## WILEY

## Computation of maximal output admissible sets for linear systems with polynomial constraints

# Daniel R. Ramirez<sup>3</sup> | Daniel Limon<sup>3</sup>

Andres Cotorruelo<sup>1</sup> | Emanuele Garone<sup>1</sup> | Ilya V. Kolmanovsky<sup>2</sup>

<sup>1</sup>Department of Automation and System Analysis, Université Libre de Bruxelles, Brussels, Belgium

<sup>2</sup>College of Engineering, University of Michigan, Michigan, Ann Arbor, USA

<sup>3</sup>Department of Systems Engineering and Automation, Universidad de Sevilla, Seville, Spain

#### Correspondence

Andres Cotorruelo, Department of Automation and System Analysis, Université Libre de Bruxelles, Brussels, Belgium. Email:

Andres.Cotorruelo.Jimenez@ulb.be

#### **Funding information**

Agencia Estatal de Investigación, Grant/Award Number: PID2019-106212RB-C41/AEI/10.13039/ 501100011033: Fédération Wallonie-Bruxelles; National Science Foundation, Grant/Award Number: 1931738

#### Abstract

In this technical note we study the computation of the Maximal Output Admissible Set for linear systems subject to polynomial constraints. The computation of an inner approximation of the Maximal Output Admissible Sets requires the determination of constraint redundancy. We use a procedure to determine polynomial constraint redundancy based on a consequence of Putinar's Positivstellensatz. Further, we present a modification of the algorithm to compute the Maximal Output Admissible Set with improved performance. Lastly, demonstrate the potential for practical applications in two case studies of spacecraft rendezvous and control of an electromagnetic actuator.

#### **KEYWORDS**

constrained control, maximal output admissible set, nonlinear systems, redundant constraints, sum of squares

#### **INTRODUCTION** 1

With applications ranging from estimating regions of attraction,<sup>1,2</sup> to the analysis of uncertain systems,<sup>3</sup> and to constrained control,<sup>4,5</sup> set-theoretic methods keep proving themselves very useful to characterize and control dynamic systems.6

This note focuses on a particularly useful family of sets: the Maximal Output Admissible Sets (MOASs). These sets were considered in Reference 7, where a MOAS was defined as "the set of initial states such that the output will fulfill the constraints in the future." More recently, this definition was extended from considering only outputs to considering every arbitrary combination of states and constant input signals.<sup>8,9</sup> The MOAS is very important for the design of many constrained control schemes. In particular, it is the key ingredient for the formulation of Reference/Command Governor schemes,<sup>9</sup> of multimode control schemes,<sup>10</sup> and is also a commonly used terminal set in Model Predictive Control (MPC) schemes.<sup>11</sup> The use of these schemes is quite common, for instance, in aerospace control applications, see for example, References 12-18.

However, at the current stage, the availability of algorithms to compute efficiently this MOASs is mainly restricted to linear systems subject to linear constraints.<sup>9</sup> This note develops a method to compute an efficient representation <sup>2 of 10</sup> WILEY

of the MOAS for linear systems subject to general polynomial constraints. This method significantly enlarges the classes of applications where MOAS is needed, for example, to systems which are feedback linearized in which case the dynamics become linear while constraints become nonlinear after the transformations and can be approximated as polynomial.

The note is organized as follows. We first give a brief introduction to the algorithmic determination of the MOAS based on certificates of constraint redundancy. We then provide a systematic way to generate these certificates in the case of polynomial constraints. Further, we modify the algorithm to determine the MOAS to improve its performance based on several observations. Finally, we demonstrate the methodology on two relevant examples of spacecraft rendezvous and electromagnetic actuator control that suggest significant potential for its practical impact.

## 2 | PROBLEM STATEMENT

Consider a prestabilized<sup>1</sup> closed loop discrete-time linear system,

$$x(t+1) = Ax(t) + Bv(t),$$
 (1)

where  $x(t) \in \mathbb{R}^n$ ,  $v(t) \in \mathbb{R}^m$ ,  $A \in \mathbb{R}^{n \times n}$  is Schur,  $B \in \mathbb{R}^{n \times p}$ , and  $t \in \mathbb{Z}_{\geq 0}$  designates the discrete-time instant. This system is subject to constraints in the form  $(x, v) \in C$ , where

$$C = \{ (x, v) : c_i(x, v) \ge 0, \ i = 1, \dots, n_c \},$$
(2)

with  $c_i \in \mathbb{R}[x, v]$ , where by  $\mathbb{R}[x, v]$  we denote the set of all polynomials in the variables  $x \in \mathbb{R}^n$  and  $v \in \mathbb{R}^m$ . Since system (1) describes a closed-loop system, the input v(t) can be seen as a set-point reference for the closed loop system. Furthermore, the set *C* is determined by constraints in x(t) and v(t) that characterize state and input constraints on the original open-loop system.

The objective of this paper is to present a procedure to compute the MOAS for this class of systems and constraints. For a prestabilized system, the MOAS can be defined as the set of states x and references v such that, if x is the initial state and the set-point v is kept constant, constraints will not be violated. More formally the MOAS can be defined as

**Definition 1** (Maximal Output Admissible Set). Consider a prestabilized system (1) subject to  $(x, v) \in C$ . The MOAS is defined as  $\mathcal{O}_{\infty} = \{(x, v) : (\hat{x}(k|x, v), v) \in C, k \in \mathbb{Z}_{\geq 0}\}$ , where  $\hat{x}(k|x, v) = A^k x + \sum_{j=0}^{k-1} A^j B v$  is the prediction of the state at time *k* given the initial state *x* and constant applied reference *v*.

## 3 | GENERAL CONSIDERATIONS ON THE COMPUTATION OF $ilde{\mathcal{O}}_\infty$

Since  $\mathcal{O}_{\infty}$  is, in general, described by an infinite number of constraints, it is a common practice to use  $\tilde{\mathcal{O}}_{\infty}$ , which is a slightly tightened version of  $\mathcal{O}_{\infty}$ , defined as  $\tilde{\mathcal{O}}_{\infty} = \mathcal{O}_{\infty} \cap \mathcal{O}^{\epsilon}$  with  $\mathcal{O}^{\epsilon} = \{(x, v) : (\bar{x}_{v}, v) \in (1 - \epsilon)C\}$ , where  $\bar{x}_{v} = (I_{n} - A)^{-1}Bv$  is the steady-state associated to the constant applied reference v and  $\epsilon > 0$  is a small constant. The set  $\tilde{\mathcal{O}}_{\infty}$  has a number of very desirable properties:<sup>7</sup> it is positively invariant, it can approximate arbitrary well  $\mathcal{O}_{\infty}$  by decreasing  $\epsilon$ , and, most notably, if  $\tilde{\mathcal{O}}_{\infty}$  is compact, then it is also *finitely determined, that is*, and there exists a  $k^{*}$  such that

$$\tilde{\mathcal{O}}_{\infty} = \tilde{\mathcal{O}}_{k^*} = \{(x, v) : (\hat{x}(k|x, v), v) \in \mathcal{C}, \forall k = 1, \dots, k^*\} \cap \mathcal{O}^{\epsilon}.$$

In general, using Lyapunov arguments<sup>20</sup> and possibly solving two (in general nonconvex) optimization problems it is possible to compute an upper bound on  $k^*$ ,  $\overline{k} \ge k^*$  for general class of systems and constraints.<sup>9</sup> However, this upper bound is typically very conservative. As a consequence, to describe the MOAS as  $\tilde{\mathcal{O}}_{\infty} = \tilde{\mathcal{O}}_{\overline{k}}$  results in an unnecessarily large number of constraints which might sensibly slow down the control algorithms making use of such a set.

<sup>&</sup>lt;sup>1</sup>It is important to remark that this is not a limiting assumption; in fact, the RG literature is based on this assumption as well as a significant part of the MPC literature, for example, semi-feedback MPC.<sup>19</sup>

In Reference 7 it was proven that  $k^*$  can be defined as the smallest k such that  $\tilde{\mathcal{O}}_k^* = \tilde{\mathcal{O}}_{k^*+1}$ . Accordingly, an iterative algorithm able to determine  $\tilde{\mathcal{O}}_{\infty}$  (Algorithm 1) and  $k^*$  was presented, which, in principle, allows to find an optimal<sup>2</sup> representation of  $\tilde{\mathcal{O}}_{\infty}$ .

#### Algorithm 1. Computation of $\tilde{\mathcal{O}}_{\infty}$ 7

$$\begin{split} \tilde{\mathcal{O}} \leftarrow \mathcal{O}^{\epsilon} \\ k \leftarrow 0 \\ \rho \leftarrow 1 \\ \textbf{while } \rho \neq 0 \land k \leq \bar{k} \textbf{ do} \\ \rho \leftarrow 0 \\ \textbf{for } i = 1, ..., n_c \textbf{ do} \\ & \textbf{if } c_i(\hat{x}(k|x, v), v) \geq 0 \text{ is not redundant with respect to } \tilde{\mathcal{O}} \textbf{ then} \\ & \rho \leftarrow 1 \\ & \tilde{\mathcal{O}} \leftarrow \tilde{\mathcal{O}} \cap \{(x, v) | c_i(\hat{x}(k|x, v), v) \geq 0\} \\ & \textbf{end if} \\ & \textbf{end for} \\ & k \leftarrow k + 1 \\ \textbf{end while} \\ \tilde{\mathcal{O}}_{\infty} = \tilde{\mathcal{O}} \end{split}$$

The main difficulty in the implementation of Algorithm 1 is to check if the constraint  $c_i(\hat{x}(k|x, v), v) \ge 0$  is redundant or not with respect to to the set  $\tilde{\mathcal{O}}$ . By definition, given a general constraint  $c(z) \ge 0$  with  $z \in \mathbb{R}^{n_z}$ , c(z) > 0 is redundant with respect to a set  $\mathcal{K} \subset \mathbb{R}^{n_z}$  if and only if  $c(z) \ge 0$ ,  $\forall z \in \mathcal{K}$ . Using this fact, a common way to build a redundancy certificate is to compute

$$\gamma^* = \min_{z \in \mathcal{X}} (z), \tag{3}$$

and to note that  $\gamma^* \ge 0$  if and only if the constraint is redundant.

The main problem with this formulation is that it provides a redundancy certificate only in the case one can compute the exact optimal solution of (3). If instead one can compute only a suboptimal solution  $\gamma \leq \gamma^*$ , the method only provides a nonredundancy certificate in the case  $\gamma < 0$ , otherwise nothing can be said. Accordingly, the above approach can be used to remove redundant constraints only when it is reasonable to solve exactly problem (3), for example, when (3) is convex. Note that for (3) to be convex the constraint to be checked must be concave and the set  $\mathcal{K}$  needs to be convex. This means that this method can be used for the computation of  $\tilde{\mathcal{O}}_{\infty}$  only in the case of linear systems subject to linear constraints. A number of efficient implementations of this idea of certifying the redundancy of linear constraints have been proposed in the literature, including Linear Programming approaches,<sup>21</sup> various heuristics<sup>22</sup> and deterministic methods.<sup>23</sup> For an extensive survey on the subject, the reader is referred to Reference 24. In the Reference Governor literature, the Linear Programming approach is customarily used in conjunction with Algorithm 1 to compute  $\tilde{\mathcal{O}}_{\infty}$  for systems subject to linear constraints.<sup>9</sup>

For what concerns the case of nonlinear constraints, to the best of the authors' knowledge, the existing literature does not provide systematic methods for the certification of redundancy, and as a consequence for the computation of  $\tilde{\mathcal{O}}_{\infty}$ . Indeed, so far most of the literature dealing with nonlinear systems and/or constraints has used either the upper bound  $\bar{k}$  derived by Lyapunov arguments that can be very conservative or, more often, empirically estimated "sufficiently long" prediction horizons.<sup>20,25</sup> In this paper we propose a novel procedure that allows to ascertain whether a polynomial constraint is redundant with respect to a set defined by polynomial constraints, thus allowing to compute for efficient representations of  $\tilde{\mathcal{O}}_{\infty}$ . The proposed solution makes use of the Sum of Squares framework.

<sup>&</sup>lt;sup>2</sup>For the sake of precision it is worth remarking that, in principle, the representation output by Algorithm 1 may not be minimal: once a constraint is added to  $\tilde{O}_k$  it is not checked for redundancy at a later stage and it might have become redundant in later iterations of the algorithm. Consequently, to obtain a minimal representation of  $\tilde{O}_{\infty}$  it is necessary to assess the redundancy of every constraint that defines this set, eliminating those which are determined to be so.

## 4 of 10 WILEY-

## 4 | REDUNDANCY CERTIFICATE FOR POLYNOMIAL CONSTRAINTS

In order to build a redundancy certificate for nonlinear constraints we recall the Putinar's formulation of the Positivstellensatz (P-satz)<sup>26</sup>

**Theorem 1** (Putinar, 1993). Consider a polynomial  $p(z) \in \mathbb{R}[z]$  and a set  $\mathcal{K} = \{z : q_i(z) \ge 0, i = 1, ..., N\}$ , where  $q_i(z) \in \mathbb{R}[z], i = 1, ..., N$ . Then every  $p(z) > 0 \ \forall z \in \mathcal{K}$  can be written as

$$p(z) - \sum_{i=1}^{N} s_i(z) q_i(z) \in \Sigma[z],$$

$$\tag{4}$$

for some  $s_i \in \Sigma[z]$ , i = 1, ..., N in which  $\Sigma[z]$  represents the set of all Sum of Squares polynomials.<sup>27</sup>

It is important to remark that this theorem is nonconstructive and that the implication it states is

$$p(z) > 0 \ \forall z \in \mathcal{K} \Rightarrow (4)$$
 holds.

Instead, we will use the converse result, which is summarized in the following corollary:

**Corollary 1.** Let  $p(x) \in \mathbb{R}[x]$  and  $\mathcal{K} = \{x \in \mathbb{R}^n : \{f_i(x)\}_{i=1}^m \ge 0\}$ , then p(x) is nonnegative over  $\mathcal{K}$  if there exist  $\{s_i(x)\}_{i=0}^m \in \Sigma[x]$  such that

$$p(x) = s_0(x) + \sum_{i=1}^m s_i(x) f_i(x).$$

Note that, by construction, the polynomial p(x) is positive over  $\mathcal{K}$  since it is the sum of strictly positive terms over  $\mathcal{K}$ . This corollary can be used to obtain a *redundancy certificate* of a polynomial constraint with respect to a set consisting of polynomial inequalities. In fact, if there exist  $s_1, \ldots, s_n \in \Sigma[z]$  such that  $p(z) - \sum_{i=1}^n s_i(z)q_i(z)$  is a sum of squares polynomial,  $p(z) \ge 0$  is certified to be redundant with respect to  $\mathcal{K}$ . This condition can be checked using the following Sum of Squares Programming (SOSP) feasibility test

find 
$$s_i(z), i = 1, ..., n_c$$
  
s.t.  
 $p(z) - \sum_{i=1}^{n_c} s_i(z)q_i(z) \in \Sigma[z]$   
 $s_i \in \Sigma[z], \quad i = 1, ..., n_c.$  (5)

In many cases it is also of interest to quantify "how much" a constraint is redundant. A possible way to do so is to maximize on a slack variable  $\gamma$  as follows

$$\gamma^* = \max \gamma$$
  
s.t  
$$p(z) - \sum_{i=1}^{n_c} s_i(z) q_i(z) - \gamma \in \Sigma[z]$$
$$s_i(z) \in \Sigma[z], \quad i = 1, \dots, n_c.$$
 (6)

In this formulation, if  $\gamma^*$  is positive, the constraint  $p(z) \ge 0$  is redundant with respect to  $\mathcal{K}$ . This slack variable approach can also be used to assess which constraints are "almost redundant." Almost redundant constraints are those whose associated  $\gamma^*$  is negative and small in absolute value, and can be potentially eliminated using inner approximations. This concept is explored in Reference 28 where it was shown that the elimination of almost redundant constraints and a pull-in transformation can be used to compute simple inner approximations of  $\tilde{\mathcal{O}}_{\infty}$  and implement reference governors with a significant reduction in online computational time and effort.

As proven in Reference 29, it is possible to determine whether a polynomial is SOS through a Semi-Definite Programming (SDP) optimization problem, which means that optimization problem (6) can be efficiently solved by off-the-shelf solvers, for example, Reference 30. Additionally, there exist software suites that provide an interface between SOSP and SDP, for example Reference 31, thus transforming the optimization problem (6) into an LMI problem.

## 5 | EFFICIENT COMPUTATION OF $\tilde{\mathcal{O}}_{\infty}$

Note that using Algorithm 1 and the proposed redundancy certificate to compute  $\tilde{\mathcal{O}}_{\infty}$  requires performing  $(k^* + 1) \cdot n_c$ LMI feasibility tests as in (5). Accordingly, every iteration will take longer than the previous one, since the number of inequalities describing  $\tilde{\mathcal{O}}_k$  grows by up of  $n_c$  every iteration. Furthermore, the number of variables of the optimization problem increases as well, since we need to declare a new SOS multiplier  $s_i$  for every new inequality. Interestingly, it is possible to reduce the number of redundancy checks and consequently the computational time by using the following proposition.

**Proposition 1.** If a constraint  $c_j(x, v) \ge 0$  is redundant at iteration k' with respect to  $\tilde{\mathcal{O}}_{k'}$ , then it will be redundant for any iterations k > k'.

*Proof.* Let  $C_i = \{(x, v) : c_i(x, v) \ge 0\}, i = 1, ..., n_c$  and let us assume that constraint *j* becomes redundant at iteration k'. By definition

$$(\hat{x}(k|x,\nu),\nu) \in \bigcap_{i=1}^{n_c} C_i, \quad k = \{1, \dots, k'-1\},$$
(7)

5 of 10

and since the *j*th constraint is redundant at iteration k'

$$(\hat{x}(k'|x,\nu),\nu) \in C_j.$$

Then for any  $(x, v) \in \tilde{\mathcal{O}}_{k'}$ ,  $(\hat{x}(k'|x, v), v) \in C_j$ . Finally, since  $\tilde{\mathcal{O}}_{k'+1} \subseteq \tilde{\mathcal{O}}_{k'}$ ,  $(x, v) \in \mathcal{O}_{k'}$  implies  $(f(x, v), v) \in C_j$ , therefore, the *j*th constraint is redundant at iteration number k' + 1.

Using Proposition 1 we can refine Algorithm 1 into the more computationally efficient Algorithm 2 which only checks the redundancy of constraints that have not been redundant so far.

Algorithm 2. More efficient computation of  $\tilde{\mathcal{O}}_{\infty}$ 

```
\tilde{\mathcal{O}} \leftarrow \mathcal{O}^{\varepsilon}
k \leftarrow 0
\rho \leftarrow 1
\rho_i \leftarrow 1, i = 1, \ldots, n_c
while \rho \neq 0 \land k \leq \bar{k} do
       \rho \leftarrow 0
       for i = 1, ..., n_c do
               if \rho_i \neq 0 then
                      if c_i(\hat{x}(k|x, v), v) \ge 0 is redundant with respect to \tilde{\mathcal{O}} then
                              \rho_i \leftarrow 0
                      else
                              \rho \leftarrow 1
                              \tilde{\mathcal{O}} \leftarrow \tilde{\mathcal{O}} \cap \{(x, v) | c_i(\hat{x}(k|x, v), v) \ge 0\}
                       end if
               end if
       end for
       k \leftarrow k + 1
end while
\tilde{\mathcal{O}}_{\infty} \leftarrow \tilde{\mathcal{O}}
```

## **6** | ILLUSTRATIVE EXAMPLES

## 6.1 | Application: satellite rendezvous

In this example we apply the proposed methodology to compute  $\tilde{\mathcal{O}}_{\infty}$  for a deputy-chief satellite rendezvous. In this setting, the control objective is for the deputy satellite to approach the chief while staying in line of sight.<sup>32</sup> A noninertial Hill frame is attached to the chief spacecraft where the three axes are the radial direction toward earth, the along-track direction toward the chief spacecraft, and the cross-track direction along the chief's angular momentum vector, respectively.

The orbit can be represented by the following linearized model,

$$\dot{x}(t) = \begin{bmatrix} 0 & 0 & 0 & 1 & 0 & 0 \\ 0 & 0 & 0 & 0 & 1 & 0 \\ 0 & 0 & 0 & 0 & 0 & 1 \\ 3v^2 & 0 & 0 & 0 & 2v & 0 \\ 0 & 0 & 0 & -2v & 0 & 0 \\ 0 & 0 & -v^2 & 0 & 0 & 0 \end{bmatrix} x(t) + \begin{bmatrix} 0 & 0 & 0 \\ 0 & 0 & 0 \\ 0 & 0 & 0 \\ 0 & 1 & 0 \\ 0 & 0 & 1 \end{bmatrix} u(t),$$
(8)

where  $v = 0.0011 \text{ rads}^{-1}$ ,  $x = [x_1 x_2 x_3 \dot{x}_1 \dot{x}_2 \dot{x}_3]^T$  are the relative positions in km and velocities in km s<sup>-1</sup> along the three axes of the Hill frame, and  $u = [u_1 u_2 u_3]^T$  are the three components of the thrust in the aforementioned axes. System (8) is subject to the following constraints:

$$(\tan^{2}\gamma)(x_{2} + 0.01)^{2} - x_{1}^{2} - x_{3}^{2} \ge 0$$

$$u_{\max}^{2} - u_{1}^{2} - u_{2}^{2} - u_{3}^{2} \ge 0$$

$$x_{2} \ge 0,$$
(9)

where  $\gamma = 15^{\circ}$  is the Line of Sight cone half angle, and  $u_{\text{max}} = 0.001 \text{ kms}^{-2}$  is the maximum thrust acceleration. Discretizing (8) with  $T_s = 30$  s and controlling it with the control law u(t) = Fx(t) + Gv(t), with LQR gain *F* and feedforward gain *G* such that *v* becomes the reference position of the deputy spacecraft yields

$$x(t+1) = Ax + Bv, \tag{10}$$

where

$$A = \begin{bmatrix} 0.0111 & 0.0007 & 0 & 0.1676 & 0.0129 & 0 \\ 0.0611 & 0.1247 & 0 & 0.8913 & 1.9383 & 0 \\ 0 & 0 & 0.0138 & 0 & 0 & 0.2079 \\ -0.0659 & -0.0006 & 0 & -0.9879 & -0.0205 & 0 \\ 0.0048 & -0.0583 & 0 & 0.0822 & -0.8705 & 0 \\ 0 & 0 & -0.0657 & 0 & 0 & -0.9861 \end{bmatrix}, \quad B = \begin{bmatrix} 0.9889 & -0.0007 & 0 \\ -0.0611 & 0.8753 & 0 \\ 0 & 0 & 0.9862 \\ 0.0659 & 0.0006 & 0 \\ -0.0048 & 0.0583 & 0 \\ 0 & 0 & 0.0657 \end{bmatrix}.$$

A 3D slice of the  $\tilde{\mathcal{O}}_{\infty}$  computed applying Algorithm 2 is depicted in Figure 1. For this system,  $k^*$  was determined to be  $k^* = 19$  and  $\tilde{\mathcal{O}}_{\infty}$  is defined by 31 inequalities. This represents a very significant decrease compared to the 30 step prediction horizon (and the resulting 90 constraints) which were empirically estimated in Reference 32. The computation of  $\tilde{\mathcal{O}}_{\infty}$  took 2.07 s on an Intel Core i7-7500 at 2.7 GHz with 16 GB of RAM. All optimization problems were solved using MOSEK<sup>33</sup> interfaced in Julia 1.5.3.



**FIGURE 1** Three-dimensional slice of the nine-dimensional  $\tilde{\mathcal{O}}_{\infty}$  of system (10) corresponding to  $x_1 = 0.1, x_2 = 5, x_3 = -1.2, x_4 = x_5 = x_6 = 0$ 

## 6.2 | Electromagnetically actuated mass-spring damper

In this example we apply the proposed methodology to an electromagnetically actuated mass-spring damper system,<sup>25</sup> depicted in Figure 2. This system has been used to describe many electromagnetic actuators, for example, injectors, valves,<sup>34,35</sup> and is modeled by the following equations:

$$\dot{x}_1 = x_2, \dot{x}_2 = -\frac{k}{m}x_1 - \frac{c}{m}x_2 + \frac{\alpha}{m}\frac{u}{(d_0 - x_1)^{\gamma}},$$
(11)

where  $x_1$  and  $x_2$  are the position in m and velocity in m s<sup>-1</sup> of the armature, respectively, k = 38.94 N m<sup>-1</sup>, m = 1.54 kg, c = 0.65 N s m<sup>-1</sup>,  $\alpha = 4.5 \cdot 10^{-5}$  C<sup>2</sup> kg<sup>-1</sup> m<sup>-3</sup>,  $d_0 = 0.0102$  m, and  $\gamma = 2$ . This system can be feedback linearized using the control law

$$u = \frac{1}{\alpha} (d_0 - x_1)^{\gamma} (kv - c_d x_2), \tag{12}$$



FIGURE 2 Electromagnetically actuated mass-spring damper system



**FIGURE 3**  $\tilde{\mathcal{O}}_{\infty}$  for system (14) viewed from two different angles

with  $c_d = 4$ . The closed loop system becomes

$$\dot{x}_1 = x_2 \dot{x}_2 = -\frac{k}{m}x_1 - \frac{c+c_d}{m}x_2 + \frac{k}{m}v.$$
(13)

System (13) is subject to the following constraints:

 $x_1 \le 0.008$ ,  $kv - c_d x_2 \ge 0$ ,  $u \le 0.3$ ,

where *u* is given by (12). Finally, the system is discretized with a sampling time of  $T_s = 0.05$  s, yielding

$$x(t+1) = \begin{bmatrix} 0.9701 & 0.0459\\ -1.1610 & 0.8312 \end{bmatrix} x(t) + \begin{bmatrix} 0.0299\\ 1.1610 \end{bmatrix} v(t).$$
(14)

The resulting  $\tilde{\mathcal{O}}_{\infty}$  is depicted in Figure 3. For its computation we considered  $\varepsilon = 10^{-2}$ .  $\tilde{\mathcal{O}}_{\infty}$  was finitely determined after  $k^* = 35$  iterations and it is described by 87 inequalities. The elapsed time to compute  $\tilde{\mathcal{O}}_{\infty}$  was 0.58 s. Out of the total 105 inequalities resulting from this horizon, 18 were determined to be redundant. It must be remarked that the upper bound  $\overline{k}$  computed using the Lyapunov approach as in Reference 20 would give an unreasonably long horizon of  $\overline{k} = 786$  and that, in absence of a sound methodology, in previous publications reporting this example the horizon k was estimated empirically.

## 7 | CONCLUSION

In this note we presented a systematic procedure to compute MOASs for linear systems subject to polynomial constraints. To do so, we introduced a redundancy certificate for polynomial constraints based on SOSP. Further, we provided a modification of the traditional algorithm for the computation of the MOAS for increased computational efficiency. Finally, we demonstrated the effectiveness of the proposed methodology by computing the MOASs for two relevant examples of spacecraft rendezvous and an electromagnetic actuator control, indicating a significant potential for practical impact of the proposed methodology.

#### AUTHOR CONTRIBUTIONS

Every author has contributed substantially to the development of both the research presented in this manuscript and the manuscript itself.

## FUNDING INFORMATION

This research has been funded by the Fédération Wallonie-Bruxelles Actions de Recherche Concertées (ARC) under project "Guaranteed by Optimization" (GbO). Ilya V. Kolmanovsky acknowledges the support of the National Science Foundation grant 1931738. Daniel R. Ramirez and Daniel Limon acknowledge the funding of Agencia Estatal de Investigación (AEI) under Grant PID2019-106212RB-C41/AEI/10.13039/501100011033.

## **CONFLICT OF INTEREST**

The authors declare no potential conflict of interests.

## DATA AVAILABILITY STATEMENT

The data that support the findings of this study are available from the corresponding author upon reasonable request.

## ORCID

Andres Cotorruelo De https://orcid.org/0000-0003-1464-6702 Ilya V. Kolmanovsky De https://orcid.org/0000-0002-7225-4160 Daniel R. Ramirez De https://orcid.org/0000-0002-8962-5984 Daniel Limon De https://orcid.org/0000-0001-9334-7289

## REFERENCES

- 1. Henrion D, Korda M. Convex computation of the region of attraction of polynomial control systems. IEEE Trans Automat Contr. 2013;59(2):297-312.
- 2. Pandita R, Chakraborty A, Seiler P, Balas G. Reachability and region of attraction analysis applied to GTM dynamic flight envelope assessment. Proceedings of the AIAA Guidance, Navigation, and Control Conference; 2009:6258.
- 3. Zhang L, Wang C, Chen L. Stability and stabilization of a class of multimode linear discrete-time systems with polytopic uncertainties. IEEE Trans Ind Electron. 2009;56(9):3684-3692.
- 4. Cotorruelo A, Limon D, Garone E. Output admissible sets and reference governors: saturations are not constraints! IEEE Trans Automat Contr. 2019;65(3):1192-1196.
- 5. Yu S, Maier C, Chen H, Allgöwer F. Tube MPC scheme based on robust control invariant set with application to Lipschitz nonlinear systems. Syst Control Lett. 2013;62(2):194-200.
- 6. Blanchini F, Miani S. Set-Theoretic Methods in Control. Springer; 2008.
- 7. Gilbert EG, Tan KT. Linear systems with state and control constraints: the theory and application of maximal output admissible sets. IEEE Trans Automat Contr. 1991;36(9):1008-1020.
- 8. Gilbert EG, Kolmanovsky I, Tan KT. Nonlinear control of discrete-time linear systems with state and control constraints: a reference governor with global convergence properties. Proceedings of 1994 33rd IEEE Conference on Decision and Control; 1994:144-149; IEEE.
- 9. Garone E, Di Cairano S, Kolmanovsky I. Reference and command governors for systems with constraints: a survey on theory and applications. Automatica. 2017;75:306-328.
- 10. Kolmanovsky I, Gilbert EG. Multimode regulators for systems with state & control constraints and disturbance inputs. Control Using Logic-Based Switching. Springer; 1997:104-117.
- 11. Mayne D. Nonlinear model predictive control: challenges and opportunities. Nonlinear Model Predict Control. 2000;26:23-44.
- 12. Tian Y, Kolmanovsky I. Reduced order and prioritized reference governors for limit protection in aircraft gas turbine engines. Proceedings of the AIAA Guidance, Navigation, and Control Conference; 2014:1149.
- 13. Simon D, Härkegård O, Löfberg J. Angle of attack and load factor limiting in fighter aircraft using command governors. Proceedings of the AIAA Guidance, Navigation, and Control Conference; 2017:1257.
- 14. Zinnecker A, Serrani A, Bolender M, Doman D. Combined reference governor and anti-windup design for constrained hypersonic vehicles models. Proceedings of the AIAA Guidance, Navigation, and Control Conference; 2009:6283.
- 15. Yucelen T, Johnson E. Command governor-based adaptive control. Proceedings of the AIAA Guidance, Navigation, and Control Conference; 2012:4618.
- 16. Magree D, Yucelen T, Johnson E. Command governor-based adaptive control of an autonomous helicopter. Proceedings of the AIAA Guidance, Navigation, and Control Conference; 2012:4830.
- 17. Wilcher K, Yucelen T. On performance improvement of gain-scheduled model reference adaptive control Laws. AIAA Scitech 2020 Forum; 2020:1118.
- 18. Dillsaver M, Cesnik C, Kolmanovsky I. Gust load alleviation control for very flexible aircraft. Proceedings of the AIAA Atmospheric Flight Mechanics Conference; 2011:6368.
- 19. Gruber J, Ramirez D, Alamo T, Camacho E. Min–max MPC based on an upper bound of the worst case cost with guaranteed stability. Application to a pilot plant. J Process Control. 2011;21(1):194-204.
- 20. Bemporad A. Reference governor for constrained nonlinear systems. IEEE Trans Automat Contr. 1998;43(3):415-419.

WILEY 9 of 10

## 10 of 10 WILEY

- 21. Caron R, McDonald J, Ponic C. A degenerate extreme point strategy for the classification of linear constraints as redundant or necessary. J Optim Theory Appl. 1989;62(2):225-237.
- 22. Paulraj S, Chellappan C, Natesan T. A heuristic approach for identification of redundant constraints in linear programming models. Int J Comput Math. 2006;83(8–9):675-683.
- 23. Telgen J. Identifying redundant constraints and implicit equalities in systems of linear constraints. Manag Sci. 1983;29(10):1209-1222.
- 24. Paulraj S, Sumathi P. A comparative study of redundant constraints identification methods in linear programming problems. Math Probl Eng. 2010;2010:1748-1763.
- 25. Gilbert E, Kolmanovsky I. Nonlinear tracking control in the presence of state and control constraints: a generalized reference governor. Automatica. 2002;38(12):2063-2073.
- 26. Putinar M. Positive polynomials on compact semi-algebraic sets. Ind Univ Math J. 1993;42(3):969-984.
- 27. Bochnak J, Coste M, Roy MF. Real Algebraic Geometry. Vol 36. Springer Science & Business Media; 2013.
- 28. Gilbert EG, Kolmanovsky I. Fast reference governors for systems with state and control constraints and disturbance inputs. Int J Robust Nonlinear Control IFAC-Affiliat J. 1999;9(15):1117-1141.
- 29. Powers V, Wörmann T. An algorithm for sums of squares of real polynomials. J Pure Appl Algebra. 1998;127(1):99-104.
- 30. Tütüncü RH, Toh KC, Todd MJ. Solving semidefinite-quadratic-linear programs using SDPT3. Math Program. 2003;95(2):189-217.
- 31. Löfberg J. YALMIP: a toolbox for modeling and optimization in MATLAB. Proceedings of the CACSD Conference; 2004; Taipei, Taiwan.
- Kalabić U, Kolmanovsky I, Gilbert E. Reference governors for linear systems with nonlinear constraints. Proceedings of the 2011 50th IEEE Conference on Decision and Control and European Control Conference; 2011:2680-2686; IEEE.
- 33. ApS M. MOSEK optimizer API for C 8.1.0.83; 2021.
- 34. Hong J, Cummings I, Washabaugh P, Bernstein D. Stabilization of a mass and spring system with electromagnetic actuation. Proceedings of the 1997 IEEE International Conference on Control Applications; 1997:189-194; IEEE.
- 35. Miller RH, Kolmanovsky I, Gilbert EG, Washabaugh PD. Control of constrained nonlinear systems: a case study. IEEE Control Syst Mag. 2000;20(1):23-32.

**How to cite this article:** Cotorruelo A, Garone E, Kolmanovsky IV, Ramirez DR, Limon D. Computation of maximal output admissible sets for linear systems with polynomial constraints. *Adv Control Appl.* 2023;5(1):e119. doi: 10.1002/adc2.119