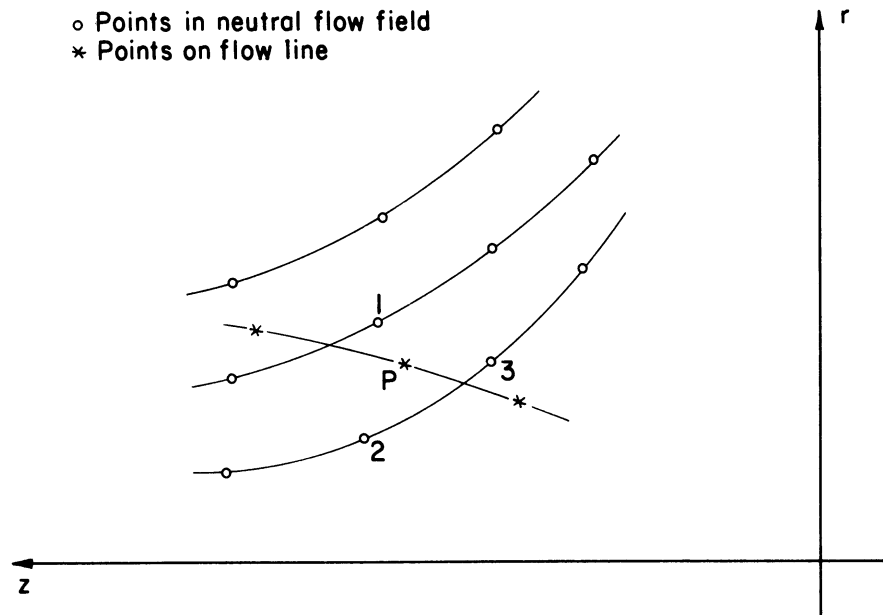


Fig. 28. The three points of tabulated velocity field which are nearest to point P.

field is tabulated be indicated by 1, 2, and 3. Making a Taylor series expansion in the components of the velocity about the point P we have to first order:

$$V_{ri} = V_P + \frac{\partial V}{\partial r} \Big|_P \delta r_i + \frac{\partial V}{\partial z} \Big|_P \delta z_i \quad i = 1, 2, 3 \quad (8.1-1)$$

where V_{ri} denotes the r component of the velocity vector at the point i and $\delta r_i, \delta z_i$ are the appropriate displacements of the points i from point P. This linear system can now be solved for the radial velocity components at the point P. A similar procedure can be used to find the required z-components of the vector \bar{V} .

It should be noted that the real difficulty in the above is the machine determination of the nearest neighbor points 1, 2, and 3. This could perhaps be done by searching the velocity table for points within a specified distance of P, which is, say, twice the maximum distance between points in the velocity table. Decrementing this distance by small amounts, some of the points so determined could be excluded until the three necessary points were found.

The contribution to the current density due to density gradients can be found by an interpolation technique similar to the above. Rearranging Eqns. (5.1-1) the current densities may be written:

$$\bar{J} = (\bar{V} - \nabla\Phi - \frac{\nabla\sigma}{\sigma}) \sigma$$

$$-\bar{L} = (\bar{V} + \nabla\Phi - \frac{\nabla\zeta}{\zeta}) \zeta. \quad (8.1-2)$$

We see then, that the contribution to the velocity of the charges due to diffusion is given by the ratio of the density gradient to the density itself (assuming the density is finite). Setting up a linear system analogous to Eqns. (8.1-1) for the charge densities, these contributions can be easily found.

Since we assume the ionized flow to be frozen, the ion densities will change directly with the neutral density increase across the shock. For computer calculations we must therefore consider the neutral flow velocity and the densities to be double valued at the shock (point P of Fig. 23), using the shock as a bounding surface between the two regions of the flow. In calculating charge density gradients ahead of the shock, the "nearest" neighbor points considered must be restricted to the region ahead of the shock or on the shock itself as approached from the front. Similarly, behind the shock calculations must use only values in the region behind the shock or values along the shock which have been corrected for the changes caused by passing through the shock.

Further corrections might also be included for such effects as the variation in diffusion (and thus mobility) constants due to the density variation behind the shock.

8.2 FURTHER EXTENSIONS AND CONCLUSIONS

The major portion of this report has been concerned with a spherical probe, and with the free space potential of such a body as the potential approximation initiating the iterative procedure. In principle the extension of the program to utilize different configurations and potentials is straightforward. Changes would be necessary in sections of the program in which the initial flow lines are computed (by changing the initial potential function).

Even if the spherical probe is retained, an extension of the process used for determining the flow line cutoff point z_{min} would be desirable. In general the charge distribution on and about the probe will be altered with each iteration. Consequently, new calculations of the charge imbalance in the plasma and of the net charge on the sphere (including for consistency the charge induced by the previous spatial distribution) should be used to obtain a new approximation to z_{min} . This would require an additional section in the program for performing the necessary numerical integrations and for obtaining the desired cutoff value from them. The numerical integrations might easily utilize the density computations mentioned in connection with the phase II extension of the program.

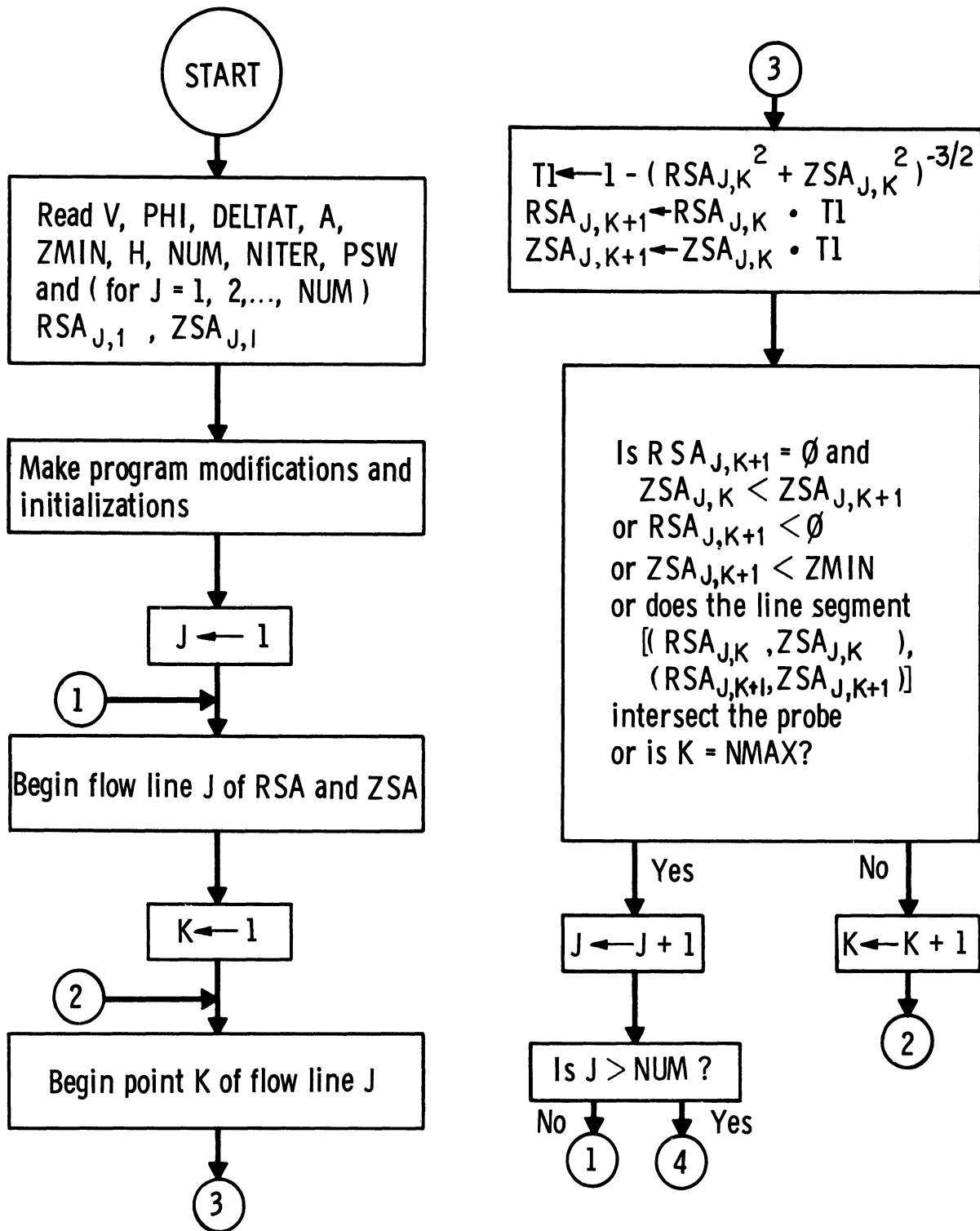
Some serious points concerning the validity of the entire procedure must be mentioned. The whole method evolves from the observation that if the potential Φ is known, then the system (4.2-3) can be solved for σ and ζ , or alternatively if the charge densities are known, the Poisson equation can be solved for the potential. This suggests solving the system by successive approximations, as has been done in this report. This process has the advantage of resolving the problem into the solution of a sequence of linear equations, rather than of a coupled system. However, this approach immediately raises the following questions: do there exist functions which can be used as initial potentials for which the process will ultimately converge to the solution of the system (4.2-3) with its boundary conditions, and if so, what are they? These questions remain unanswered; results of running the computer program evolved in this report (see Section 7.1) are not sufficient to indicate possible resolution of these important points at this time.

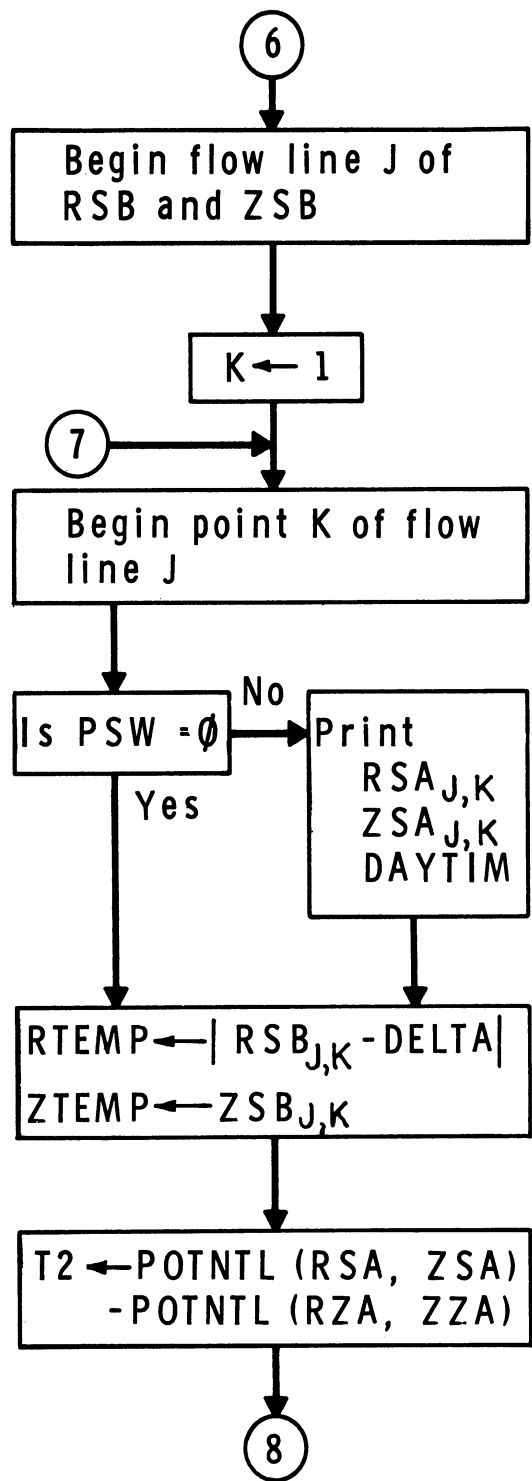
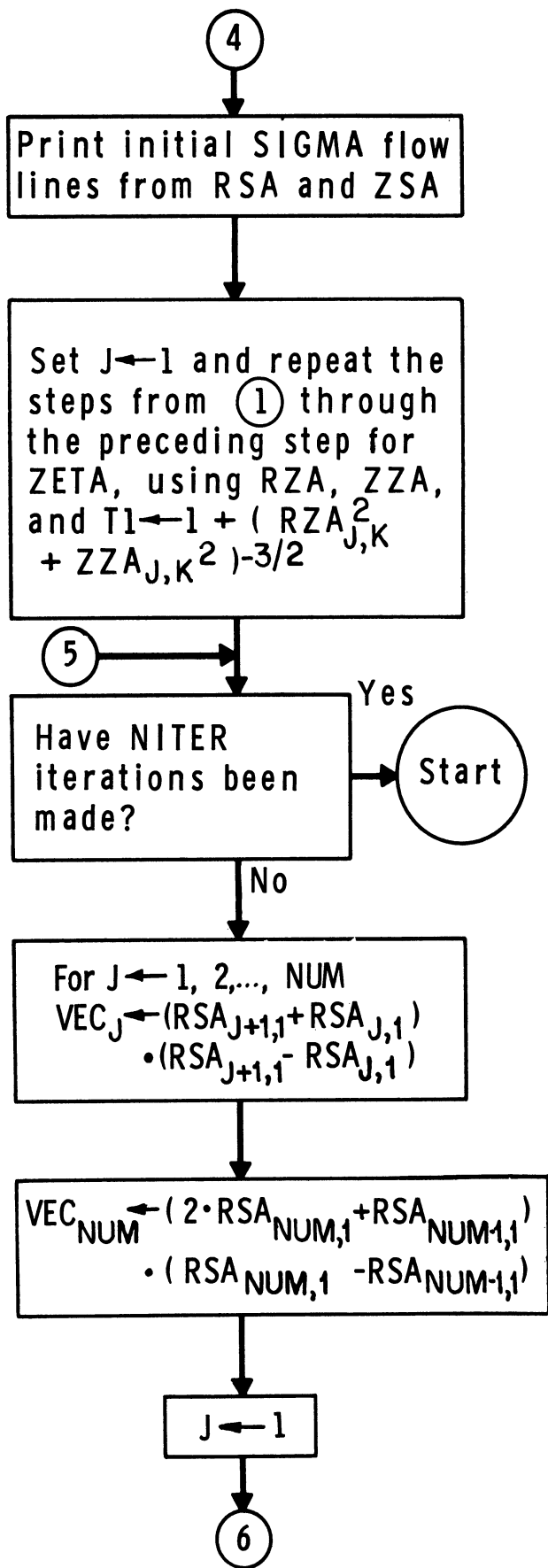
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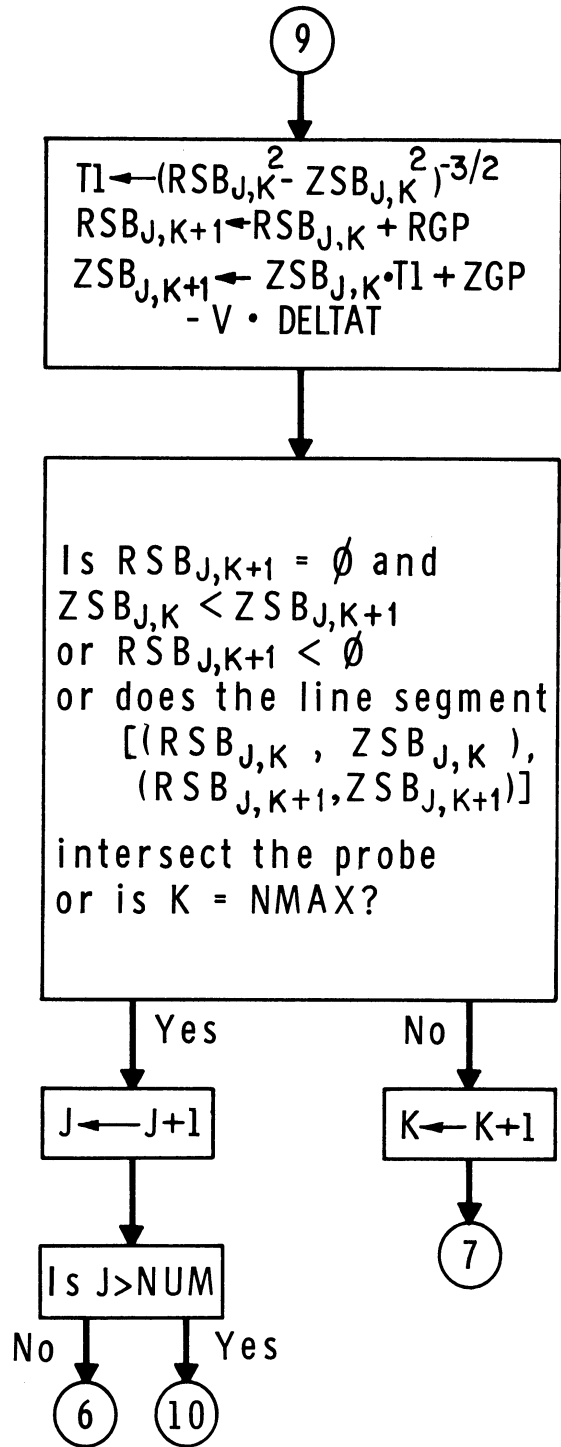
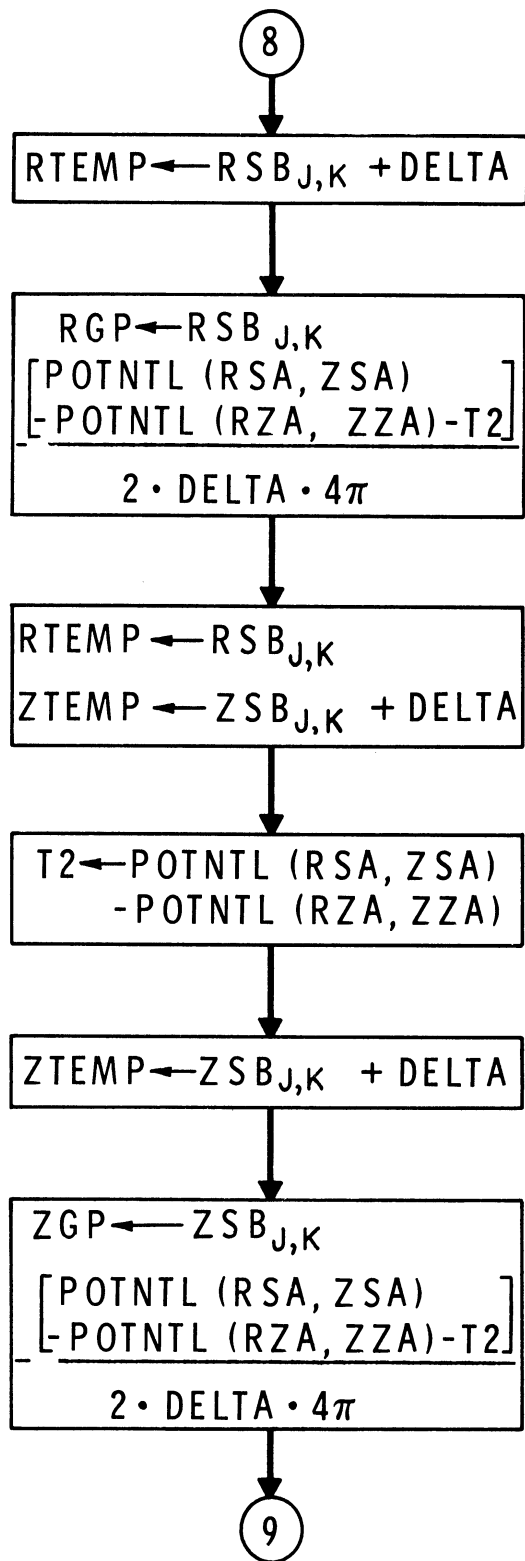
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APPENDIX

FLOW DIAGRAM AND PROGRAM LISTING







10

Set $J \leftarrow 1$ and repeat the steps from 6 through the preceding step for RZB and ZZB using $RGP \leftarrow RZB_{J,K}$

$$+ \frac{\begin{bmatrix} \text{POTNTL} (\text{RSA}, \text{ZSA}) \\ - \text{POTNTL} (\text{RZA}, \text{ZZA}) - \text{T2} \end{bmatrix}}{2 \cdot \text{DELTA} \cdot 4\pi}$$

$ZGP \leftarrow ZZB_{J,K}$

$$+ \frac{\begin{bmatrix} \text{POTNTL} (\text{RSA}, \text{ZSA}) \\ - \text{POTNTL} (\text{RZA}, \text{ZZA}) - \text{T2} \end{bmatrix}}{2 \cdot \text{DELTA} \cdot 4\pi}$$

$RZB_{J,K+1} \leftarrow -RZB_{J,K} \cdot T1 + RGP$
 $ZZB_{J,K+1} \leftarrow -ZZB_{J,K} \cdot T1 + ZGP - V \cdot \text{DELTAT}$

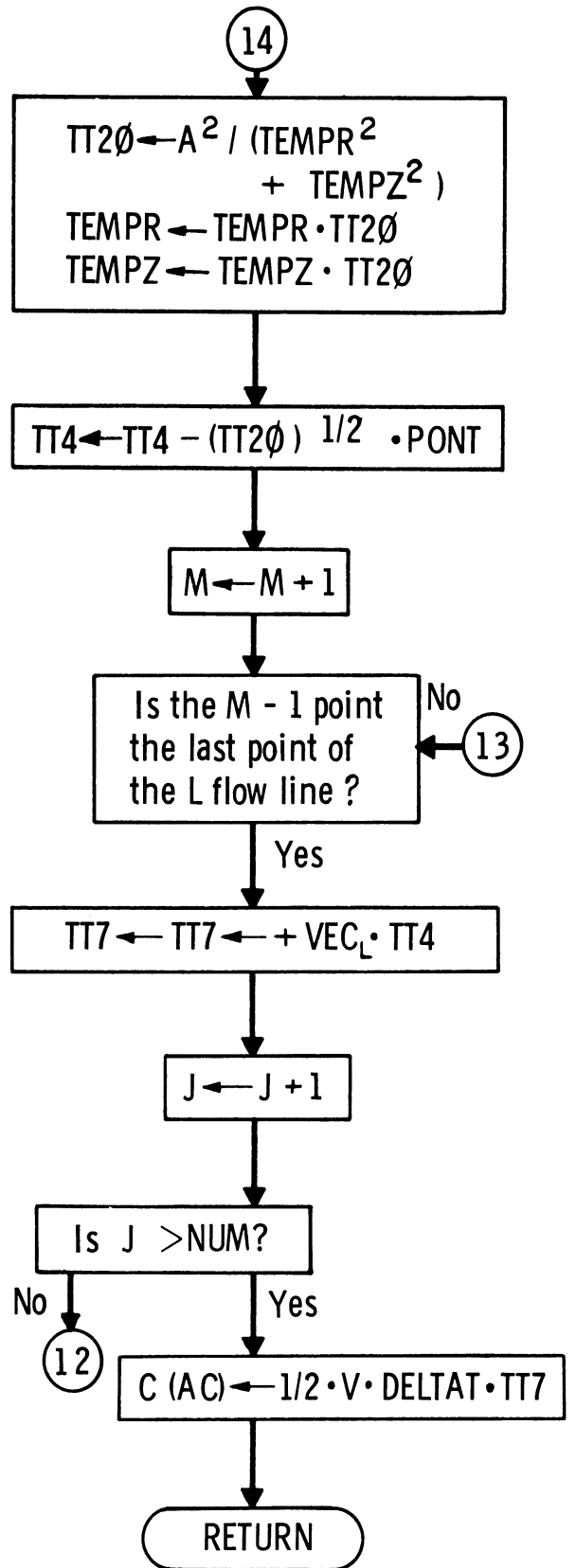
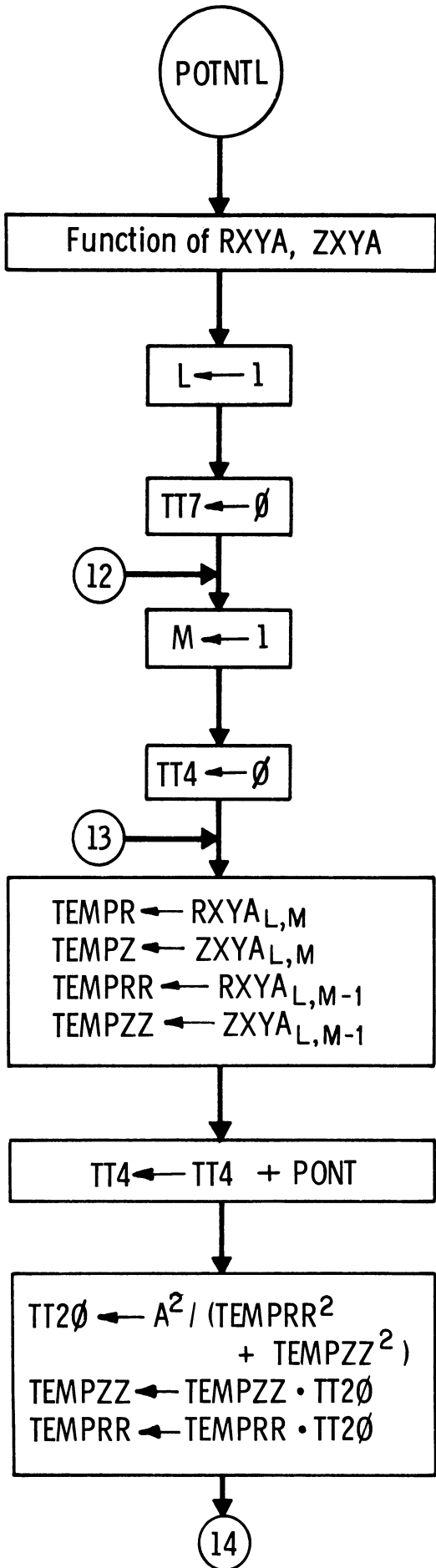
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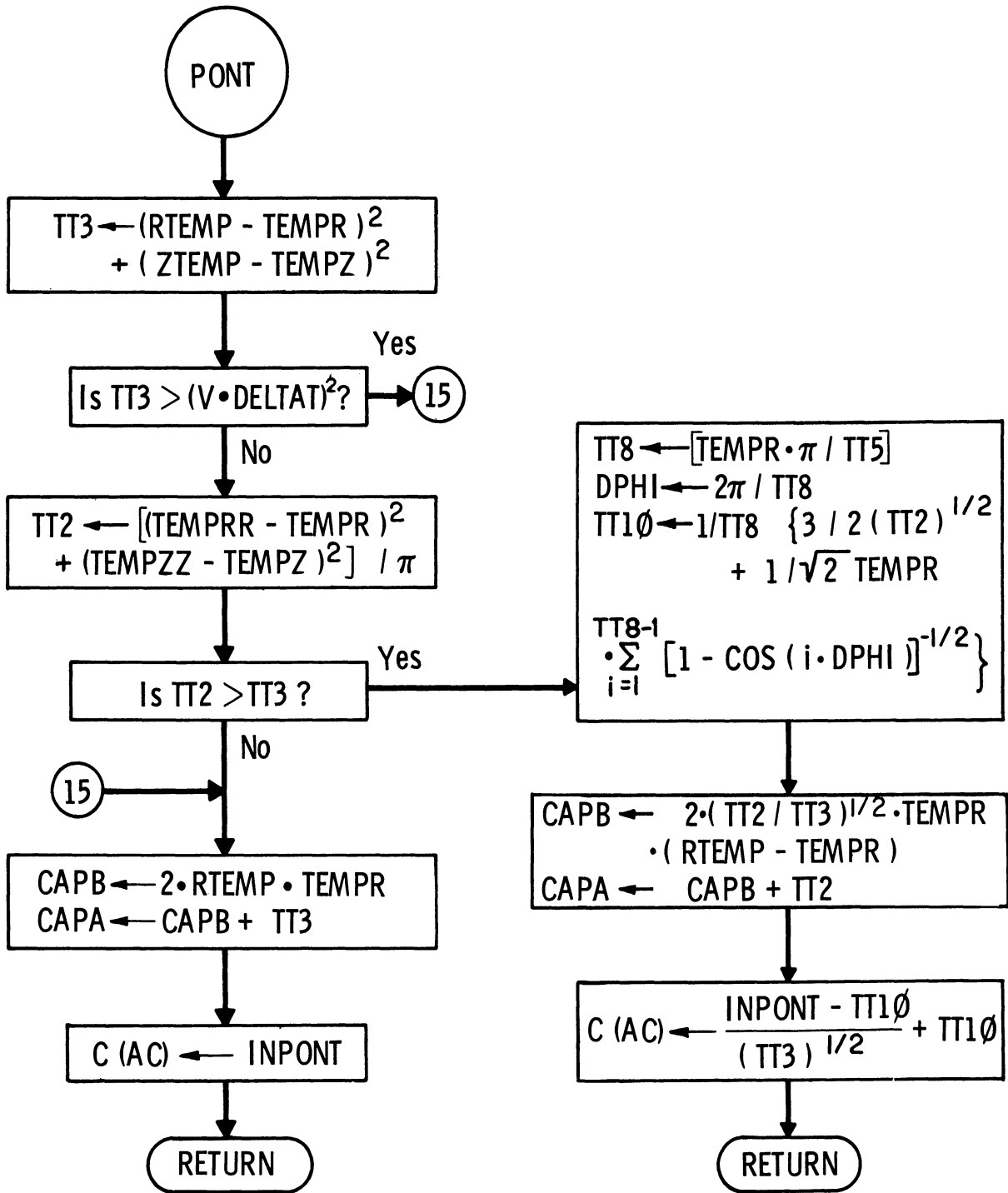
11

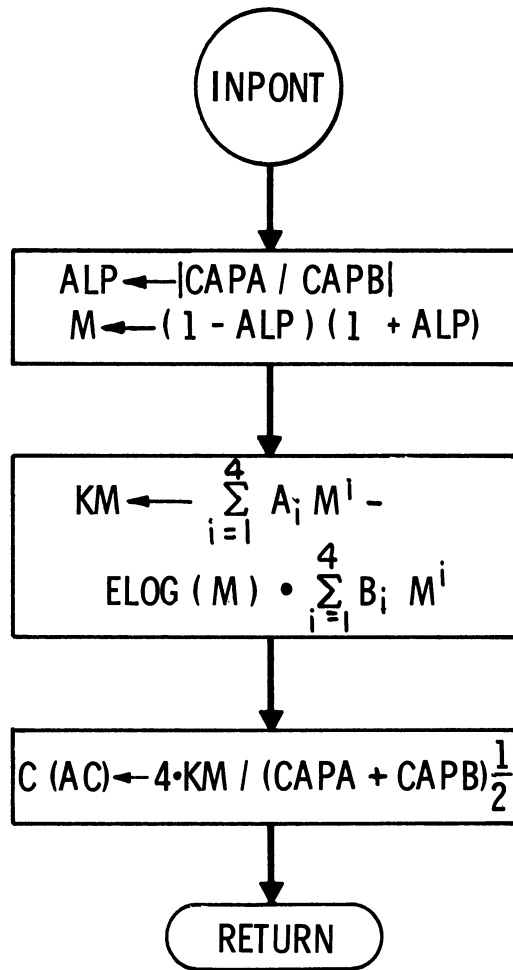
For $J \leftarrow 1, 2, \dots, \text{NUM}$
 and $K \leftarrow 1, 2, \dots, \text{NMAX}$
 $\text{RSA}_{J,K} \leftarrow \text{RSB}_{J,K}$
 $\text{ZSA}_{J,K} \leftarrow \text{ZSB}_{J,K}$
 $\text{RZA}_{J,K} \leftarrow \text{RZB}_{J,K}$
 $\text{ZZA}_{J,K} \leftarrow \text{ZZB}_{J,K}$

Print the new flow line iteration from the "A" matrices

5







PROGRAM LENGTH (CTAL) 522C7
INITIAL TRANSFER VECTOR CCCCC

PROGRAM COMMON LENGTH CCCCC
TRANSFER VECTOR LENGTH CCO11

LOWEST ERASABLE DEFINED OCCOO
FIRST LOCATION EXECUTED OOO11

EXTERNAL ROUTINES CALLED BY THIS PROGRAM
CCCCC 622563254626 CC SETECF
CCCC1 33512521246C CC •READ
CCCC2 33CCC103CCCC CC •C13CO
CCCC3 334751314563 CC •PRINT
CCCC4 242176633144 CC DAYTIM
CCCC5 625051636C6C CC SQRT
CCCC6 234662606C6C CC COS
CCCC7 254346276C6C CC ELCG
CCCCC 25515146516C CC ERRCR

PROGRAM LISTING (NUMBERS IN CTAL)

CCCC11	0074	CC	4	CCCCC	C1C	CALL	CN	BRIEF	START	SETECF,END	READ PARAMETERS AND STARTING POINTS
CCCC13	0074	CC	4	CCCO1	C1C	READ	CF	PM			VELOCITY
CCCC15	-1	CCCCC	0	C267C	C1C	ICP	FIN	CALL			PRCBE POTENTIAL
CCCC16	-1	CCCCC	C	C2671	C1C	ICP	V	ICP			TIME INCREMENT
CCCC17	-1	CCCCC	C	C2672	C1C	ICP	PHI	ICP			PRCBE RADIUS
CCCC20	-1	CCCCC	C	C2673	C1C	ICP	DELTA1	ICP			
CCCC21	C560	CC	C	C2673	C1C	ICP	A	ICP			
CCCC22	C260	CC	C	C2673	C1C	FMP	A	FMP			
CCCC23	C601	CC	C	C2674	C1C	STO	A2	STO			
CCCC24	C560	CC	C	C2670	C1C	LDQ	V	LDQ			
CCCC25	C260	CC	C	C2672	C1C	FMP	DELTA1	FMP			
CCCC26	C601	CC	C	C2676	C1C	STO	VD1	STO			
CCCC27	C131	CC	C	CCCCC	CC	XCA		XCA			
CCCC30	C26C	CC	C	C2676	C1C	FMP	VD1	FMP			
CCCC31	C6C1	CC	C	C2675	C1C	STO	V2	STO			
CCCC32	C560	CC	C	C2671	C1C	LDQ	PHI	LDQ			
CCCC33	C260	CC	C	C2672	C1C	FMP	DELTA1	FMP			
CCCC34	C601	CC	C	C27C0	C1C	STO	K	STO			
CCCC35	-1	CCCCC	C	C27C1	C1C	ICP	ZMIN	ICP			MINIMUM Z COORDINATE
CCCC36	-1	CCCCC	C	C27C3	C1C	ICP	H	ICP			FRACTION OF TCP OF TRAJECTORY SPACING FOR POINTS FOR FINDING GRADIENTS
CCCC37	-1	CCCCC	C	C2717	C1C	ICP	NUM	ICP			NUMBER OF TRAJECTORIES
CCCC40	-1	CCCCC	C	C2720	C1C	ICP	NMAX	ICP			MAXIMUM NUMBER OF POINTS PER TRAJECTORY
CCCC41	-1	CCCCC	C	C2721	C1C	ICP	NITER	ICP			NUMBER OF INTERACTIONS TO BE MADE
CCCC42	-1	CCCCC	C	C2641	C1C	ICP	PSW	ICP			
CCCC43	05C0	CC	0	C2720	C1C	CLA	NMAX	CLA			
CCCC44	04C0	CC	0	522C6	C1C	ACD	=1	ACD			
CCCC45	C601	CC	C	C272C	C1C	STO	NMAX	STO			
CCCC46	0534	CC	1	C2720	C1C	LXA	NMAX,1	LXA			
CCCC47	-0634	CC	1	CC222	C1C	SXD	S5,1	SXD			SET
CCCC50	-0634	CC	1	CC241	C1C	SXD	S6,1	SXD			UP
CCCC51	-0634	CC	1	CC373	C1C	SXD	S7,1	SXD			LCCP
CCCC52	-0634	CC	1	CC6C6	C1C	SXD	S8,1	SXD			TERMINATIONS
CCCC53	-0634	CC	1	CC727	C1C	SXD	S9,1	SXD			AND
CCCC54	-0634	CC	1	CC744	C1C	SXD	L2CAC,1	SXD			INCREMENTS
CCCC55	-0634	CC	1	C1513	C1C	SXD	S31,1	SXD			
CCCC56	-0634	CC	1	C1224	C1C	SXD	S1C,1	SXD			
CCCC57	-0634	CC	1	C2052	C1C	SXD	S11,1	SXD			

CC152	C634	CC 1	CC452	C1C	SXA	S25B+1,1	CCMPLE
CC153	0634	CC 1	C164C	C1C	SXA	S27B+1,1	TCTAL
CC154	1 11612	1	CC155	C1C	TXI	*+1,1,ZZA-ZSA	PAGES PER
CC155	0634	CC 1	CC665	C1C	SXA	S26B+1,1	TRAJECTRY
CC156	0634	CC 1	C1717	C1C	SXA	S28B+1,1	MATRIX
CC157	C500	CC C	C2717	C1C	CLA	NUM	ZERC
CC160	C621	CC C	CC412	C1C	STA	L5A	CUT
CC161	0621	CC C	CC625	C1C	STA	L15A	TRAJECTRY
CC162	C621	CC C	C1577	C1C	STA	L25A	MATRICES
CC163	C621	CC C	C1656	C1C	STA	L35A	
CC164	0400	CC C	52205	C1C	ADD	= 3	
CC165	0131	CC C	CC000	CC	XCA	= 4	
CC166	-0754	CC C	CC000	CC	ZAC	TP	
CC167	0221	CC C	52204	C1C	DVP	L3C,C	
CC170	-0600	CC C	C2725	C1C	STQ	L13C,C	
CC171	-0634	CC C	CC371	C1C	SXD	SIZE,1	
CC172	-0634	CC C	CC604	C1C	SXD	RSA,1	
CC173	0774	CC 1	C4704	CC	AXT	ZSA,1	
CC174	0600	CC 1	C7636	C1C	STZ	RZA,1	
CC175	0600	CC 1	14543	C1C	STZ	ZZA,1	
CC176	0600	CC 1	21450	C1C	STZ	RSB,1	
CC177	0600	CC 1	26355	C1C	STZ	ZSB,1	
CC200	0600	CC 1	32262	C1C	STZ	RZB,1	
CC201	0600	CC 1	40166	C1C	STZ	ZZB,1	
CC202	0600	CC 1	45072	C1C	STZ	*-8,1,1	
CC203	0600	CC 1	51776	C1C	STZ	1,2	
CC204	2 0000	1	CC174	C1C	TIX	NUM,1	
CC205	C774	CC 2	CC001	CC	AXT	RSA,2	
CC206	0534	CC 1	C2717	C1C	LXA	ZSA,2	
CC207	-1 0000	2	C7636	C1C	IGP	RSA,2	
CC210	-1 0000	2	14543	C1C	LOP	RSA,2	
CC211	0500	CC 2	C7636	C1C	CLA	RZA,2	
CC212	0601	CC 2	21450	C1C	STO	RZA,2	
CC213	0601	CC 2	C7637	C1C	STO	RZA+1,2	
CC214	0601	CC 2	21451	C1C	STO	RZA+1,2	
CC215	C500	CC 2	14543	C1C	CLA	ZSA,2	
CC216	C601	CC 2	26355	C1C	STU	ZZA,2	
CC217	C300	CC C	C2676	C1C	FAD	VDT	
CC220	0601	CC 2	14544	C1C	STO	ZSA+1,2	
CC221	0601	CC 2	26356	C1C	STO	ZZA+1,2	
CC222	1 0000	2	CC223	C1C	TXI	*+1,2,**	
CC223	2 0000	1	CC207	C1C	TIX	S4,1,1	
CC224	-1 0000	C	CC000	CC	ENDIO		
CC225	2 0000	2	C0226	C1C	TIX	*+1,2,1	
CC226	-0634	CC 2	CC374	C1C	SXD	L4,2	
CC227	-0634	CC 2	CC457	C1C	SXD	L8,2	
CC230	-0634	CC 2	CC607	C1C	SXD	L14,2	
CC231	-0634	CC 2	CC672	C1C	SXD	L18,2	
CC232	-0634	CC 2	C1226	C1C	SXD	L24A,2	
CC233	-0634	CC 2	C1547	C1C	SXD	L24E,2	
CC234	-0634	CC 2	C1645	C1C	SXD	L28,2	
CC235	-0634	CC 2	C1515	C1C	SXD	L34A,2	
CC236	-0634	CC 2	C1563	C1C	SXD	L34E,2	
CC237	-0634	CC 2	C1724	C1C	SXD	L38,2	
CC240	-0634	CC 2	C2060	C1C	SXD	LL17,2	
CC241	2 0000	2	CC242	C1C	TIX	*+1,2,**	
CC242	-0634	CC 2	CC745	C1C	SXD	L2CAE,2	
CC243	-0634	CC 2	C1225	C1C	SXD	L24,2	

CC244	-0634	CC 2	C1514	C1C	SXID	L34,2	SET LP
CC245	0774	CC 2	CC001	CC	AXT	1,2	
CC246	0754	CC 2	CC000	CC	PXA	,2	CCMPLTE
CC247	0734	CC 1	CC000	CC	PAX	,1	INITIAL
CC250	0774	CC 4	CC001	CC	AXT	1,4	APPRXIMATICNS
CC251	0634	CC 4	CC360	C1C	SXA	L2C,4	FCR
CC252	0634	CC 4	CC370	C1C	SXA	L3B,4	SIGMA
CC253	0560	CC 1	C7636	C1C	LCQ	RSA,1	TRAJECTORIES
CC254	0260	CC 1	C7636	C1C	FMP	RSA,1	
CC255	0601	CC C	C2642	C1C	STC	T1	
CC256	0560	CC 1	14543	C1C	LCQ	ZSA,1	
CC257	0260	CC 1	14543	C1C	FMP	ZSA,1	
CC260	0300	CC C	C2642	C1C	FAD	T1	
CC261	0500	CC C	52203	C1C	LCQ	=-1.5	
CC262	0074	CC 4	CC002	C1C	CALL	.C13CC	EXPONENTIATION SUBROUTINE
CC263	0131	CC C	CC000	CC	XCA		
CC264	0260	CC C	C2700	C1C	FMP	K	
CC265	0601	CC C	C2642	C1C	STU	T1	
CC266	0500	CC C	52202	C1C	CLA	=1.	
CC267	0300	CC C	C2642	C1C	FAD	T1	
CC270	0601	CC C	C2642	C1C	STO	T1	
CC271	0131	CC C	CC000	CC	XCA	RSA,1	
CC272	0260	CC 1	C7636	C1C	FMP	RSA-1,1	
CC273	0601	CC 1	C7635	C1C	STC	L3A	
CC274	-0120	CC C	CC367	C1C	TMI	L3	
CC275	0560	CC C	C2642	C1C	LCQ	T1	
CC276	0260	CC 1	14543	C1C	FMP	ZSA,1	
CC277	0302	CC C	C2676	C1C	FSB	VDT	
CC300	0601	CC 1	14542	C1C	STO	ZSA-1,1	
CC301	0402	CC C	C2701	C1C	SUB	ZMIN	
CC302	-0120	CC C	CC366	C1C	TMI	L3	
CC303	0500	CC 1	C7636	C1C	CLA	RSA,1	
CC304	-0100	CC C	CC310	C1C	TNZ	**4	
CC305	0500	CC 1	14542	C1C	CLA	ZSA-1,1	
CC306	0302	CC 1	14543	C1C	FSB	ZSA,1	
CC307	0120	CC C	CC366	C1C	TPL	L3	
CC310	0560	CC 1	C7635	C1C	LCQ	RSA-1,1	
CC311	0260	CC 1	C7635	C1C	FMP	RSA-1,1	
CC312	0601	CC C	C2711	C1C	STO	TEMP	
CC313	0560	CC 1	14542	C1C	LCQ	ZSA-1,1	
CC314	0260	CC 1	14542	C1C	FMP	ZSA-1,1	
CC315	0300	CC C	C2711	C1C	FAD	TEMP	
CC316	0402	CC C	C2674	C1C	SUB	A2	
CC317	-0120	CC C	CC366	C1C	TMI	L3	
CC320	0500	CC 1	14543	C1C	CLA	ZSA,1	
CC321	0302	CC 1	14542	C1C	FSB	ZSA-1,1	
CC322	0601	CC C	C2711	C1C	STO	TEMP	
CC323	0500	CC 1	C7636	C1C	CLA	RSA,1	
CC324	0302	CC 1	C7635	C1C	FSR	RSA-1,1	
CC325	0241	CC C	C2711	C1C	FCP	TEMP	
CC326	-0600	CC C	C2726	C1C	STQ	R	
CC327	0260	CC C	C2726	C1C	FMP	B	
CC328	0601	CC C	C2727	C1C	STC	B2	
CC331	0560	CC 1	14543	C1C	LCQ	ZSA,1	
CC332	0260	CC C	C2726	C1C	FMP	R	
CC333	0302	CC 1	C7636	C1C	FSB	RSA,1	
CC334	0601	CC C	C2730	C1C	STO	B1	
CC335	0131	CC C	CC000	CC	XCA		


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CC326 0260 CC C C2730 C1C
CC337 0601 CC C C2731 C1C
CC340 0302 CC C C2674 C1C
CC341 0241 CC C C2731 C1C
CC342 -0600 CC C C2711 C1C
CC343 0500 CC C 52202 C1C
CC344 0241 CC C C2727 C1C
CC345 0131 CC C C0000 CC
CC346 0300 CC C 52202 C1C
CC347 0131 CC C C0000 CC
CC350 0260 CC C C2711 C1C
CC351 0402 CC C 52202 C1C
CC352 0100 CC C C0366 C1C
CC353 0120 CC C C0357 C1C
CC354 0560 CC 1 14543 C1C
CC355 0260 CC 1 14542 C1C
CC356 -0120 CC C C0366 C1C
CC357 1 C0001 1 C0360 C1C
CC360 0774 CC 4 C0000 CC
CC361 1 C0001 4 C0362 C1C
CC362 0634 CC 4 C0360 C1C
CC363 0634 CC 4 C0370 C1C
CC364 -3 C0000 4 C0253 C1C
CC365 0020 CC C C0371 C1C
CC366 0600 CC 1 14542 C1C
CC367 0600 CC 1 14542 C1C
CC370 0774 CC 4 C0000 CC
CC371 -3 C0000 4 C0373 C1C
CC372 -0634 CC 4 C0371 C1C
CC373 1 C0000 2 C0374 C1C
CC374 -3 C0000 2 C0246 C1C
CC375 -0534 CC 4 C0371 C1C
CC376 0634 CC 4 C0406 C1C
CC377 -0535 CC 4 C0371 C1C
CC400 1 C0000 4 C0401 C1C
CC401 -0634 CC 4 C0456 C1C
CC402 0500 CC 0 52204 C1C
CC403 0601 CC C C2723 C1C
CC404 0774 CC 2 C0001 CC
CC405 0600 CC C C2724 C1C
CC406 0774 CC 1 C0000 CC
CC407 0500 CC C C2724 C1C
CC410 0400 CC C 52206 C1C
CC411 0601 CC C C2724 C1C
CC412 0774 CC 4 C0000 CC
CC413 2 C0004 4 C0415 C1C
CC414 0634 CC 4 C2723 C1C
CC415 0634 CC 4 C0412 C1C
CC416 0074 CC 4 C0003 C1C
CC431 0100 CC C C0453 C1C
CC432 -1 C0000 2 07636 C1C
CC433 -1 C0000 2 14543 C1C
CC434 0500 CC C C2723 C1C
CC436 0100 CC C 52206 C1C
CC437 -1 C0000 2 C0000 CC
CC440 -1 C0000 2 C0000 CC

```

FMP STU
 FSB A2
 FCP B12
 STQ TEMP
 CLA =1.
 FCP B2
 XCA =1.
 FAD
 XCA TEMP
 FMP =1.
 SUB L3
 TZE L2B
 TPL ZSA,1
 LDG ZSA-1,1
 FMP L3
 TMI **1,1,1
 AXI **4
 TXI **1,4,1
 SXA L2C,4
 SXA L3B,4
 TXL L2,4,**
 L3C
 STZ ZSA-1,1
 STZ RSA-1,1
 AXI **4
 TXL **2,4,**
 SXD L3C,4
 TXI **1,2,**
 TXL L1,2,**
 LXD L3C,4
 SXA L5,4
 LDC L3C,4
 TXI **1,4,**
 SXD L7,4
 CLA =4
 STO VAR
 AXI 1,2
 STZ P
 AXI **1
 CLA P
 ACC =1
 STO P
 AXI **4
 TIX **2,4,4
 SXA VAR,4
 SXA L5A,4
 PRINT FVAR,.....,FCUTL,P,TP,V,PHI,DELTA,A,H,ZMIN
 CLA VAR
 TZE L7-3
 ICP RSA,2
 ICP ZSA,2
 CLA VAR
 SUB =1
 TZE L7-3
 ICP **2
 ICP **4,2

***=INTERMEDIATE IMA XSA
 ***=NMAX-1
 ***=INTERMEDIATE IMA XSA
 ***=I MAXSA
 ***=NMAX
 ***=NMAX*NUM
 ***=4*NMAX
 FCRMAT VARIABLE
 ***=I MAXSA
 PAGE NUMBER
 ***=SIGMA TRAJECTORIES STILL TO BE PRINTED

PRINT INITIAL
 SIGMA TRAJECTORIES
 ***=RSA-NMAX
 ***=ZSA-NMAX

CC441	0500	CC	C	C2723	CIC	CLA	VAR
CC442	04C2	CC	C	522C1	CIC	SUB	=2
CC443	0100	CC	C	C6453	CIC	TZE	L7-3
CC444	-1	CC000	2	CC000	CC	ICP	**2
CC445	-1	CC000	2	CC000	CC	ICP	**2
CC446	0500	CC	C	C2723	CIC	CLA	VAR
CC447	04C2	CC	C	52205	CIC	SUB	=3
CC450	0100	CC	C	C0453	CIC	TZE	L7-3
CC451	-1	CC000	2	CC000	CC	ICP	**2
CC452	-1	CC000	2	CC000	CC	ICP	**2
CC453	1	CC001	2	C6454	CIC	TXI	**2
CC454	2	CC001	1	C0432	CIC	TXI	**2
CC455	-1	CC000	2	CC000	CC	TIX	L6,1,1
CC456	1	CC000	2	CC457	CIC	TXI	**2
CC457	-3	CC000	2	CC046	CIC	TXL	**2
CC460	0774	CC	2	CC001	CC	AXT	1,2
CC461	0754	CC	2	CC000	CC	PXA	2
CC462	0734	CC	1	CC000	CC	PXA	1
CC463	0774	CC	4	CC001	CC	AXT	1,4
CC464	0634	CC	4	CC0573	CIC	SXA	L12C,4
CC465	0634	CC	4	CC603	CIC	SXA	L13B,4
CC466	0560	CC	1	21450	CIC	LDQ	RZA,1
CC467	0260	CC	1	21450	CIC	FMP	RZA,1
CC470	0601	CC	C	C2642	CIC	STO	TI
CC471	0560	CC	1	26355	CIC	LDQ	ZZA,1
CC472	0260	CC	1	26355	CIC	FMP	ZZA,1
CC473	0300	CC	C	C2642	CIC	FAD	TI
CC474	0560	CC	0	52203	CIC	LDQ	=-1.5
CC475	CC74	CC	4	CC002	CIC	GALL	.C130C
CC476	0131	CC	C	CC000	CC	XCA	K
CC477	0260	CC	C	C2700	CIC	FMP	TI
CC500	0601	CC	C	C2642	CIC	STO	TI
CC501	0500	CC	0	52202	CIC	CLA	=1.
CC502	0302	CC	0	C2642	CIC	FSB	TI
CC503	0601	CC	C	C2642	CIC	STO	TI
CC504	0131	CC	C	CC000	CC	XCA	RZA,1
CC505	0260	CC	1	21450	CIC	FMP	RZA,1,1
CC506	0601	CC	1	21447	CIC	STO	RZA,1,1
CC507	-0120	CC	0	CC602	CIC	TMI	L13A
CC510	0560	CC	0	C2642	CIC	LDQ	TI
CC511	0260	CC	1	26355	CIC	FMP	ZZA,1
CC512	0302	CC	C	C2676	CIC	FSB	VCT
CC513	0601	CC	1	26354	CIC	STO	ZZA,1,1
CC514	0402	CC	C	C2701	CIC	SUB	ZMIN
CC515	-0120	CC	C	CC601	CIC	TMI	L13
CC516	0500	CC	1	21450	CIC	CLA	RZA,1
CC517	-0100	CC	C	CC523	CIC	TNZ	**4
CC520	0500	CC	1	26354	CIC	CLA	ZZA,1,1
CC521	0302	CC	1	26355	CIC	FSB	ZZA,1
CC522	0120	CC	C	CC601	CIC	TPL	L13
CC523	0560	CC	1	21447	CIC	LDQ	RZA,1,1
CC524	0260	CC	1	21447	CIC	FMP	RZA,1,1
CC525	0601	CC	C	C2711	CIC	STO	TEMP
CC526	0560	CC	1	26354	CIC	LDQ	ZZA,1,1
CC527	0260	CC	1	26354	CIC	FMP	ZZA,1,1
CC530	0300	CC	C	C2711	CIC	FAD	TEMP
CC531	0402	CC	C	C2674	CIC	SUB	A2
CC532	-0120	CC	C	CC601	CIC	TMI	L13

**=RSA-2*NVAX
**=ZSA-2*NVAX

**=RSA-3*NVAX
**=ZSA-3*NVAX

**=4*NVAX-1MAXSA
**=NVAX*NUM

CCMPUTE
INITIAL
ZETA TRAJECTORIES

CC533	0500	CC	1	26355	C1C	CLA	ZZA,1
CC534	0302	CC	1	26354	C1C	FSB	ZZA-1,1
CC535	0601	CC	C	C2711	C1C	STO	TEMP
CC536	0500	CC	1	21450	C1C	CLA	RZA,1
CC537	0302	CC	1	21447	C1C	FSB	ZZA-1,1
CC540	0241	CC	C	C2711	C1C	FDP	TEMP
CC541	-0600	CC	C	C2726	C1C	STQ	B
CC542	0260	CC	C	C2726	C1C	FMP	B
CC543	0601	CC	C	C2727	C1C	STO	B2
CC544	0560	CC	1	26355	C1C	LDG	ZZA,1
CC545	0260	CC	C	C2726	C1C	FMP	B
CC546	0302	CC	1	21450	C1C	FSB	RZA,1
CC547	0601	CC	C	C2730	C1C	STO	B1
CC550	0131	CC	C	CC000	CC	XCA	
CC551	0260	CC	C	C2730	C1C	FMP	B1
CC552	0601	CC	C	C2731	C1C	STO	B12
CC553	0302	CC	C	C2674	C1C	FSB	A2
CC554	0241	CC	C	C2731	C1C	FDP	B12
CC555	-0600	CC	C	C2711	C1C	STQ	TEMP
CC556	0500	CC	C	52202	C1C	CLA	=1.
CC557	0241	CC	C	C2727	C1C	FDP	B2
CC560	0131	CC	C	CC000	CC	XCA	=1.
CC561	0300	CC	C	52202	C1C	FAD	
CC562	0131	CC	C	CC000	CC	XCA	
CC563	0260	CC	C	C2711	C1C	FMP	TEMP
CC564	0402	CC	C	52202	C1C	SUB	=1.
CC565	0100	CC	C	CC601	C1C	TZE	L13
CC566	0120	CC	C	CC572	C1C	TPL	L12B
CC567	0560	CC	1	26355	C1C	LDG	ZZA,1
CC570	0260	CC	1	26354	C1C	FMP	ZZA-1,1
CC571	-0120	CC	C	CC601	C1C	TMI	L13
CC572	1	CC	CC	1	CC573	C1C	L12B
CC573	0774	CC	C	CC000	CC	AXT	**1,1,1
CC574	1	CC	CC	1	CC575	C1C	**4
CC575	0634	CC	4	CC573	C1C	TXI	**1,4,1
CC576	0634	CC	4	CC603	C1C	SXA	L12C,4
CC577	-3	CC	CC	4	CC466	C1C	L13B,4
CC600	0220	CC	C	CC604	C1C	TRA	L12,4,**
CC601	0600	CC	1	21447	C1C	STZ	L13C
CC602	0600	CC	1	26354	C1C	STZ	RZA-1,1
CC603	0774	CC	4	CC000	CC	AXT	**4
CC604	-3	CC	CC	4	CC606	C1C	**2,4,**
CC605	-0634	CC	4	CC604	C1C	SXD	L13C,4
CC606	1	CC	CC	2	CC607	C1C	**1,2,**
CC607	-3	CC	CC	2	CC461	C1C	**1,2,**
CC610	-0534	CC	4	CC604	C1C	LXD	L13C,4
CC611	0634	CC	4	CC621	C1C	SXA	L15,4
CC612	-0535	CC	4	CC604	C1C	LDC	L13C,4
CC613	1	CC	CC	4	CC614	C1C	**1,4,**
CC614	-0634	CC	4	CC671	C1C	SXD	L17,4
CC615	0500	CC	C	52204	C1C	CLA	=4
CC616	0601	CC	C	C2723	C1C	STO	VAR
CC617	0774	CC	2	CC001	CC	AXT	1,2
CC620	0600	CC	C	C2724	C1C	STZ	P
CC621	0774	CC	1	CC000	CC	AXT	**1
CC622	0500	CC	C	C2724	C1C	CLA	P
CC623	0400	CC	C	52206	C1C	ADD	=1
CC624	0601	CC	C	C7724	C1C	STO	P

***=INTERMEDIATE IMAXZA

***=NPAX-1

***=INTERMEDIATE IMAXZA

***=IPAXZA

***=NPAX

***=NPAX*NUM

***=4**NPAX

***=IPAXZA

***ZETA TRAJECTORIES STILL TO BE PRINTED

***4
**2,4,4
VAR,4
L15A,4
FVAR,....,FCUT2,F,TP,V,PHI,DELTA,T,A,H,Z,MIN
VAR
L17-3
RZA,2
ZZA,2
VAR
=1
L17-3
**2
**2
**2
=2
L17-3
**2
**2
**2
VAR
=3
L17-3
**2
**2
**2,1
L16,1,1

***RZA-NMAX
***ZZA-NMAX

***RZA-2*NMAX
***ZZA-2*NMAX

***RZA-3*NMAX
***ZZA-3*NMAX

***4*MAX-1MAXZA
***MAX*NUM

PCINTS
FCR NEXT
APPRXIMATICN
TC TRAJECTORIES

***NMAX

00625	C774	CC	4	CC000	CC	L15A	AXT	**4
00626	2	CC0C4	4	CC630	C1C		TIX	**2,4,4
00627	0634	CC	4	C2723	C1C		SXA	VAR,4
00628	C634	CC	4	CC625	C1C		SXA	L15A,4
00629	C074	CC	4	CC003	C1C		PRINT	FVAR,....,FCUT2,F,TP,V,PHI,DELTA,T,A,H,Z,MIN
00630	0500	CC	0	C2723	C1C		CLA	VAR
00631	010C	CC	C	CC666	C1C	L16	TZE	L17-3
00632	-1	CC0CC	2	21450	C1C		IOP	RZA,2
00633	-1	CC0CC	2	26355	C1C		IOP	ZZA,2
00634	0500	CC	C	C2723	C1C		CLA	VAR
00635	0402	CC	C	52206	C1C		SUB	=1
00636	010C	CC	C	CC666	C1C		TZE	L17-3
00637	-1	CC0CC	2	CC000	CC	S26	IOP	**2
00638	-1	CC0CC	2	CC000	CC		IOP	**2
00639	-1	CC0CC	2	CC000	CC		CLA	VAR
00640	0500	CC	C	C2723	C1C		SUB	=2
00641	0402	CC	0	52201	C1C		TZE	L17-3
00642	010C	CC	C	CC666	C1C		IOP	**2
00643	-1	CC0CC	2	CC000	CC	S26A	IOP	**2
00644	-1	CC0CC	2	CC000	CC		IOP	**2
00645	0500	CC	C	C2723	C1C		CLA	VAR
00646	0402	CC	0	52205	C1C		SUB	=3
00647	010C	CC	C	CC666	C1C		TZE	L17-3
00648	-1	CC0CC	2	CC000	CC		IOP	**2
00649	-1	CC0CC	2	CC000	CC	S26B	IOP	**2
00650	-1	CC0CC	2	CC000	CC		IOP	**2
00651	2	CC0C1	1	CC645	C1C		TIX	**1,2,1
00652	-1	CC0CC	2	CC000	CC		TIX	L16,1,1
00653	-1	CC0CC	2	CC000	CC		ENDIO	
00654	-3	CC0C0	2	CC621	C1C	L17	TIX	**1,2,**
00655	0600	CC	C	C2714	C1C	L18	TXL	L15,2,**
00656	-0534	CC	4	CC371	C1C		STZ	CCUNT
00657	-0634	CC	4	CC604	C1C		LXD	L3C,4
00658	-0534	CC	4	CC604	C1C		SXD	L24AA,4
00659	0500	CC	C	C2714	C1C		LXD	L13C,4
00660	0500	CC	C	C2714	C1C	L2C	SXD	L34AA,4
00661	0402	CC	0	C2721	C1C		CLA	CCUNT
00662	010C	CC	C	CC013	C1C		SUB	NITER
00663	-1	CC0CC	2	CC000	CC		TZE	START
00664	-1	CC0CC	2	CC000	CC		SXD	L23C,C
00665	-1	CC0CC	2	CC000	CC		SXD	L33C,C
00666	2	CC0C1	1	CC645	C1C		AXT	1,2
00667	-1	CC0CC	2	CC000	CC		LXA	NUM,1
00668	-1	CC0CC	2	CC000	CC		CLA	RSA,2
00669	-3	CC0C0	2	CC621	C1C	S8.5	STO	RSB,2
00670	0600	CC	C	C2714	C1C		CLA	ZSA,2
00671	-0534	CC	4	CC371	C1C		STO	ZSB,2
00672	-0634	CC	4	CC604	C1C		CLA	RZA,2
00673	-0534	CC	4	CC604	C1C		STO	RZB,2
00674	0500	CC	C	C2714	C1C		CLA	ZZA,2
00675	0402	CC	0	C2721	C1C		STO	ZZB,2
00676	010C	CC	C	CC013	C1C		CLA	RSA+1,2
00677	-0634	CC	C	C1222	C1C		STO	RSB+1,2
00678	0774	CC	2	CC001	CC		CLA	ZSA+1,2
00679	C534	CC	1	C2717	C1C		STO	ZSB+1,2
00680	0500	CC	2	C7636	C1C		CLA	KZA+1,2
00681	0601	CC	2	33262	C1C		STO	ZZA+1,2
00682	0500	CC	2	14543	C1C		CLA	ZZB+1,2
00683	0601	CC	2	40166	C1C		STO	**1,2,**
00684	0500	CC	2	21450	C1C		CLA	
00685	0601	CC	2	45072	C1C		STO	
00686	0500	CC	2	26355	C1C		CLA	
00687	0601	CC	2	51776	C1C		STO	
00688	0500	CC	2	C7637	C1C		CLA	
00689	0601	CC	2	33263	C1C		STO	
00690	0500	CC	2	14544	C1C		CLA	
00691	0601	CC	2	40167	C1C		STO	
00692	0500	CC	2	21451	C1C		CLA	
00693	0601	CC	2	45073	C1C		STO	
00694	0500	CC	2	26356	C1C		CLA	
00695	0601	CC	2	51777	C1C		STO	
00696	1	CC0CC	2	CC730	C1C	S5	TXI	

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CC730 2 CCCC1 1 CC7C7 C1C
CC731 0774 CC 1 CCCC1 CC
CC732 0774 CC 2 CCCC1 CC
CC733 C5CC CC 2 CCCC CC
CC734 C3C2 CC 2 C7636 C1C
CC735 C601 CC C C27C4 C1C
CC736 C5CC CC 2 CCCC CC
CC737 C3CC CC 2 C7636 C1C
CC740 C131 CC C CCCC CC
CC741 C26C CC C C27C4 C1C
CC742 C6C1 CC 1 52166 C1C
CC743 1 CCCC1 1 C0744 C1C
CC744 1 CCCC 2 C0745 C1C
CC745 -3 CCCC 2 C0733 C1C
CC746 C5CC CC 2 C7636 C1C
CC747 C4CC CC C 522CC C1C
CC750 C3CC CC C C27C4 C1C
CC751 C131 CC C CCCC CC
CC752 C26C CC C C27C4 C1C
CC753 C6C1 CC 1 52166 C1C
CC754 C774 CC 2 CCCC1 CC
CC755 C5CC CC 2 CCCC CC
CC756 C3C2 CC 2 33262 C1C
CC757 C131 CC C CCCC CC
CC760 C26C CC C C27C3 C1C
CC761 C6C1 CC C C2677 C1C
CC762 C754 CC 2 CCCC CC
CC763 C734 CC 1 CCCC CC
CC764 C774 CC 4 CCCC1 CC
CC765 C634 CC 4 C1211 C1C
CC766 C634 CC 4 C1221 C1C
CC767 -0520 CC C C2641 C1C
CC770 CC2C CC C C1CC1 C1C
CC771 CC74 CC 4 CCCC3 C1C
CC772 C6C1 CC C C264C C1C
CC773 CC74 CC 4 CCCC3 C1C
CC775 -1 CCCC 1 33262 C1C
CC776 -1 CCCC 1 4C166 C1C
CC777 -1 CCCC C C264C C1C
C1CCC -1 CCCC C CCCC CC
C1CC1 C5CC CC 1 33262 C1C
C1CC2 C1CC CC C C1C40 C1C
C1CC3 C3C2 CC C C2677 C1C
C1CC4 C6C2 CC C C2660 C1C
C1CC5 C5CC CC 1 4C166 C1C
C1CC6 C6C1 CC C C2661 C1C
C1CC7 CC74 CC 4 C1726 C1C
C1CC 3 CCCC C 2145C C1C
C1C11 3 CCCC C 26355 C1C
C1C12 C6C1 CC C C2642 C1C
C1C13 C074 CC 4 C1726 C1C
C1C14 3 CCCC C C7636 C1C
C1C15 3 CCCC C 14543 C1C
C1C16 C3C2 CC C C2642 C1C
C1C17 C6C1 CC C C2643 C1C
C1C20 C5CC CC 1 33262 C1C
C1C21 C3CC CC C C2677 C1C
C1C22 C6C1 CC C C264C C1C

TIX
AXT
AXT
AXT
CLA
FSB
STO
GLA
FAD
XCA
FMP
STO
TXI
TXI
TXL
GLA
ADD
FAD
XCA
FMP
STO
AXT
CLA
FSB
XCA
FMP
STO
PXA
PAX
AXT
SXA
SXA
NXT
TRA
CALL
STO
PRINT
ICP
ICP
ICP
ENDIO
CLA
TZE
FSB
SLW
CLA
STC
TSX
PAR
PAR
STO
TSX
PAR
PAR
FSB
STO
CLA
FAD
STO

L2CA
L2CA
S29
L2CAC
L2CAE
L2CC
L21
L22
L22.1

SE,S,1,1
1,1
1,2
**2
RSA,2
DA
**2
RSA,2
DA
VEC,1
**1,1,1
**1,2,***
L2CA,2,***
=KIKS
DA
DA
VEC,1
1,2
**2
RSB,2
H
DELTA
,2
,1
1,4
L22C,4
L23B,4
PSW
L22.1
DAYTIM
X2
F1
RSB,1
ZSB,1
X2
RSB,1
L22.2
DELTA
RTEMP
ZSB,1
ZTEMP
PCNTNL,4
RZA
ZZA
T1
PCNTNL,4
RSA
ZSA
T1
T2
RSB,1
DELTA
RTEMP

```

**=RSA-NMAX

**=RSA-NMAX

**=NMAX

**=NMAX*(NUM-1)

**=RSB-NMAX

CMPUTE

NEXT

APPROXIMATIONS

TC TRAJECTORIES

01C23	GC74	CC	4	C1726	C1C	TSX	PCNTNL,4
01C24	3	CCCC	C	21450	C1C	PAR	RZA
01C25	3	CCCC	C	26355	C1C	PAR	ZZA
01C26	0601	CC	C	C2642	C1C	TSX	PCNTNL,4
01C27	0601	CC	C	C1726	C1C	PAR	RZA
01C30	3	CCCC	C	C7636	C1C	PAR	ZSA
01C31	3	CCCC	C	14543	C1C	FSB	T1
01C32	03C2	CC	C	C2642	C1C	FSB	T2
01C33	03C2	CC	C	C2643	C1C	FSB	DELTA
01C34	0241	CC	C	C2677	C1C	FDP	=.C357EE7
01C35	0260	CC	C	52177	C1C	FMP	
01C36	076C	CC	C	CCCC2	CC	CHS	
01C37	03CC	CC	1	33262	C1C	FAC	RSB,1
01C40	0601	CC	C	C2666	C1C	STU	RGP
01C41	05CC	CC	1	33262	C1C	CLA	RSB,1
01C42	0601	CC	C	C2660	C1C	STU	RTEMP
01C43	05CC	CC	1	40166	C1C	CLA	ZSB,1
01C44	03C2	CC	C	C2677	C1C	FSB	DELTA
01C45	0601	CC	C	C2661	C1C	STU	ZTEMP
01C46	0C74	CC	4	C1726	C1C	TSX	PCNTNL,4
01C47	3	CCCC	C	21450	C1C	PAR	RZA
01C50	3	CCCC	C	26355	C1C	PAR	ZZA
01C51	0601	CC	C	C2642	C1C	STU	T1
01C52	0C74	CC	4	C1726	C1C	TSX	PCNTNL,4
01C53	3	CCCC	C	C7636	C1C	PAR	RZA
01C54	3	CCCC	C	14543	C1C	PAR	ZSA
01C55	03C2	CC	C	C2642	C1C	FSB	T1
01C56	0601	CC	C	C2643	C1C	FSB	T2
01C57	05CC	CC	1	40166	C1C	CLA	ZSB,1
01C60	03CC	CC	C	C2677	C1C	FAD	DELTA
01C61	0601	CC	C	C2661	C1C	STU	ZTEMP
01C62	0C74	CC	4	C1726	C1C	TSX	PCNTNL,4
01C63	3	CCCC	C	21450	C1C	PAR	RZA
01C64	3	CCCC	C	26355	C1C	PAR	ZZA
01C65	0601	CC	C	C2642	C1C	STU	T1
01C66	0C74	CC	4	C1726	C1C	TSX	PCNTNL,4
01C67	3	CCCC	C	C7636	C1C	PAR	RZA
01C70	3	CCCC	C	14543	C1C	PAR	ZSA
01C71	03C2	CC	C	C2642	C1C	FSB	T1
01C72	03C2	CC	C	C2643	C1C	FSB	T2
01C73	0241	CC	C	C2677	C1C	FDP	DELTA
01C74	0260	CC	C	52177	C1C	FMP	=.C357EE7
01C75	076C	CC	C	CCCC2	CC	CHS	
01C76	03CC	CC	1	40166	C1C	FAD	ZSB,1
01C77	0601	CC	C	C2667	C1C	STU	ZGP
01C80	056C	CC	1	33262	C1C	LDQ	RSB,1
01C81	C260	CC	1	33262	C1C	FMP	RSB,1
01C82	0601	CC	C	C2642	C1C	STU	T1
01C83	0560	CC	1	40166	C1C	LDQ	ZSB,1
01C84	C260	CC	1	40166	C1C	FMP	ZSB,1
01C85	03CC	CC	C	C2642	C1C	FAD	T1
01C86	0601	CC	C	C2642	C1C	STU	T1
01C87	0131	CC	C	CCCC	CC	XCA	
01C88	C26C	CC	C	C2642	C1C	FMP	T1
01C89	0601	CC	C	C2643	C1C	STU	T2
01C90	0C74	CC	4	CCCC5	C1C	CALL	SCR1,T1
01C91	0241	CC	C	C2643	C1C	FDP	T2
01C92	C26C	CC	C	C27CC	C1C	FMP	K

L22.2

=.5/(4*PI)

01116	06C1	CC	C	C2642	C1C	STO	T1
01117	0131	CC	C	CC0CC	CC	XCA	RSB,1
01120	C26C	CC	1	33262	C1C	FMP	RGP
01121	03C0	CC	C	C2666	C1C	FAD	RSB-1,1
01122	06C1	CC	1	33261	C1C	STO	**2
01123	C1CC	CC	C	C1125	C1C	TZE	L23A
01124	-0120	CC	C	C122C	C1C	TMI	T1
01125	05C0	CC	C	C2642	C1C	LDQ	ZSB,1
01126	C26C	CC	1	40166	C1C	FMP	ZGP
01127	C3C0	CC	0	C2667	C1C	FAD	VLT
01130	03C2	CC	C	C2676	C1C	FSB	ZSB-1,1
01131	06C1	CC	1	4C165	C1C	STO	ZMIN
01132	04C2	CC	C	C2701	C1C	SUB	L23
01133	-0120	CC	0	C1217	C1C	TMI	RSB,1
01134	05C0	CC	1	33262	C1C	CLA	**4
01135	-C1CC	CC	C	C1141	C1C	TNZ	ZSB-1,1
01136	05CC	CC	1	4C165	C1C	CLA	ZSB,1
01137	03C2	CC	1	4C166	C1C	FSB	L23
01140	012C	CC	C	C1217	C1C	TPL	RSB-1,1
01141	05C0	CC	1	33261	C1C	LDQ	RSB-1,1
01142	02C0	CC	1	33261	C1C	FMP	RSB-1,1
01143	06C1	CC	C	C2711	C1C	STO	TEMP
01144	C5C0	CC	1	4C165	C1C	LDQ	ZSB-1,1
01145	02C0	CC	1	4C165	C1C	FMP	ZSB-1,1
01146	03CC	CC	C	C2711	C1C	FAD	TEMP
01147	C4C2	CC	C	C2674	C1C	SUB	A2
01150	-0120	CC	C	C1217	C1C	TMI	L23
01151	05C0	CC	1	4C166	C1C	CLA	ZSB,1
01152	03C2	CC	1	4C165	C1C	FSB	ZSB-1,1
01153	06C1	CC	C	C2711	C1C	STO	TEMP
01154	05C0	CC	1	33262	C1C	CLA	RSB,1
01155	03C2	CC	1	33261	C1C	FSB	RSB-1,1
01156	0241	CC	0	C2711	C1C	FDP	TEMP
01157	-C6C0	CC	C	C2726	C1C	STQ	B
01160	C2C0	CC	C	C2726	C1C	FMP	B
01161	06C1	CC	C	C2727	C1C	STO	B2
01162	05C0	CC	1	4C166	C1C	LDQ	ZSB,1
01163	C2C0	CC	C	C2726	C1C	FMP	B
01164	C3C2	CC	1	33262	C1C	FSB	RSB,1
01165	06C1	CC	C	C2730	C1C	STO	B1
01166	0131	CC	C	CC0CC	CC	XCA	B1
01167	C2C0	CC	C	C273C	C1C	FMP	B12
01170	06C1	CC	C	C2731	C1C	STO	A2
01171	03C2	CC	C	C2674	C1C	FSB	B12
01172	0241	CC	C	C2731	C1C	FDP	TEMP
01173	-C6C0	CC	C	C2711	C1C	STQ	=1.
01174	C5C0	CC	C	522C2	C1C	CLA	B2
01175	0241	CC	C	C2727	C1C	FDP	=1.
01176	0131	CC	C	CC0CC	CC	XCA	
01177	03C0	CC	C	522C2	C1C	FAD	
0120C	0131	CC	C	CC0CC	CC	XCA	TEMP
01201	C2C0	CC	C	C2711	C1C	FMP	=1.
01202	04C2	CC	0	522C2	C1C	SUB	L23
01203	C1C0	CC	C	C1217	C1C	TZE	L22B
01204	0120	CC	C	C121C	C1C	TPL	ZSB,1
01205	05C0	CC	1	4C166	C1C	LDQ	ZSB-1,1
01206	C2C0	CC	1	4C165	C1C	FMP	L23
01207	-0120	CC	C	C1217	C1C	TMI	

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0121C 1 CCCC1 1 C1211 C1C
01211 0774 CC 4 CCCC CC
01212 1 CCCC1 4 C1213 C1C
01213 0634 CC 4 C1211 C1C
01214 0634 CC 4 C1221 C1C
01215 -3 CCCC 4 C0767 C1C
01216 CC2C CC C1222 C1C
01217 C6CC CC 1 40165 C1C
01220 C6CC CC 1 33261 C1C
01221 0774 CC 4 CCCC CC
01222 -3 CCCC 4 C1224 C1C
01223 -0634 CC 4 C1222 C1C
01224 1 CCCC 2 C1225 C1C
01225 -3 CCCC 2 C0755 C1C
01226 -3 CCCC 2 C0762 C1C
01227 -0534 CC 2 C1222 C1C
0123C -0534 CC 4 C1222 C1C
01231 3 CCCC 4 C1233 C1C
01232 -0534 CC 4 C1231 C1C
01233 0634 CC 4 C1535 C1C
01234 0634 CC 4 C1573 C1C

01235 0754 CC 4 C0000 CC
01236 -0760 CC C C0003 CC
01237 C400 CC C C2720 C1C
0124C 0734 CC 4 C0000 CC
01241 -0634 CC 4 C1546 C1C
01242 1 CCCC 4 C1243 C1C
01243 -0634 CC 4 C1644 C1C
01244 -0634 CC 2 C1231 C1C
01245 0774 CC 2 C0001 CC
01246 0500 CC 2 C0000 CC
01247 0302 CC 2 45072 C1C
0125C 0131 CC C C0000 CC
01251 0260 CC C C2703 C1C
01252 0601 CC C C2677 C1C
01253 0754 CC 2 C0000 CC
01254 0734 CC 1 C0000 CC
01255 0774 CC 4 C0001 CC
01256 0634 CC 4 C1500 C1C
01257 0634 CC 4 C1510 C1C
0126C -0520 CC C C2641 C1C
01261 CC74 CC 4 C0004 C1C
01262 C601 CC C C2640 C1C
01263 CC74 CC 4 C0003 C1C
01265 -1 C0000 1 45072 C1C
01266 -1 C0000 1 51776 C1C
01267 -1 C0000 C C2640 C1C
0127C -1 C0000 C C0000 CC

01271 0500 CC 1 45072 C1C
01272 C100 CC C C1327 C1C
01273 C302 CC C C2677 C1C
01274 0602 CC C C2660 C1C
01275 0500 CC 1 51776 C1C
01276 0601 CC C C2661 C1C
01277 0C74 CC 4 C1726 C1C
0130C 3 C0000 C 21450 C1C
01301 3 C0000 C 26355 C1C

L22B TXI
L22C AXT
TXI
TXI
SXA
L22C,4
L23B,4
L22,4,**
L23C
ZSB-1,1
RSB-1,1
**4
**2,4,**
L23C,4
**1,2,**
L2CC,2,**
L21,2,**
L23C,2
L23C,4
**2,4,**
L24AA,4
L24B,4
L25,4
PXA
,4
SSM
ACD
PAX
,4
SXC
TXI
SXD
SXD
AXT
,2
L24AA,2
RZB,2
H
DELTA
,2
PAX
,1
AXT
1,4
SXA
L32C,4
SXA
L33B,4
NZT
PSW
DAYTIM
X2
STD
X2
PRINT
F1
RZB,1
ZRB,1
X2
ENDIO
RZB,1
L32,2
TZE
DELTA
FSB
RTEMP
SLW
ZRB,1
CLA
ZTEMP
STX
PCINTL,4
PAR
RZA
ZZA

**=INTERMEDIATE IMAXSB
**=NPAX-1
**=INTERMEDIATE IMAXSB
**=IMAXSB
**=NPAX
**=NPAX*(NUM-1)
**=NPAX*NUM
**=PREVCLUS IMAXSB
**=3*NPAX
**=RZB-NPAX

```


01302	06C1	CC	C	C2642	C1C	STO	T1	PCNTNL,4
01303	0C74	CC	4	C1726	C1C	TSX	RSA	
01304	3	CCCC	C	C7636	C1C	PAR	ZSA	
01305	3	CCCC	C	14543	C1C	FSB	T1	
01306	03C2	CC	C	C2642	C1C	STU	T2	
01307	06C1	CC	C	C2643	C1C	CLA	RZB,1	
01310	05C0	CC	1	45C72	C1C	FAD	DELTA	
01311	03C0	CC	C	C2677	C1C	STO	RTEMP	
01312	06C1	CC	C	C2660	C1C	TSX	PCNTNL,4	
01313	0C74	CC	4	C1726	C1C	PAR	RZA	
01314	3	CCCC	0	21450	C1C	PAR	ZZA	
01315	3	CCCC	C	26355	C1C	STO	T1	
01316	06C1	CC	C	C2642	C1C	TSX	PCNTNL,4	
01317	0C74	CC	4	C1726	C1C	PAR	RSA	
01320	3	CCCC	C	C7636	C1C	PAR	ZSA	
01321	3	CCCC	C	14543	C1C	FSB	T1	
01322	03C2	CC	C	C2642	C1C	FSB	T2	
01323	03C2	CC	C	C2643	C1C	FCP	DELTA	
01324	0241	CC	C	C2677	C1C	FMP	=.C357E87	
01325	0260	CC	C	52177	C1C	FAD	RZB,1	
01326	03C0	CC	1	45C72	C1C	STO	RGP	
01327	06C1	CC	C	C2666	C1C	CLA	RZB,1	
01330	05C0	CC	1	45C72	C1C	STU	RTEMP	
01331	06C1	CC	C	C2660	C1C	STU	ZZB,1	
01332	05C0	CC	1	51776	C1C	FSB	DELTA	
01333	03C2	CC	C	C2677	C1C	STO	ZTEMP	
01334	06C1	CC	C	C2661	C1C	TSX	PCNTNL,4	
01335	0C74	CC	4	C1726	C1C	PAR	RZA	
01336	3	CCCC	C	21450	C1C	PAR	ZZA	
01337	3	CCCC	C	26355	C1C	STO	T1	
01340	06C1	CC	C	C2642	C1C	TSX	PCNTNL,4	
01341	0C74	CC	4	C1726	C1C	PAR	RSA	
01342	3	CCCC	C	C7636	C1C	PAR	ZSA	
01343	3	CCCC	C	14543	C1C	FSB	T1	
01344	03C2	CC	C	C2642	C1C	STO	T2	
01345	06C1	CC	C	C2642	C1C	CLA	ZZR,1	
01346	05C0	CC	1	51776	C1C	FAD	DELTA	
01347	03C0	CC	C	C2677	C1C	TSX	ZTEMP	
01350	06C1	CC	C	C2661	C1C	PAR	PCNTNL,4	
01351	0C74	CC	4	C1726	C1C	PAR	RZA	
01352	3	CCCC	C	21450	C1C	STO	T1	
01353	3	CCCC	C	26355	C1C	PAR	ZZA	
01354	06C1	CC	C	C2642	C1C	STO	T1	
01355	0C74	CC	4	C1726	C1C	TSX	PCNTNL,4	
01356	3	CCCC	C	C7636	C1C	PAR	RSA	
01357	3	CCCC	C	14543	C1C	PAR	ZSA	
01360	03C2	CC	C	C2642	C1C	FSB	T1	
01361	03C2	CC	C	C2643	C1C	FSB	T2	
01362	0241	CC	C	C2677	C1C	FDP	DELTA	
01363	0260	CC	0	52177	C1C	FMP	=.C357E87	
01364	03C0	CC	1	51776	C1C	FAD	ZZB,1	
01365	06C1	CC	C	C2667	C1C	STC	ZGP	
01366	0560	CC	1	45C72	C1C	LDG	RZB,1	
01367	0260	CC	1	45C72	C1C	FMP	RZB,1	
01370	0601	CC	C	C2642	C1C	STO	T1	
01371	0560	CC	1	51776	C1C	LDQ	ZZB,1	
01372	0260	CC	1	51776	C1C	FMP	ZZB,1	
01373	03C0	CC	0	C2642	C1C	FAD	T1	

L32.2

01374	0601	CC	C	C2642	C1C	STO	T1
01375	0131	CC	C	CC000	CC	XCA	T1
01376	0260	CC	C	C2642	C1C	FMP	T1
01377	0601	CC	C	C2643	C1C	STO	T2
01400	0074	CC	4	CC005	C1C	CALL	SGRT,T1
01402	0241	CC	C	C2643	C1C	FDP	T2
01403	0260	CC	C	C2700	C1C	FMP	K
01404	0601	CC	C	C2642	C1C	STO	T1
01405	0131	CC	C	CC000	CC	XCA	RZB,1
01406	0260	CC	1	45072	C1C	FMP	RZB,1
01407	0760	CC	C	CC002	CC	CHS	RGP
01410	0300	CC	C	C2666	C1C	FAD	RZB-1,1
01411	0601	CC	1	45071	C1C	STO	L33A
01412	-0120	CC	C	C1507	C1C	TMI	L33
01413	0560	CC	C	C2642	C1C	LDQ	T1
01414	0260	CC	1	51776	C1C	FMP	ZZB,1
01415	0760	CC	C	CC002	CC	CHS	ZGP
01416	0300	CC	C	C2667	C1C	FAD	VCT
01417	0302	CC	C	C2676	C1C	FSB	ZZB-1,1
01420	0601	CC	1	51775	C1C	STO	ZMIN
01421	0402	CC	C	C2701	C1C	SUB	L33
01422	-0120	CC	C	C1506	C1C	TMI	RZB,1
01423	0500	CC	1	45072	C1C	CLA	ZZB-1,1
01424	-0100	CC	C	C1430	C1C	TNZ	ZZB,1
01425	0500	CC	1	51775	C1C	CLA	L33
01426	0302	CC	1	51776	C1C	FSB	RZB-1,1
01427	0120	CC	C	C1506	C1C	TPL	L33
01430	0560	CC	1	45071	C1C	LDQ	RZB-1,1
01431	0260	CC	1	45071	C1C	FMP	RZB-1,1
01432	0601	CC	C	C2711	C1C	STO	TEMP
01433	0560	CC	1	51775	C1C	LDQ	ZZB-1,1
01434	0260	CC	1	51775	C1C	FMP	ZZB-1,1
01435	0300	CC	C	C2711	C1C	FAD	TEMP
01436	0402	CC	C	C2674	C1C	SUB	A2
01437	-0120	CC	C	C1506	C1C	TMI	L33
01440	0500	CC	1	51776	C1C	CLA	ZZB,1
01441	0302	CC	1	51775	C1C	FSB	ZZB-1,1
01442	0601	CC	C	C2711	C1C	STO	TEMP
01443	0500	CC	1	45072	C1C	CLA	RZB,1
01444	0302	CC	1	45071	C1C	FSR	RZB-1,1
01445	0241	CC	C	C2711	C1C	FDP	TEMP
01446	-0600	CC	C	C2726	C1C	STQ	B
01447	0260	CC	C	C2726	C1C	FMP	B
01450	0601	CC	C	C2727	C1C	STU	B2
01451	0560	CC	1	51776	C1C	LDQ	ZZB,1
01452	0260	CC	C	C2726	C1C	FMP	B
01453	0302	CC	1	45072	C1C	FSB	RZB,1
01454	0601	CC	C	C2730	C1C	STO	B1
01455	0131	CC	C	CC000	CC	XCA	B1
01456	0260	CC	C	C2730	C1C	FMP	B1
01457	0601	CC	C	C2731	C1C	STO	B12
01460	0302	CC	C	C2674	C1C	FSB	A2
01461	0241	CC	C	C2731	C1C	FDP	B12
01462	-0600	CC	C	C2711	C1C	STQ	TEMP
01463	0500	CC	C	52202	C1C	CLA	=1.
01464	0241	CC	C	C2727	C1C	FDP	B2
01465	0131	CC	C	CC000	CC	XCA	=1.
01466	0300	CC	C	52202	C1C	FAD	=1.

01467	0131	CC	CCCC	CC	XCA	TEMP	
01470	0260	CC	C2711	C1C	FMP	=1,	
01471	0402	CC	C 52202	C1C	SUB	L33	
01472	0100	CC	C 01506	C1C	TZE	L32B	
01473	0120	CC	C 01477	C1C	TPL	ZZB,1	
01474	0560	CC	1 51776	C1C	LDQ	ZZB-1,1	
01475	0260	CC	1 51775	C1C	FMP	L33	
01476	-0120	CC	0 01506	C1C	TMI	**1,1,1	**=INTERMEDIATE IMAXB
01477	1 00001	1	01500	C1C	TXI	**4	
01500	0774	CC	4 00000	C0	AXT	**1,4,1	
01501	1 00001	4	01502	C1C	TXI	L32C,4	
01502	0634	CC	4 01500	C1C	SXA	L33B,4	**=NMAX-1
01503	0634	CC	4 01510	C1C	SXA	L32,4,**	
01504	-3 00000	4	01260	C1C	TXL	L33C	
01505	0020	CC	C 01511	C1C	TRA	ZZB-1,1	**=INTERMEDIATE IMAXB
01506	0600	CC	1 51775	C1C	STZ	ZZB-1,1	**=IMAXB
01507	0600	CC	1 45071	C1C	STZ	**4	
01511	-3 00000	4	01513	C1C	TXL	**2,4,**	
01512	-0634	CC	4 01511	C1C	SXD	L33C,4	**=NMAX
01513	1 00000	4	01514	C1C	TXI	**1,2,**	**=NMAX*(NUM-1)
01514	-3 00000	2	01246	C1C	TXL	L30C,2,**	**=NMAX*NUM
01515	-3 00000	2	01253	C1C	TXL	L31,2,**	
01516	-0534	CC	2 01511	C1C	LXD	L33C,2	
01517	-0534	CC	4 01511	C1C	LXD	L33C,4	**=PREVICUS IMAXB
01520	3 00000	4	01522	C1C	TXH	**2,4,**	
01521	-0534	CC	4 01520	C1C	LXD	L34AA,4	
01522	0634	CC	4 01551	C1C	SXA	L34B,4	
01523	0634	CC	4 01652	C1C	SXA	L35,4	
01524	0754	CC	4 00000	CC	PXA	,4	
01525	-0760	CC	C 0003	CC	SSM	NMAX	
01626	0400	CC	C 02720	C1C	ACD	,4	
01527	0734	CC	4 00000	CC	PAX	L34C,4	
01530	-0634	CC	4 01562	C1C	SXD	**1,4,**	**=3*NMAX
01531	1 00000	4	01532	C1C	TXI	L37,4	
01632	-0634	CC	4 01723	C1C	SXD	L34AA,2	
01534	0774	CC	2 00001	CC	AXT	1,2	**=MAXIMUM IMAXB
01535	0774	CC	1 00000	CC	AXT	**1	SHIFT NEW
01536	0500	CC	2 33262	C1C	CLA	RSB,2	SIGMA
01537	0601	CC	2 07636	C1C	STO	RSA,2	TRAJECTORIES
01540	0600	CC	2 32262	C1C	STZ	RSB,2	TC A
01541	0500	CC	2 40166	C1C	CLA	ZSB,2	MATRICES
01542	0601	CC	2 14543	C1C	STO	ZSA,2	
01543	0600	CC	2 40166	C1C	STZ	ZSB,2	
01544	1 00001	2	01545	C1C	TXI	**1,2,1	
01545	2 00001	1	01536	C1C	TIX	L24C,1,1	**=NMAX-(MAXIMUM IMAXB)
01546	1 00000	2	01547	C1C	TXI	**1,2,**	**=NMAX*NUM
01547	-3 00000	2	01535	C1C	TXL	L24B,2,**	
01550	0774	CC	2 00001	CC	AXT	1,2	**=MAXIMUM IMAXB
01551	0774	CC	1 00000	CC	AXT	**1	
01552	0500	CC	2 45072	C1C	CLA	RZB,2	
01553	0601	CC	2 21450	C1C	STO	RZA,2	
01554	0600	CC	2 45072	C1C	STZ	RZB,2	
01555	0500	CC	2 51776	C1C	CLA	ZZB,2	
01556	0601	CC	2 26355	C1C	STC	ZZA,2	
01557	0600	CC	2 51776	C1C	STZ	ZZB,2	
01560	1 00001	2	01561	C1C	TXI	**1,2,1	

```

01561 2 CCCC 1 C1552 C1C
01562 1 CCCC 2 C1563 C1C
01563 -3 CCCC 2 C1551 C1C
01564 0500 C C 2714 C1C
01565 0400 C C 52206 C1C
01566 0601 C C 02714 C1C
01567 0500 C C 0 52204 C1C
01570 0601 C C 0 2723 C1C
01571 0774 C C 2 CCCC C0
01572 0600 C C 0 2724 C1C
01573 0774 C C 1 CCCC C0
01574 0500 C C 2724 C1C
01575 0400 C C 52206 C1C
01576 0601 C C 0 2724 C1C
01577 0774 C C 4 CCCC C0
01600 2 CCCC 4 C1602 C1C
01601 0634 C C 4 2723 C1C
01602 0634 C C 4 C1577 C1C
01603 0074 C C 4 CCCC C1C
01616 0500 C C 0 2723 C1C
01617 0100 C C C1641 C1C
01620 -1 CCCC 2 07636 C1C
01621 -1 CCCC 2 14543 C1C
01622 0500 C C 0 2723 C1C
01623 0402 C C 0 52206 C1C
01624 0100 C C C1641 C1C
01625 -1 CCCC 2 CCCC C0
01626 -1 CCCC 2 CCCC C0
01627 0500 C C 0 2723 C1C
01630 0402 C C 0 52201 C1C
01631 0100 C C C1641 C1C
01632 -1 CCCC 2 CCCC C0
01633 -1 CCCC 2 CCCC C0
01634 0500 C C 0 2723 C1C
01635 0402 C C 0 52205 C1C
01636 0100 C C C1641 C1C
01637 -1 CCCC 2 CCCC C0
01640 -1 CCCC 2 CCCC C0
01641 1 CCCC 2 C1642 C1C
01642 2 CCCC 1 C1620 C1C
01643 -1 CCCC C CCCC C0
01644 1 CCCC 2 C1645 C1C
01645 -3 CCCC 2 C1573 C1C
01646 0500 C C 52204 C1C
01647 0601 C C 0 2723 C1C
01650 0774 C C 2 CCCC C0
01651 0600 C C 0 2724 C1C
01652 0774 C C 1 CCCC C0
01653 0500 C C 0 2724 C1C
01654 0400 C C 0 52206 C1C
01655 0601 C C 0 2724 C1C
01656 0774 C C 4 CCCC C0
01657 2 CCCC 4 C1661 C1C
01660 0634 C C 4 2723 C1C
01661 0634 C C 4 C1656 C1C
01662 0074 C C 4 CCCC C1C
01675 0500 C C 0 2723 C1C
01676 C100 C C 0 C1720 C1C

```

L34C,1,1
+1,2,
L34B,2,**
CCUNT
=1
COLUNT
=4
VAR
1,2
P
**+1
**=MAXIMUM IMAXSB
P
=1
P
**+4
**+2,4,4
VAR,4
L25A,4
FVAR,***,FCLT3,CCLNT,P,TP,V,PHI,DELTAT,A,H,ZMIN
VAR
L27-3
RSA,2
ZSA,2
VAR
=1
L27-3
**+2
**+2
**+2
VAR
=2
L27-3
**+2
**+2
=3
L27-3
**+2
**+2
VAR
=3
L27-3
**+2
**+2
**+2
TXI
L26,1,1
+1,2,
L25,2,**
=4
VAR
1,2
P
**+1
**=MAXIMUM IMAXSB
P
=1
P
**+4
**+2,4,4
VAR,4
L35A,4
FVAR,***,FCLT4,CCLNT,P,TP,V,PHI,DELTAT,A,H,ZMIN
VAR
L37-3

**=NPAX-(MAXIMUM IMAXZB)
**=NPAX*NUM
**=SIGMA TRAJECTORIES STILL TO BE PRINTED
**=RSA-NMAX
**=ZSA-NMAX
\$*=RZA-2*NMAX
**=ZSA-2*NMAX
**=RSA-3*NMAX
**=ZSA-3*NMAX
**=4*NMAX-(MAXIMUM IMAXSB)
**=NPAX*NUM
**=MAXIMUM IMAXZB
**=ZETA TRAJECTORIES STILL TO BE PRINTED

```

C1677 -1 CCCCC 2 2145C C1C ICP RZA*2
C1700 -1 CCCCC 2 26355 C1C ZZA*2
C1701 0500 CC C 2723 C1C VAR
C1702 0402 CC C 52206 C1C SUB =1
C1703 1100 CC C 1720 C1C TZE L37-3
C1704 -1 CCCCC 2 C0000 CC IGP **2
C1705 -1 CCCCC 2 C0000 CC IOP **2
C1706 0500 CC C 2723 C1C GLA VAR
C1707 0402 CC C 52201 C1C =2
C1710 1100 CC C 1720 C1C TZE L37-3
C1711 -1 CCCCC 2 C0000 CC IGP **2
C1712 -1 CCCCC 2 C0000 CC ICP **2
C1713 0500 CC C 2723 C1C GLA VAR
C1714 0402 CC C 52205 C1C SUB =3
C1715 1100 CC C 1720 C1C TZE L37-3
C1716 -1 CCCCC 2 C0000 CC IGP **2
C1717 -1 CCCCC 2 C0000 CC ICP **2
C1720 1 CCCCC 2 C1721 C1C TXI **1,2,1
C1721 2 CCCC1 1 C1677 C1C TIX L36,1,1
C1722 -1 CCCCC C C0000 CC ENDIO
C1723 1 CCCCC 2 C1724 C1C TXI **1,2,**
C1724 -3 CCCCC 2 C1652 C1C TXL L35,2,**
C1725 0200 CC C C0700 C1C TRA L2C
C1726 0634 CC 4 C2064 C1C SXA LL18,4
C1727 0634 CC 1 G2065 C1C SXA LL19,1
C1730 0634 CC 2 C2066 C1C SXA LL20,2
C1731 0500 CC 4 C0001 CC GLA 1,4
C1732 0621 CC C C1754 C1C STA LL6
C1733 0621 CC C C2045 C1C STA LL14
C1734 0500 CC 4 C0002 CC GLA 2,4
C1735 0621 CC C C1756 C1C STA LL7
C1736 0621 CC C C2047 C1C STA LL15
C1737 0774 CC 2 C0001 CC AXT 1,2
C1740 0634 CC 2 C1747 C1C SXA LL2,2
C1741 0634 CC 2 C2051 C1C SXA LL16,2
C1742 0600 CC C C2652 C1C STZ TT7
C1743 0754 CC 2 C0000 CC LL1 PAX ,2
C1744 0734 CC 1 C0000 CC PAX ,1
C1745 0600 CC C C2647 C1C STZ TT4
C1746 0600 CC C C2715 C1C CCLAT2
C1747 0774 CC 2 C0000 CC **2
C1750 0500 CC 2 52166 C1C VEC,2
C1751 0601 CC C C2644 C1C TTI
C1752 1 CCCCC 2 C1753 C1C TXI **1,2,1
C1753 0634 CC 2 C1747 C1C SXA LL2,2
C1754 0500 CC 1 C0000 CC **1
C1755 0601 CC C C2662 C1C STO TEMPR
C1756 0500 CC 1 C0000 CC **1
C1757 0601 CC C C2663 C1C STO TEMPZ
C1760 1 77777 1 C1761 C1C TXI **1,1,-1
C1761 0522 CC C C1754 C1C XEC LL6
C1762 0601 CC C C2664 C1C STO TEMPRR
C1763 0522 CC C C1756 C1C XEC LL7
C1764 0601 CC C C2665 C1C STC TEMPZ
C1765 0674 CC 4 C2070 C1C TSX PCNT,4
C1766 0300 CC C C2647 C1C FAD TT4
C1767 0601 CC C C2647 C1C STO TT4
C1770 0560 CC C C2664 C1C LDG TEMPRR

```

***4*MAX-(MAXIMUM I MAXZB)
 **=NMAX*NUM

SAVE
 INDEX
 REGISTERS
 TRANSFER
 ARGUMENTS

***1ST VEC SUBSCRIPT CN CURRENT TRAJECTORY

***ADDRESS COMPLEMENT FOR TSX TO POINTL
***CONTENTS OF XR1 BEFORE TSX TO POINTL
***CONTENTS OF XR2 BEFORE TSX TO POINTL

02064	0774	CC 4	CC000	CC	AXT	***,4
02065	0774	CC 1	CC000	CC	AXT	***,1
02066	0774	CC 2	CC000	CC	AXT	***,2
02067	CC20	CC 4	CC003	CO	TRA	3,4
02070	0634	CC 4	C2260	C10	SXA	RET,4
02071	0500	CC 0	C2660	C10	CLA	RTEMP
02072	0302	CC 0	C2662	C10	FSB	TEMPR
02073	0601	CC 0	C2646	C10	STO	TT3
02074	0131	CC C	CC000	CC	XCA	TT3
02075	0260	CC C	C2646	C10	FMP	TT3
02076	0601	CC C	C2646	C10	STO	TT3
02077	0500	CC C	C2661	C10	CLA	ZTEMP
02100	0302	CC C	C2663	C10	FSB	TEMPZ
02101	0601	CC C	C2645	C10	STO	TT2
02102	0131	CC C	CC000	CO	XCA	TT2
02103	0260	CC C	C2645	C10	FMP	TT2
02104	0300	CC C	C2646	C10	FAD	TT3
02105	0601	CC C	C2646	C10	STO	TT3
02106	0402	CC 0	C2675	C10	SUB	V2
02107	0120	CC C	C2131	C10	TPL	LL2CA
02110	0500	CC C	C2664	C10	CLA	TEMPRR
02111	0302	CC C	C2662	C10	FSB	TEMPR
02112	0601	CC C	C2645	C10	STO	TT2
02113	0131	CC C	CC000	CO	XCA	TT2
02114	0260	CC C	C2645	C10	FMP	TT2
02115	0601	CC C	C2645	C10	STO	TT2
02116	0500	CC C	C2665	C10	CLA	TEMPZZ
02117	0302	CC C	C2663	C10	FSB	TEMPZ
02120	0601	CC C	C2702	C10	STO	ALP
02121	0131	CC C	CC000	CO	XCA	ALP
02122	0260	CC C	C2702	C10	FMP	ALP
02123	0300	CC C	C2645	C10	FAD	TT2
02124	0241	CC C	52175	C10	FDP	=3.1415927
02125	0131	CC C	CC000	CO	XCA	TT2
02126	0601	CC C	C2645	C10	STO	TT2
02127	0402	CC C	C2646	C10	SUB	TT3
02130	0120	CC C	C2141	C10	TPL	LL12
02131	0500	CC C	C2660	C10	CLA	RTEMP
02132	0241	CC C	52174	C10	FDP	=.5
02133	0260	CC C	C2662	C10	FMP	TEMPR
02134	0601	CC C	C2713	C10	STO	CAPB
02135	0300	CC C	C2646	C10	FAD	TT3
02136	0601	CC C	C2712	C10	STO	CAPA
02137	0074	CC 4	C2262	C10	TSX	INPCNT,4
02140	CC20	CC C	C2260	C10	TRA	RET
02141	0074	CC 4	CC005	C10	CALL	SQRT,TT2
02143	0601	CC C	C2650	C10	STO	TT5
02144	0600	CC C	C2655	C10	STZ	TTIC
02145	0500	CC C	C2662	C10	CLA	TEMPR
02146	0241	CC C	C2650	C10	FDP	TT5
02147	0260	CC C	52175	C10	FMP	=3.1415927
02150	0300	CC C	52173	C10	UFA	=K233K9
02151	0601	CC C	C2653	C10	STO	TT8
02152	0322	CC 0	52173	C10	ERA	=K233K9
02153	0402	CC C	52206	C10	SUB	=1
02154	0340	CC C	52206	C10	CAS	=1
02155	CC20	CC C	C2162	C10	TRA	**+5
02156	CC20	CC C	C2157	C10	TRA	**+1

NORMAL
POTENTIAL
CALCULATION

CALCULATION
FOR PCINTS

SPHERE CF
EXCLUSION

02157	0500	CC	0	52202	C1C	CLA	=1.
02160	0601	CC	0	02653	C10	STO	TT8
02161	0C20	CC	0	C2223	C1C	TRA	SS2+4
02162	0601	CC	0	C2654	C1C	STO	TT9
02163	0500	CC	0	C2653	C1C	CLA	TT8
02164	0300	CC	0	52173	C10	FAD	=K233K9
02165	0601	CC	0	C2653	C1C	STO	TT8
02166	0500	CC	0	52172	C1C	CLA	=6.2831853
02167	0241	CC	0	C2653	C1C	FDP	TT8
02170	0600	CC	0	C2706	C10	STQ	DPHI
02171	0131	CC	0	CC000	CC	XCA	PHIN
02172	0601	CC	0	C2705	C1C	STO	PHIN
02173	0C74	CC	4	CC006	C1C	CALL	CGS,PHIN
02175	0760	CC	0	CC002	CC	CHS	
02176	0300	CC	0	52202	C10	FAD	=1.
02177	0601	CC	0	C2657	C1C	STO	TT4C
02200	0074	CC	4	CC005	C10	CALL	SQRT,TT4C
02202	0601	CC	0	C2657	C1C	STO	TT4C
02203	0500	CC	0	52202	C10	CLA	=1.
02204	0241	CC	0	C2657	C10	FDP	TT40
02205	0131	CC	0	CC000	CC	XCA	
02206	0300	CC	0	02655	C10	FAD	TTIC
02207	0601	CC	0	02655	C10	STO	TTIC
02210	0500	CC	0	C2654	C10	CLA	TT9
02211	0402	CC	0	52206	C10	SUB	=1
02212	0100	CC	0	C2217	C10	TZE	SS2
02213	0601	CC	0	C2654	C10	STO	TT9
02214	0500	CC	0	C2705	C1C	CLA	PHIN
02215	0300	CC	0	02706	C10	FAD	DPHI
02216	0020	CC	0	C2172	C10	TRA	SS3
02217	0500	CC	0	C2655	C10	CLA	TTIC
02220	0241	CC	0	C2662	C1C	FDP	TEMPR
02221	0260	CC	0	52171	C10	FMP	=.70711
02223	0500	CC	0	52170	C10	CLA	TTIC
02224	0241	CC	0	C2650	C1C	FDP	=1.5
02225	0131	CC	0	CC000	CC	XCA	TT5
02226	0300	CC	0	02655	C10	FAD	TTIC
02227	0131	CC	0	CC000	CC	XCA	
02230	0241	CC	0	C2653	C1C	FDP	TT8
02231	0260	CC	0	52167	C10	FMP	=6.2831853072
02232	0601	CC	0	C2655	C10	STO	TTIC
02233	0074	CC	4	CC005	C1C	CALL	SQRT,TT3
02235	0601	CC	0	C2646	C1C	STO	TT3
02236	0500	CC	0	02650	C1C	CLA	TT5
02237	0241	CC	0	C2646	C1C	FDP	TT3
02240	0600	CC	0	C2646	C1C	STQ	RTEMP
02241	0500	CC	0	C2660	C1C	CLA	TEMPR
02242	0302	CC	0	02662	C10	FSB	TEMPR
02243	0131	CC	0	CC000	CC	XCA	
02244	0260	CC	0	C2646	C1C	FMP	TT3
02245	0300	CC	0	C2662	C1C	FAD	TEMPR
02246	0241	CC	0	52174	C10	FDP	=5
02247	0260	CC	0	C2662	C1C	FMP	TEMPR
02250	0601	CC	0	C2713	C1C	STO	CAPB
02251	0300	CC	0	C2645	C1C	FAD	TT2
02252	0601	CC	0	02712	C1C	STO	CAPA
02253	0C74	CC	4	C2262	C1C	TSX	INPCNT,4

52202 2014CCCCCCCC CC
52203 6016CCCCCCCC CC
52204 CCCCCCCCCCCC04 CC
52205 CCCCCCCCCCCC03 CO
52206 CCCCCCCCCCCC01 CO

*** 52207 IS THE FIRST LOCATION NOT USED BY THIS PROGRAM.



