

RCS MEASUREMENT OF POLARIMETRIC ACTIVE RADAR CALIBRATORS

by

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Abstract

Practical aspects of polarimetric measurement of active radar calibrators (PARC) are discussed in this report. A new polarimetric calibration technique (STCT) is employed for polarimetric radar cross section measurement of L- and C-band PARCs. The amplitude and phase of the scattering matrix elements are given over a wide range of incidence angle for azimuth, elevation, 45° , and 135° planes. Tables for scattering matrix elements and the polarization signatures at boresight are also provided.

Contents

1	Introduction	1
2	System Configuration and Measurement Setup	2
3	Calibration Procedure	6
4	Experimental Results	7
5	Conclusions	9
A	Appendix A	A-1
B	Appendix B	B-1

List of Figures

1	Block diagram of the scatterometer system.	3
2	Block diagram of the pulsing network.	3
3	Block diagram of the L-band microwave circuitry.	5
4	Block diagram of the C-band microwave circuitry.	5
5	Automatic radar cross section measurement setup.	14
6	Time domain response of an L-band PARC.	15
7	Time domain response of a C-band PARC.	15
8	The front panel of L- and C-band PARC's as seen by a radar. . . .	16
9	The orientation of L- and C-band PARC's for pattern measurement.	16
10	The azimuth and elevation polarization signature for L1 PARC. . .	17
11	The 45° and 135° polarization signature for L1 PARC.	18
12	The azimuth and elevation polarization signature for C1 PARC. . .	19
13	The 45° and 135° polarization signature for C1 PARC.	20
14	Azimuth pattern of amplitude of scattering matrix elements for L1 PARC.	21
15	Elevation pattern of amplitude of scattering matrix elements for L1 PARC.	22
16	45° pattern of amplitude of scattering matrix elements for L1 PARC.	23
17	45° pattern of phase of scattering matrix elements for L1 PARC. . .	24
18	135° pattern of amplitude of scattering matrix elements for L1 PARC.	25
19	135° pattern of phase of scattering matrix elements for L1 PARC. .	26

20	Azimuth pattern of amplitude of scattering matrix elements for L2 PARC.	27
21	Elevation pattern of amplitude of scattering matrix elements for L2 PARC.	28
22	45° pattern of amplitude of scattering matrix elements for L2 PARC.	29
23	45° pattern of phase of scattering matrix elements for L2 PARC. . .	30
24	135° pattern of amplitude of scattering matrix elements for L2 PARC.	31
25	135° pattern of phase of scattering matrix elements for L2 PARC. .	32
26	Azimuth pattern of amplitude of scattering matrix elements for L3 PARC.	33
27	Elevation pattern of amplitude of scattering matrix elements for L3 PARC.	34
28	45° pattern of amplitude of scattering matrix elements for L3 PARC.	35
29	45° pattern of phase of scattering matrix elements for L3 PARC. . .	36
30	135° pattern of amplitude of scattering matrix elements for L3 PARC.	37
31	135° pattern of phase of scattering matrix elements for L3 PARC. .	38
32	Azimuth pattern of amplitude of scattering matrix elements for C1 PARC.	39
33	Azimuth pattern of phase of scattering matrix elements for C1 PARC.	40
34	Elevation pattern of amplitude of scattering matrix elements for C1 PARC.	41
35	Elevation pattern of phase of scattering matrix elements for C1 PARC.	42

36	45° pattern of amplitude of scattering matrix elements for C1 PARC.	43
37	135° pattern of amplitude of scattering matrix elements for C1 PARC.	44
38	Azimuth pattern of amplitude of scattering matrix elements for C2 PARC.	45
39	Azimuth pattern of phase of scattering matrix elements for C2 PARC.	46
40	Elevation pattern of amplitude of scattering matrix elements for C2 PARC.	47
41	Elevation pattern of phase of scattering matrix elements for C2 PARC.	48
42	45° pattern of amplitude of scattering matrix elements for C2 PARC.	49
43	135° pattern of amplitude of scattering matrix elements for C2 PARC.	50
44	Azimuth pattern of amplitude of scattering matrix elements for C3 PARC.	51
45	Azimuth pattern of phase of scattering matrix elements for C3 PARC.	52
46	Elevation pattern of amplitude of scattering matrix elements for C3 PARC.	53
47	Elevation pattern of phase of scattering matrix elements for C3 PARC.	54
48	45° pattern of amplitude of scattering matrix elements for C3 PARC.	55
49	135° pattern of amplitude of scattering matrix elements for C3 PARC.	56

List of Tables

- 1 Elements of scattering matrix for L-band PARCs. The entries are in the form $A_{ij}\angle\phi_{ij}$, where A_{ij} is the RCS σ_{ij} in dBsm ($\sigma_{ij} = 4\pi|s_{ij}|^2$) and ϕ_{ij} is the phase of s_{ij} in degrees. 12
- 2 Elements of scattering matrix for L-band PARCs. The entries are in the form $A_{ij}\angle\phi_{ij}$, where A_{ij} is the RCS σ_{ij} in dBsm ($\sigma_{ij} = 4\pi|s_{ij}|^2$) and ϕ_{ij} is the phase of s_{ij} in degrees. 13

1 Introduction

Airborne and space-borne polarimetric imaging SAR has been shown to be an important remote sensing tool for the acquisition of quantitative information about the earth's environment on a global scale. In order to utilize the quantitative data scientifically, external calibration of the radar system is necessary. The external calibration of radar systems is performed by using a target with a specified radar cross section (RCS) known as a calibration target.

Calibration targets, in general, can be categorized into two major groups: 1) passive calibrators and 2) active calibrators [1]. Although passive calibrators are more stable and reliable than the active ones, they are less desirable because of their large physical dimensions. In recent years, polarimetric active radar calibrators (PARC) have been used extensively and are planned to be employed for external calibration of SAR systems [2] in support of future spaceborne missions.

The success of external calibration relies on the knowledge of the scattering matrix of the calibration target(s). Although it may be possible to estimate the elements of the scattering matrix for a calibration target to a reasonable extent, the manufacturing tolerances always leave a great deal of uncertainty in the estimation. Therefore it is imperative to measure the calibrators against a precise calibration target (such as a metallic sphere) under laboratory condition. This also reveals another drawback for passive calibrators with large physical dimensions, namely that the far-field condition and uniform illumination criteria cannot be met in the laboratory.

Until recently, it has been very difficult to measure the scattering matrices of targets over a wide range of incidence angle and frequency with the desired accuracy. Advances in technology and calibration methods have made it possible to measure the elements of the scattering matrix with an accuracy of 0.5 dB in magnitude and 5° in phase [3].

The purpose of this report is to document measurements of JPL's L- and C-band PARCs using the new calibration technique (STCT) given in Appendix A. The measurements were performed in the UM anechoic chamber using a fully automated polarimetric scatterometer[4].

2 System Configuration and Measurement Setup

The polarimetric measurements of the PARCs were performed by L- and C-band scatterometers. A simplified block diagram of the scatterometer system is shown in Fig. (1). The scatterometer is an HP 8753A-based system with both phase and amplitude measurement capability and 100 dB dynamic range. The ability of the network analyzer to generate the time domain response of the frequency measurement allows us to separate the unwanted short-range signals from the desired target response (known as software gating). The sequence of polarization selection, data collection, and target orientation is performed via an HP 9000 series computer. The relay actuator energizes the frequency and polarization switches. The amplifier and pulsing network eliminates the short-range returns from the antenna and circulators to increase the dynamic range for RCS measurements [5].

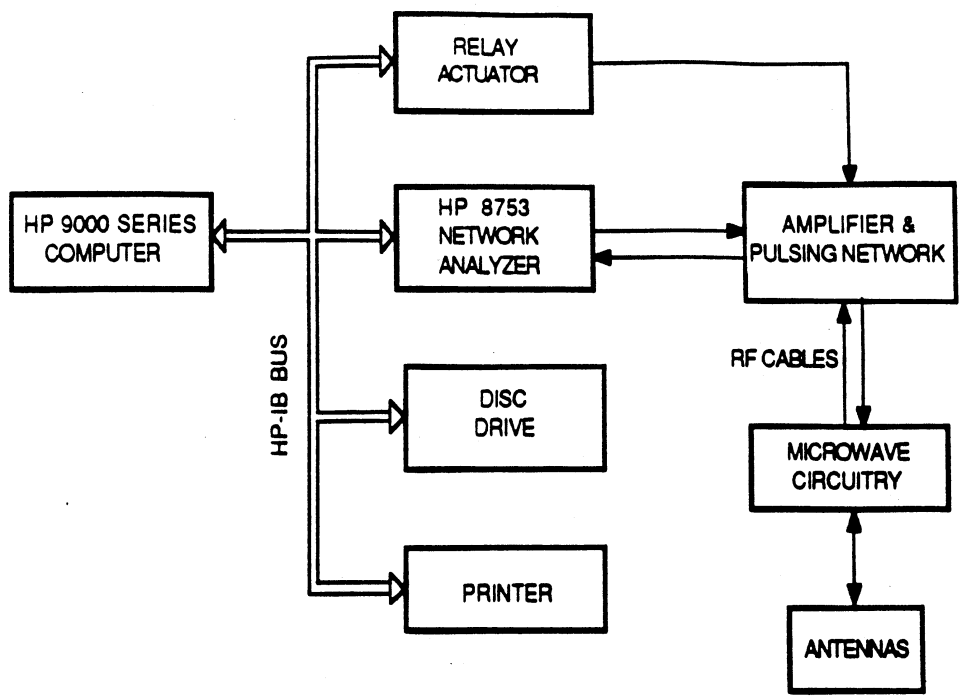


Figure 1: Block diagram of the scatterometer system.

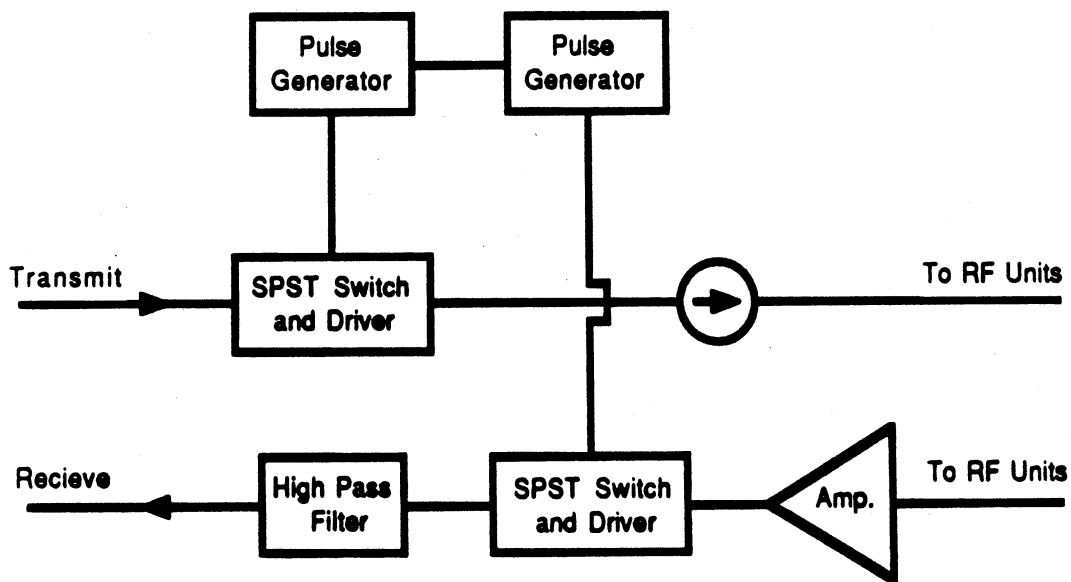


Figure 2: Block diagram of the pulsing network.

In this scheme, the receiver is switched off during transmission and then reconnected when the target return is expected to arrive at the receiver. Since the switching is done at a much higher rate than the receiver's bandwidth the network analyzer does not sense that the incoming signal is pulsed and it is measured as if it were a CW signal. The block diagram of the amplifier and pulsing network is shown in Fig. (2).

The synthesized source of the network analyzer spans the frequency range 300 KHz to 3 GHz and therefore for C-band up- and down-convertors are used. The block diagrams of the L- and C-band microwave circuitry are given in Figs. (3) and (4), respectively. The up-convertor of the C-band unit is a very stable microwave source operating at 6.5 GHz. The frequency range of the network analyzer for C-band measurement must be set to 1-1.5 GHz in order to operate the C-band scatterometer at 5-5.5 GHz. The operating frequency of the L-band system is 1.1-1.4 GHz. A new orthomode transducer (OMT) and dual-polarized antenna were designed for this project at L-band. The overall cross-polarization isolation of the new OMT and antenna is better than 35 dB.

Because PARC has a high gain amplifier, reflection from nearby objects might increase the feedback which would cause oscillation. To avoid this problem and also to have a very good signal to background ratio, the PARCs were mounted on a styrofoam pedestal in an anechoic chamber. The correct position of the PARCs with respect to the antenna coordinate system was accomplished by an azimuth-over-elevation positioner as depicted in Fig. (5). The azimuth turntable

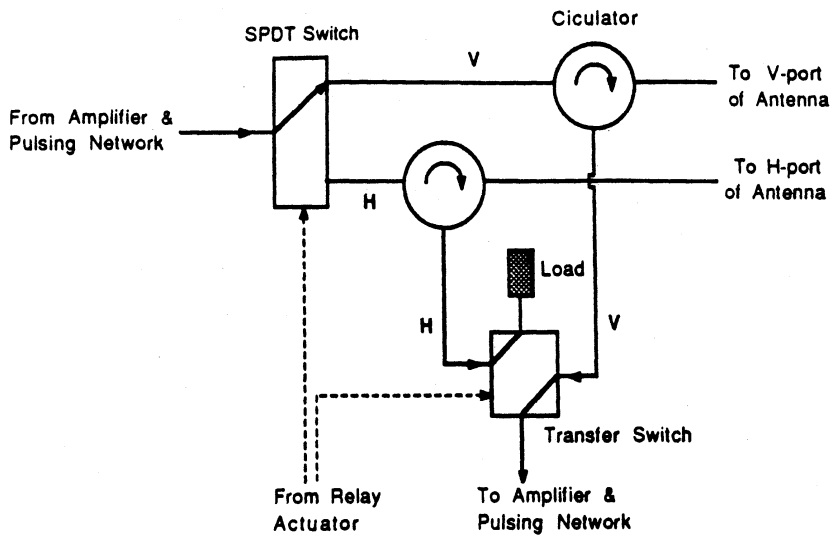


Figure 3: Block diagram of the L-band microwave circuitry.

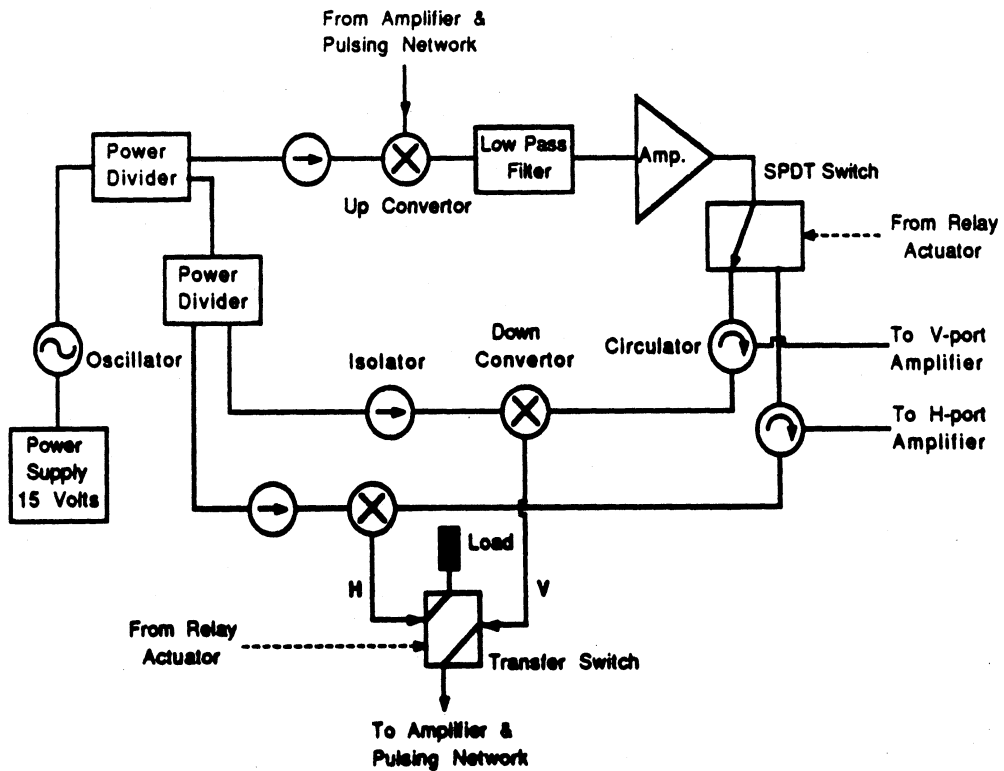


Figure 4: Block diagram of the C-band microwave circuitry.

is a computer controlled stepper motor with an accuracy of a fraction of a tenth of a degree and the elevation controller is a precise analog positioner.

3 Calibration Procedure

To calibrate the PARCs, we used the single-target calibration technique (STCT) described in Appendix A. The error in measurement of scattering matrix using this technique is less than 0.5 dB in amplitude and 5° in phase. With the STCT the antenna cross-talk contamination and channel imbalances are obtained by measuring only a single calibration target, namely a metallic sphere. This technique is immune to errors caused by target alignment with respect to the antenna coordinate system.

Using a four-port network approach it is shown that the measured scattering matrix of a target with scattering matrix \mathbf{s} is given by

$$\mathbf{M} = \begin{bmatrix} R_1 & 0 \\ 0 & R_2 \end{bmatrix} \begin{bmatrix} 1 & C \\ C & 1 \end{bmatrix} \mathbf{s} \begin{bmatrix} 1 & C \\ C & 1 \end{bmatrix} \begin{bmatrix} T_1 & 0 \\ 0 & T_2 \end{bmatrix} + \mathbf{N}$$

where the \mathbf{R} and \mathbf{T} matrices are the receive and transmit channel imbalances, C is the antenna cross-talk contamination factor, and \mathbf{N} is a matrix representing thermal noise and background reflections. The background contribution can be obtained by measuring the empty chamber and the effect of the thermal noise can be minimized by an integration process.

By denoting the measured scattering matrix elements of the sphere and the PARC, respectively, by m_{ij}^o and m_{ij}^u , the unknown scattering matrix elements of

the PARC can be obtained from

$$\begin{aligned}
s_{vv} &= \frac{1}{(1-C^2)^2} \left[-2C^2 \left(\frac{m_{12}^u}{m_{12}^o} + \frac{m_{21}^u}{m_{21}^o} \right) + (1+C^2) \left(\frac{m_{11}^u}{m_{11}^o} + C^2 \frac{m_{22}^u}{m_{22}^o} \right) \right] s^o \\
s_{hh} &= \frac{1}{(1-C^2)^2} \left[-2C^2 \left(\frac{m_{12}^u}{m_{12}^o} + \frac{m_{21}^u}{m_{21}^o} \right) + (1+C^2) \left(\frac{m_{22}^u}{m_{22}^o} + C^2 \frac{m_{11}^u}{m_{11}^o} \right) \right] s^o \\
s_{vh} &= \frac{C}{(1-C^2)^2} \left[2 \frac{m_{12}^u}{m_{12}^o} + 2C^2 \frac{m_{21}^u}{m_{21}^o} - (1+C^2) \left(\frac{m_{11}^u}{m_{11}^o} + \frac{m_{22}^u}{m_{22}^o} \right) \right] s^o \\
s_{hv} &= \frac{C}{(1-C^2)^2} \left[2 \frac{m_{21}^u}{m_{21}^o} + 2C^2 \frac{m_{12}^u}{m_{12}^o} - (1+C^2) \left(\frac{m_{11}^u}{m_{11}^o} + \frac{m_{22}^u}{m_{22}^o} \right) \right] s^o
\end{aligned}$$

where s^o is the theoretical value for the diagonal elements of the sphere's scattering matrix. The cross-talk contamination factor is given by

$$C = \pm \frac{1}{\sqrt{a}} \left(1 - \sqrt{1-a} \right)$$

where $a \triangleq \frac{m_{12}^o m_{21}^o}{m_{11}^o m_{22}^o}$ and the branch of the square root is chosen such that $\text{Re} \left[\sqrt{1-a} \right] >$

0. The uncertainty on the sign of C can be removed when measuring PARCs because the general trend of phase behavior is known.

4 Experimental Results

The measurements were performed in a 14-meter long anechoic chamber. The PARC under test was mounted on a styrofoam pedestal on an azimuth-over-elevation positioner. To assure that the PARC is not oscillating in the chamber, the energy flow between the receive and transmit antenna was monitored by inserting a 20 dB directional coupler between the amplifier and the transmit antenna. Also to avoid saturating the PARC's amplifier and scatterometer's receiver, the scatterometer transmit power was adjusted such that the PARC output power was around 5 dBm. For the actual RCS measurements the directional coupler was removed.

A 12-inch metallic sphere was used as the calibration target, and the signal to noise ratio was better than 30 dB in all cases. The L- and C-band measurements were performed over the frequency range 1.1-1.4 GHz and 5-5.5 GHz, respectively. All the data presented in this report are measurements at the center frequencies, namely 1.25 GHz for L-band and 5.3 GHz for the C-band. The backscattered radar cross section patterns were measured over the range of incidence angle from -40 to +40 degrees relative to the boresight direction in the azimuth, elevation, 45°, and 135° planes.

The radar cross section of a PARC can be decomposed into two components. The first component is the contribution of the front panel and the antennas and the second component is the contribution of the delay line and amplifier. Using the range-gating capability of the scatterometers, the two responses can be separated. Figs. (6) and (7) represent the time-domain responses of an L-band and a C-band PARC, respectively, where the relative amplitudes of the two components can be compared. In all of the results shown in this report, the contribution of the front panel and the antennas has been gated out.

In this study three L-band and three C-band PARCs were measured. The L-band PARCs are designated according to their serial number and the C-band PARCs designations are according to Fig. (8). In order to measure the RCS patterns in the desired planes, the PARCs were oriented as shown in Fig. (9). Tables 1 and 2 give the measured elements of the scattering matrices for L- and C-band at boresight for quick reference.

The polarization signature of PARCs L1 and C1 are plotted in Figs. (10) - (13). It is shown that co-polarized signatures for the case where the antennas are oriented 45° with respect to vertical direction (45° and 135° for L-band and azimuth and elevation for C-band) are slightly different from the theoretical response due to small errors in the phase measurement. However, for the case where the antennas are parallel and perpendicular to the vertical direction (azimuth and elevation for L-band and 45° and 135° for C-band) the errors in phase measurement do not affect the polarization signatures.

In Figs. (14) - (49) the measured RCS patterns of L- and C-band PARCs are shown. The C-band patterns seem reasonable but the L-band patterns are not quite symmetric, the maximum RCS appears at about 10° from boresight, and the measured phases are very much different from the expected values. These problems are due to the fact that the transmit and receive antennas are very close to each other (touching). The antennas are in the near field of each other and therefore the amplitude and phase patterns are different from the patterns of isolated antennas.

5 Conclusions

The radar cross section patterns of three L-band and three C-band PARCs were measured polarimetrically. The measurements were performed in an anechoic chamber and accurate orientation of the PARCs was achieved by a very precise azimuth-over-elevation positioner.

The measurements show that the RCS is not very sensitive to changes in the

azimuth and elevation angles, but it is very sensitive to the rotation angle about the boresight direction. It is found that when the PARCs transmit and receive antennas are parallel and perpendicular to the vertical direction, small errors in the orientation angles or errors in phase measurement have a minor effect on calibration accuracy. It is also found that the close proximity of the L-band antennas has caused significant problems for the patterns and considerable errors in the scattering matrix elements. We recommend that the antennas be separated by at least one wavelength.

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		vv	vh	hv	hh
L_1	Az	1.9 \angle 0.0	-31.1 \angle 91.3	36.72 \angle 121.4	1.4 \angle -120.0
	El	-6.1 \angle 0.0	35.0 \angle 14.2	-53.2 \angle -27.6	-2.5 \angle -84.6
	45°	31.2 \angle 0.0	30.6 \angle -13.4	30.6 \angle 197.0	29.9 \angle 184.5
	135°	31.3 \angle 0.0	30.9 \angle 177.8	30.9 \angle 19.4	30.8 \angle 185.2
L_2	Az	-3.4 \angle 0.0	-24.6 \angle -9.1	35.8 \angle 153.0	-3.3 \angle -34.5
	El	-3.0 \angle 0.0	36.4 \angle 184.9	-47.6 \angle -81.2	-2.7 \angle 20.0
	45°	29.9 \angle 0.0	29.5 \angle 9.0	29.5 \angle 175.7	29.2 \angle 184.9
	135°	30.8 \angle 0.0	30.6 \angle 187.8	30.6 \angle -6.5	30.2 \angle 182.4
L_3	Az	-2.7 \angle 0.0	-57.6 \angle -74.1	36.6 \angle 173.4	-2.6 \angle 186.6
	El	-17.4 \angle 0.0	38.8 \angle -56.6	-44.2 \angle 202.4	-1.7 \angle 101.9
	45°	32.2 \angle 0.0	31.6 \angle 8.0	31.8 \angle 179.1	31.9 \angle 182.7
	135°	32.6 \angle 0.0	32.2 \angle 186.7	32.2 \angle 0.3	32.3 \angle 183.3

Table 1: Elements of scattering matrix for L-band PARCs. The entries are in the form $A_{ij}\angle\phi_{ij}$, where A_{ij} is the RCS σ_{ij} in dBsm ($\sigma_{ij} = 4\pi|s_{ij}|^2$) and ϕ_{ij} is the phase of s_{ij} in degrees.

		vv	vh	hv	hh
C_1	Az	27.2∠0.0	27.1∠187.2	27.1∠-1.7	27.1∠185.5
	El	27.2∠0.0	27.1∠-3.1	27.0∠183.7	26.8∠180.8
	45°	-1.9∠0.0	-24.2∠119.0	32.8∠-59.0	-7.1 ∠182.2
	135°	-1.2∠0.0	33.0∠-52.3	-46.0∠122.9	-3.1∠-163.6
C_2	Az	26.7∠0.0	26.7∠-2.7	26.7∠181.6	26.6∠179.3
	El	26.6∠0.0	26.7∠180.1	26.7∠-0.2	26.8∠180.3
	45°	-19.6∠0.0	32.5∠-148.8	-46.6∠52.2	-10.9 ∠145.8
	135°	-40.2∠0.0	-55.2∠6.3	32.4 ∠-55.7	-15.6∠186.5
C_3	Az	26.5∠0.0	26.4∠1.0	26.4∠178.6	26.2∠179.6
	El	26.3∠0.0	26.1∠180.0	26.1∠-1.3	25.9∠178.9
	45°	-5.2∠0.0	32.3∠145.3	-51.0∠-18.6	-8.3∠179.3
	135°	-8.5∠0.0	-44.3∠-141.4	31.5∠46.8	-3.8∠184.4

Table 2: Elements of scattering matrix for L-band PARCs. The entries are in the form $A_{ij} \angle \phi_{ij}$, where A_{ij} is the RCS σ_{ij} in dBsm ($\sigma_{ij} = 4\pi|s_{ij}|^2$) and ϕ_{ij} is the phase of s_{ij} in degrees.

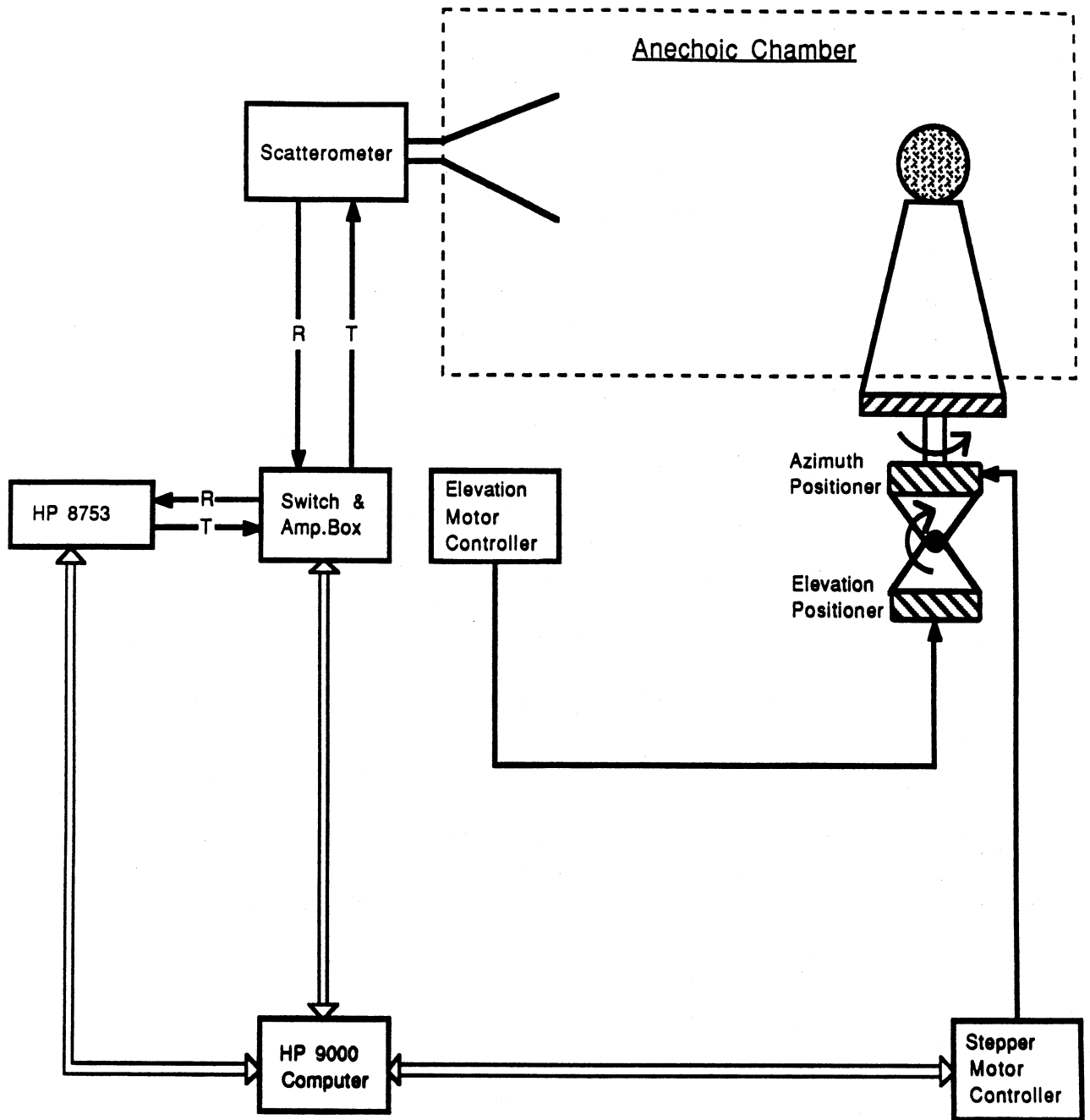


Figure 5: Automatic radar cross section measurement setup.

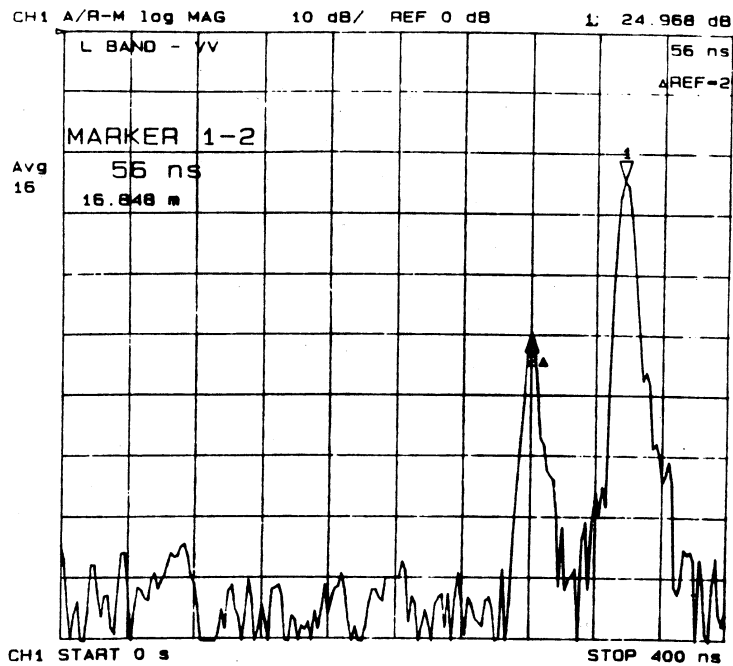


Figure 6: Time domain response of an L-band PARC.

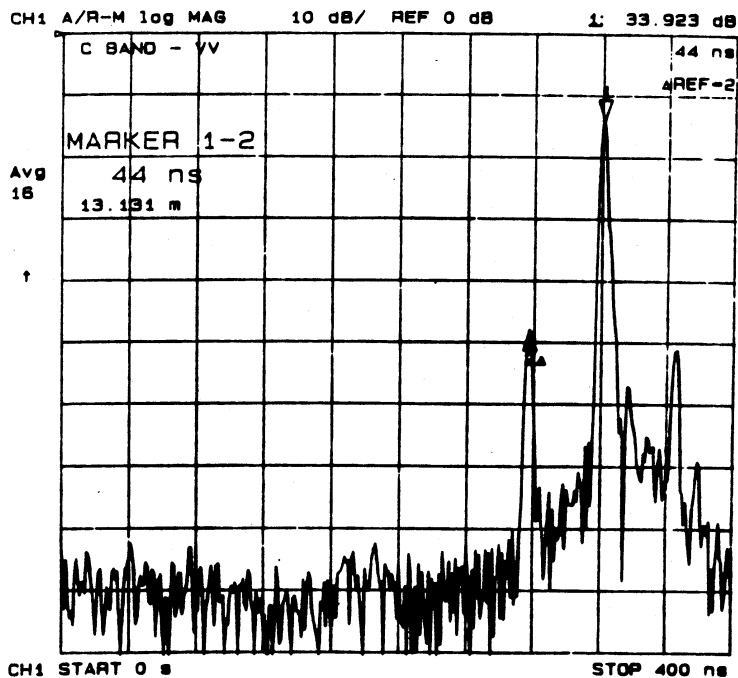


Figure 7: Time domain response of a C-band PARC.

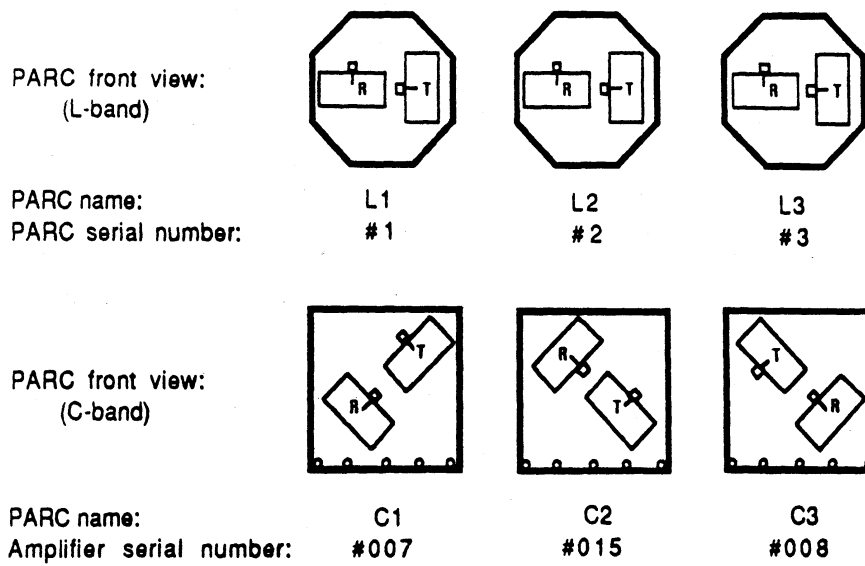


Figure 8: The front panel of L- and C-band PARC's as seen by a radar.

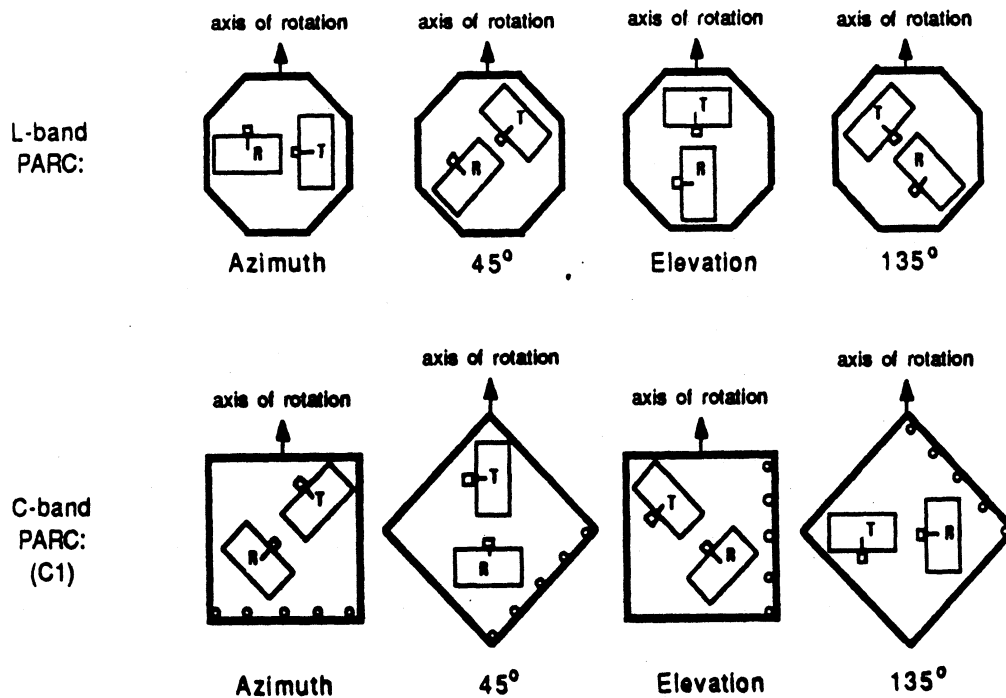
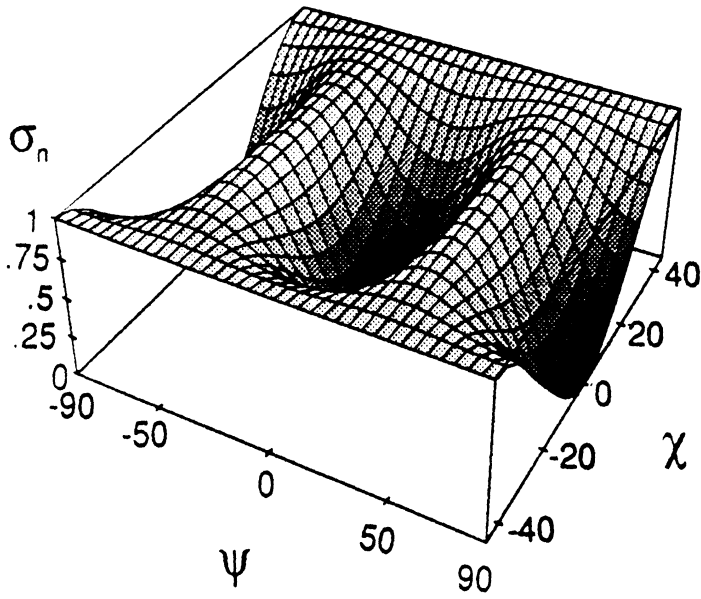
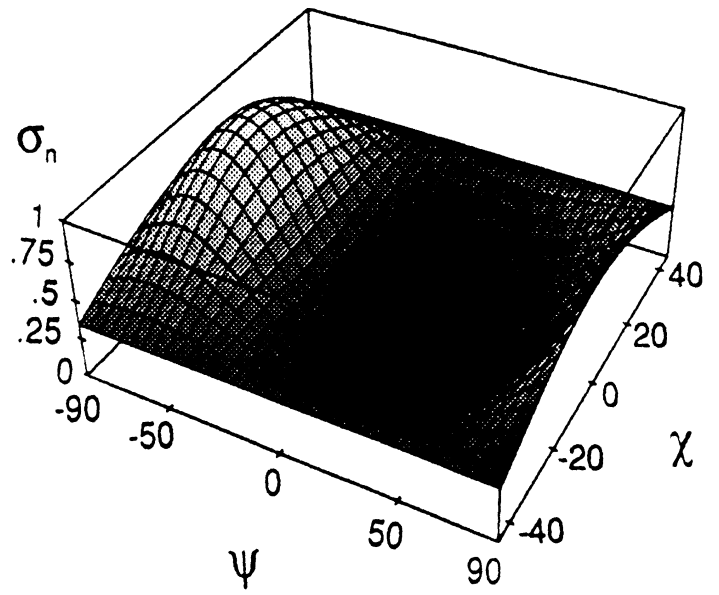


Figure 9: The orientation of L- and C-band PARC's for pattern measurement.

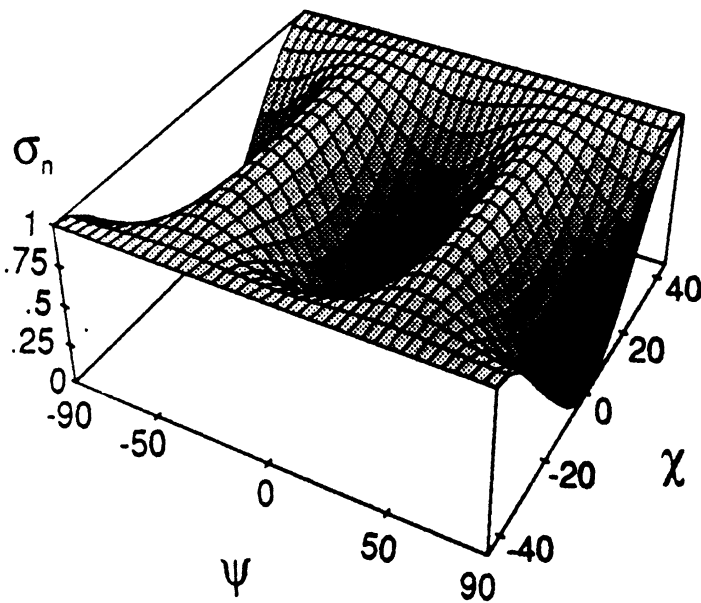
Co-pol. L1 Azimuth



Cross-pol. L1 Azimuth



Co-pol. L1 Elevation



Cross-pol. L1 Elevation

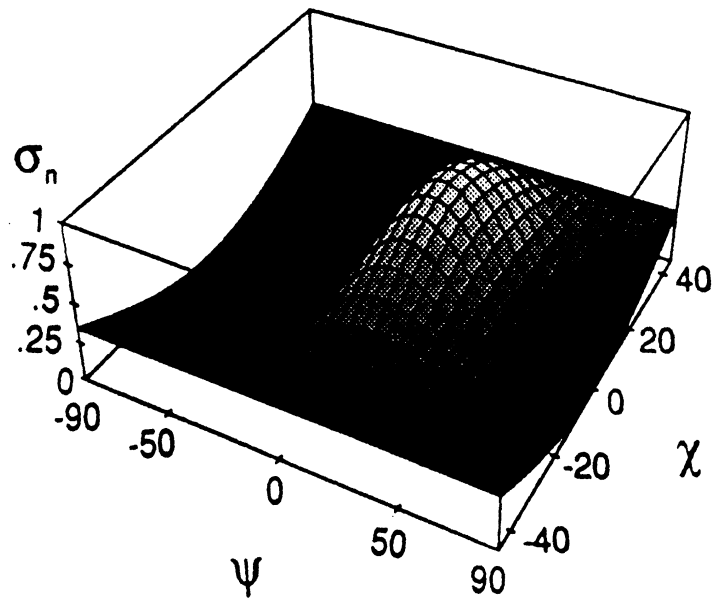
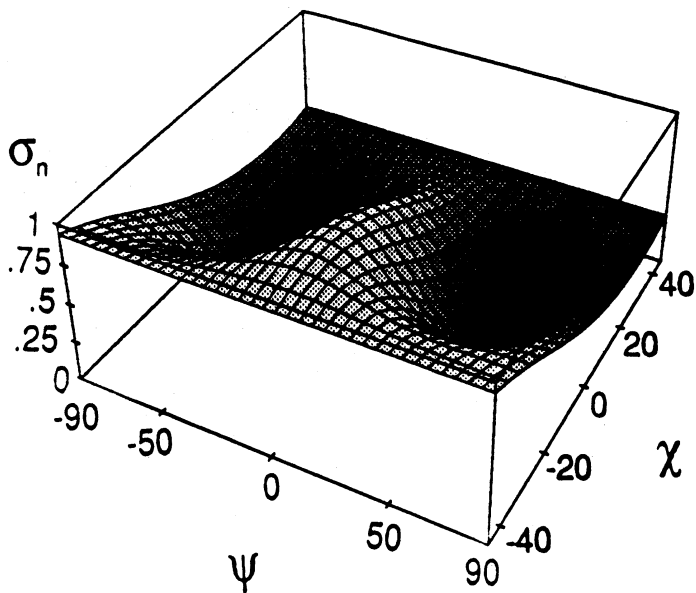
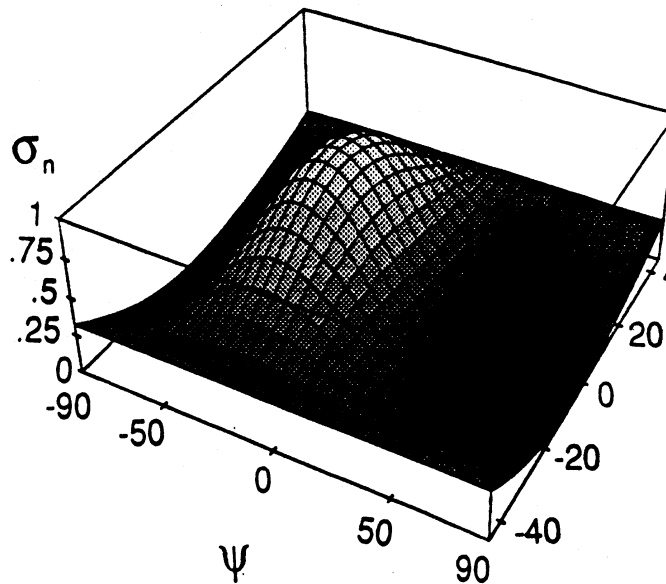


Figure 10: The azimuth and elevation polarization signature for L1 PARC.

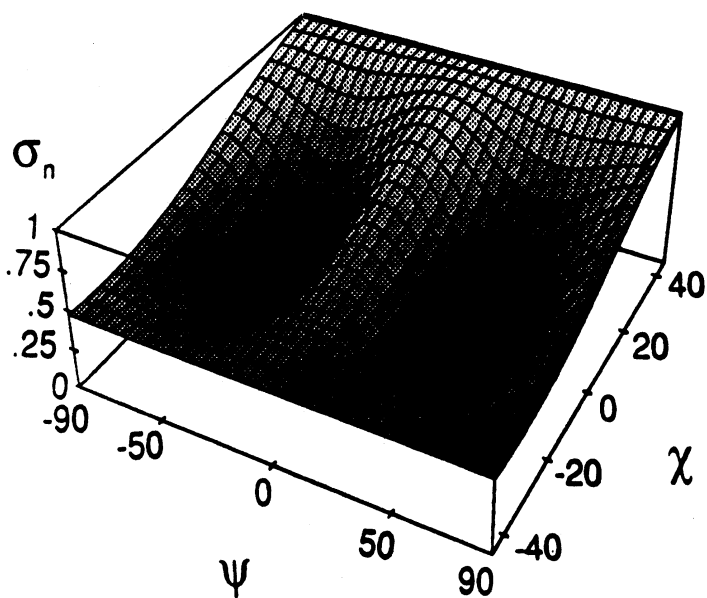
Co-pol. L1 45°



Cross-pol. L1 45°



Co-pol. L1 135°



Cross-pol. L1 135°

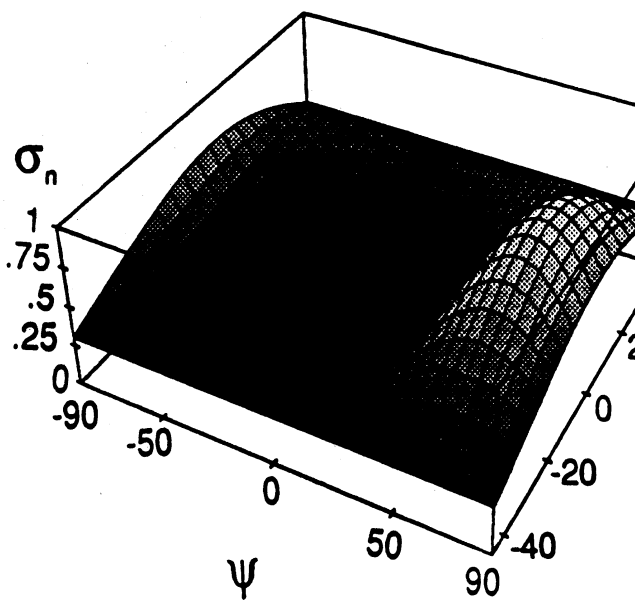
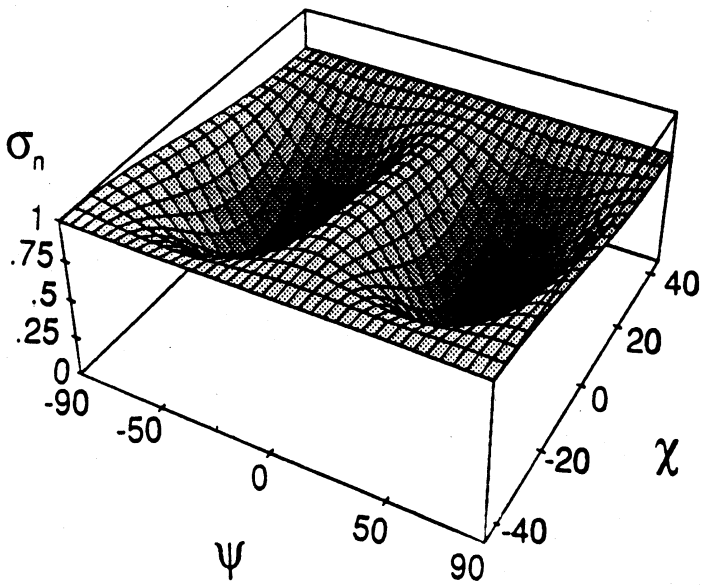
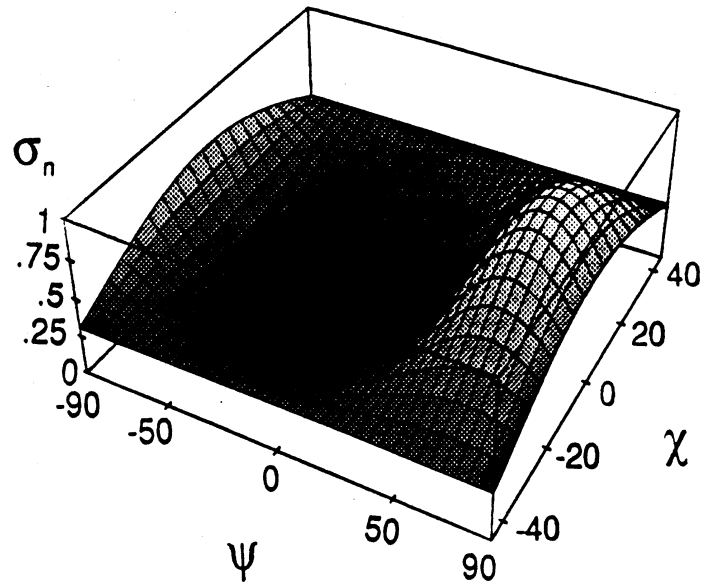


Figure 11: The 45° and 135° polarization signature for L1 PARC.

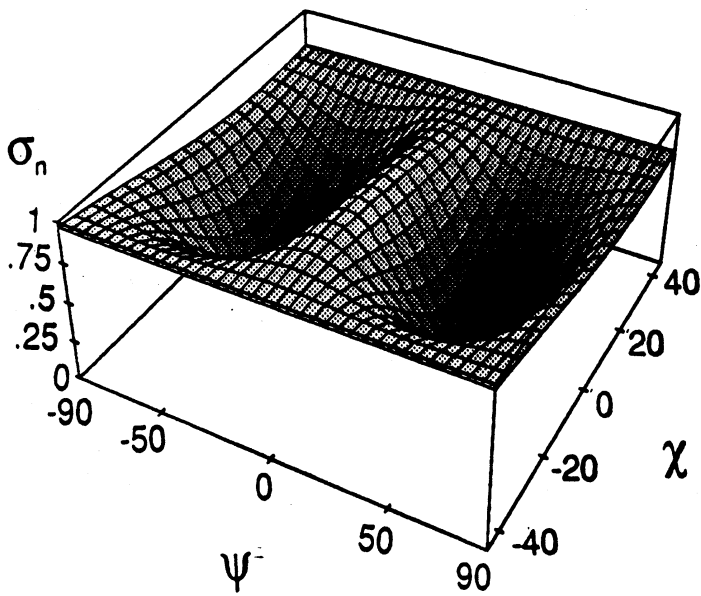
Co-pol. C1 Azimuth



Cross-pol. C1 Azimuth



Co-pol. C1 Elevation



Cross-pol. C1 Elevation

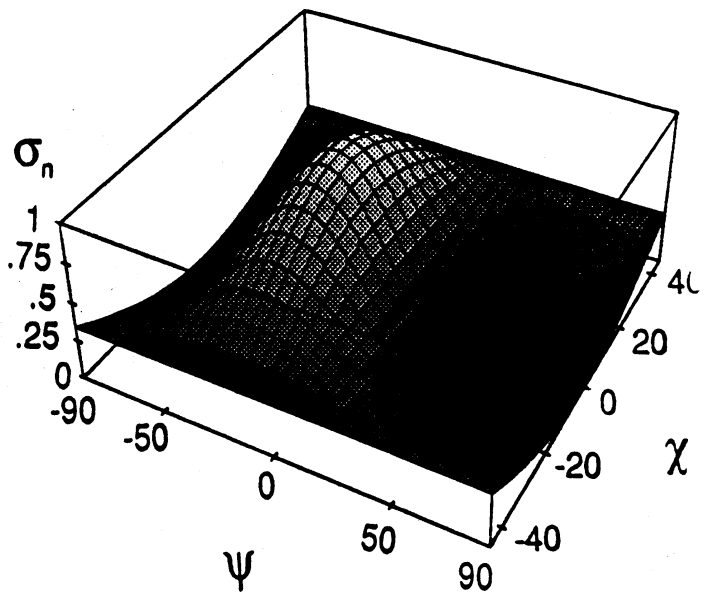
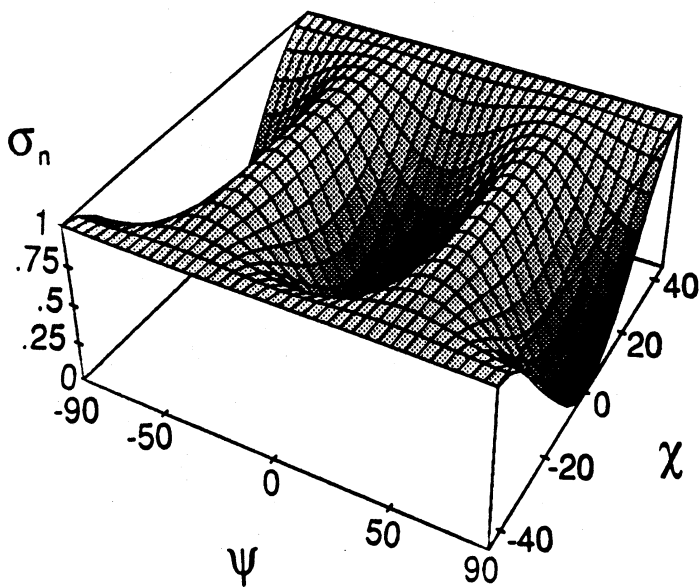
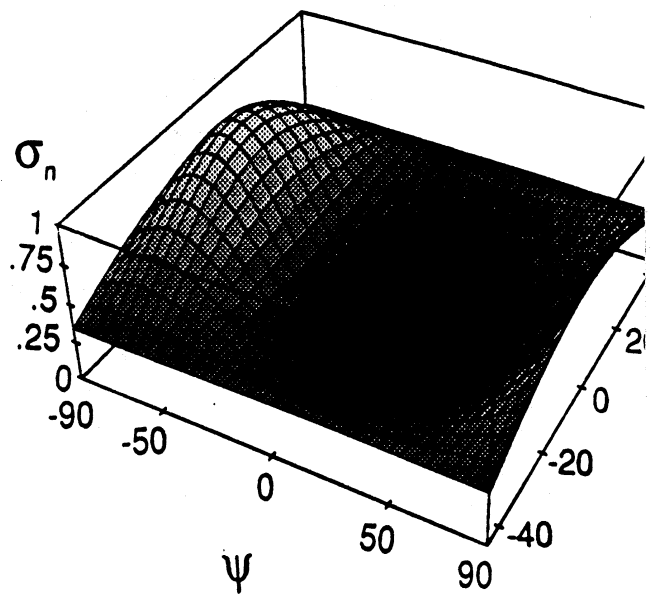


Figure 12: The azimuth and elevation polarization signature for C1 PARC.

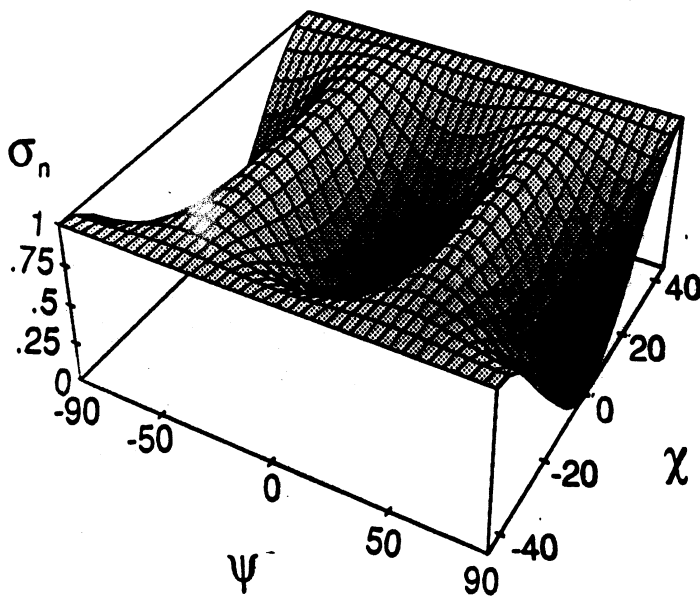
Co-pol. C1 45°



Cross-pol. C1 45°



Co-pol. C1 135°



Cross-pol. C1 135°

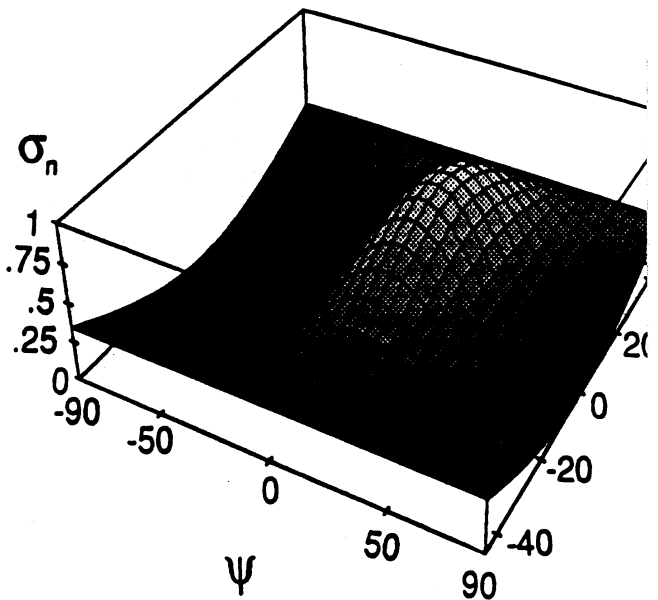


Figure 13: The 45° and 135° polarization signature for C1 PARC.

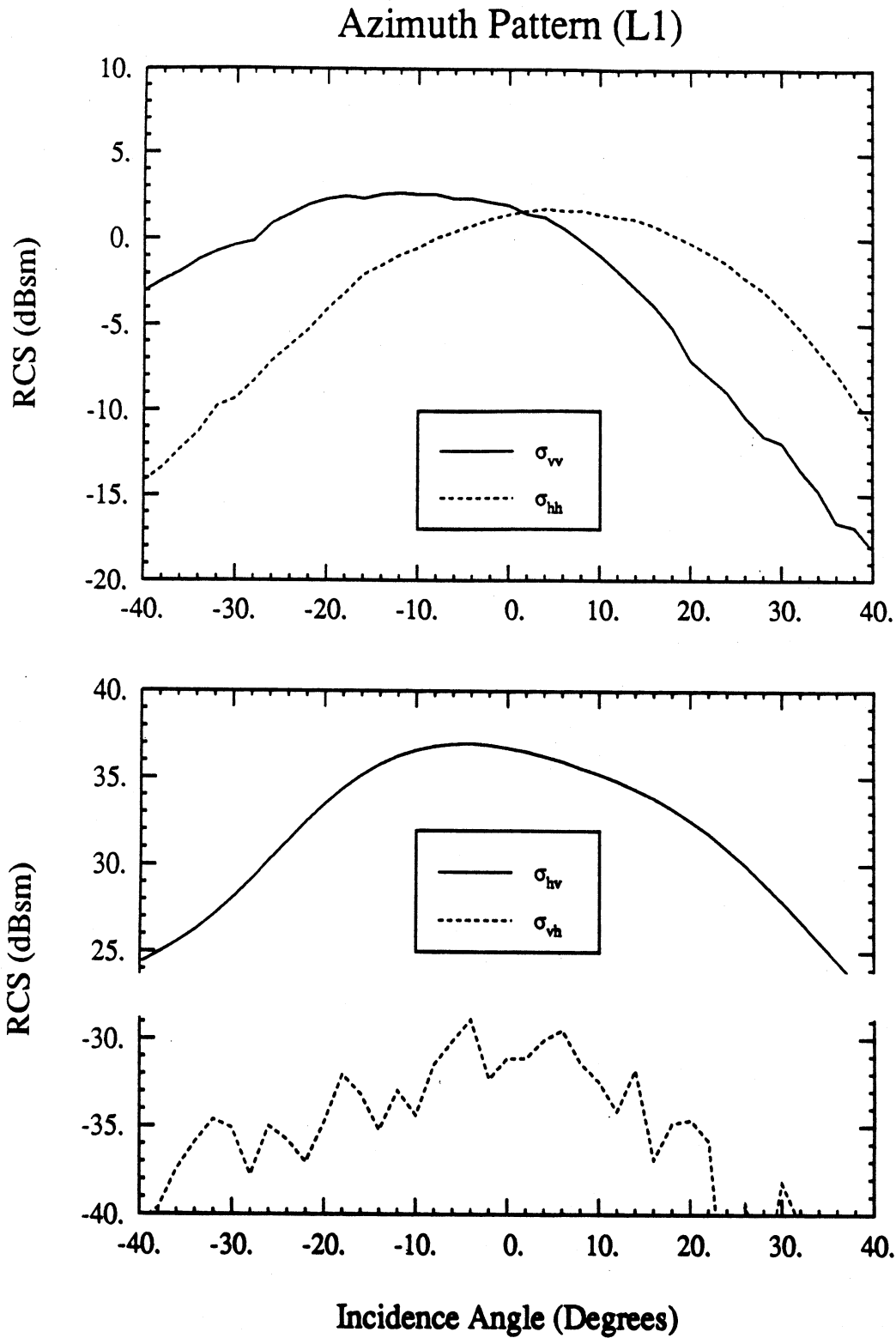


Figure 14: Azimuth pattern of amplitude of scattering matrix elements for L1 PARC.

Elevation Pattern (L1)

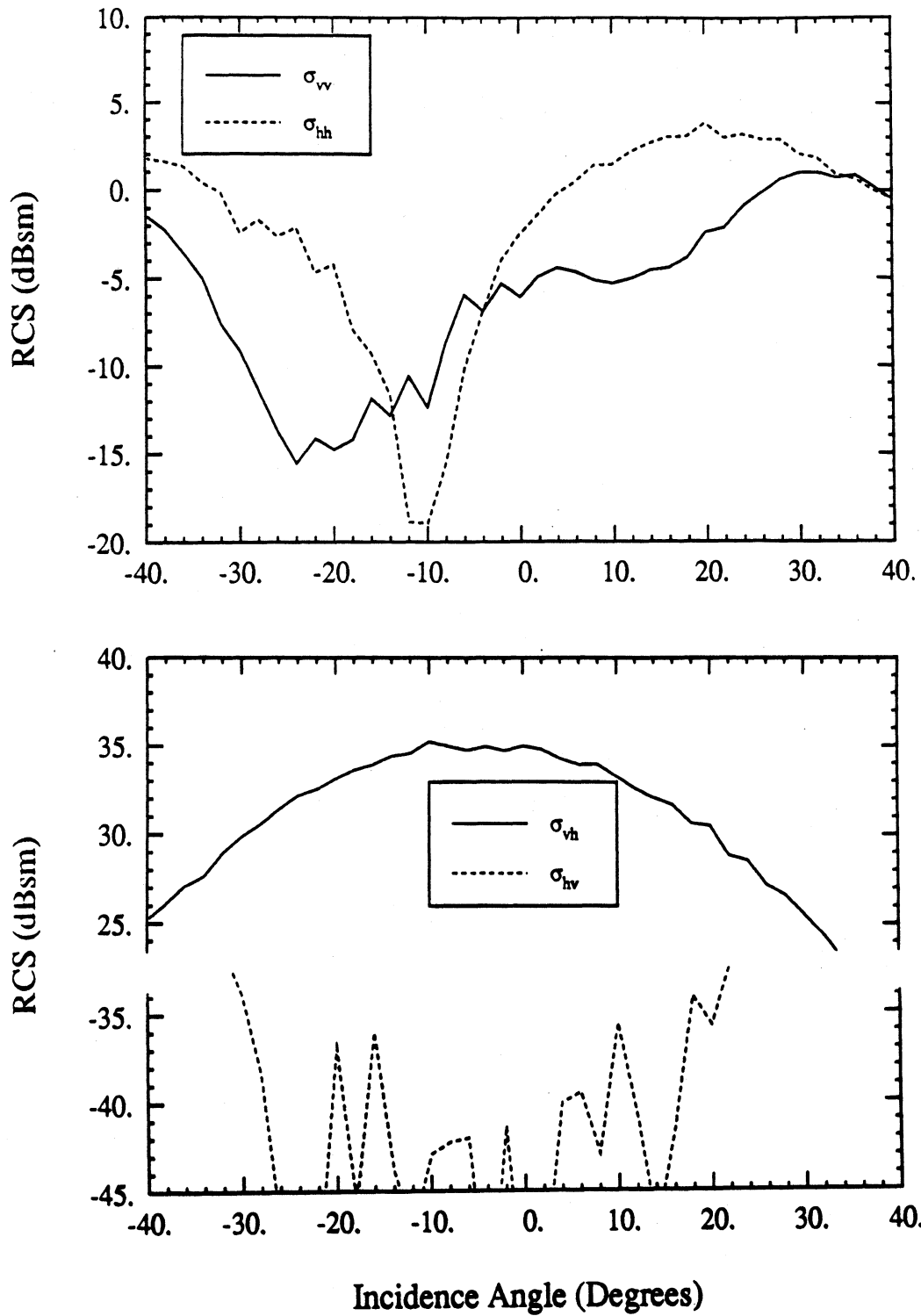


Figure 15: Elevation pattern of amplitude of scattering matrix elements for L1 PARC.

45° Pattern (L1)

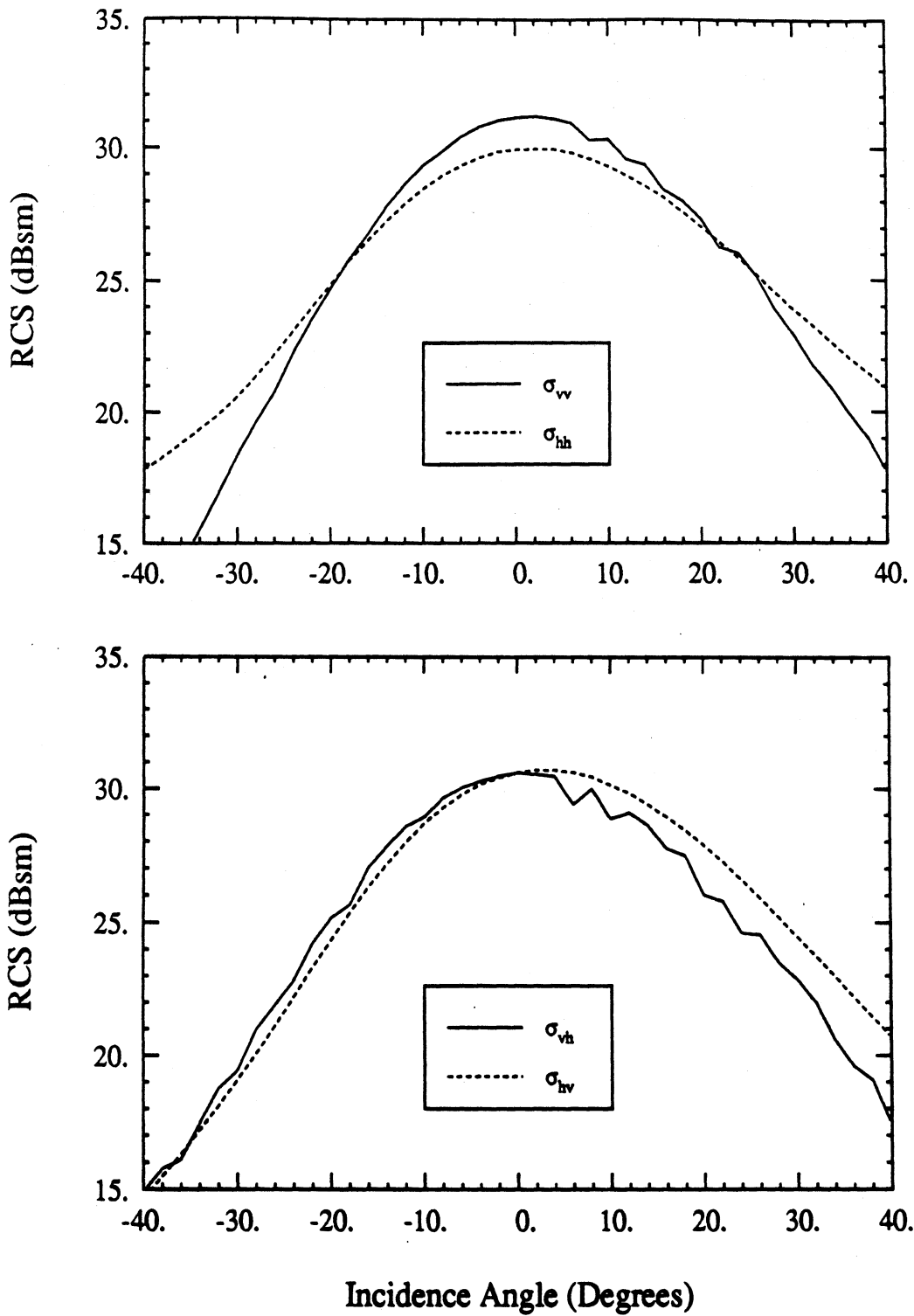


Figure 16: 45° pattern of amplitude of scattering matrix elements for L1 PARC.

45° Pattern (L1)

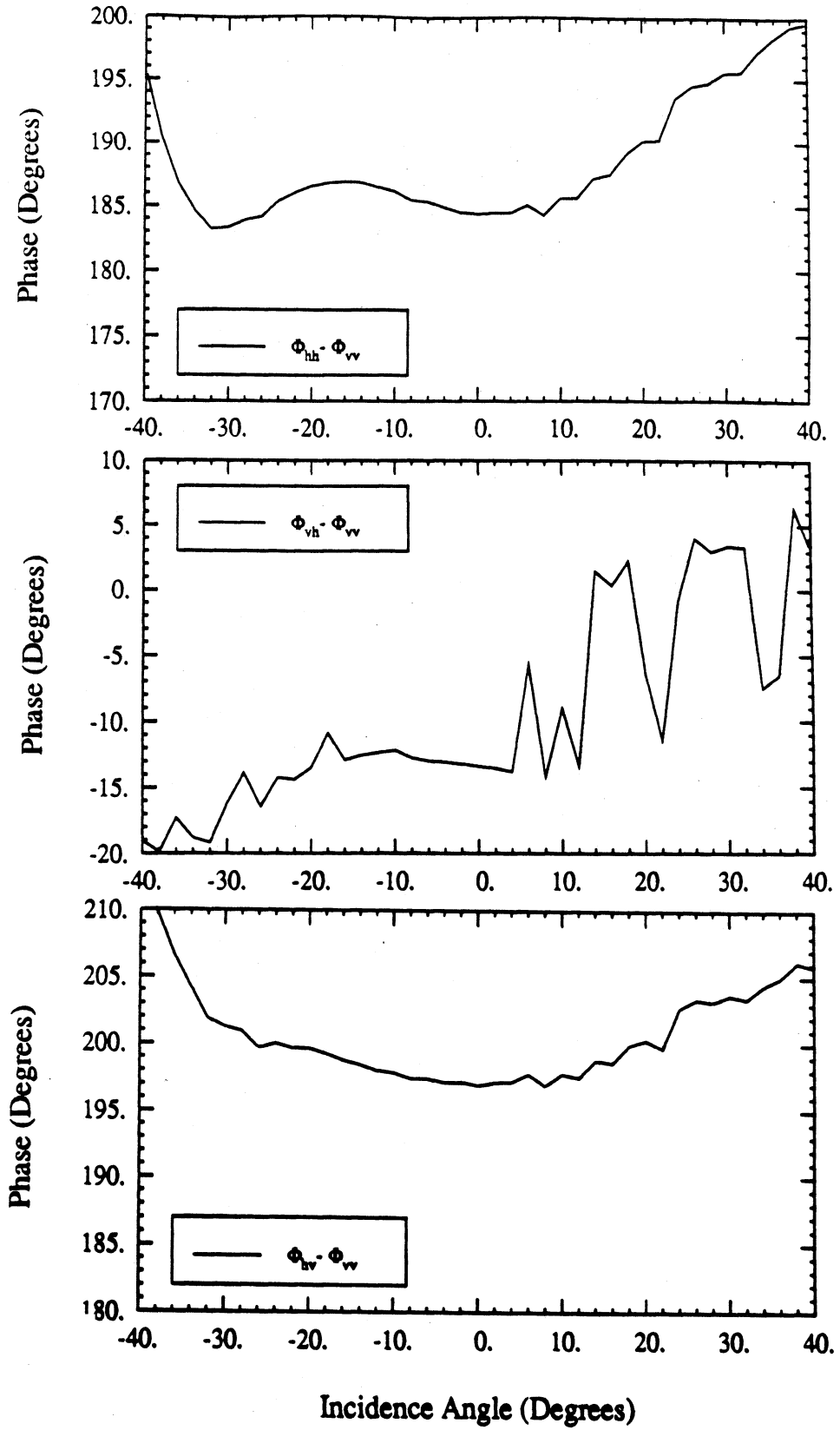


Figure 17: 45° pattern of phase of scattering matrix elements for L1 PARC.

135° Pattern (L1)

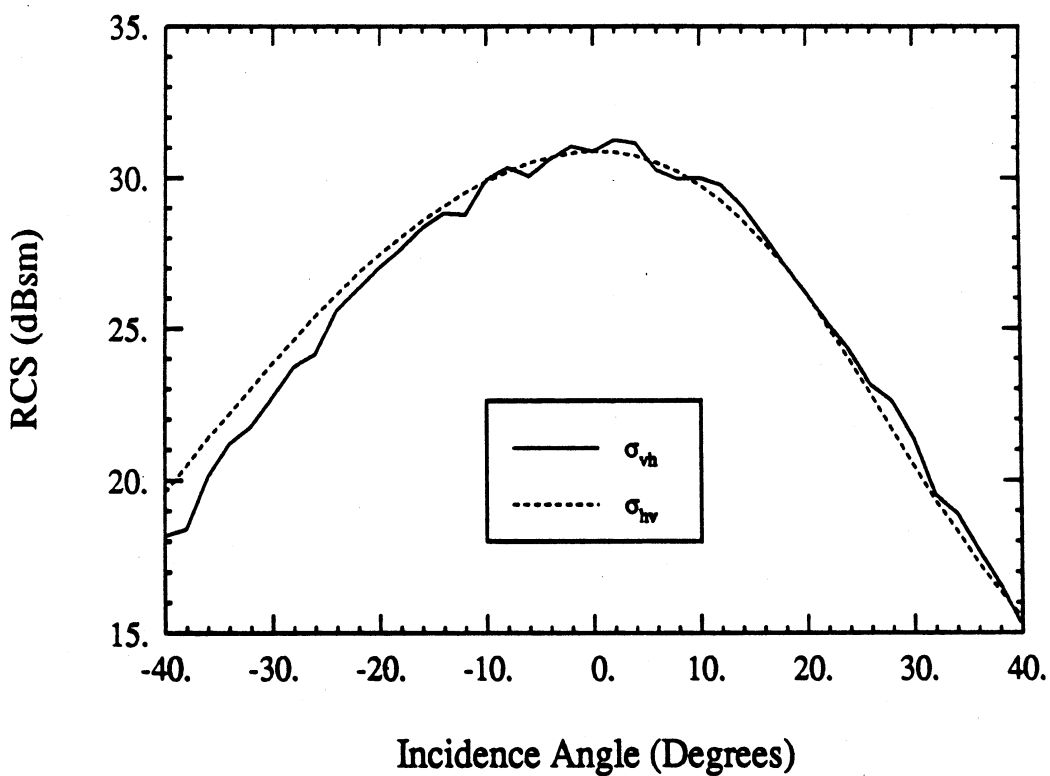
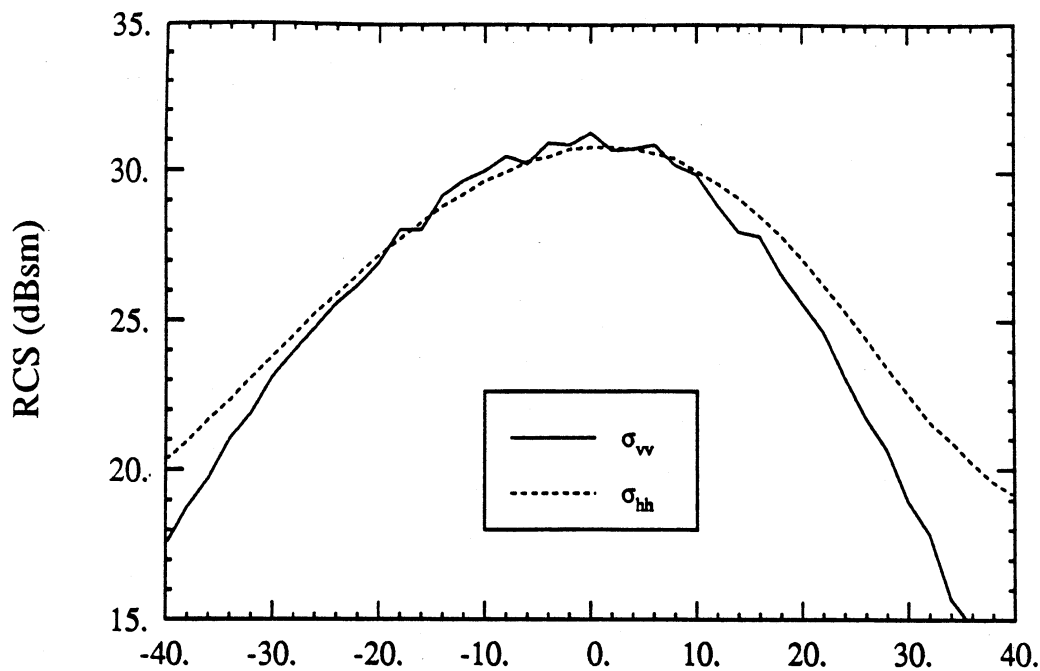


Figure 18: 135° pattern of amplitude of scattering matrix elements for L1 PARC.

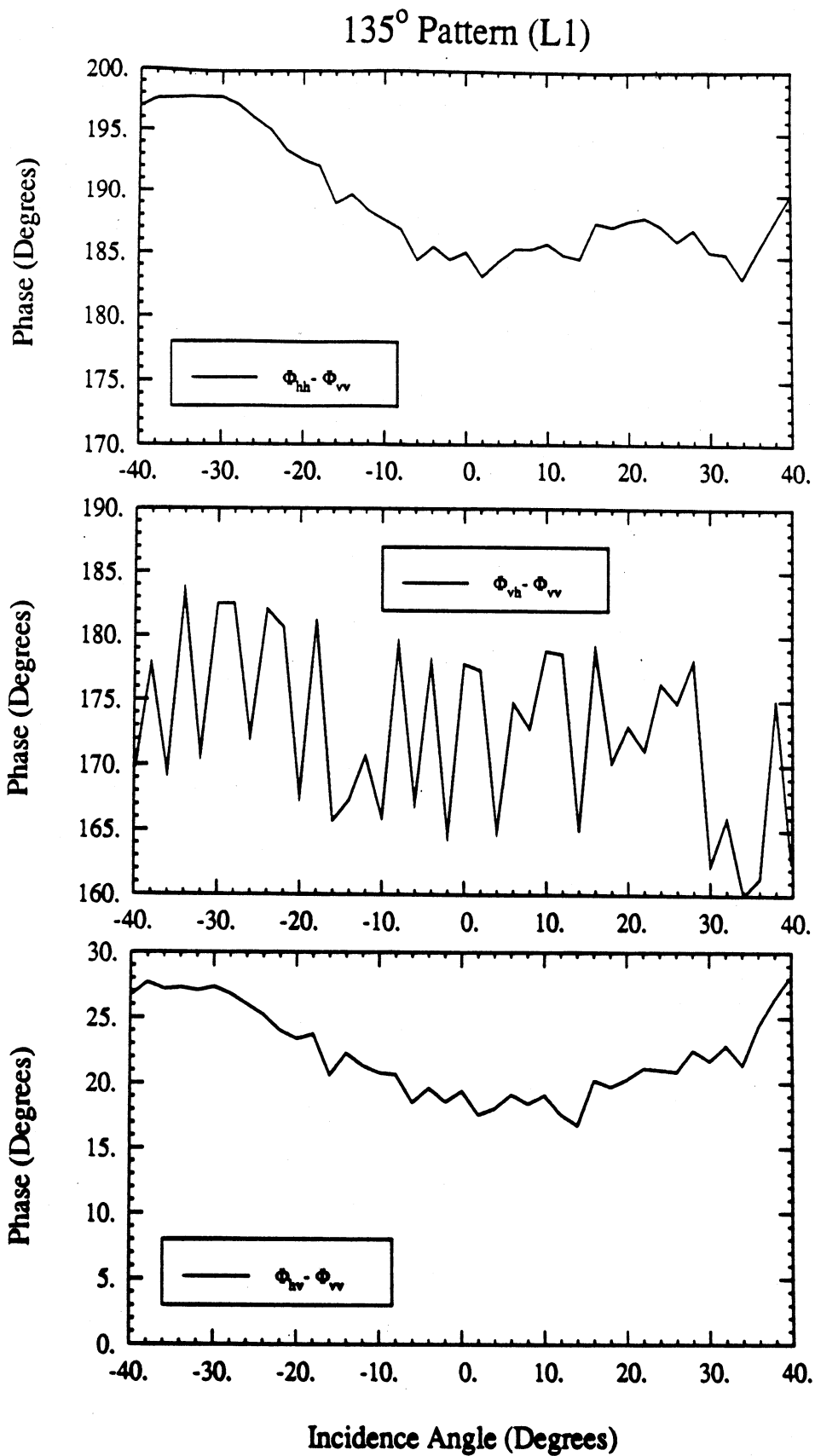


Figure 19: 135° pattern of phase of scattering matrix elements for L1 PARC.

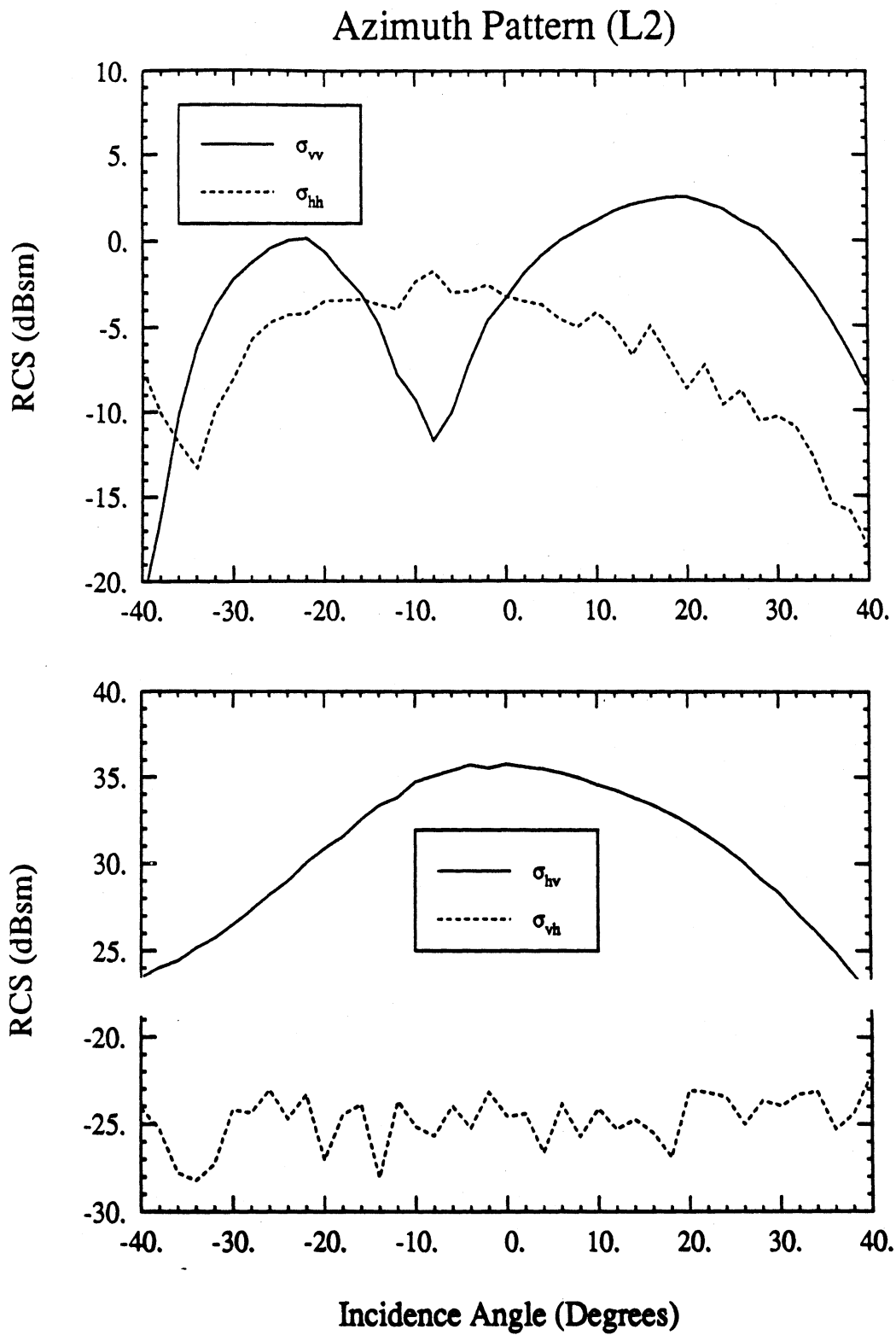


Figure 20: Azimuth pattern of amplitude of scattering matrix elements for L2 PARC.

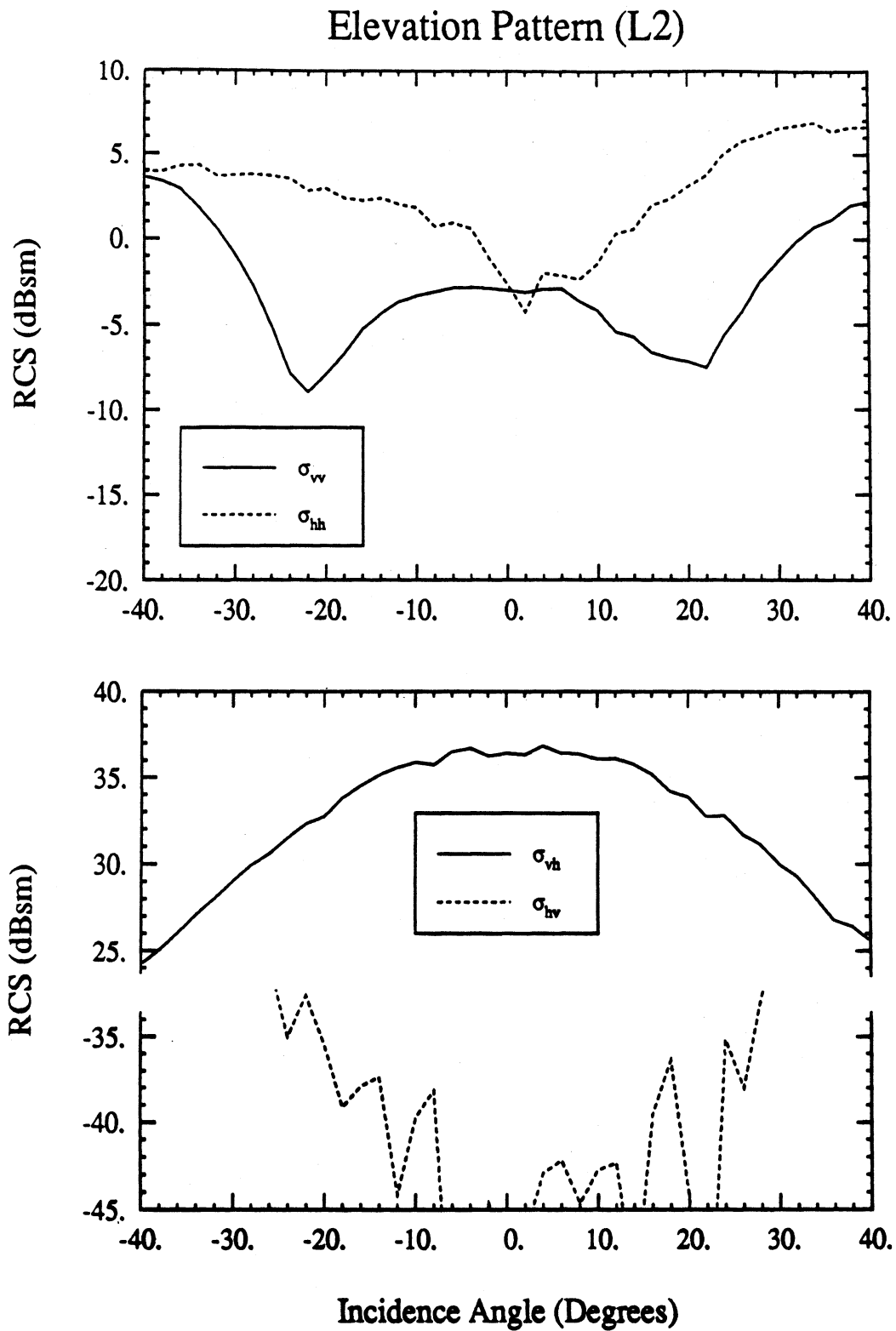


Figure 21: Elevation pattern of amplitude of scattering matrix elements for L2 PARC.

45° Pattern (L2)

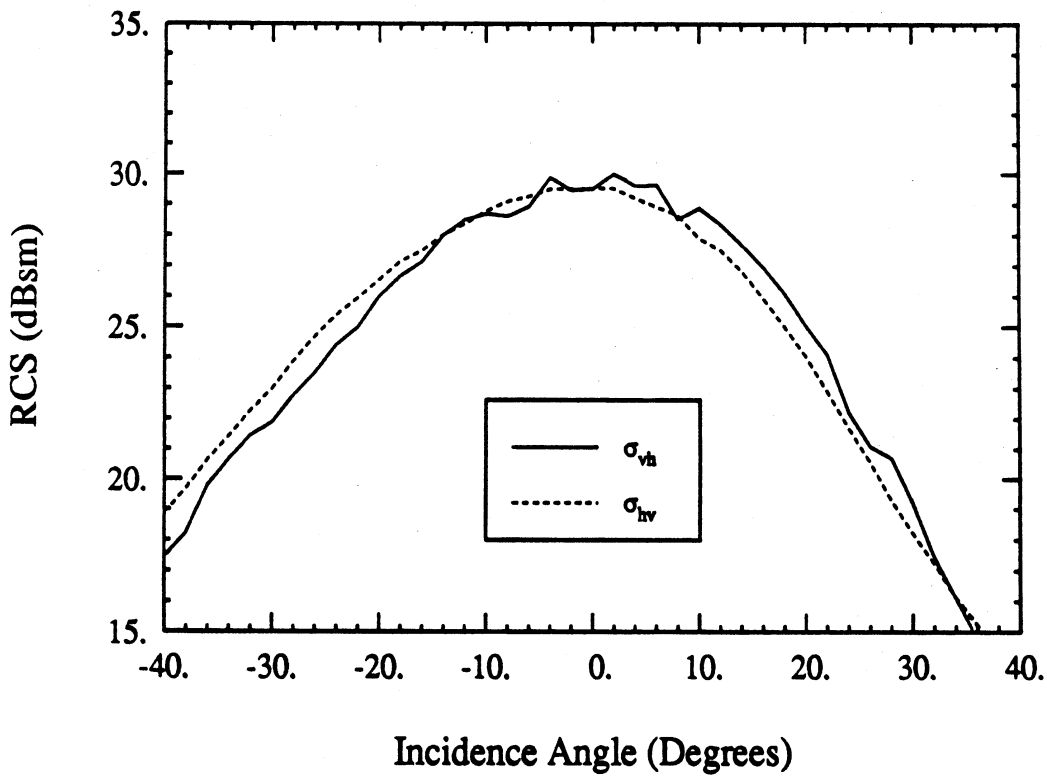
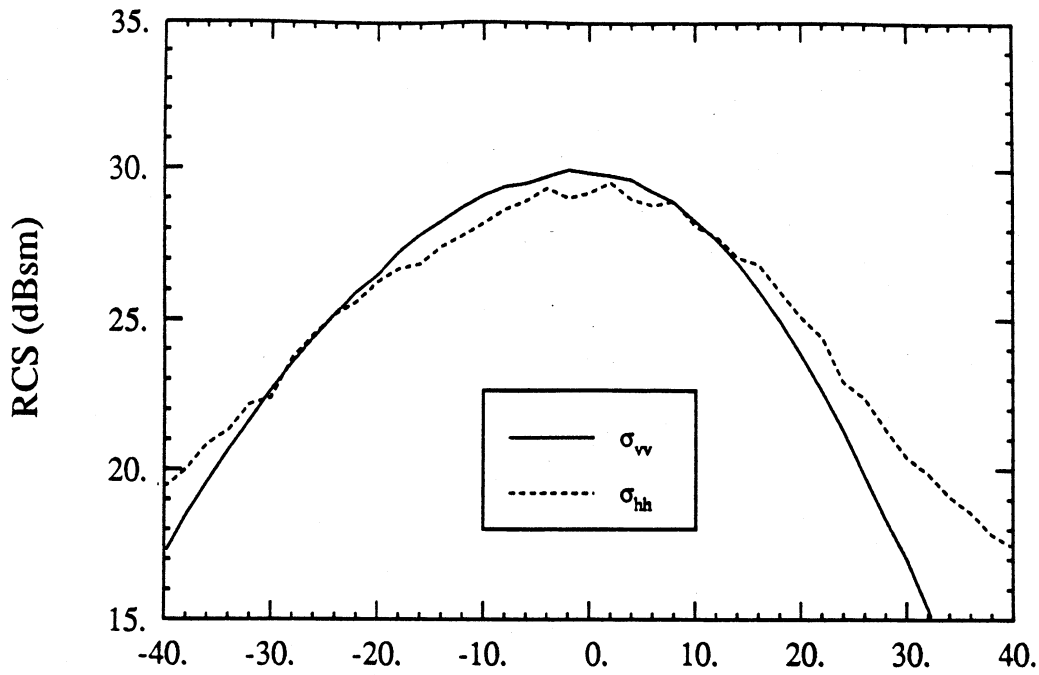


Figure 22: 45° pattern of amplitude of scattering matrix elements for L2 PARC.

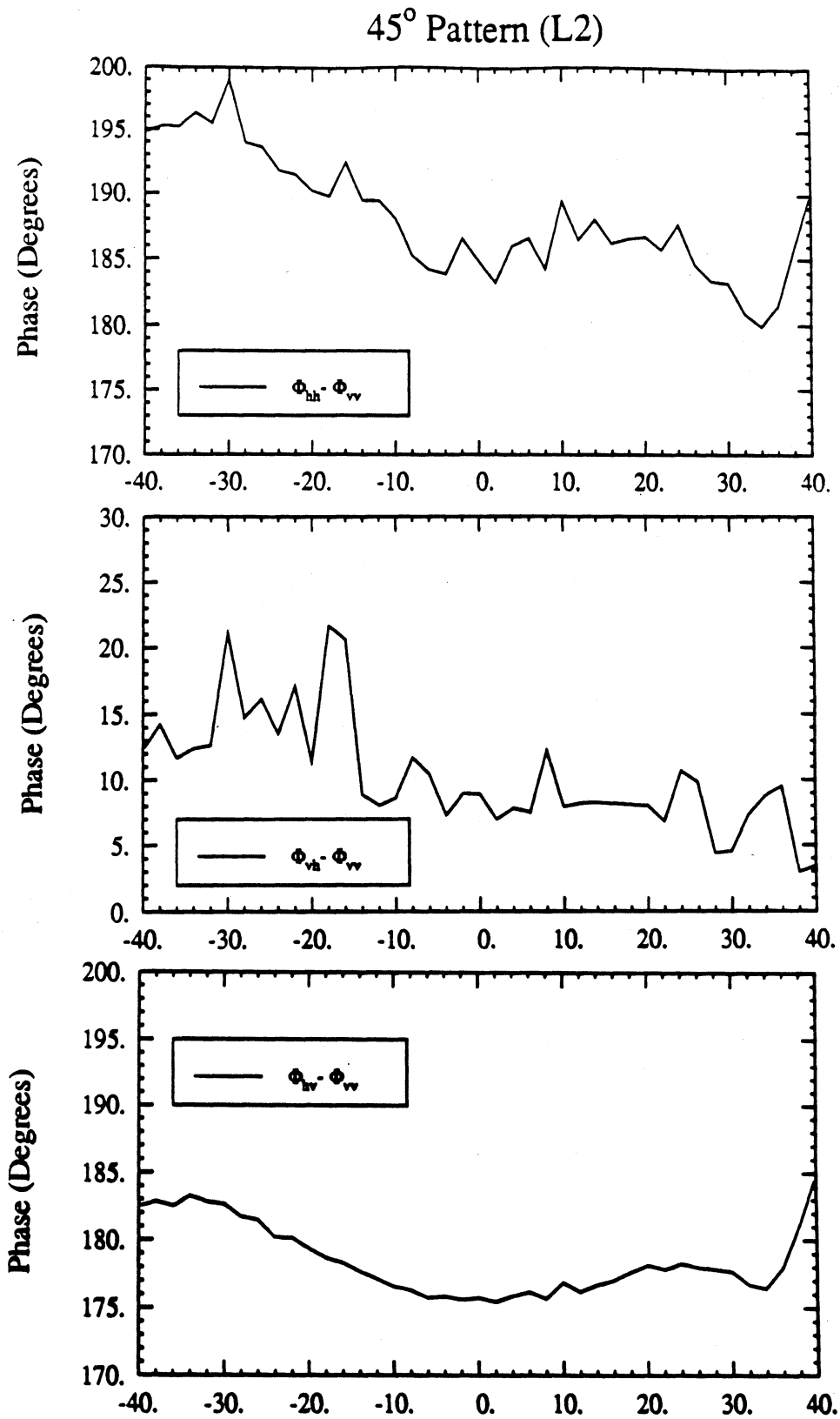


Figure 23: 45° pattern of phase of scattering matrix elements for L2 PARC.

135° Pattern (L2)

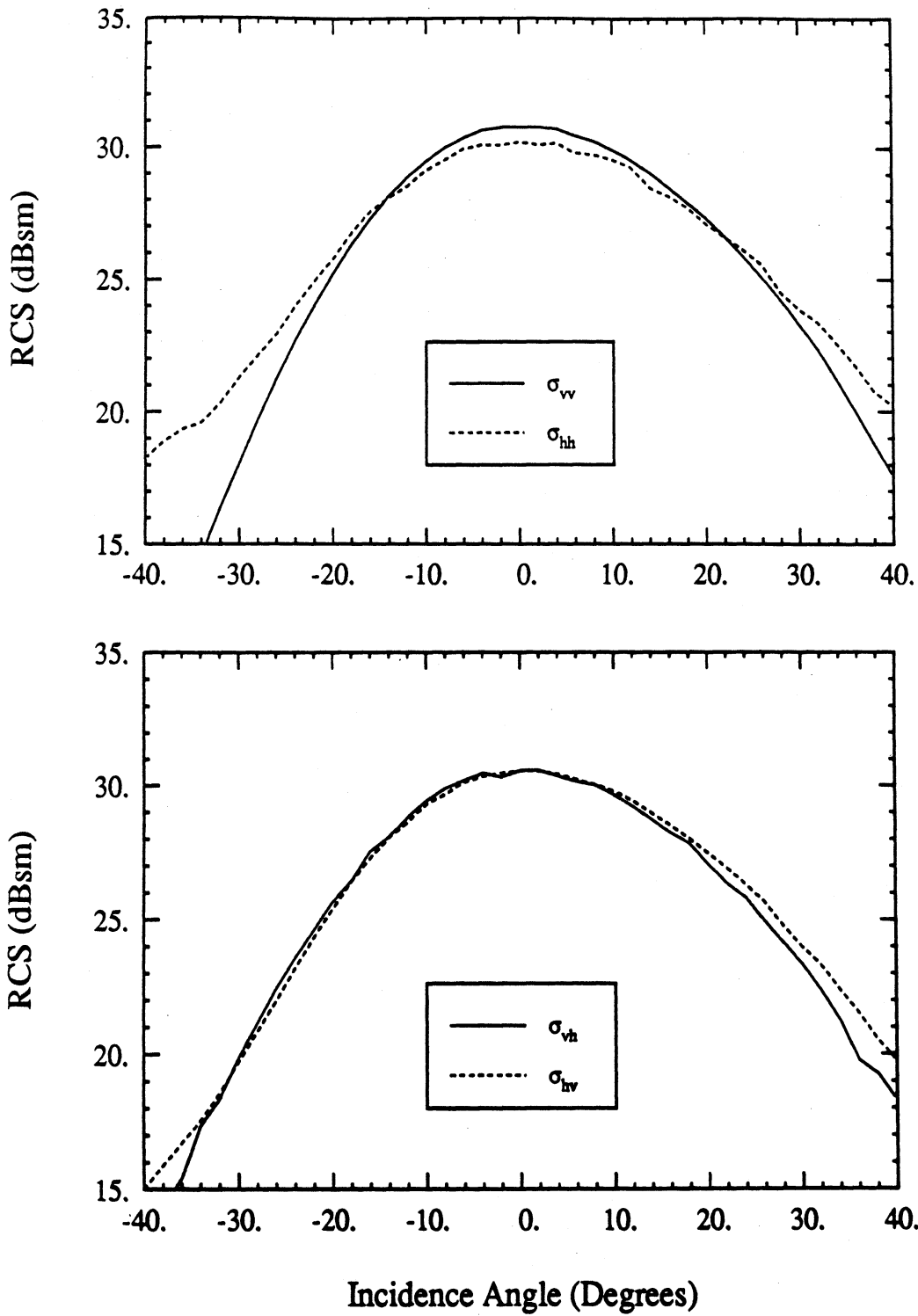


Figure 24: 135° pattern of amplitude of scattering matrix elements for L2 PARC.

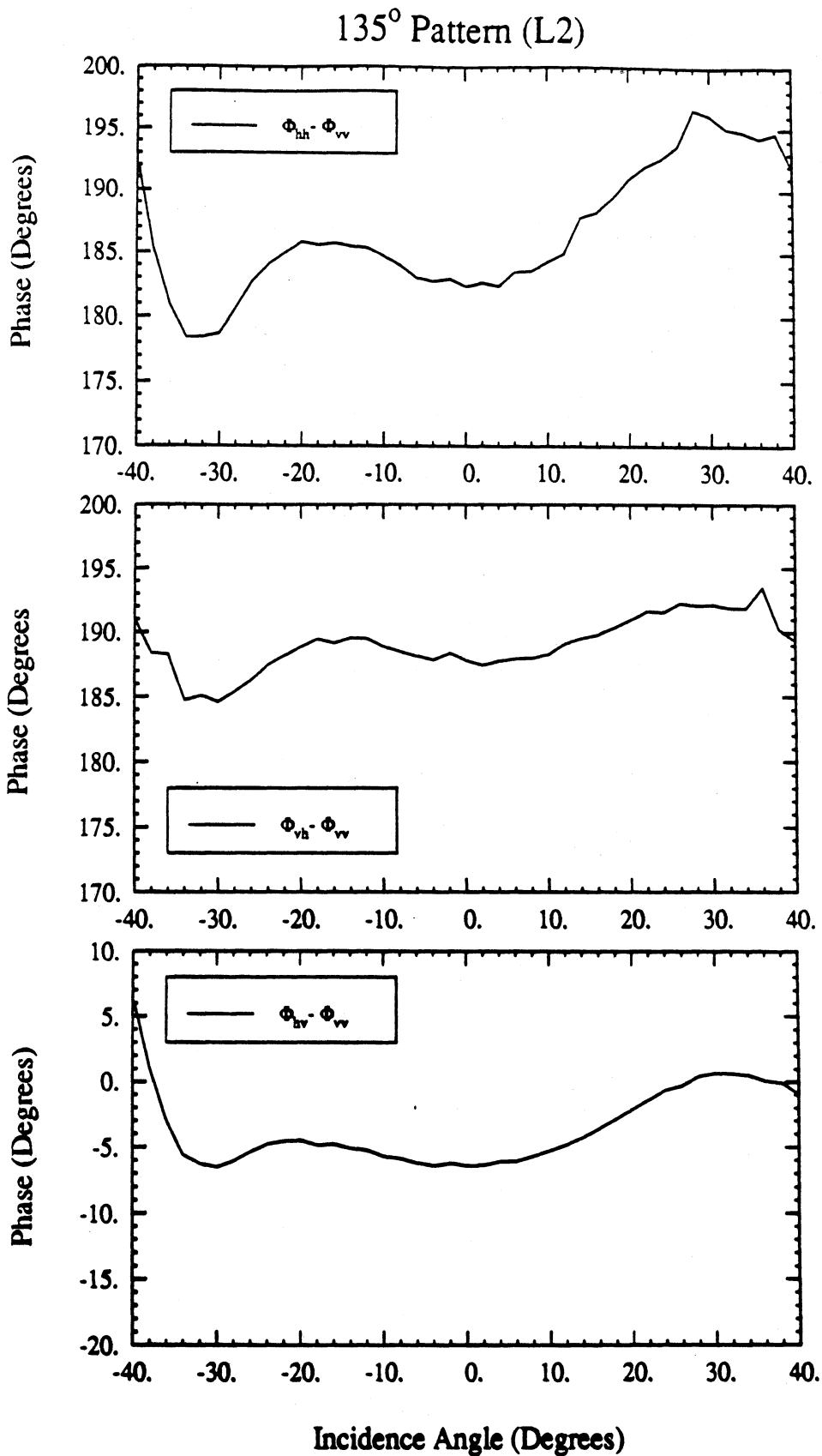


Figure 25: 135° pattern of phase of scattering matrix elements for L2 PARC.

Azimuth Pattern (L3)

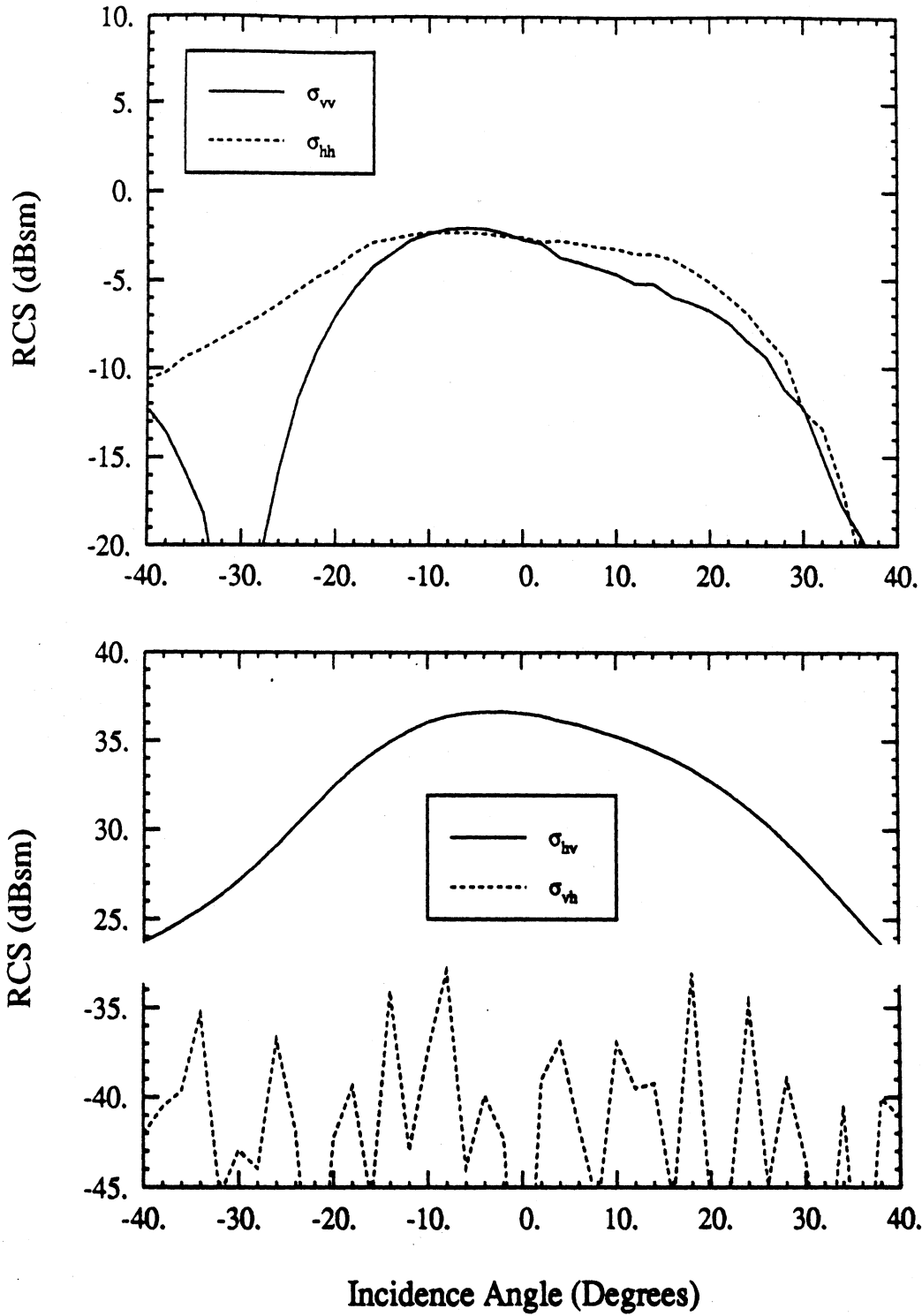


Figure 26: Azimuth pattern of amplitude of scattering matrix elements for L3 PARC.

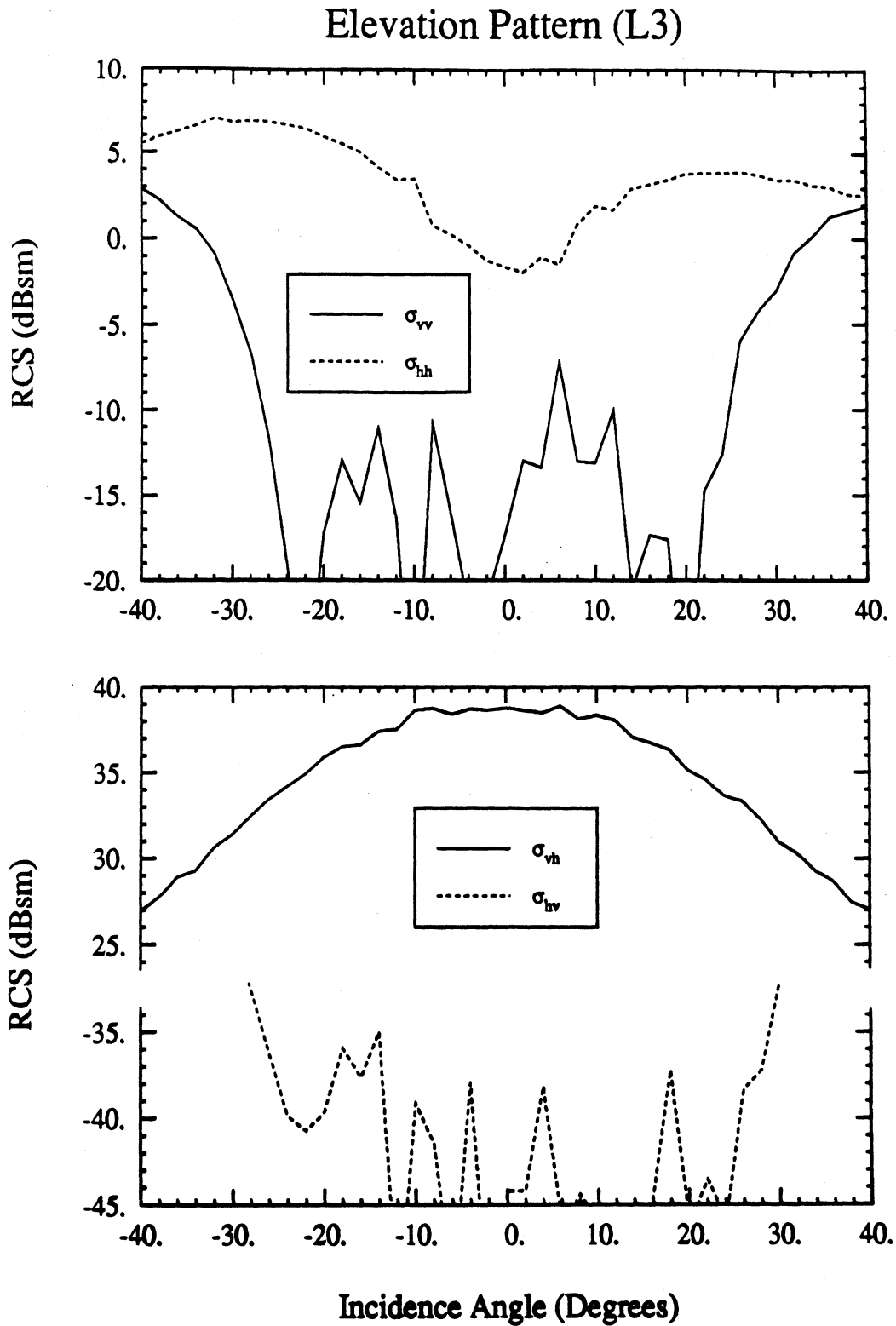


Figure 27: Elevation pattern of amplitude of scattering matrix elements for L3 PARC.

45° Pattern (L3)

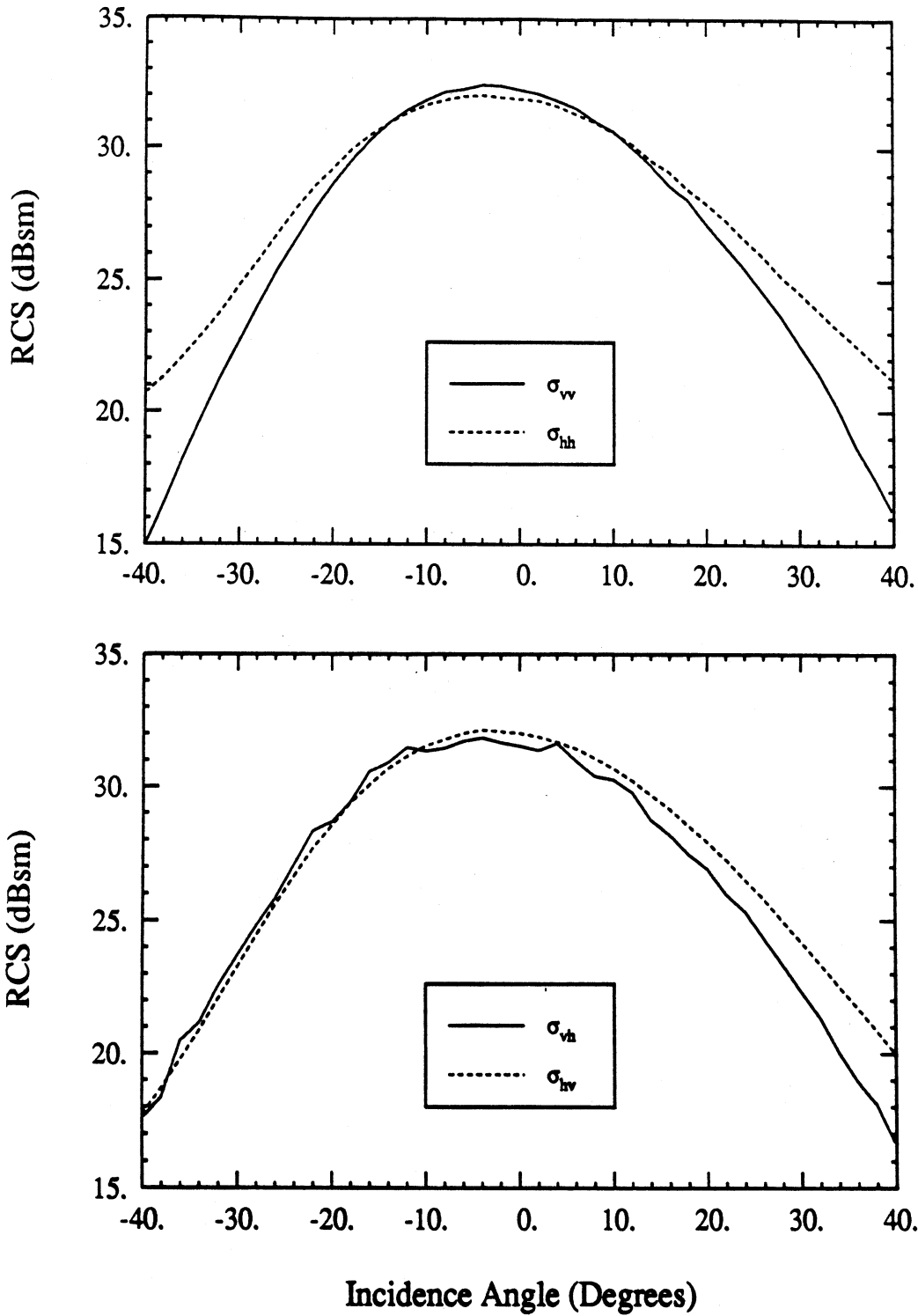


Figure 28: 45° pattern of amplitude of scattering matrix elements for L3 PARC.

45° Pattern (L3)

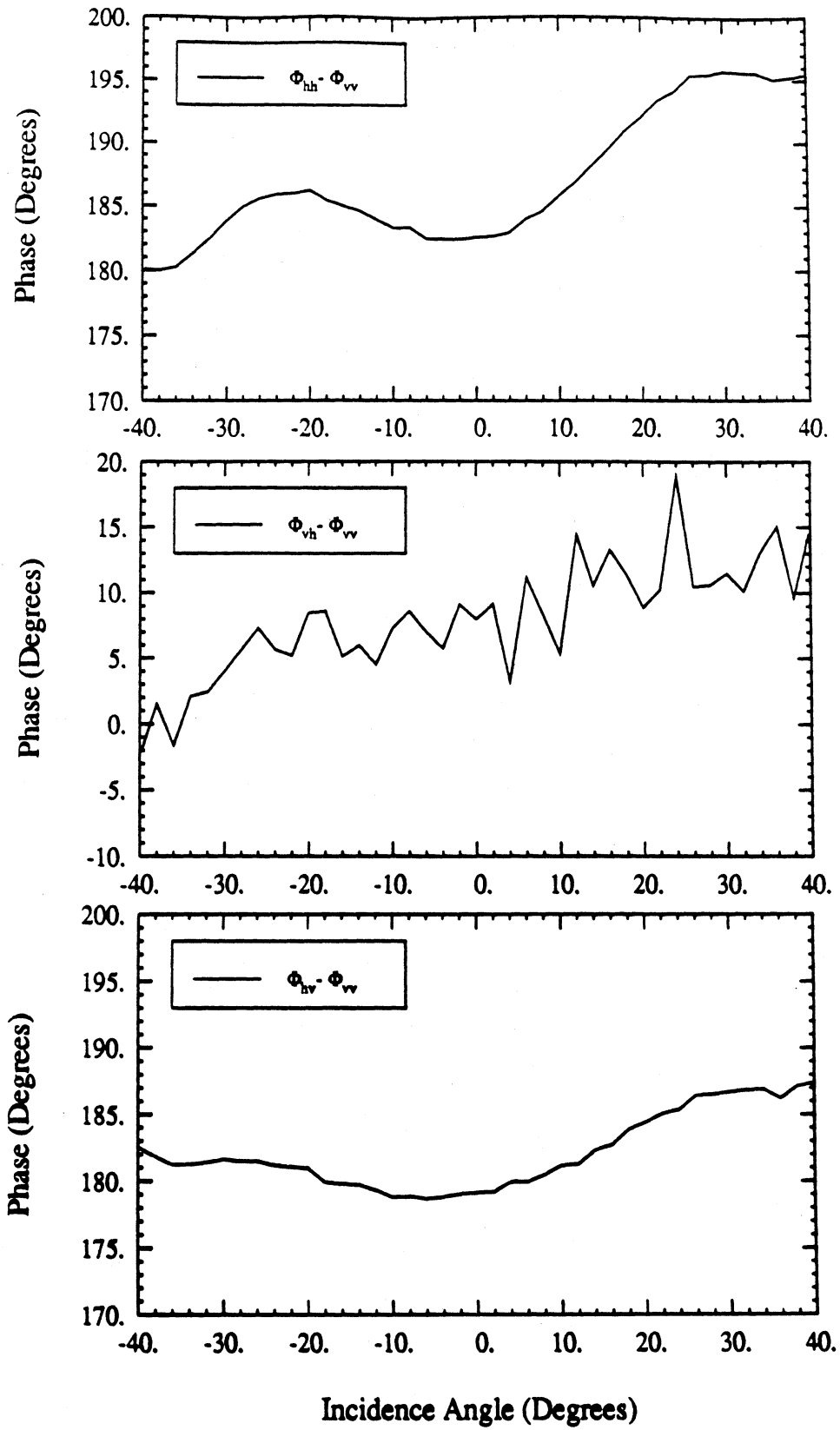


Figure 29: 45° pattern of phase of scattering matrix elements for L3 PARC.

135° Pattern (L3)

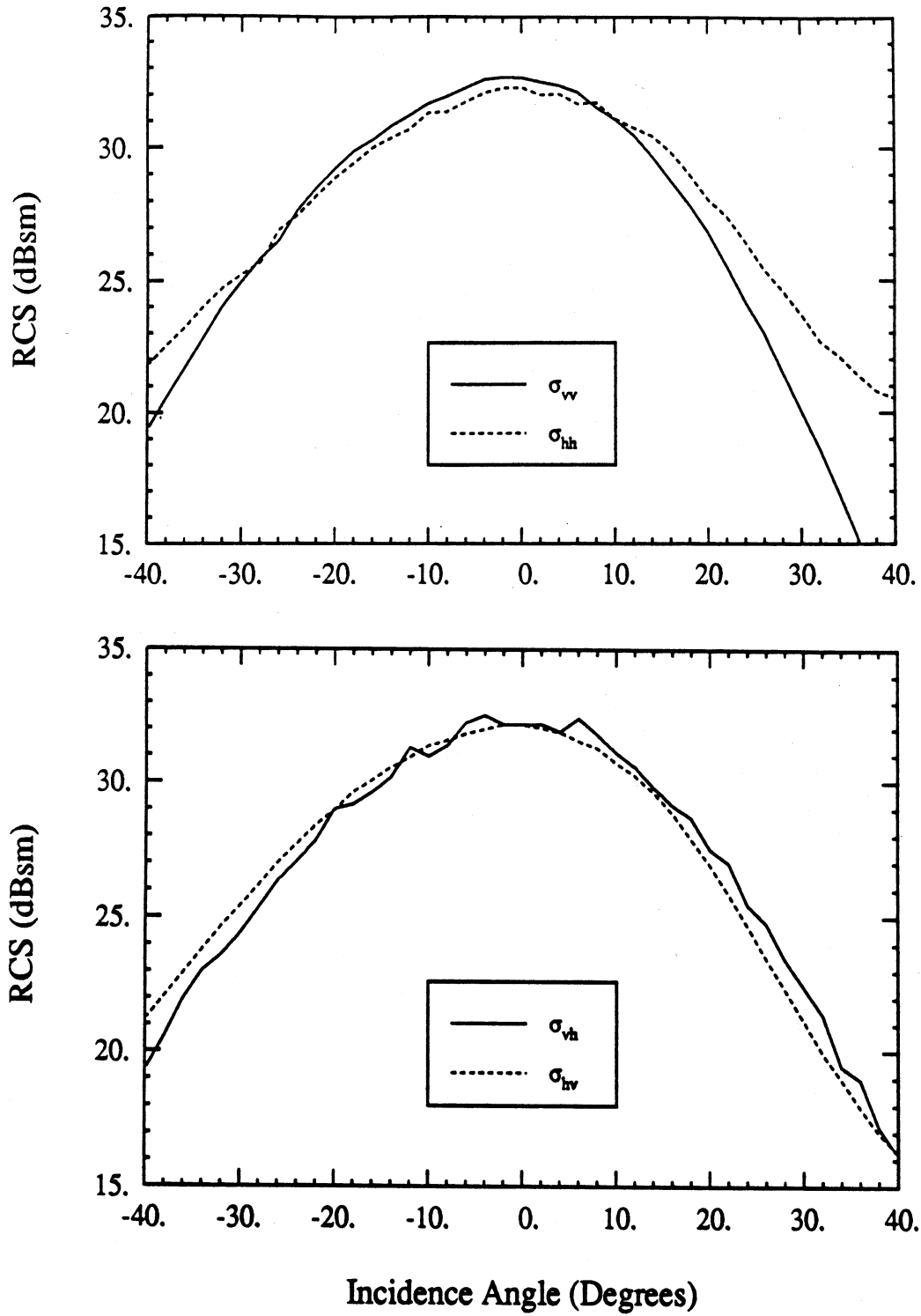


Figure 30: 135° pattern of amplitude of scattering matrix elements for L3 PARC.

135° Pattern (L3)

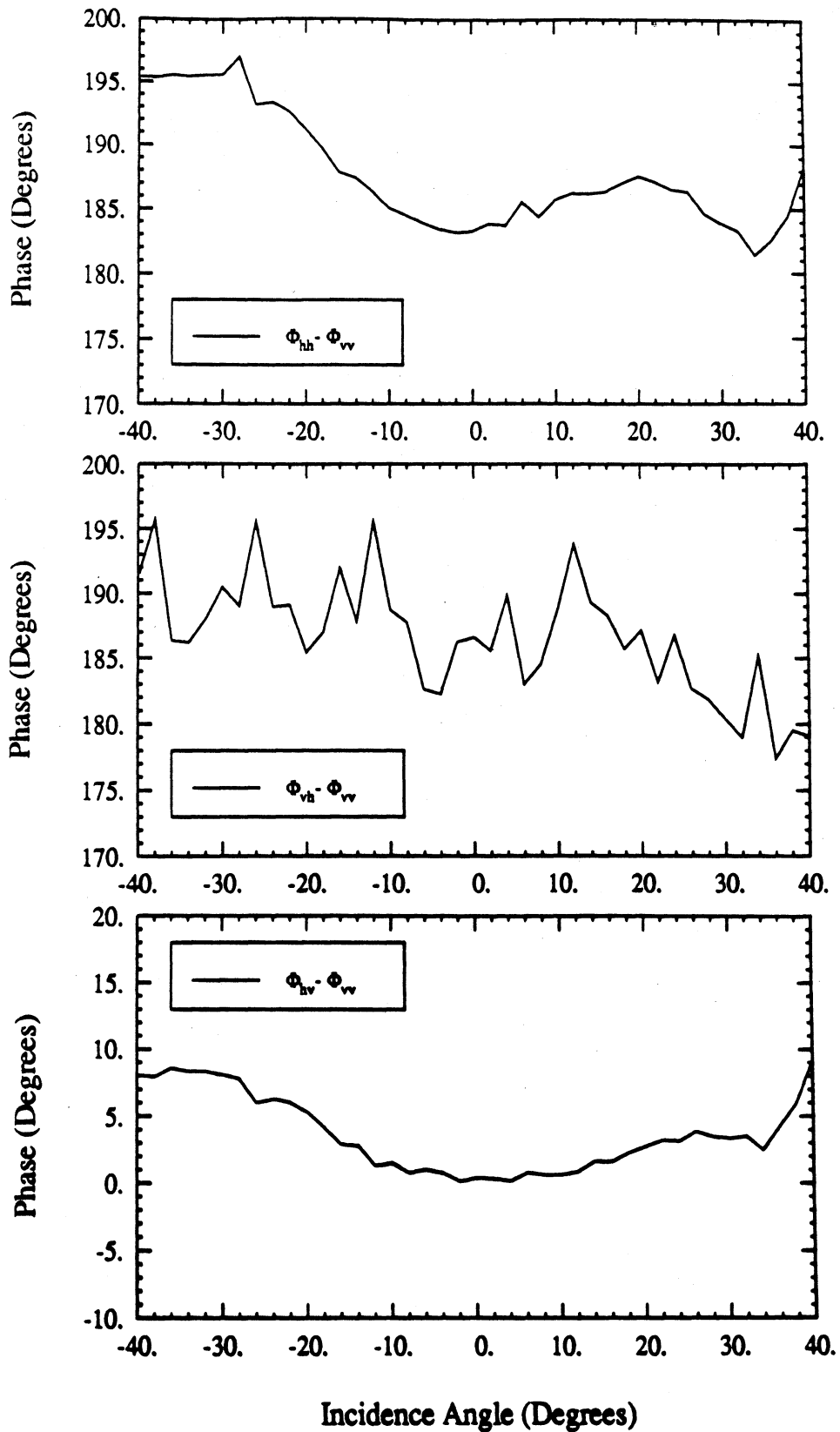


Figure 31: 135° pattern of phase of scattering matrix elements for L3 PARC.

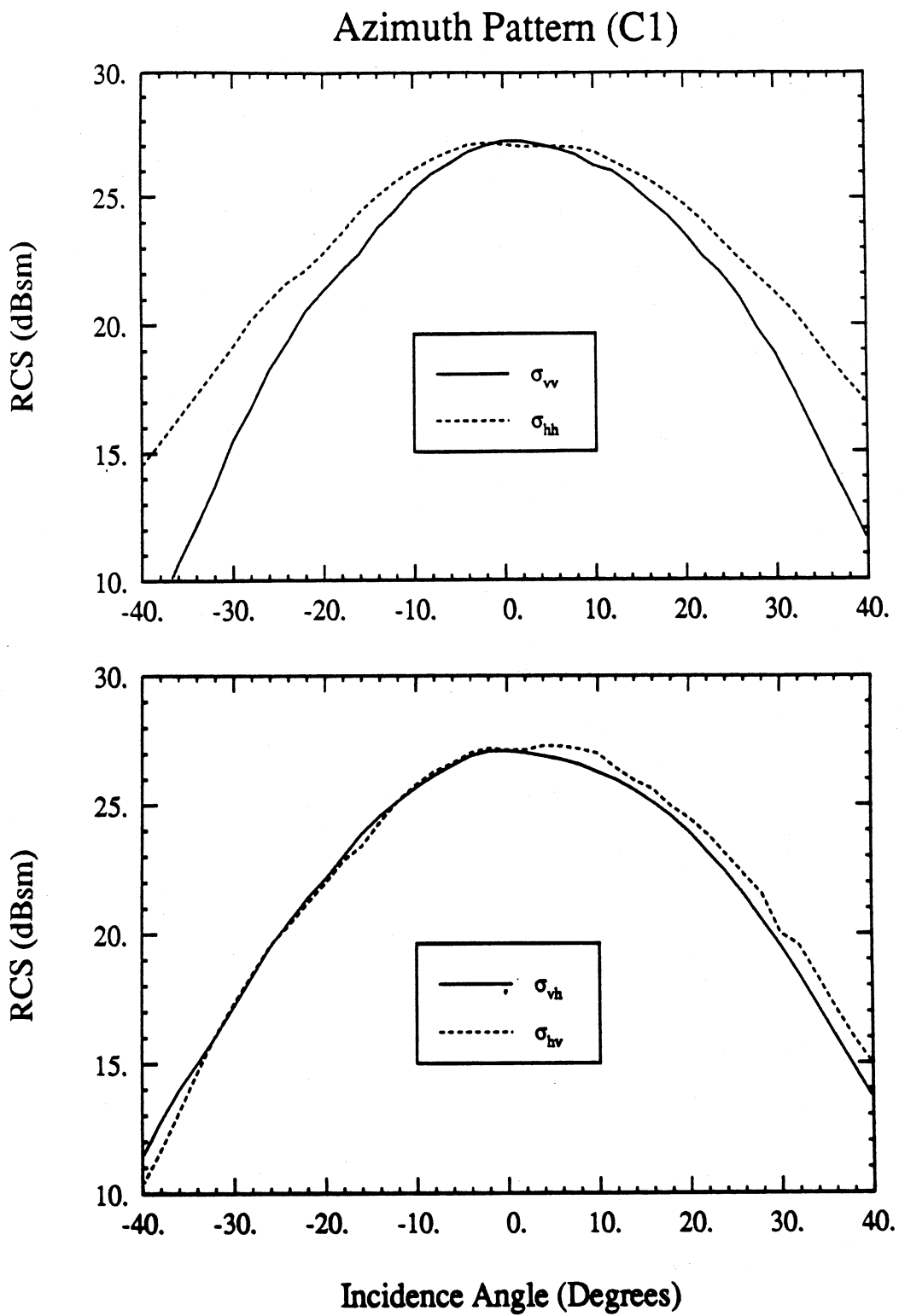


Figure 32: Azimuth pattern of amplitude of scattering matrix elements for C1 PARC.

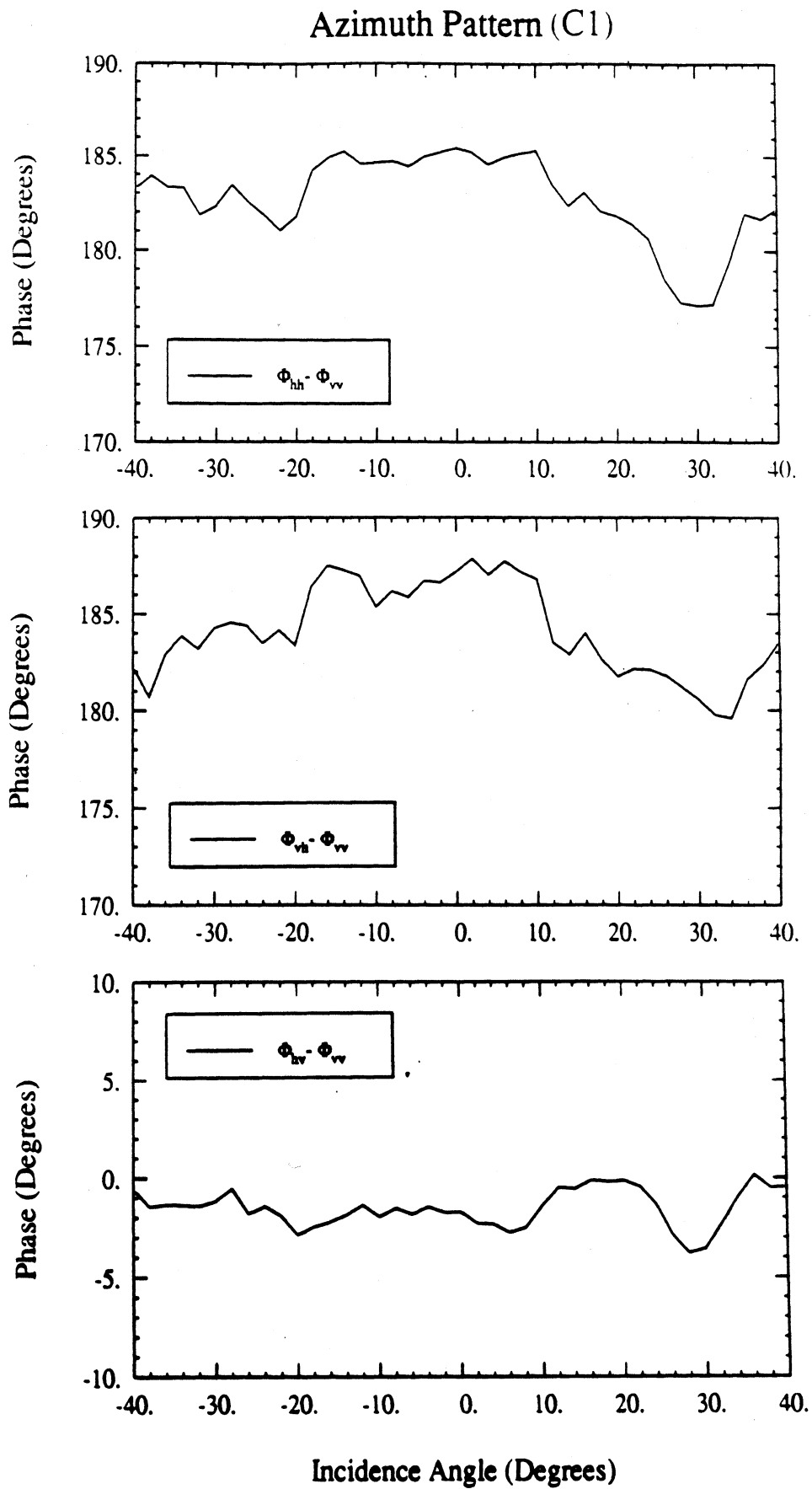


Figure 33: Azimuth pattern of phase of scattering matrix elements for C1 PARC.

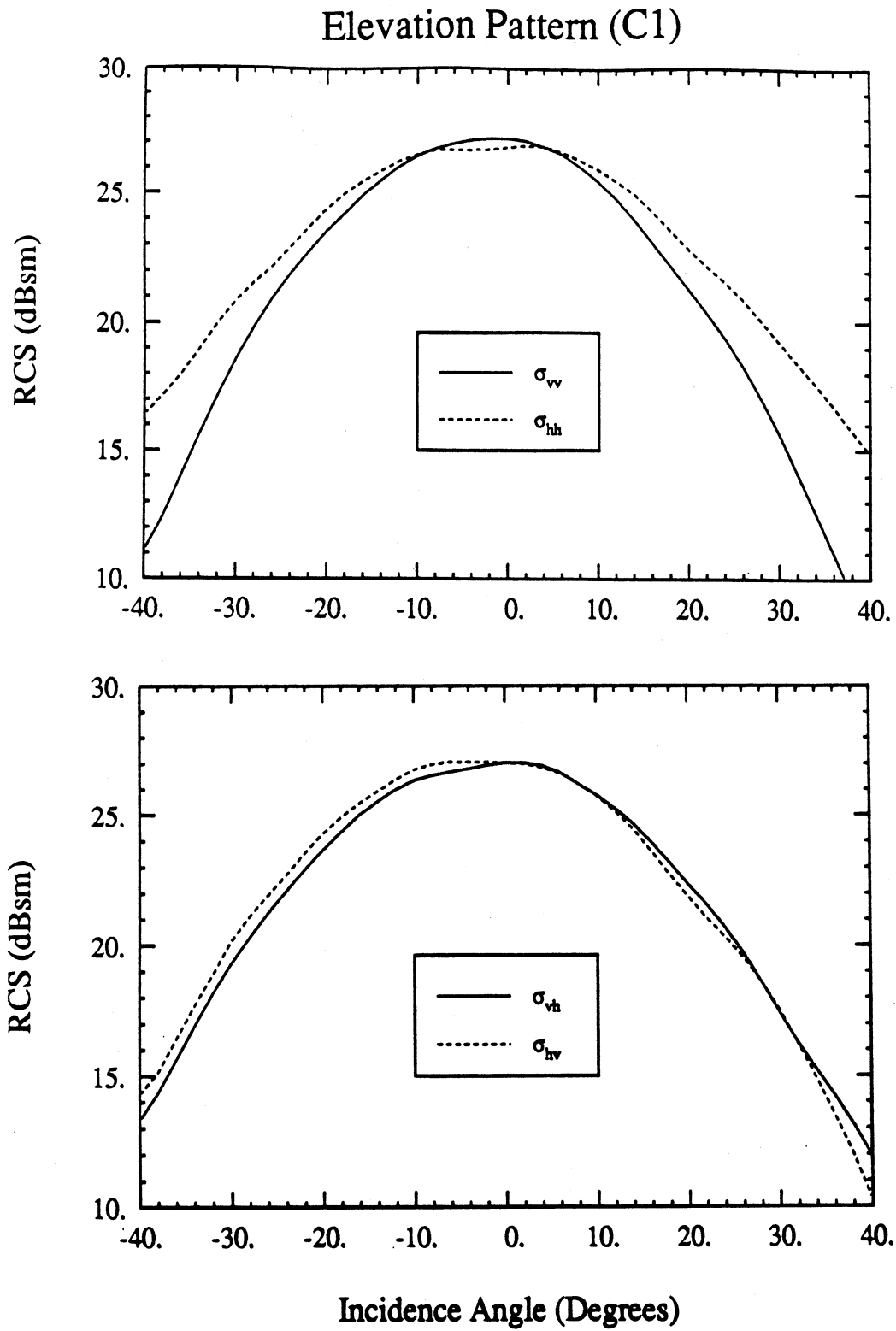


Figure 34: Elevation pattern of amplitude of scattering matrix elements for C1 PARC.

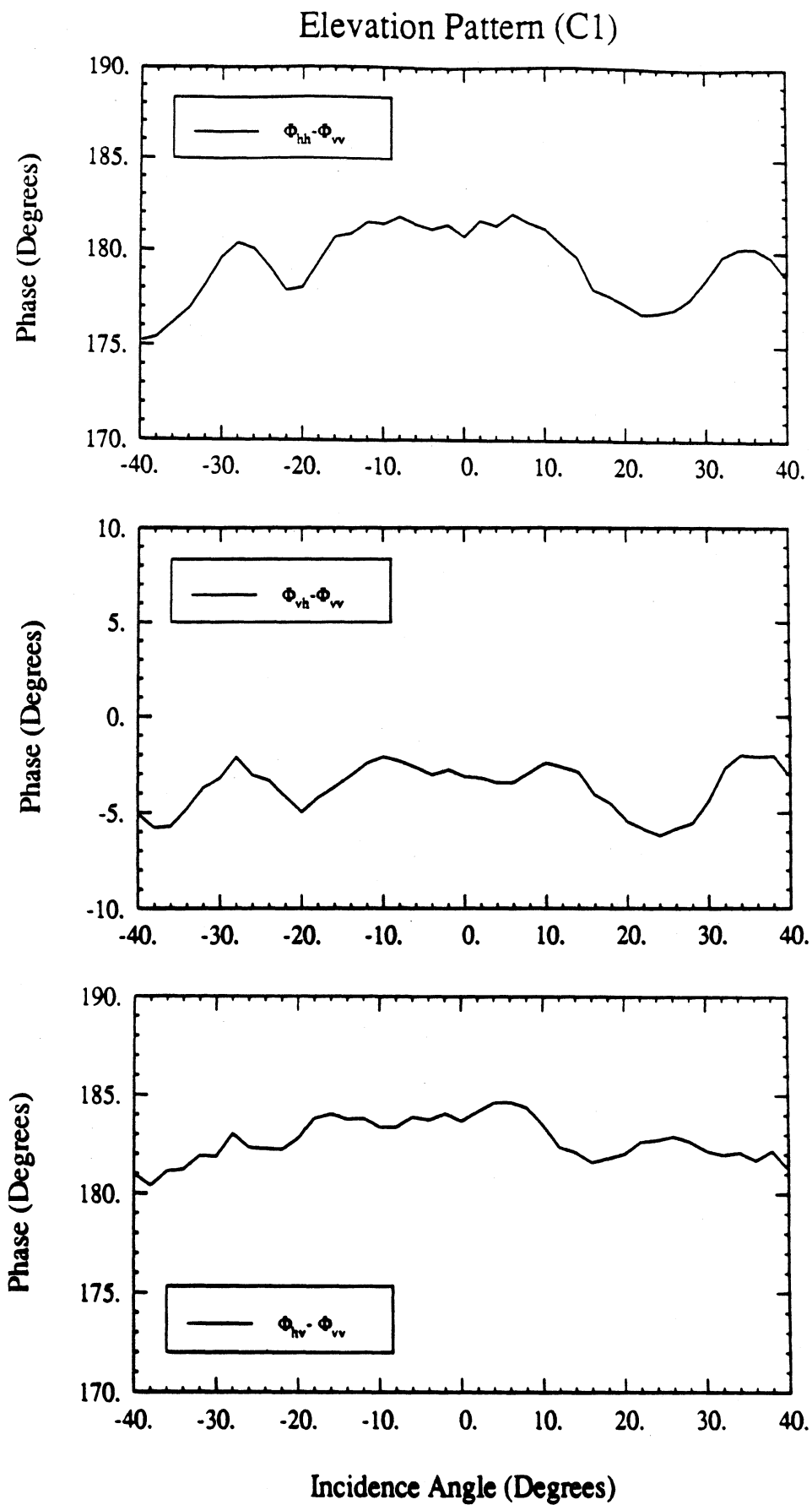


Figure 35: Elevation pattern of phase of scattering matrix elements for C1 PARC.

45° Pattern (C1)

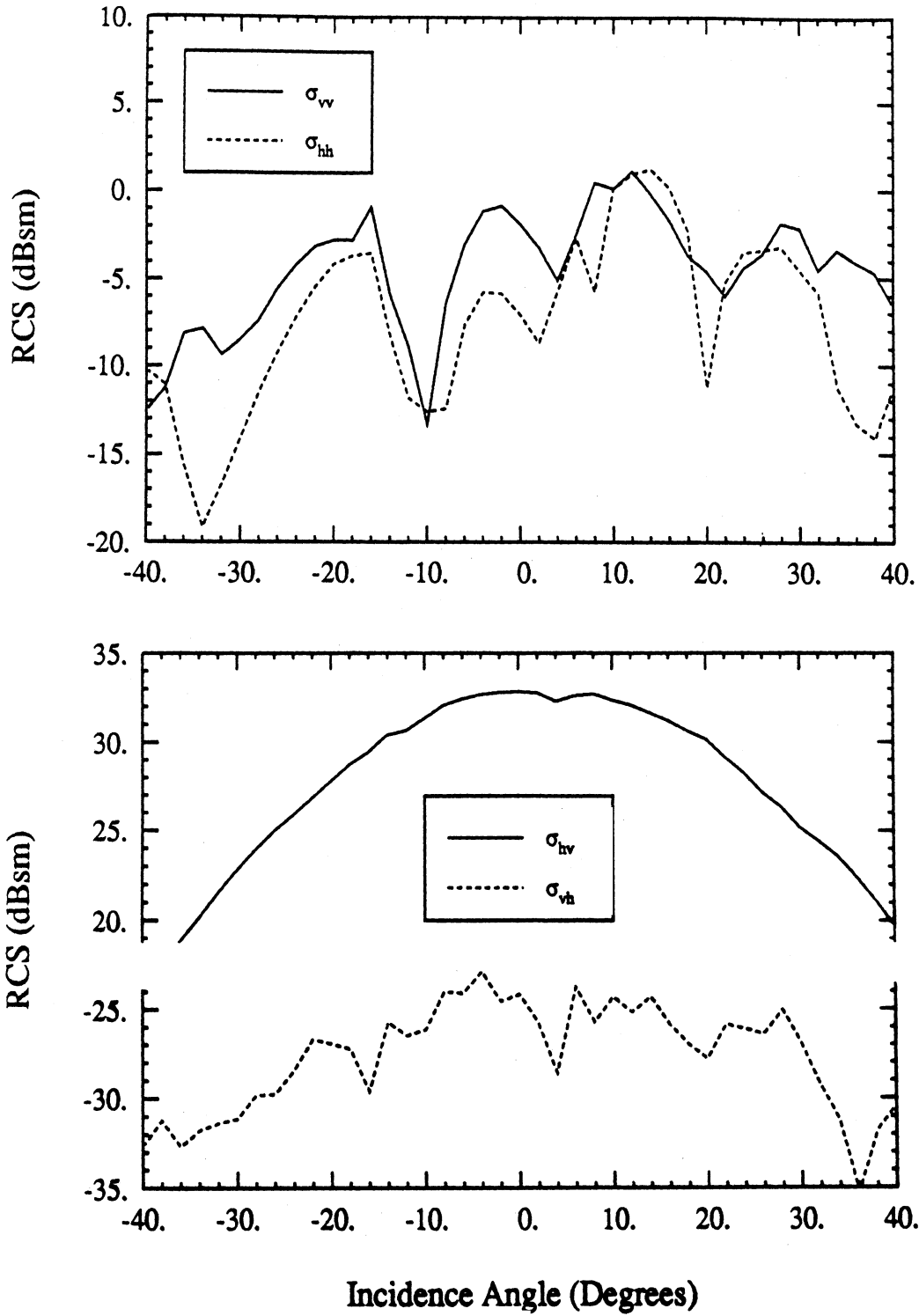


Figure 36: 45° pattern of amplitude of scattering matrix elements for C1 PARC.

135° Pattern (C1)

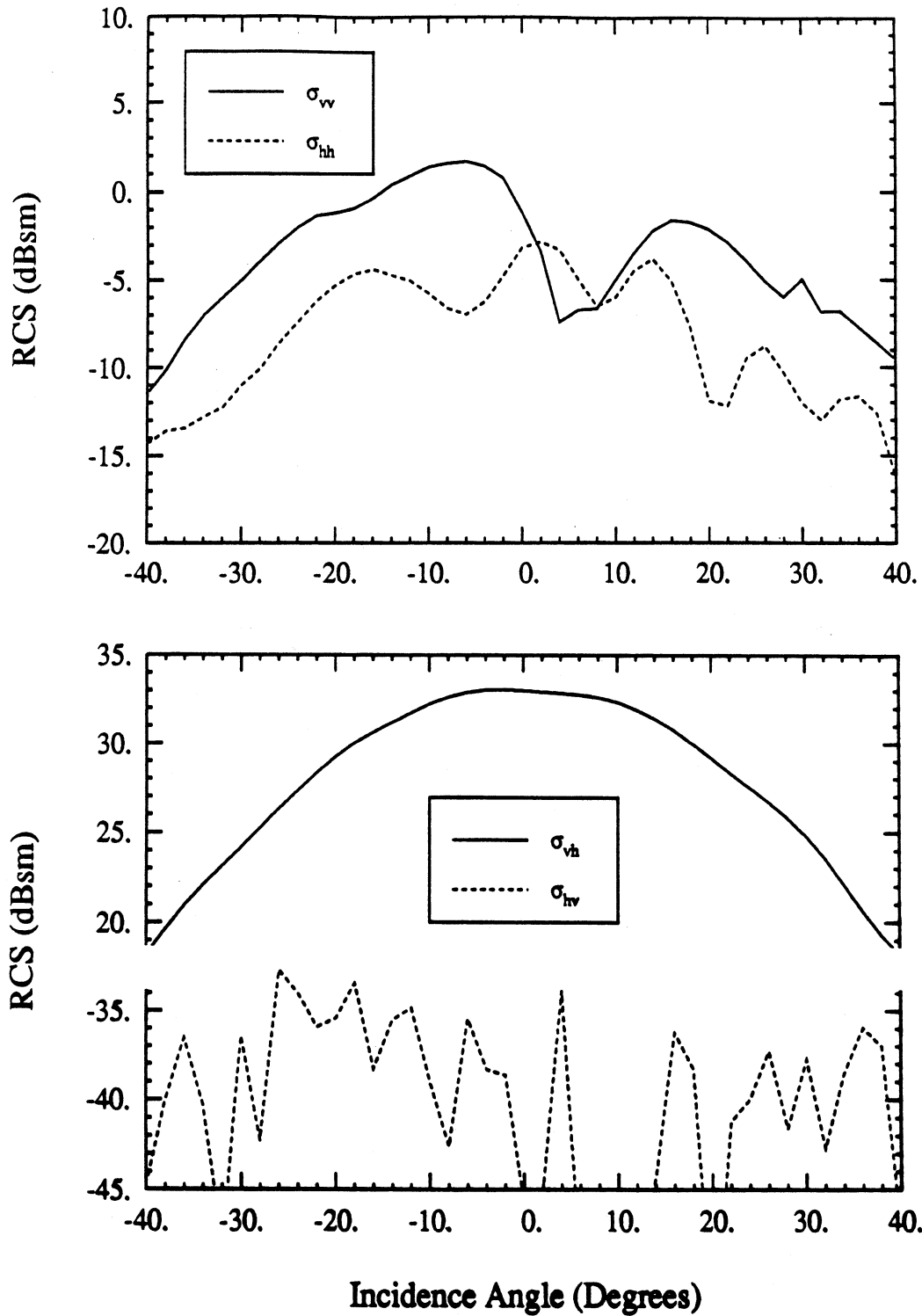


Figure 37: 135° pattern of amplitude of scattering matrix elements for C1 PARC.

Azimuth Pattern (C2)

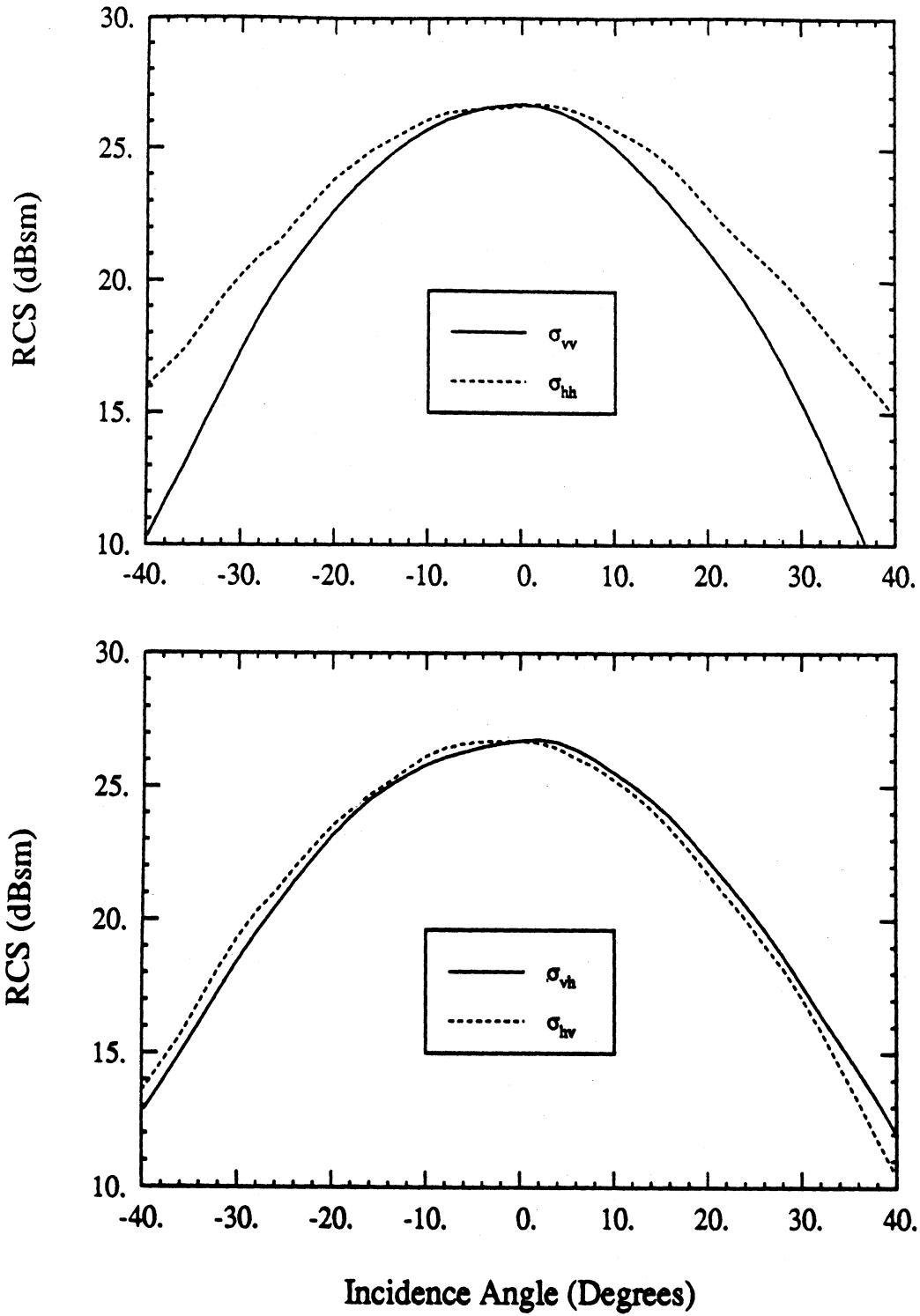


Figure 38: Azimuth pattern of amplitude of scattering matrix elements for C2 PARC.

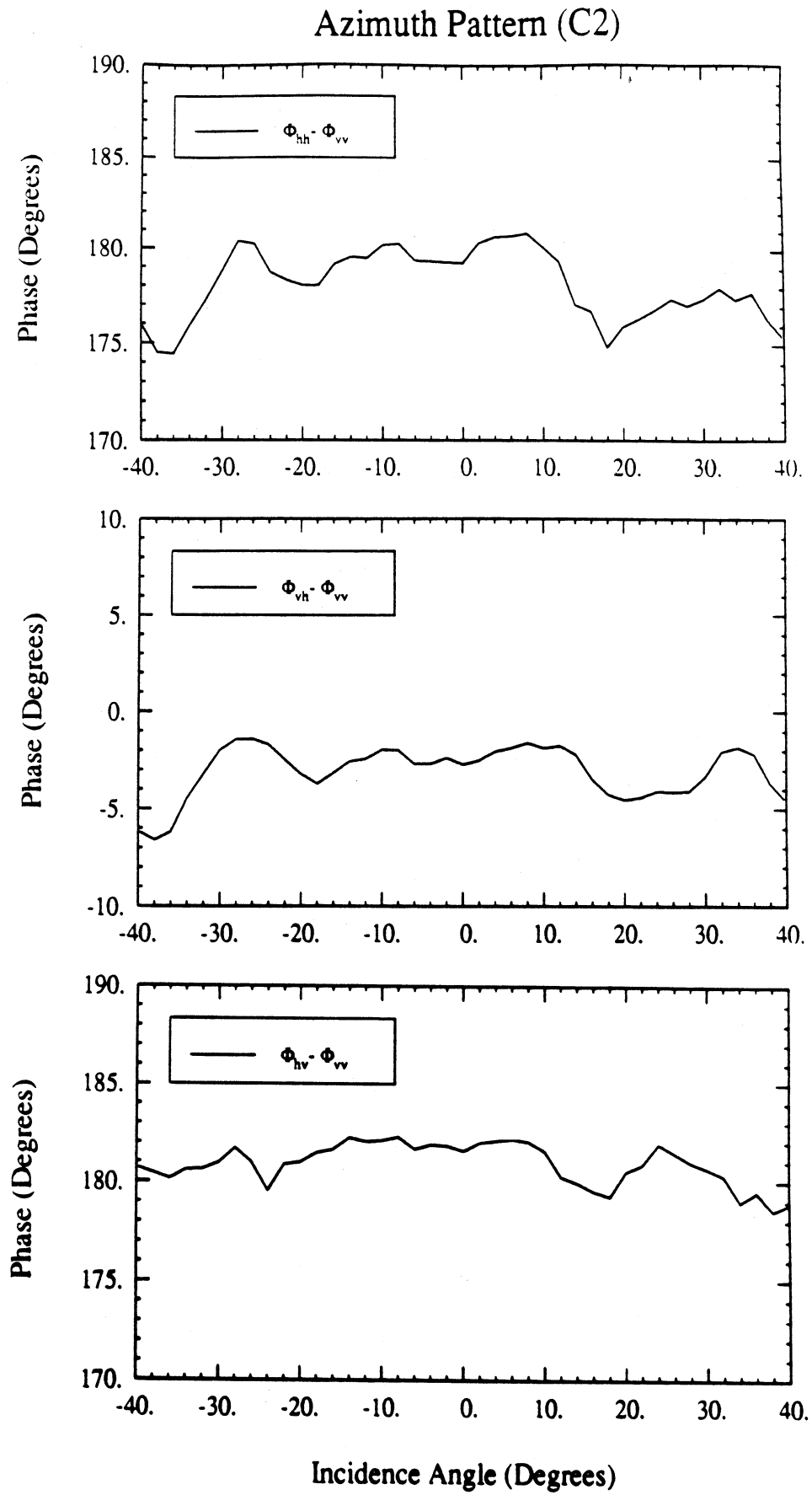


Figure 39: Azimuth pattern of phase of scattering matrix elements for C2 PARC.

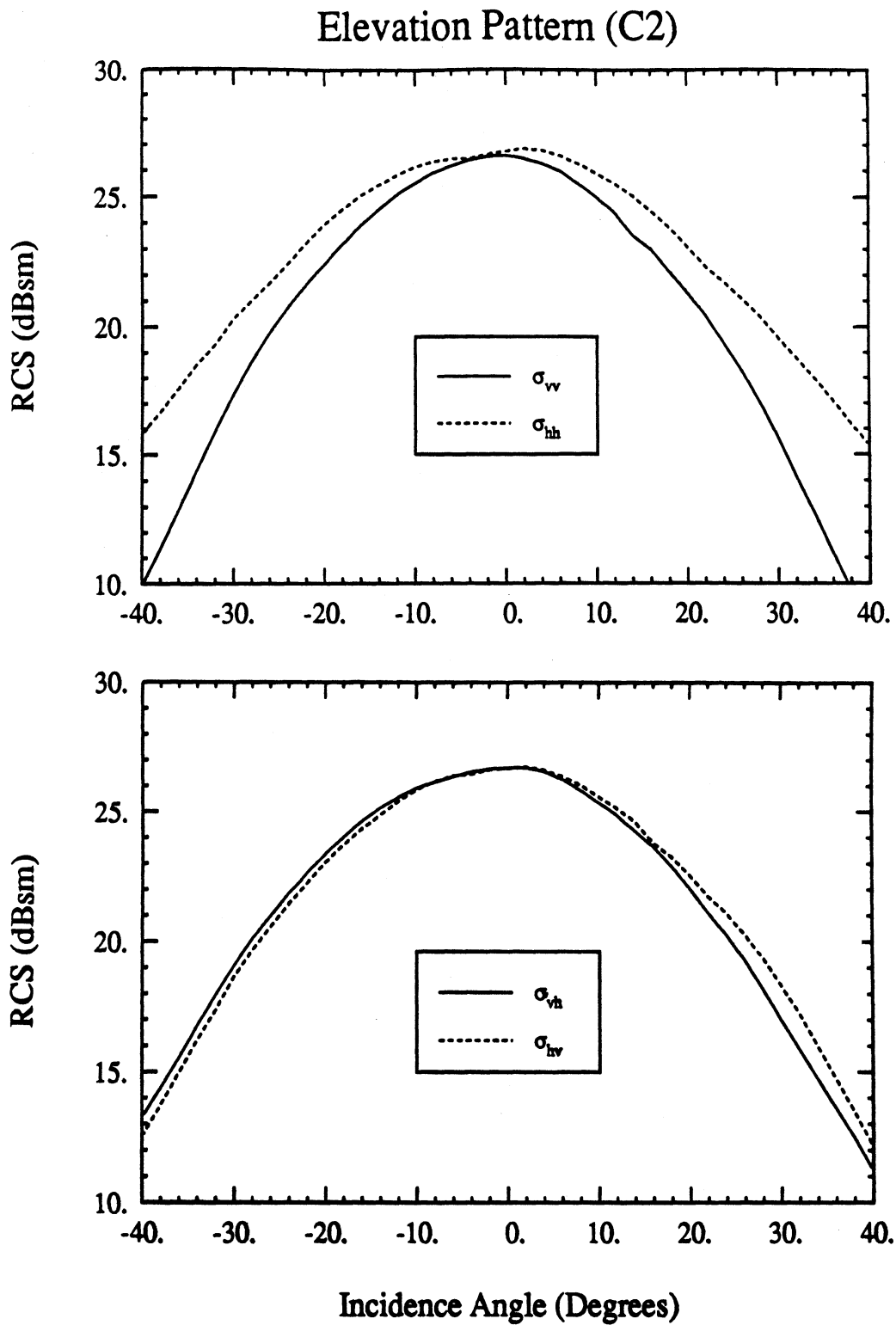


Figure 40: Elevation pattern of amplitude of scattering matrix elements for C2 PARC.

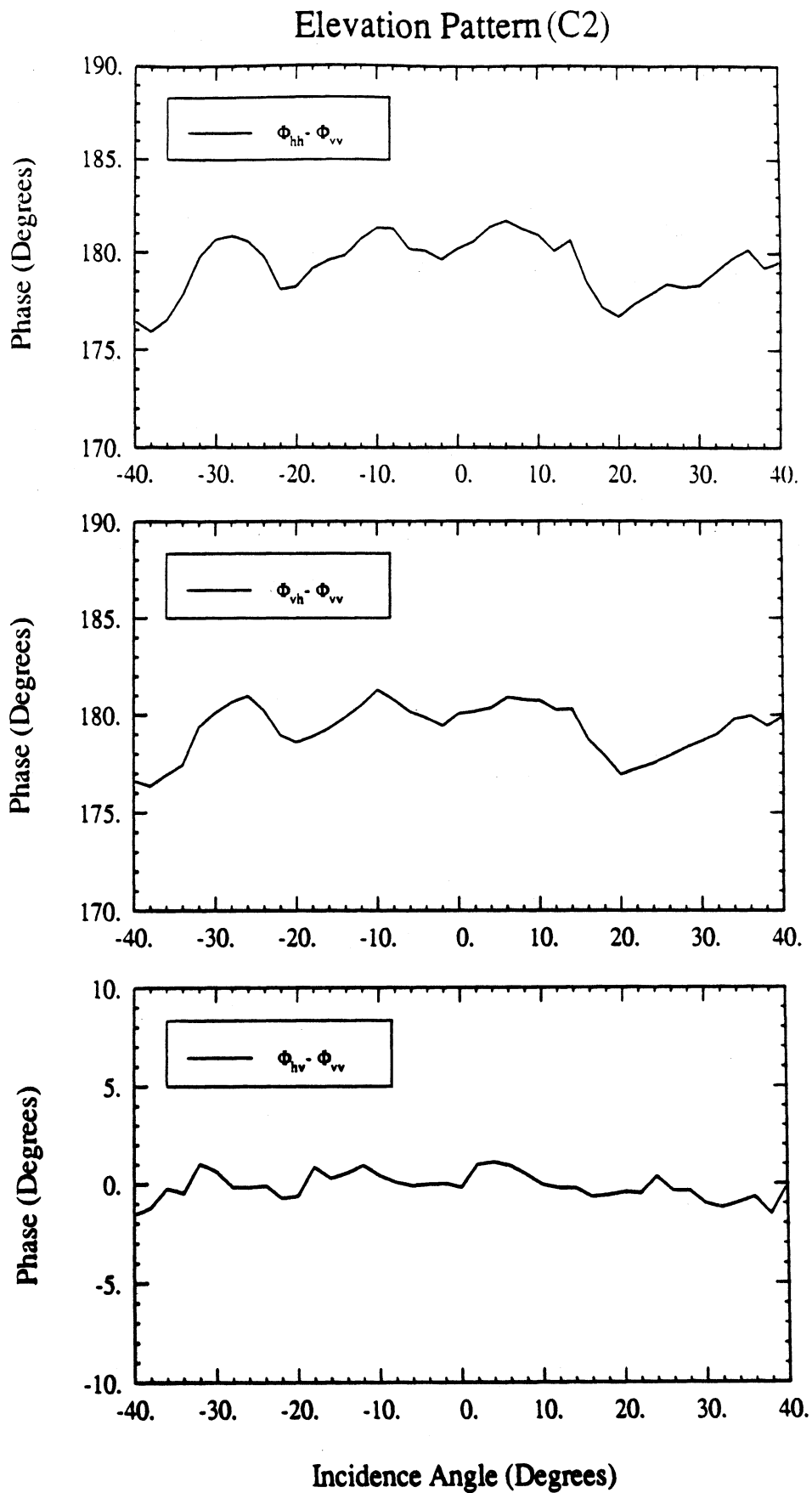


Figure 41: Elevation pattern of phase of scattering matrix elements for C2 PARC.

45° Pattern (C2)

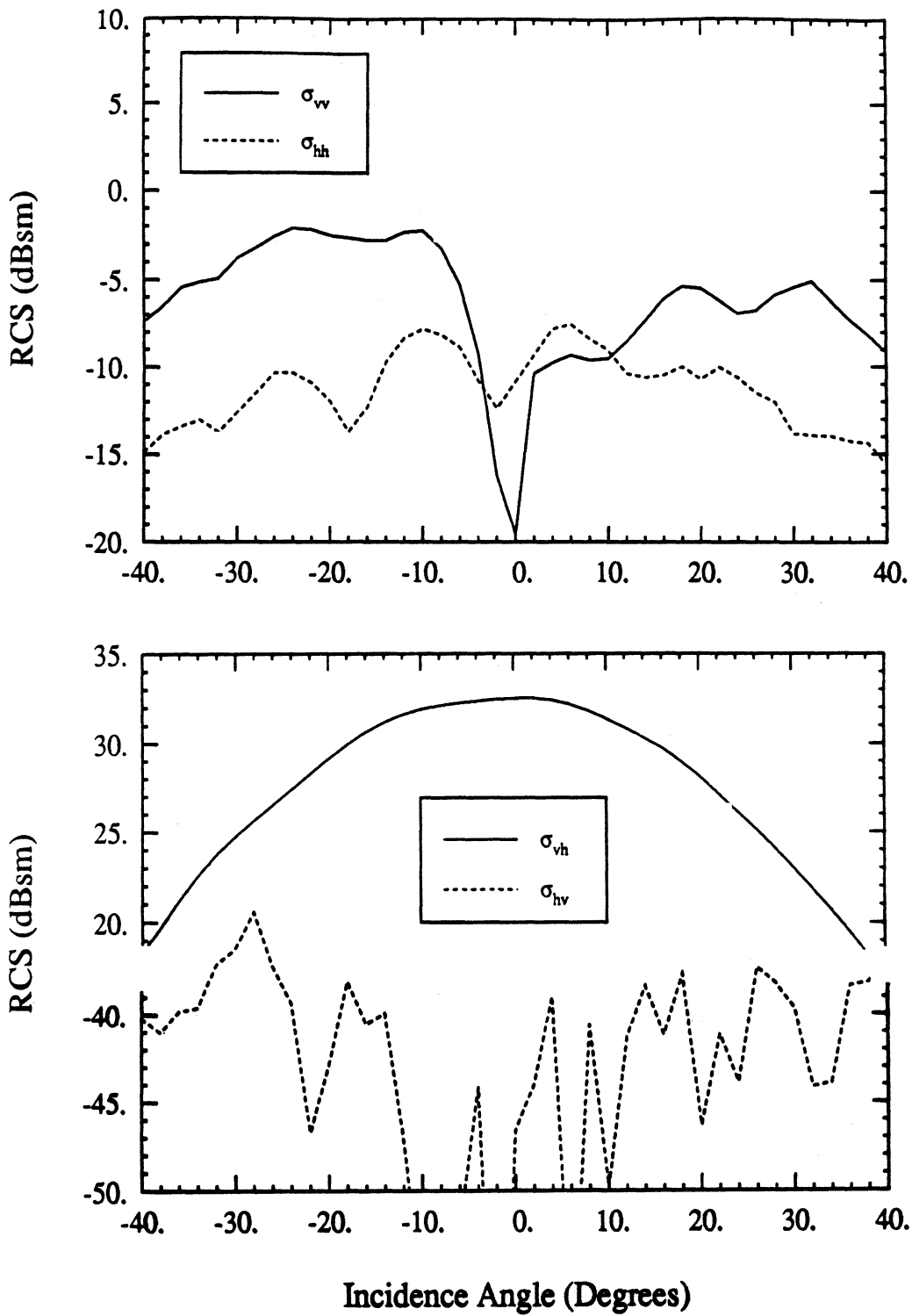


Figure 42: 45° pattern of amplitude of scattering matrix elements for C2 PARC.

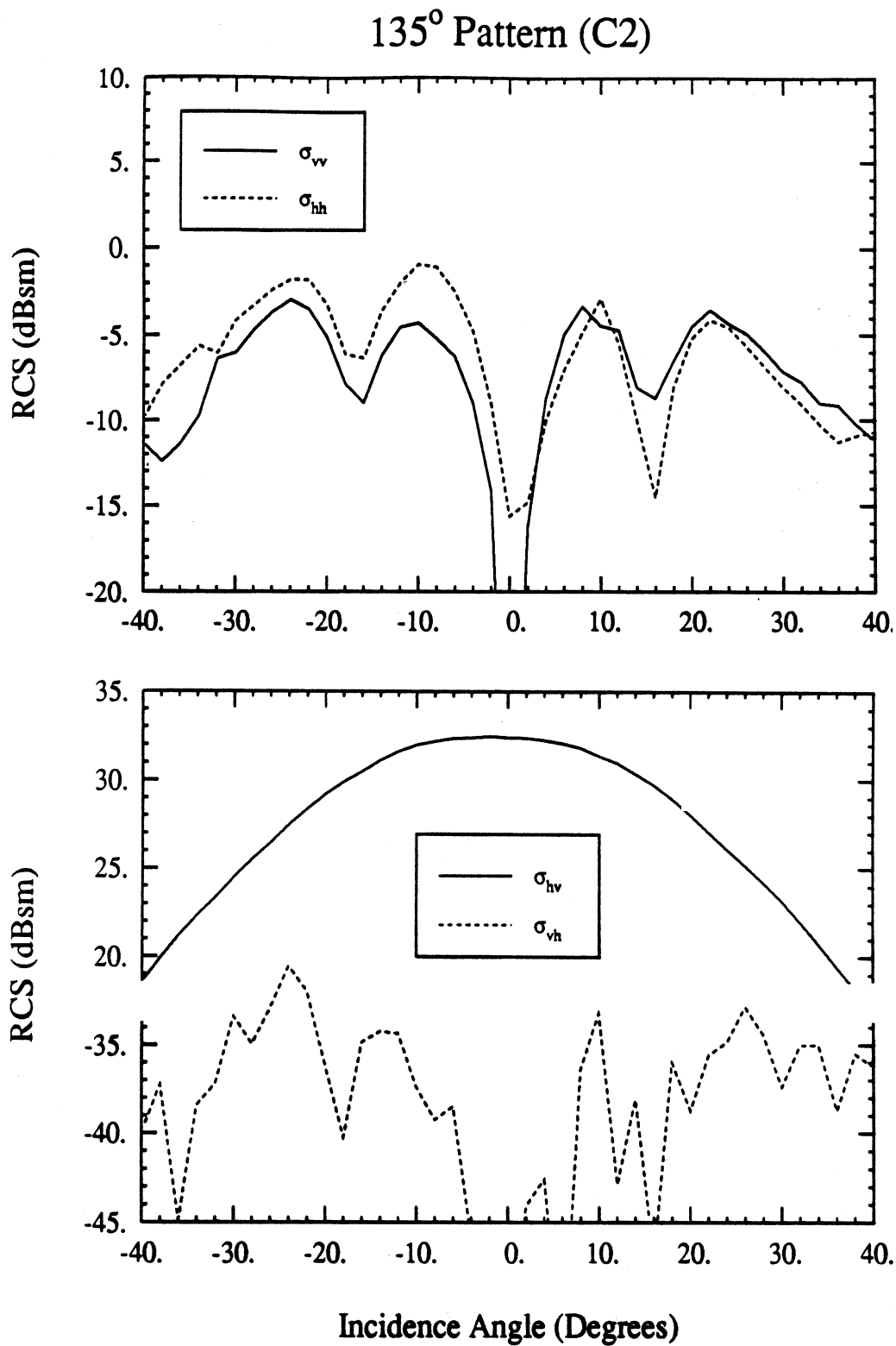


Figure 43: 135° pattern of amplitude of scattering matrix elements for C2 PARC.

Azimuth Pattern (C3)

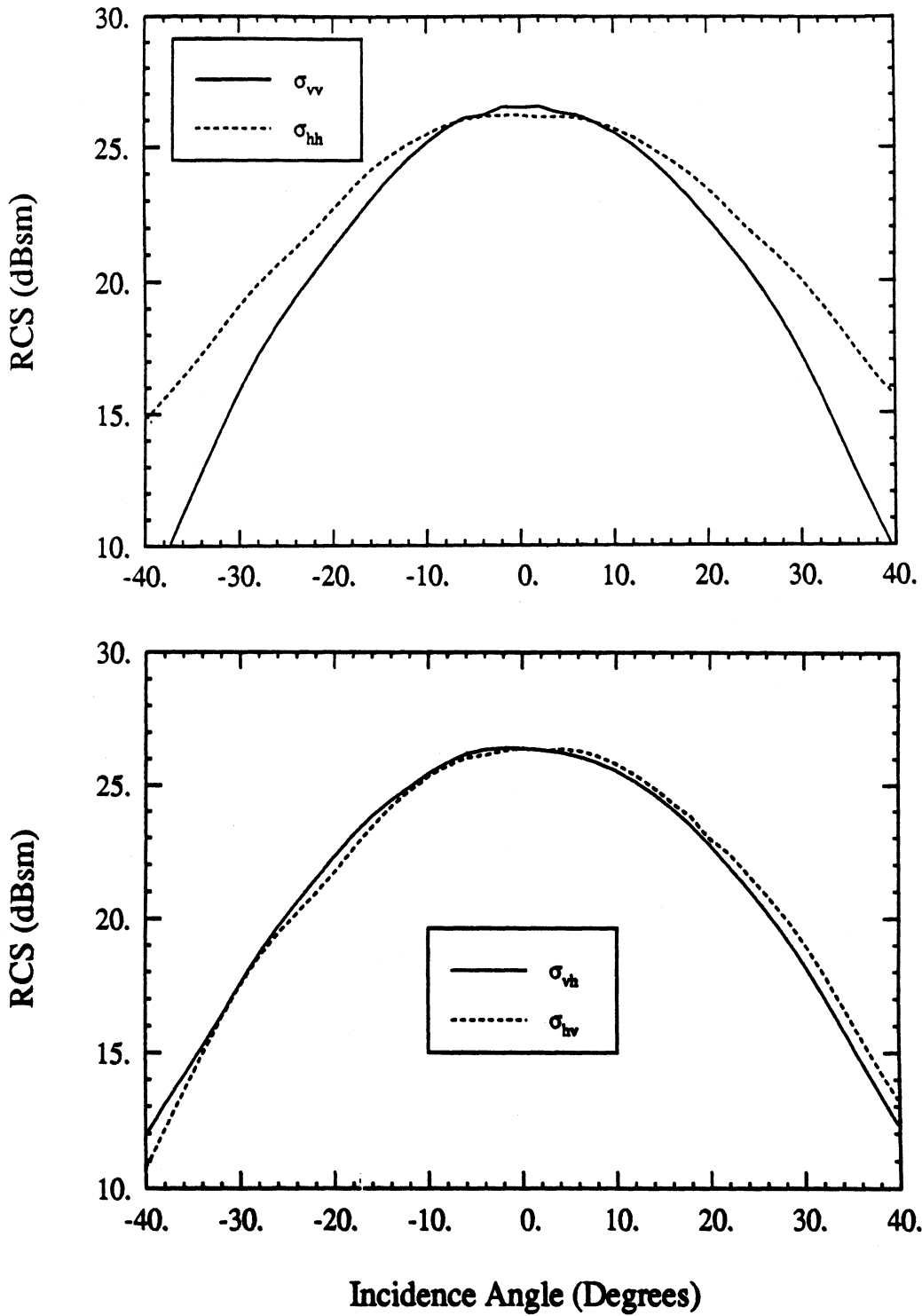


Figure 44: Azimuth pattern of amplitude of scattering matrix elements for C3 PARC.

Azimuth Pattern (C3)

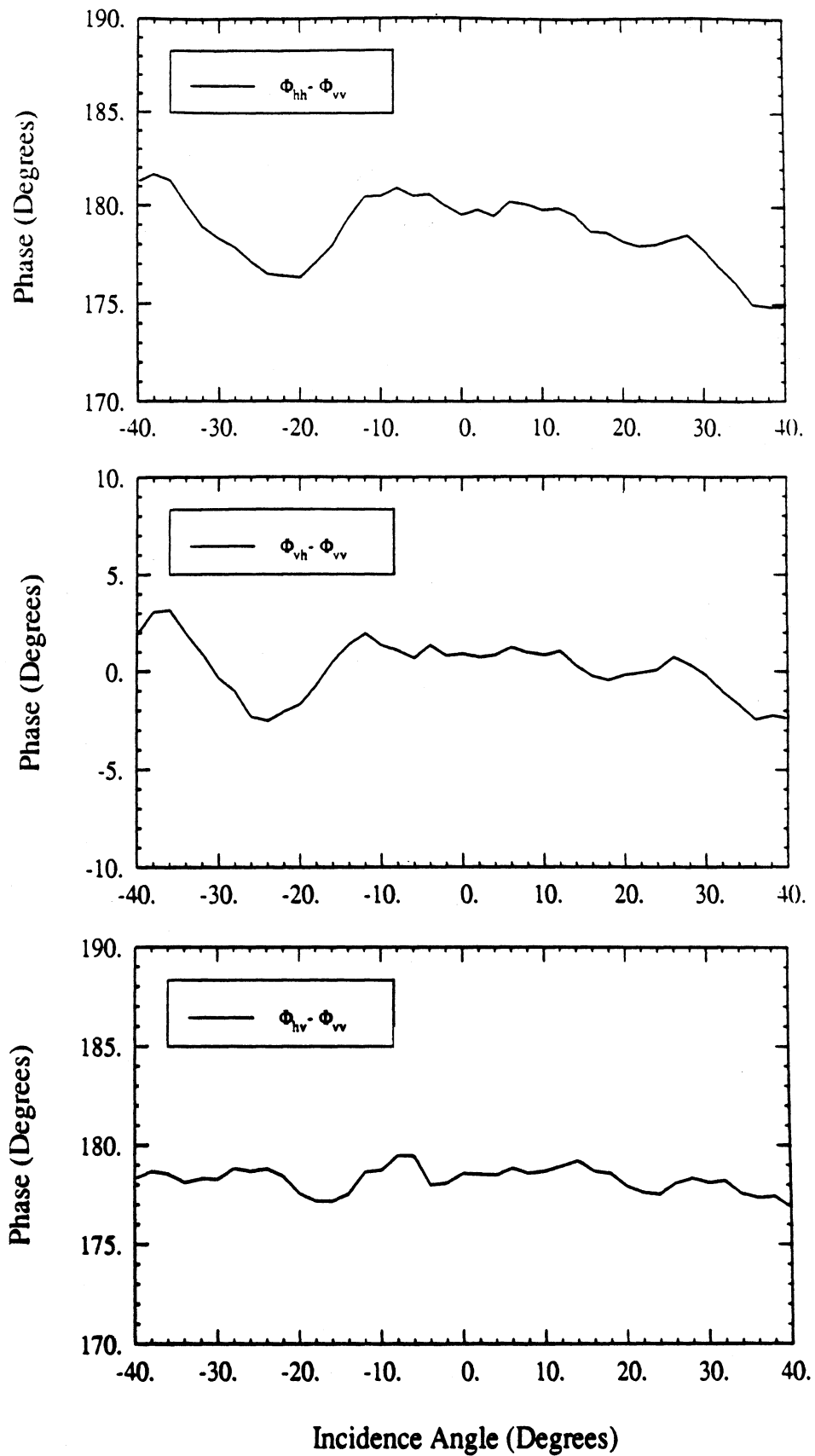


Figure 45: Azimuth pattern of phase of scattering matrix elements for C3 PARC.

Elevation Pattern (C3)

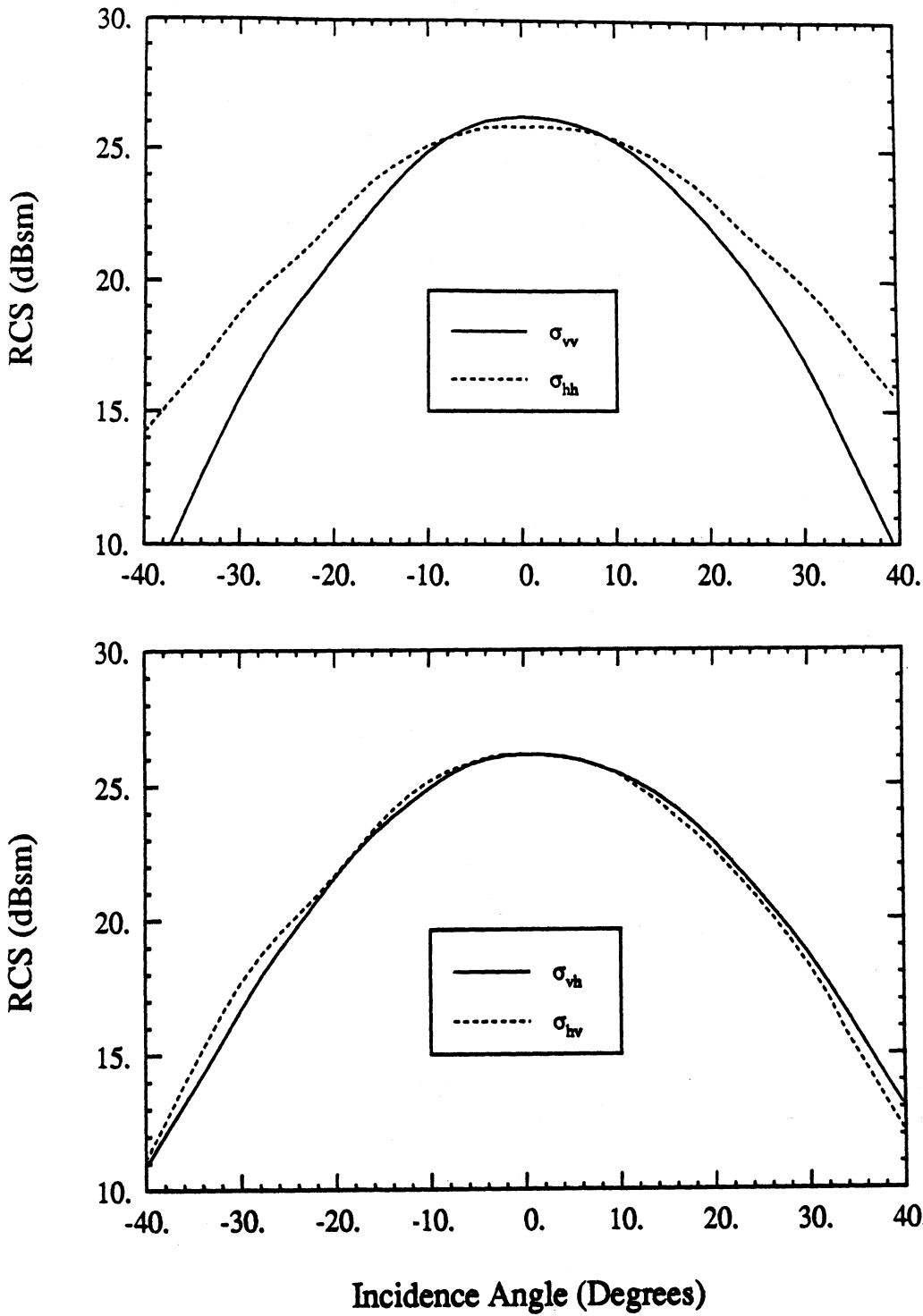


Figure 46: Elevation pattern of amplitude of scattering matrix elements for C3 PARC.

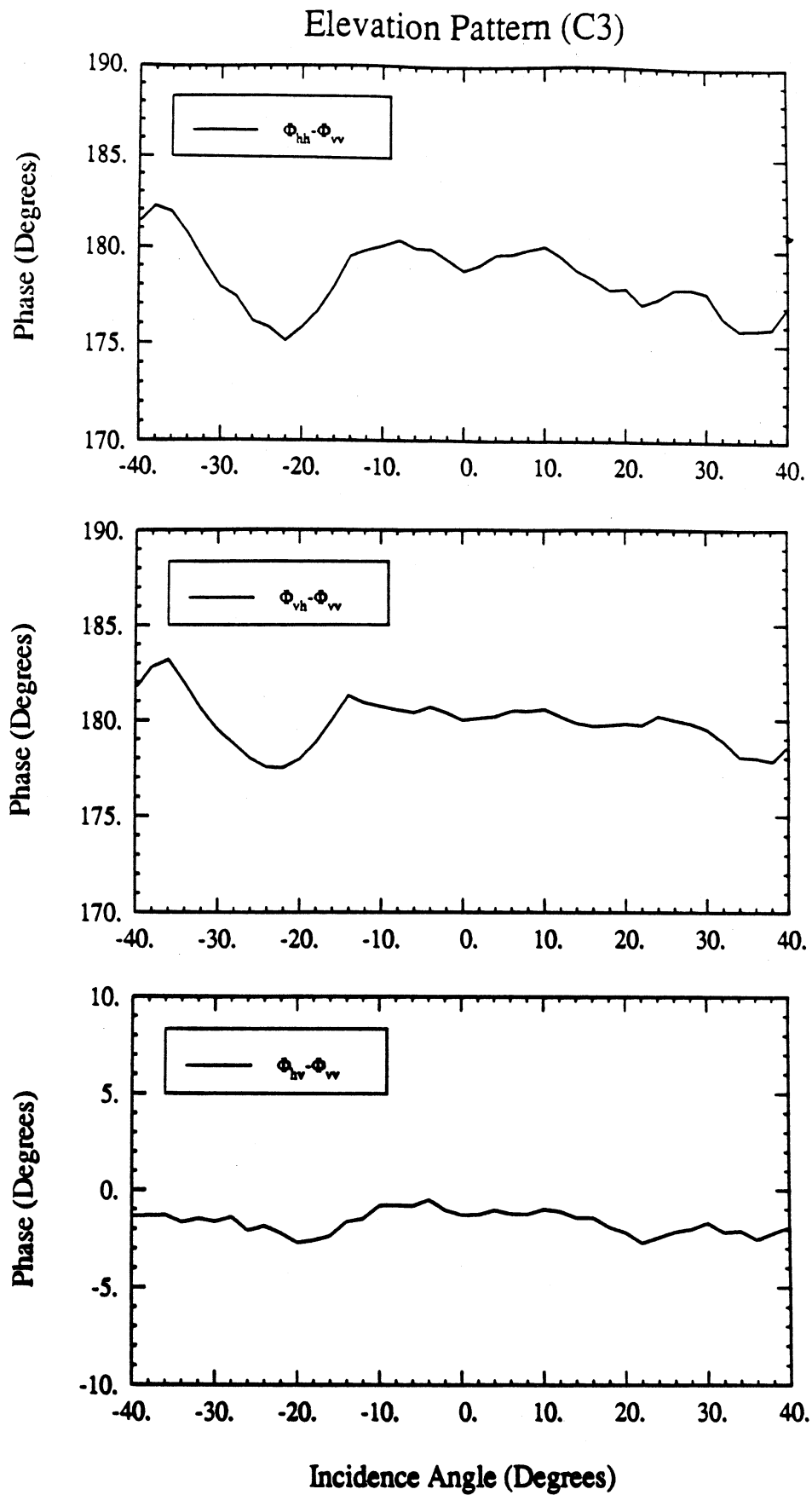


Figure 47: Elevation pattern of phase of scattering matrix elements for C3 PARC.

45° Pattern (C3)

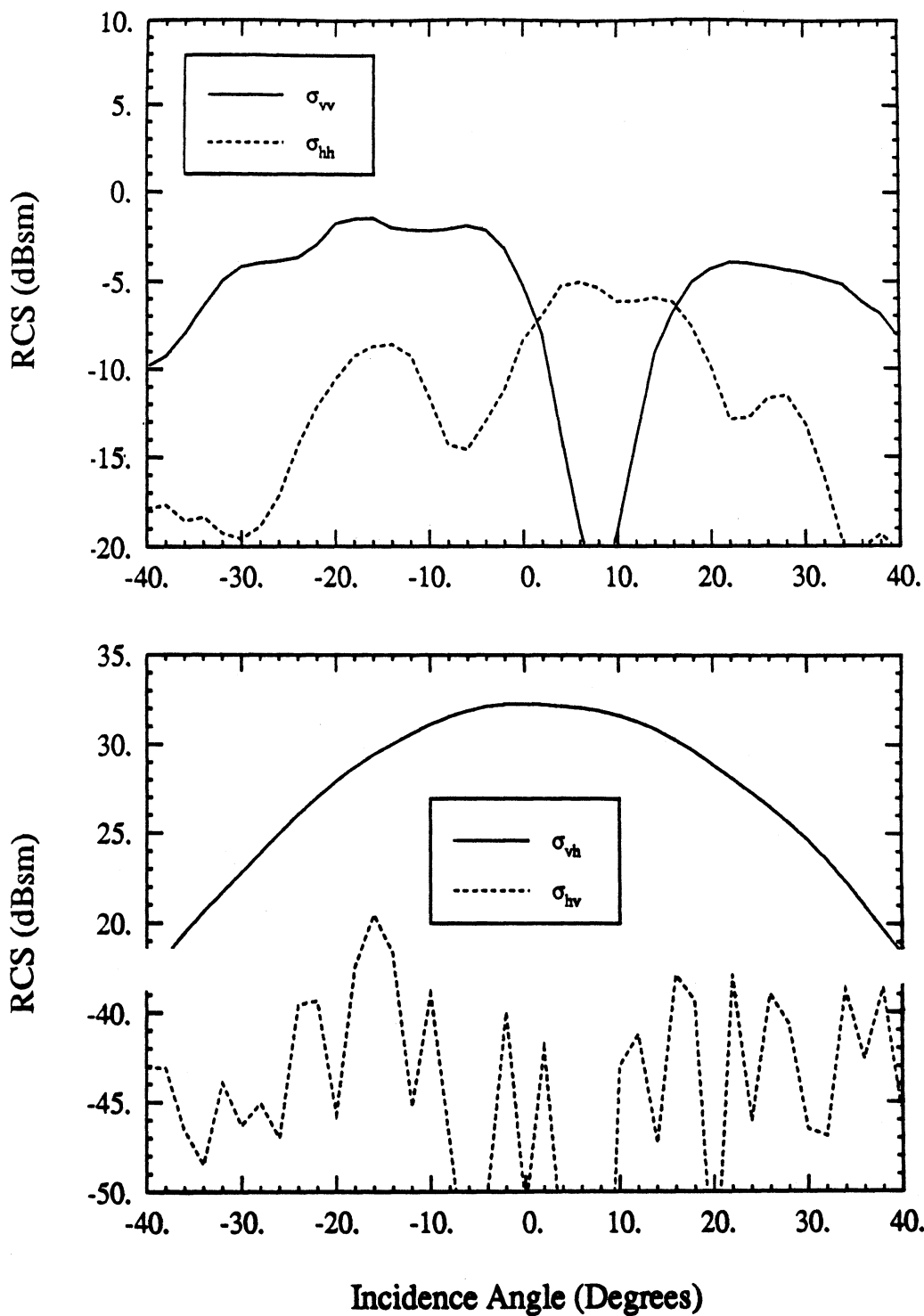


Figure 48: 45° pattern of amplitude of scattering matrix elements for C3 PARC.

135° Pattern (C3)

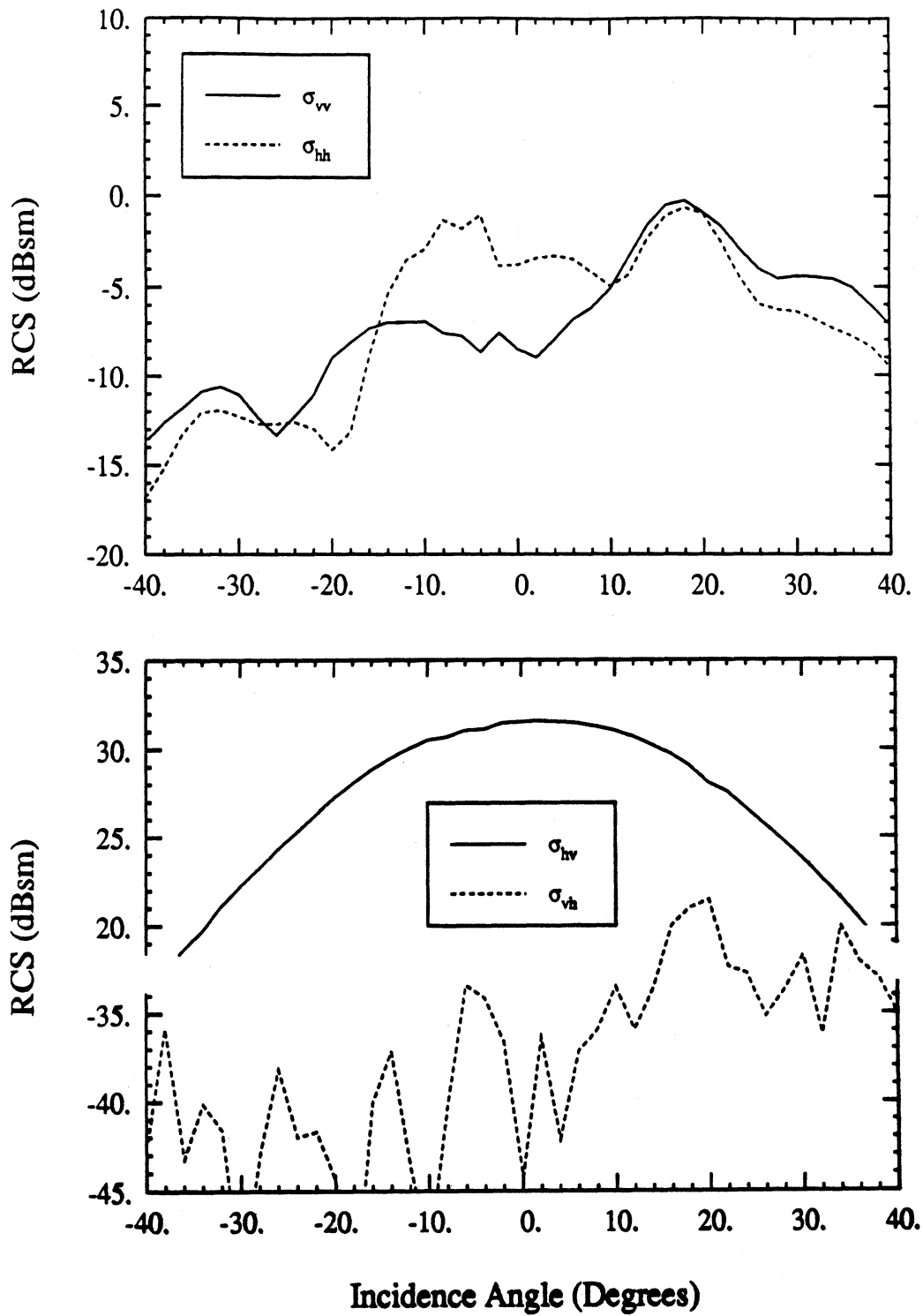


Figure 49: 135° pattern of amplitude of scattering matrix elements for C3 PARC.

APPENDIX A

A CONVENIENT TECHNIQUE FOR POLARIMETRIC CALIBRATION OF SINGLE-ANTENNA RADAR SYSTEMS

K. Sarabandi and F.T.Ulaby

Abstract- This paper introduces a practical technique for calibrating single-antenna polarimetric radar systems. With this technique, only a single calibration target, such as a conducting sphere or a trihedral corner reflector, is needed to calibrate the radar system, both in amplitude and phase, for all linear polarization configurations. By using a metal sphere, which is orientation independent, error in calibration measurement is minimized, while simultaneously calibrating the cross-polarization channels.

The antenna system and two orthogonal channels (in free space) are modeled as a four-port passive network. Upon using the reciprocity relations for the passive network, and assuming the cross-coupling terms of the antenna to be equal the cross-talk factors of the antenna system and the transmit and receive channel imbalances can be obtained from measurement of the backscatter from a metal sphere. For an X-band radar system with cross-polarization isolation of 25 dB, comparison of values measured for a sphere and a cylinder with theoretical values shows agreement within 0.4 dB in magnitude and 5° in phase. Also an effective polarization isolation of 50 dB is achieved using this calibration technique.

A1 Introduction

Accurate knowledge of the scattering matrix of a target is an important ingredient towards extracting biophysical information about the target. The scattering matrix of a target can be measured by using a set of orthogonal polarization. In practice, however, it is very difficult, if not impossible, to design an antenna system with perfect isolation between the orthogonal polarization channels, which leads to contamination of the measurements.

In recent years, considerable effort has been devoted to the development of techniques for calibrating polarimetric radar systems. Calibration techniques available in the literature can be categorized into two major groups: 1) calibration techniques for imaging radars, and 2) calibration techniques for point-target measurement systems, which may also be appropriate for imaging radars. In the first group, the scattering properties of clutter are usually employed to simplify the calibration problem [Sheen and Kasischke, 1989]. van Zyl [1989] and Klein [1989] developed a method for estimating the cross-talk contamination of the antenna by assuming that the like- and cross-polarized responses of natural targets with azimuthal symmetry are uncorrelated. Among the point-target calibration techniques, the generalized calibration technique (GCT) by Whitt and Ulaby [1989] characterizes the distortion matrices (channel imbalances and antenna cross-talk) of the receive and transmit antenna by using three calibration targets. An eigenvalue approach is employed to solve for the distortion matrices. In a similar technique by Barnes [1986] the distortion matrices are obtained by using targets with specific scattering

matrices. This technique is referred to by Ulaby et al. [1989] as the constrained calibration technique (CCT). Although, in principle, GCT and CCT can fully characterize the distortion matrices, they are very sensitive to target alignment and to the knowledge of the theoretical values of the scattering matrices of the calibration targets. A third calibration technique for point targets by Sarabandi, et al. [1989] uses a sphere and any other depolarizing calibration target (scattering matrix of this target need not to be known), and is therefore immune to errors caused by target orientation and lack of precise knowledge of the theoretical values of the calibration targets' scattering matrices. However, the drawback of this method, which is called the isolated-antenna calibration technique (IACT), is that it does not account for cross-talk contamination in the antenna. The isolated antenna assumption can lead to significant errors in the cross-polarized terms when the ratio of cross- to like-polarized terms is small and/or cross-talk contamination is large.

To remove the drawback of the IACT while maintaining insensitivity to orientation of the calibration targets, we introduce in this paper a technique for calibrating single-antenna radar systems using a four-port network approach. The antenna system and two orthogonal directions in free space are modeled as a four-port network, and channel imbalances as well as the antenna cross-talk contamination are determined by measuring the backscatter from a single calibration target, namely a conducting sphere. This technique will henceforth be referred to as STCT, or single-target calibration technique. Like IACT, STCT is is insensitive to target orientation, but it also accounts for the antenna cross-talk contamination. If the

antenna cross-talk contamination is very small (≈ 0), the STCT is not appropriate and the IACT should be used instead.

The validity and accuracy of this technique were tested using X-band and L-band scatterometers, both in an anechoic chamber and under field conditions. Cylinders and spheres are used as test targets. Excellent agreement was obtained between the measured and theoretical values of the test target. Also the effective cross-polarization isolation obtained in this method was in the order of 50 dB.

A2 Theoretical Formulation

By defining a set of orthogonal directions in a plane perpendicular to the direction of propagation, the field components of the wave scattered by a given point target can be related to the components of the incident plane wave through the scattering matrix of the target \mathbf{s} . The antenna structure of a polarimetric radar system must be designed in such a way that the transmit and receive polarizations are parallel to the specified orthogonal directions. In practice, however, it is not possible to construct antennas that are totally free of polarization contaminations, i.e., coupling between the orthogonal polarization ports of the antenna. Polarization contamination (antenna cross-talk) takes place in the orthogonal mode transducer (OMT) and in the antenna structure itself.

Suppose the two orthogonal directions in free space are viewed as two ports of a four-port passive device that includes the OMT and the antenna structure (see Fig. 1). This four-port network can be characterized by a scattering matrix \mathcal{S}

which relates the incident wave vector V^+ to the reflected wave vector V^- ,

$$V^- = SV^+$$

where

$$S = \begin{bmatrix} S_{11} & S_{12} & S_{13} & S_{14} \\ S_{21} & S_{22} & S_{23} & S_{24} \\ S_{31} & S_{32} & S_{33} & S_{34} \\ S_{41} & S_{42} & S_{43} & S_{44} \end{bmatrix} \quad (A1)$$

Since the four-port device is passive its scattering matrix must be symmetric.

Thus,

$$S_{ij} = S_{ji} \quad i, j \in \{1, 2, 3, 4\}$$

If the reference plane of the n^{th} port is translated outward by distance l_n , the new scattering matrix for the device becomes [Collins, pp. 172, 1966]

$$S' = \Theta S \Theta \quad (A2)$$

where the translation matrix Θ is given by

$$\Theta = \begin{bmatrix} e^{-i\beta_1 l_1} & 0 & 0 & 0 \\ 0 & e^{-i\beta_2 l_2} & 0 & 0 \\ 0 & 0 & e^{-i\beta_3 l_3} & 0 \\ 0 & 0 & 0 & e^{-i\beta_4 l_4} \end{bmatrix} \quad (A3)$$

In (3) β_n is the propagation constant of the n^{th} port transmission line. In this case, since ports 3 and 4 are two ports in free space, $\beta_3 = \beta_4 = k_0$ and the translation

matrix must be modified to account for spherical propagation. If the target is located at a distance r from the radar system and the reference planes at ports 3 and 4 are translated to the target location, then the translation matrix becomes

$$\Theta = \begin{bmatrix} 1 & 0 & 0 & 0 \\ 0 & 1 & 0 & 0 \\ 0 & 0 & \frac{e^{-ik_0 r}}{r} & 0 \\ 0 & 0 & 0 & \frac{e^{-ik_0 r}}{r} \end{bmatrix}$$

The scattering matrix, after translation of the reference planes of ports 3 and 4 by distance r , takes the following form

$$\mathcal{S}' = \begin{bmatrix} \mathcal{S}_{11} & \mathcal{S}_{12} & \mathcal{S}_{13} \frac{e^{-ik_0 r}}{r} & \mathcal{S}_{14} \frac{e^{-ik_0 r}}{r} \\ \mathcal{S}_{21} & \mathcal{S}_{22} & \mathcal{S}_{23} \frac{e^{-ik_0 r}}{r} & \mathcal{S}_{24} \frac{e^{-ik_0 r}}{r} \\ \mathcal{S}_{31} \frac{e^{-ik_0 r}}{r} & \mathcal{S}_{32} \frac{e^{-ik_0 r}}{r} & \mathcal{S}_{33} \frac{e^{-2ik_0 r}}{r^2} & \mathcal{S}_{34} \frac{e^{-2ik_0 r}}{r^2} \\ \mathcal{S}_{41} \frac{e^{-ik_0 r}}{r} & \mathcal{S}_{42} \frac{e^{-ik_0 r}}{r} & \mathcal{S}_{43} \frac{e^{-2ik_0 r}}{r^2} & \mathcal{S}_{44} \frac{e^{-2ik_0 r}}{r^2} \end{bmatrix} \quad (\text{A4})$$

Note that here we have ignored the gain and effective areas of the transmit and receive antennas which will be included in the channel imbalances. The signal flow-chart of the antenna system and free space ports is shown in Fig. 2. Suppose the radar is equipped with a space discriminating filter (range gating filter) which is tuned at r . The filtered scattering matrix (\mathcal{S}'') is then given by

$$\mathcal{S}'' = \begin{bmatrix} 0 & 0 & \mathcal{S}_{13} & \mathcal{S}_{14} \\ 0 & 0 & \mathcal{S}_{23} & \mathcal{S}_{24} \\ \mathcal{S}_{31} & \mathcal{S}_{32} & 0 & 0 \\ \mathcal{S}_{41} & \mathcal{S}_{42} & 0 & 0 \end{bmatrix} \frac{e^{-ik_0 r}}{r}$$

Basically, short-range reflections from the antenna system and multiple bounces between the antenna and the target have been gated out. The incident and reflected waves at each port can now be represented by two uncoupled matrix equations as follows

$$\begin{bmatrix} V_1^- \\ V_2^- \end{bmatrix} = \frac{e^{-ik_0 r}}{r} \begin{bmatrix} S_{13} & S_{14} \\ S_{23} & S_{24} \end{bmatrix} \begin{bmatrix} V_3^+ \\ V_4^+ \end{bmatrix} \quad (\text{A5})$$

$$\begin{bmatrix} V_3^- \\ V_4^- \end{bmatrix} = \frac{e^{-ik_0 r}}{r} \begin{bmatrix} S_{31} & S_{32} \\ S_{41} & S_{42} \end{bmatrix} \begin{bmatrix} V_1^+ \\ V_2^+ \end{bmatrix} \quad (\text{A6})$$

On the other hand, the incident and reflected waves at ports 3 and 4 are the scattered and incident waves, respectively, of the target, which is represented by a two-port network and thus related to each other by the scattering matrix of the target s . That is,

$$\begin{bmatrix} V_3^+ \\ V_4^+ \end{bmatrix} = \begin{bmatrix} s_{vv} & s_{vh} \\ s_{hv} & s_{hh} \end{bmatrix} \begin{bmatrix} V_3^- \\ V_4^- \end{bmatrix} \quad (\text{A7})$$

Note that the scattering matrix used here is defined in the backscattering alignment convention since the orthogonal directions are specified independent of the incident and scattering directions. After rearranging (5)-(7) in order to relate the reflected waves to the incident waves at ports 1 and 2, in addition to employing the reciprocity property of the four-port network, we get

$$\begin{bmatrix} V_1^- \\ V_2^- \end{bmatrix} = \frac{e^{-2ik_0 r}}{r^2} \begin{bmatrix} S_{13} & S_{14} \\ S_{23} & S_{24} \end{bmatrix} \begin{bmatrix} s_{vv} & s_{vh} \\ s_{hv} & s_{hh} \end{bmatrix} \begin{bmatrix} S_{13} & S_{23} \\ S_{14} & S_{24} \end{bmatrix} \begin{bmatrix} V_1^+ \\ V_2^+ \end{bmatrix} \quad (\text{A8})$$

Upon normalizing with respect to the like-polarized channels (8) becomes

$$\begin{bmatrix} V_1^- \\ V_2^- \end{bmatrix} = \frac{e^{-2ik_0r}}{r^2} \begin{bmatrix} \mathcal{S}_{13} & 0 \\ 0 & \mathcal{S}_{24} \end{bmatrix} \begin{bmatrix} 1 & C_1 \\ C_2 & 1 \end{bmatrix} \begin{bmatrix} s_{vv} & s_{vh} \\ s_{hv} & s_{hh} \end{bmatrix} \begin{bmatrix} 1 & C_2 \\ C_1 & 1 \end{bmatrix} \begin{bmatrix} \mathcal{S}_{13} & 0 \\ 0 & \mathcal{S}_{24} \end{bmatrix} \begin{bmatrix} V_1^+ \\ V_2^+ \end{bmatrix} \quad (\text{A9})$$

where $C_1 = \mathcal{S}_{14}/\mathcal{S}_{13}$ and $C_2 = \mathcal{S}_{23}/\mathcal{S}_{24}$ are the antenna cross-talk factors.

So far we have modeled the antenna system, free space channel, and the target by a two-port network. To account for the effects of active circuits on the performance of the overall radar system, let us consider the block diagram depicted in Fig. 3. The transmit (V_{tv} , V_{th}) and receive (V_{rv} , V_{rh}) voltages are the quantities measured by the radar. The channel imbalance quantities (T_v , T_h , R_v , R_h), which relate the transmit/receive voltages to the incident/reflected waves at port 1 and 2, account for variations (in both amplitude and phase) of the active circuits and the antenna gains. The transmit and receive channels of the radar system are separated by a transmit-receive switch (TR switch) or a circulator. These components can be assumed ideal because any leakage that may occur will not be sampled by the range gating process. Therefore, the transmit and receive voltages can be related to the incident and reflected voltages of ports 1 and 2 by

$$\begin{bmatrix} V_{rv} \\ V_{rh} \end{bmatrix} = \begin{bmatrix} R_v & 0 \\ 0 & R_h \end{bmatrix} \begin{bmatrix} V_1^- \\ V_2^- \end{bmatrix} \quad (\text{A10})$$

$$\begin{bmatrix} V_1^+ \\ V_2^+ \end{bmatrix} = \begin{bmatrix} T_v & 0 \\ 0 & T_h \end{bmatrix} \begin{bmatrix} V_{tv} \\ V_{th} \end{bmatrix} \quad (\text{A11})$$

Using (9) in (10) and (11) results in

$$\begin{bmatrix} V_{rv} \\ V_{rh} \end{bmatrix} = \frac{e^{-2ik_0r}}{r^2} \begin{bmatrix} R_v \mathcal{S}_{13} & 0 \\ 0 & R_h \mathcal{S}_{24} \end{bmatrix} \begin{bmatrix} 1 & C_1 \\ C_2 & 1 \end{bmatrix} \begin{bmatrix} s_{vv} & s_{vh} \\ s_{hv} & s_{hh} \end{bmatrix} \begin{bmatrix} 1 & C_2 \\ C_1 & 1 \end{bmatrix} \begin{bmatrix} T_v \mathcal{S}_{13} & 0 \\ 0 & T_h \mathcal{S}_{24} \end{bmatrix} \begin{bmatrix} V_{tv} \\ V_{th} \end{bmatrix} \quad (\text{A12})$$

which may be written in matrix notation as

$$\mathbf{V}_r = \frac{e^{-2ik_0r}}{r^2} \mathbf{RCsC}^T \mathbf{TV}_t \quad (\text{A13})$$

The matrix $\mathbf{M} = \mathbf{RCsC}^T \mathbf{T}$ represents the measured (uncalibrated) scattering matrix of the target under observation. If the matrices \mathbf{C} , \mathbf{R} , and \mathbf{T} are known, the actual scattering matrix \mathbf{s} can then be obtained. To determine \mathbf{C} , \mathbf{R} , and \mathbf{T} we note that these matrices depend on the choice of the orthogonal channels in free space. So far we have made no assumption on the direction of the orthogonal channels (v and h) except that they are perpendicular to the direction of propagation. Once the v and h directions are specified, the scattering matrix of the target can, in principle, be determined. A radar system with linear polarization configurations usually is oriented such that for a given polarization most of the transmitted energy falls into the desired channel, i.e. an orientation for which the antennas cross-talk factors (C_1 and C_2) are minimal. With available design techniques, it is easy to achieve the conditions $|C_1| \leq 0.1$ and $|C_2| \leq 0.1$, which correspond to a polarization isolation level of 20 dB, but achieving much greater isolation level is difficult. For accurate polarimetric measurements, the effective isolation level

should be on the order of 40 dB. Hence the factors C_1 and C_2 may not be ignored, but should instead be determined by the calibration technique.

To demonstrate how the choice of the coordinate frame affects the antennas' cross-talk factors, we obtain a relationship for the cross-talk factors when the coordinate frame is rotated by an angle ψ . Suppose \mathbf{s} represents the scattering matrix of a target for a particular coordinate frame and \mathbf{s}' denotes the scattering matrix of the same target when the coordinate frame is rotated around the incidence direction by an angle ψ . It is a trivial matter to show that

$$\mathbf{s}' = \begin{bmatrix} \cos \psi & -\sin \psi \\ \sin \psi & \cos \psi \end{bmatrix} \mathbf{s} \begin{bmatrix} \cos \psi & \sin \psi \\ -\sin \psi & \cos \psi \end{bmatrix} \quad (\text{A14})$$

Let us indicate the imbalance and cross-talk matrices in the rotated coordinate system by a prime sign. Since the relative orientation of the antenna system and target have not been changed, the measured scattering matrices for two coordinate systems are identical. That is, $\mathbf{M}' = \mathbf{M}$. Therefore,

$$\mathbf{M} = \mathbf{R}\mathbf{C}\mathbf{s}\mathbf{C}^T\mathbf{T} = \mathbf{R}'\mathbf{C}'\mathbf{s}'\mathbf{C}'^T\mathbf{T}' \quad (\text{A15})$$

Using (14) in (15) results in

$$\mathbf{M} = \mathbf{R}'\mathbf{P}\mathbf{s}\mathbf{P}^T\mathbf{T}' \quad (\text{A16})$$

where

$$\mathbf{P} = \begin{bmatrix} \cos \psi + C'_1 \sin \psi & -\sin \psi + C'_1 \cos \psi \\ \sin \psi + C'_2 \cos \psi & \cos \psi - C'_2 \sin \psi \end{bmatrix} \quad (\text{A17})$$

Again, by normalizing the diagonal elements of the matrix \mathbf{P} to 1 and then comparing the resultant matrices to \mathbf{C} , \mathbf{R} , and \mathbf{T} it can easily be inferred that

$$C_1 = \frac{-\sin \psi + C'_1 \cos \psi}{\cos \psi + C'_1 \sin \psi} \quad (\text{A18})$$

$$C_2 = \frac{\sin \psi + C'_2 \cos \psi}{\cos \psi - C'_2 \sin \psi} \quad (\text{A19})$$

$$\mathcal{S}_{13} = (\cos \psi + C'_1 \sin \psi) \mathcal{S}'_{13}$$

$$\mathcal{S}_{24} = (\cos \psi - C'_2 \sin \psi) \mathcal{S}'_{24}$$

If it is required that maximum energy falls into the v channel when port 1 (v port of the OMT) is energized, the condition $C_1 = 0$ should be enforced which is equivalent to setting $\psi = \arctan C'_1$. This condition, however, does not maximize the energy transfer into the h channel when the h port of the OMT is energized. Moreover the cross-talk terms usually are complex quantities and the above condition may not be achievable. In order to maximize the energy transfer for both channels and simplify the calibration procedure we look for a rotation angle such that $C_1 = C_2 \ll 1$, i.e.

$$\frac{-\sin \psi + C'_1 \cos \psi}{\cos \psi + C'_1 \sin \psi} = \frac{\sin \psi + C'_2 \cos \psi}{\cos \psi - C'_2 \sin \psi}$$

which requires that

$$\psi = \frac{1}{2} \arctan \frac{C'_1 - C'_2}{1 + C'_1 C'_2}. \quad (\text{A20})$$

This can be accomplished if both C'_1 and C'_2 are real quantities. It is relatively easy to design the antenna such that $C'_1 \approx C'_2$ to begin with, and then by adjusting the rotation angle using (20) the cross-talk factors can be made even more

similar. Therefore from here on, we shall assume that the antennas' cross-talk factors are identical and the error associated with this assumption is on the order of the difference in the imaginary parts of the cross-talk factors. In view of this approximation, the measured scattering matrix is given by

$$\mathbf{M} = \mathbf{R} \begin{bmatrix} 1 & C \\ C & 1 \end{bmatrix} \mathbf{s} \begin{bmatrix} 1 & C \\ C & 1 \end{bmatrix} \mathbf{T}. \quad (\text{A21})$$

A3 Calibration Procedure

The relationship between the measured scattering matrix \mathbf{M} and the actual scattering matrix of an unknown target is given by equation (21). If the elements of the imbalance and cross-talk matrices are known, the scattering matrix of the target can be obtained from

$$\mathbf{s} = \mathbf{C}^{-1} \mathbf{R}^{-1} \mathbf{M} \mathbf{T}^{-1} \mathbf{C}^{-1}.$$

The matrices \mathbf{R} and \mathbf{T} are diagonal and \mathbf{C} is symmetric with known diagonal elements; therefore, there is a total of five unknowns that need to be determined. The standard approach is to measure targets with known scattering matrices to establish a set of equations for the unknown elements of the \mathbf{R} , \mathbf{T} , and \mathbf{C} matrices. By measuring each calibration target, four nonlinear equations are obtained, so it seems that at least two targets are needed to find all the five unknowns. But as will be shown, a sphere or a target with similar scattering matrix (such as a trihedral) is sufficient to characterize the scattering matrix of the unknown target. In fact,

it is not required to find all the five unknowns to obtain the scattering matrix of the unknown target.

Upon expanding (21) and noting that for backscattering, the scattering matrix is symmetric ($s_{12} = s_{21}$) we get

$$\mathbf{M} = \mathbf{R} \begin{bmatrix} s_{vv} + 2Cs_{vh} + C^2s_{hh} & (1 + C^2)s_{vh} + C(s_{vv} + s_{hh}) \\ (1 + C^2)s_{vh} + C(s_{vv} + s_{hh}) & s_{hh} + 2Cs_{vh} + C^2s_{vv} \end{bmatrix} \mathbf{T}. \quad (\text{A22})$$

For the sake of simplicity, the diagonal elements of \mathbf{R} and \mathbf{T} will be denoted by R_i and T_i ($i = 1, 2$) respectively. Measuring a sphere with radar cross section $\sigma^\circ = |s^\circ|^2$, (22) provides the following set of equations

$$R_1T_1(1 + C^2) = \frac{m_{11}^\circ}{s^\circ} \quad (\text{A23})$$

$$2R_1T_2C = \frac{m_{12}^\circ}{s^\circ} \quad (\text{A24})$$

$$2R_2T_1C = \frac{m_{12}^\circ}{s^\circ} \quad (\text{A25})$$

$$R_2T_2(1 + C^2) = \frac{m_{22}^\circ}{s^\circ} \quad (\text{A26})$$

where m_{ij}° denotes the ij^{th} elements of the measured scattering matrix of the sphere.

The cross-talk term can be obtained by multiplying (23) by (26) and (24) by (25) and then eliminating the term $R_1T_1R_2T_2$ from the resultant equations. Thus

$$\frac{4C^2}{(1 + C^2)^2} = \frac{m_{12}^\circ m_{21}^\circ}{m_{11}^\circ m_{22}^\circ} \triangleq a$$

which is a biquadratic equation with four possible solutions given by

$$C = \pm \frac{1}{\sqrt{a}} \pm \sqrt{\frac{1}{a} - 1}.$$

Requiring C to be a small number, two of these solutions can be discarded (note that $|a| \ll 1$), therefore

$$C = \pm \frac{1}{\sqrt{a}}(1 - \sqrt{1-a}) \quad (\text{A27})$$

To meet the condition $|C| < 1$, the branch cut for $\sqrt{1-a}$ is chosen such that $\text{Re}[\sqrt{1-a}] > 0$. Therefore C is determined from the sphere measurement within a \pm sign.

By denoting the measured scattering matrix elements of the unknown target by m_{ij}^u and using (23)-(26) to find the products of $R_i T_j$, we obtain

$$s_{vv} + C(s_{vh} + s_{hv}) + C^2 s_{hh} = \frac{m_{11}^u}{m_{11}^o}(1 + C^2)s^o \quad (\text{A28})$$

$$C^2 s_{vv} + C(s_{vh} + s_{hv}) + s_{hh} = \frac{m_{22}^u}{m_{22}^o}(1 + C^2)s^o \quad (\text{A29})$$

$$C(s_{vv} + s_{hh}) + s_{vh} + C^2 s_{hv} = \frac{m_{12}^u}{m_{12}^o}(2C)s^o \quad (\text{A30})$$

$$C(s_{vv} + s_{hh}) + s_{hv} + C^2 s_{vh} = \frac{m_{21}^u}{m_{21}^o}(2C)s^o \quad (\text{A31})$$

Solving these equations simultaneously, the unknown scattering matrix elements can be obtained and are given by

$$s_{vv} = \frac{1}{(1-C^2)^2} \left[-2C^2 \left(\frac{m_{12}^u}{m_{12}^o} + \frac{m_{21}^u}{m_{21}^o} \right) + (1+C^2) \left(\frac{m_{11}^u}{m_{11}^o} + C^2 \frac{m_{22}^u}{m_{22}^o} \right) \right] s^o \quad (\text{A32})$$

$$s_{hh} = \frac{1}{(1-C^2)^2} \left[-2C^2 \left(\frac{m_{12}^u}{m_{12}^o} + \frac{m_{21}^u}{m_{21}^o} \right) + (1+C^2) \left(\frac{m_{22}^u}{m_{22}^o} + C^2 \frac{m_{11}^u}{m_{11}^o} \right) \right] s^o \quad (\text{A33})$$

$$s_{vh} = \frac{C}{(1-C^2)^2} \left[2 \frac{m_{12}^u}{m_{12}^o} + 2C^2 \frac{m_{21}^u}{m_{21}^o} - (1+C^2) \left(\frac{m_{11}^u}{m_{11}^o} + \frac{m_{22}^u}{m_{22}^o} \right) \right] s^o \quad (\text{A34})$$

$$s_{hv} = \frac{C}{(1-C^2)^2} \left[2 \frac{m_{21}^u}{m_{21}^o} + 2C^2 \frac{m_{12}^u}{m_{12}^o} - (1+C^2) \left(\frac{m_{11}^u}{m_{11}^o} + \frac{m_{22}^u}{m_{22}^o} \right) \right] s^o \quad (\text{A35})$$

It should be pointed out that there is no ambiguity in s_{11} and s_{22} since the the branch of $\sqrt{1-a}$ is defined, but there is a 180° phase ambiguity in s_{12} and s_{21} .

Expressions (32)-(35) give the elements of the scattering matrix when the calibration and the unknown targets are at the same range from the radar. If the range of the calibration target (r_0) is different from the range of the unknown target (r_u), (32)-(35) must be modified by a multiplying factor $(\frac{r_u}{r_0})^2 e^{-2ik_0(r_0-r_u)}$.

The complex quantity C is an inherent characteristic of the antenna system and does not change with variations in the performance of the active devices in the radar system, and therefore is less affected by environmental changes. The ambiguity in the sign of C for an antenna system may be easily resolved once by measuring a target with a known phase relationship between the elements of its scattering matrix (such as a tilted cylinder).

To investigate the accuracy of the measurement of C , we use the fact that $|a| \ll 1$ and therefore (27) becomes $C \approx \frac{1}{2}\sqrt{a}$. If the uncertainty in measurement of a is represented by Δ and $|\Delta| \ll |a|$, then

$$C + \delta C = \frac{1}{2}\sqrt{a + \Delta} \approx \frac{1}{2}\sqrt{a}\left(1 + \frac{\delta}{a}\right)$$

from which we get

$$\frac{\delta C}{C} = \frac{1}{2} \frac{\Delta}{a}.$$

It is concluded that the uncertainty in C is about 50% of the uncertainty in measuring a .

Using a sphere as the calibration target not only simplifies calculation of the unknowns significantly, but also offers two more advantages. One advantage is that the scattering matrix of a sphere is insensitive to orientation and therefore no

error will be incurred because of target orientation. The second advantage stems from the fact that spheres are the only three-dimensional structures for which an exact theoretical scattering matrix is known.

A4 Comparison with Measured Data

The validity of the STCT is now examined by measuring scattering matrices of cylinders and spheres as test targets employing a polarimetric X-band scatterometer. The results based on the IACT also is included for comparison. The measurements were performed in a 14-m long anechoic chamber and the target orientation was facilitated by a very fine-tune azimuth over elevation stepper motor positioner. Detailed description of the scatterometer and measurement setup is given in [3].

The analysis given in section 2 does not take into account the effect of noise and disturbances. In reality the measured scattering matrix includes an additive noise factor and therefore (21) becomes

$$\mathbf{M} = \mathbf{RCsCT} + \mathbf{N} \quad (\text{A36})$$

where \mathbf{N} is a matrix representation of disturbances. In order to measure \mathbf{s} accurately, all the elements of \mathbf{N} must be much smaller than the elements of \mathbf{M} . The disturbances, for a typical radar system, may include thermal and background noise. Thermal noise is a zero-mean random process with power proportional to the product of the system bandwidth and noise temperature. This effect can be minimized using an averaging process. The background noise includes the signal returns from objects at ranges comparable to that of the test target or the short-

range multiple reflections within the system. This problem can be eliminated using background subtraction from the target and background response. Another source of error in the measurement of s is the interaction of the target with its support structure (pedestal). This interaction is not linear and hence cannot be subtracted out.

A 12-inch diameter sphere was used as the calibration target, and test targets included a 6-inch sphere, a 8-inch sphere, and a conducting cylinder with a diameter of 0.8 cm and length of 27.2 cm, observed at three different orientations (vertical, horizontal, and 45°). The calibrated elements of the scattering matrix were then compared with the theoretical values computed using the exact Mie-series solution for the spheres and using a semi-exact solution for the cylinder. The semi-exact solution is based on the assumption that the current induced on the surface of the cylinder is identical with that of an infinite cylinder with the same radius. This solution is accurate in the specular direction and when the cylinder length is much larger than the wavelength.

Using averaging and background subtraction a signal to noise ratio of better than 40 dB was achieved in measuring the elements of M . For targets with $s_{12} = 0$ (such as sphere and vertical cylinder), the signal to noise ratio for the off-diagonal elements of M was better than 25 dB.

Figure 4-6 compare the theoretical and measured scattering matrix elements of the 6-inch sphere. The error in the like-polarized terms is less than 0.3 dB and the results based on the IACT and STCT are exactly identical. It is also

shown that the error in the phase is less than 2 degrees. Figure 6 shows the cross-polarized component of the sphere (theoretical value = $-\infty$ in dB scale) where there is a significant disagreement between STCT and IACT. An effective polarization isolation of 50 dB is obtained using STCT. It should be pointed out that the minimum noise-equivalent cross section of the radar system is -65 dBsm. Therefore the cross polarization isolation is limited by the system noise in this case. Similar results were also obtained for the 8-inch sphere. Figures 7-10 depict the results for the vertical cylinder and excellent agreement between the measured data and theory is achieved. As shown in Fig. 9 the measured effective polarization isolation for the vertical cylinder is -50 dB. Results for the 45° tilted cylinder are shown in Figs. 11-15 where the accuracy of the STCT is within ± 0.4 dB in magnitude and $\pm 5^\circ$ in phase. These plots demonstrate the superiority of the STCT over the IACT.

A5 Concluding Remarks

A convenient calibration technique for single-antenna polarimetric radar system with range-gating capability has been developed. The radar cross-talk contamination and channel imbalances are obtained by measuring the backscatter from a single calibration target whose scattering matrix is diagonal with equal diagonal entries. The insensitivity to alignment of calibration target offered by this technique makes it particularly useful for field operation.

Using a four-port network approach, it is shown that the cross-talk contam-

ination factor is a feature of the antenna system only, and hence is not affected by instability of active devices in the radar system. Excellent agreement between measurements and theory was obtained when a sphere was used as the calibration target and cylinders and other spheres were used as test targets. A minimum effective polarization isolation of 50 dB was achieved using this technique.

Acknowledgement This work was supported by JPL SIR-C calibration project.

References

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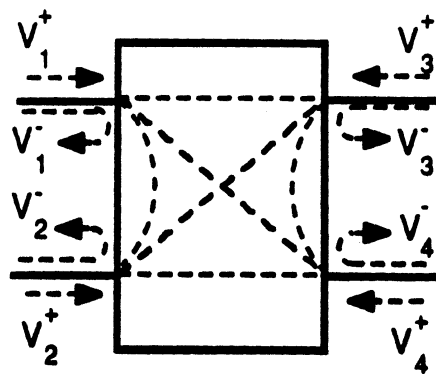
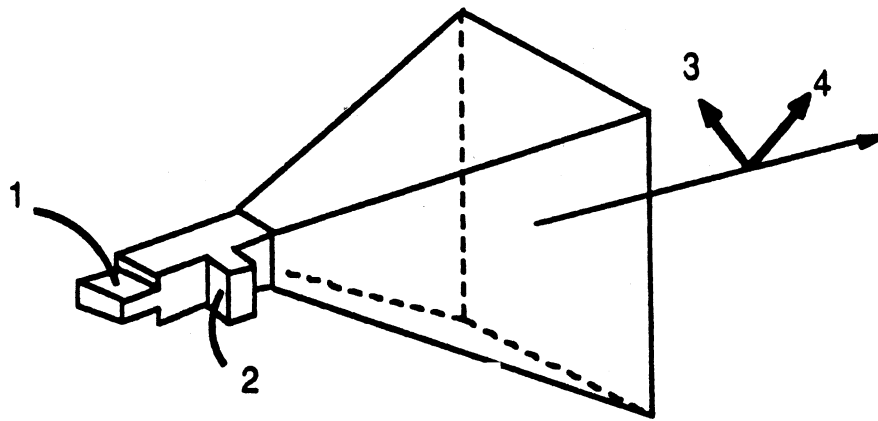


Figure A-1: Antenna system and its equivalent circuit four-port representation.

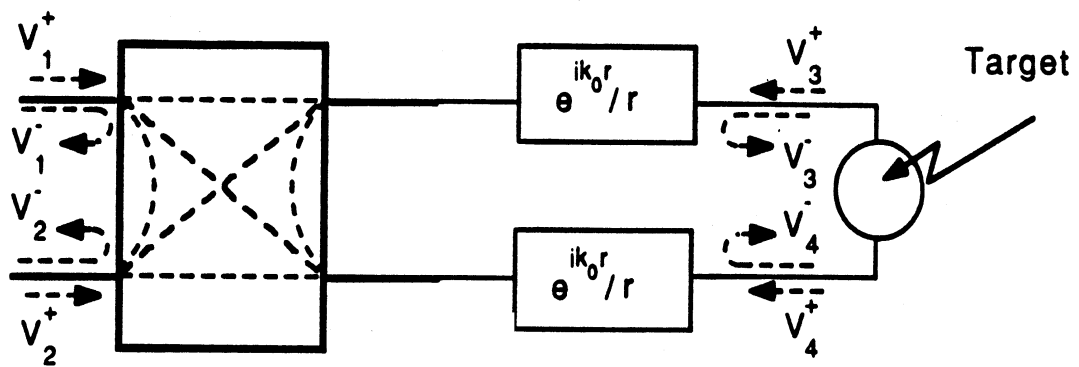


Figure A-2: Signal flow-chart of the antenna system, free space, and the target.

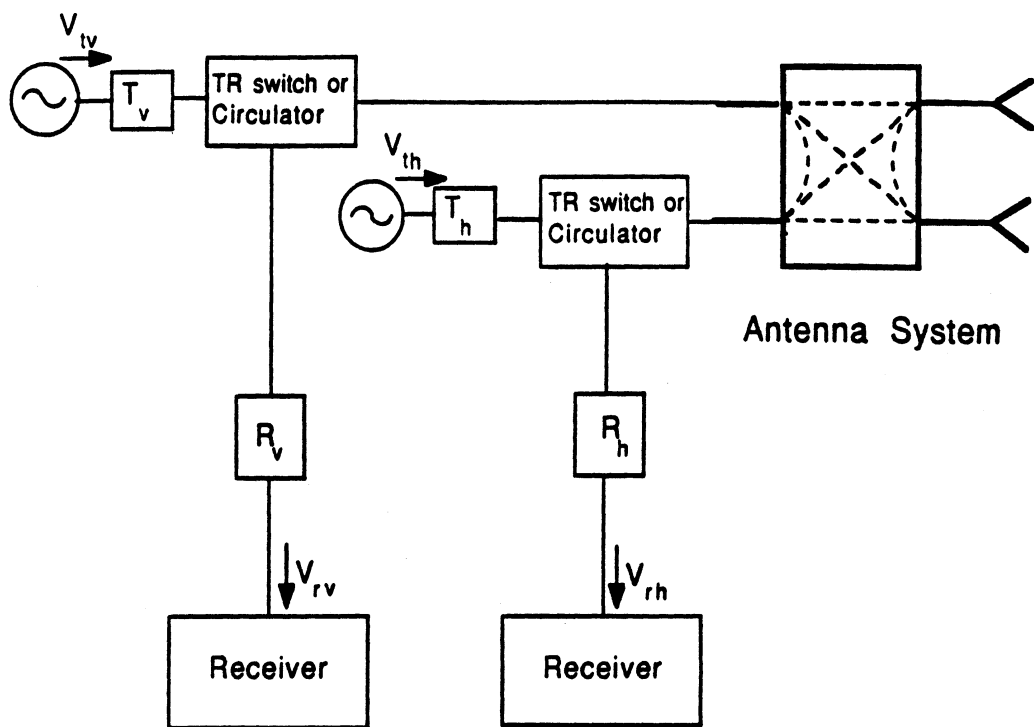


Figure A-3: Simplified block diagram of a typical polarimetric radar.

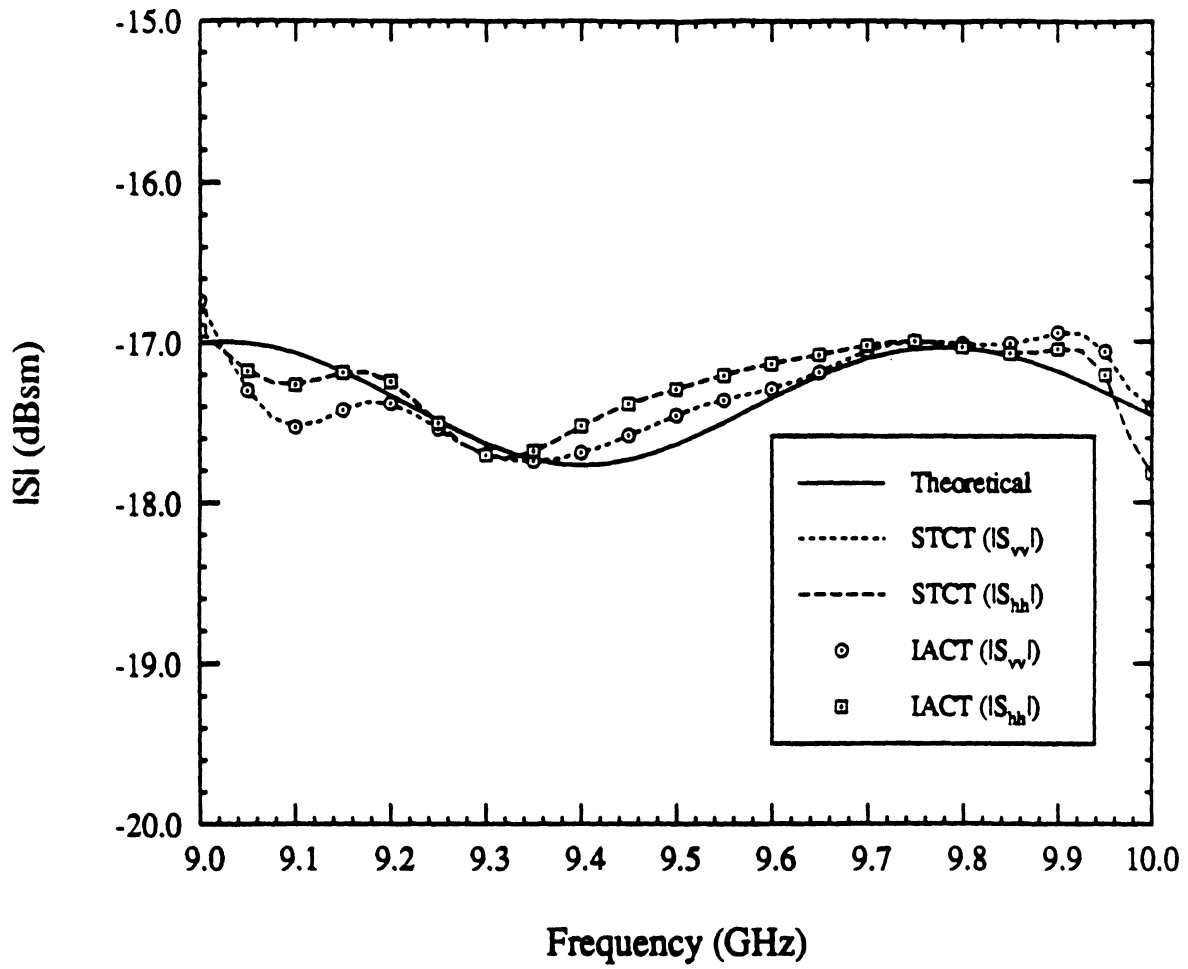


Figure A-4: Magnitude of the diagonal elements of the scattering matrix of a 6-inch sphere.

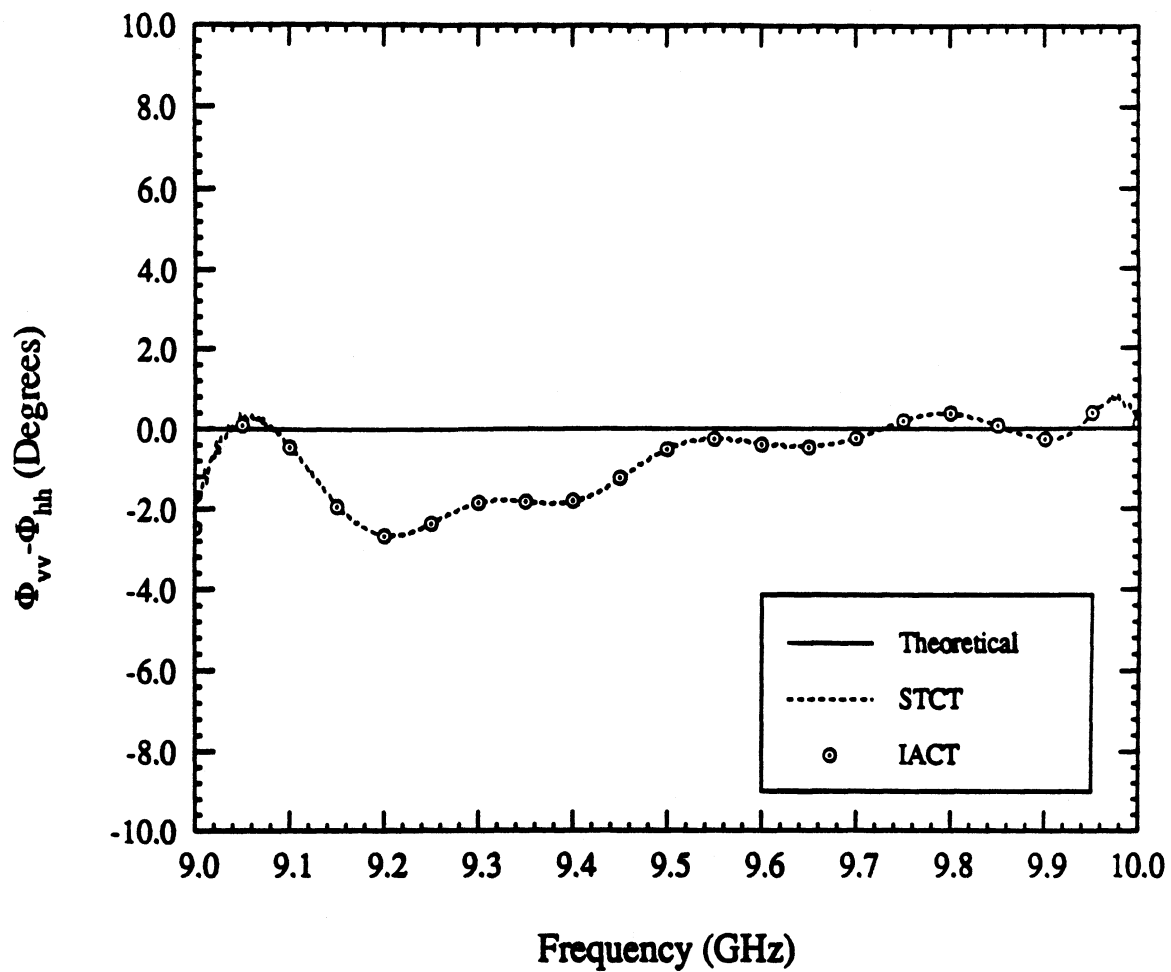


Figure A-5: Phase difference between the diagonal elements of the scattering matrix of a 6-inch sphere.

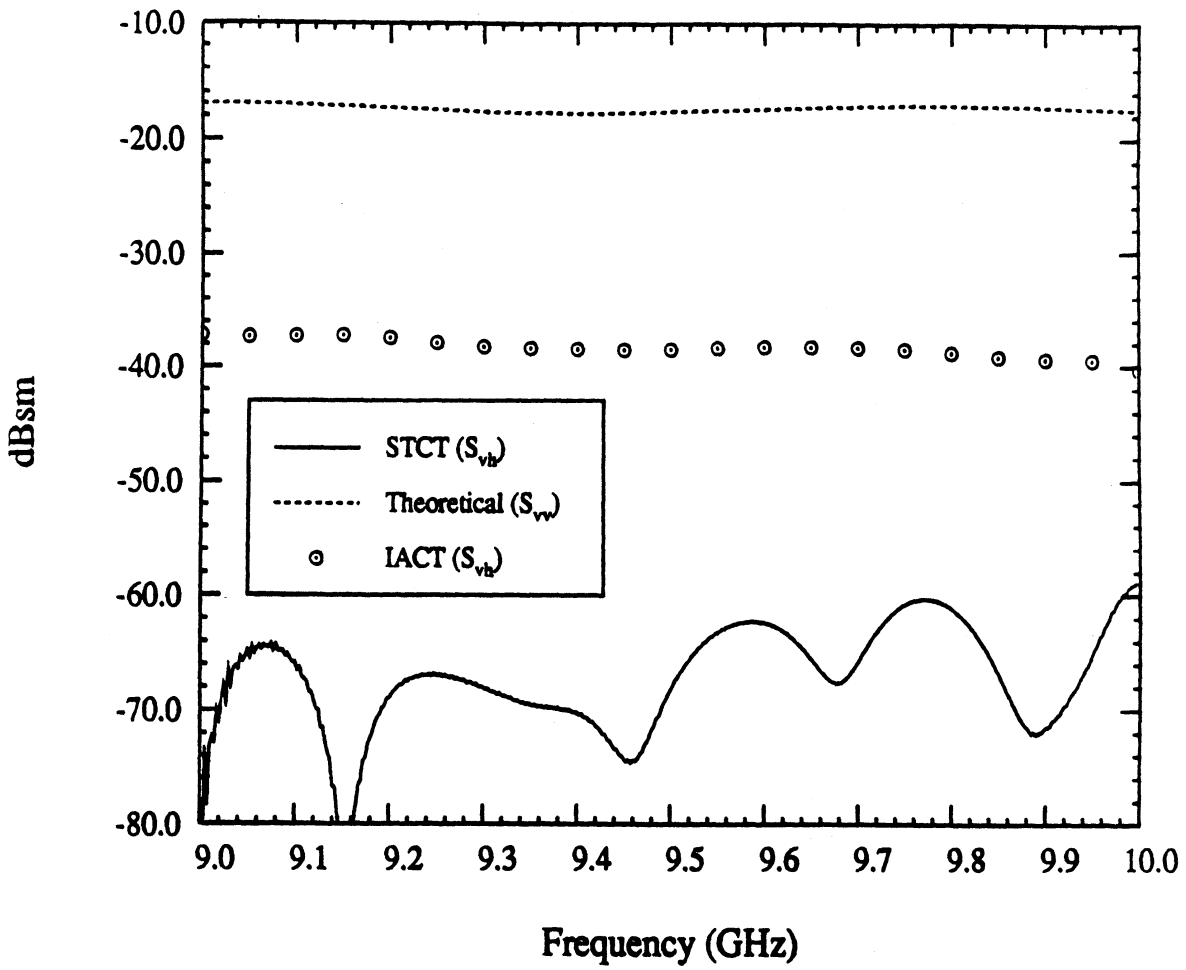


Figure A-6: Magnitude of the off-diagonal element of the scattering matrix of a 6-inch sphere compared with one of the diagonal elements.

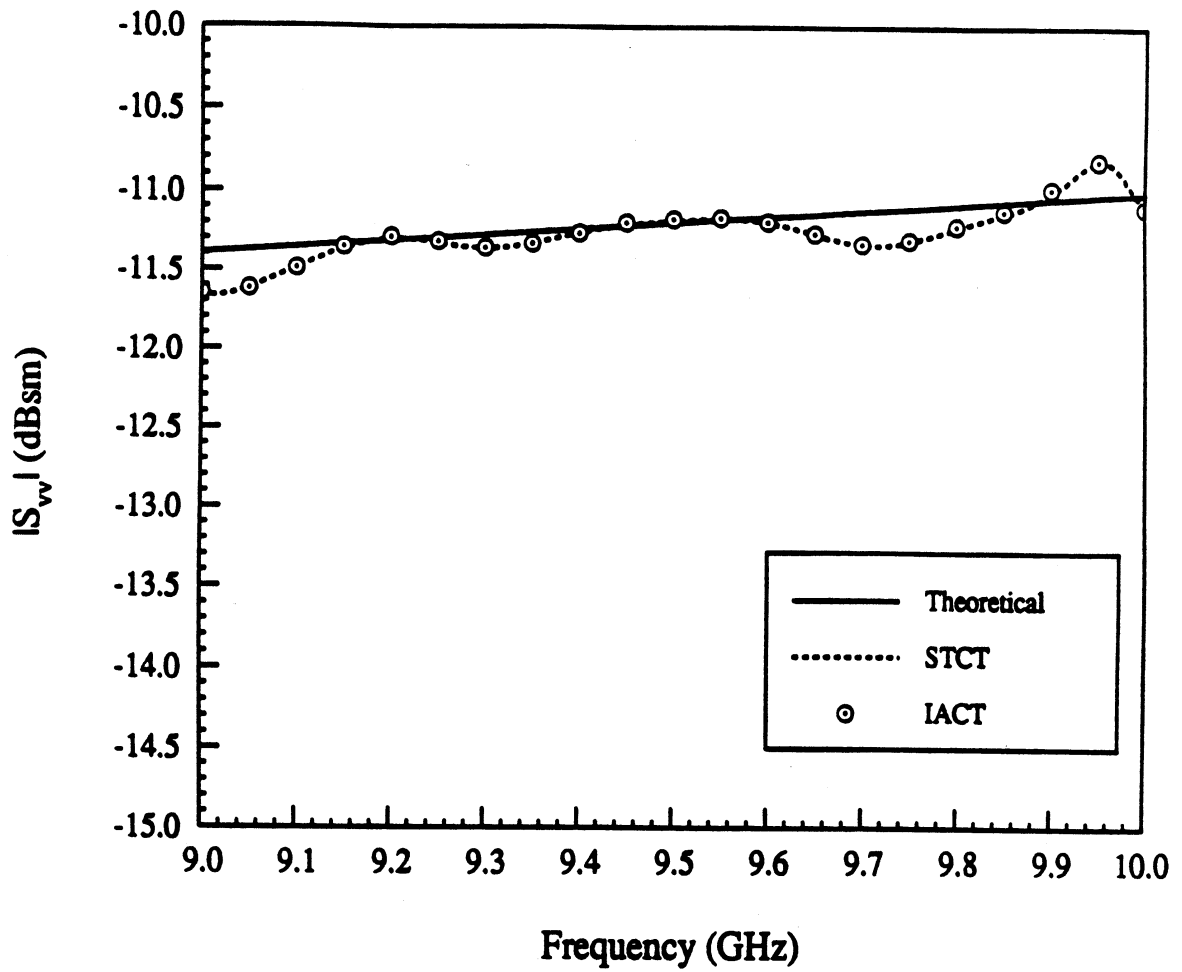


Figure A-7: Magnitude of the diagonal element (s_{vv}) of the scattering matrix of the vertical cylinder.

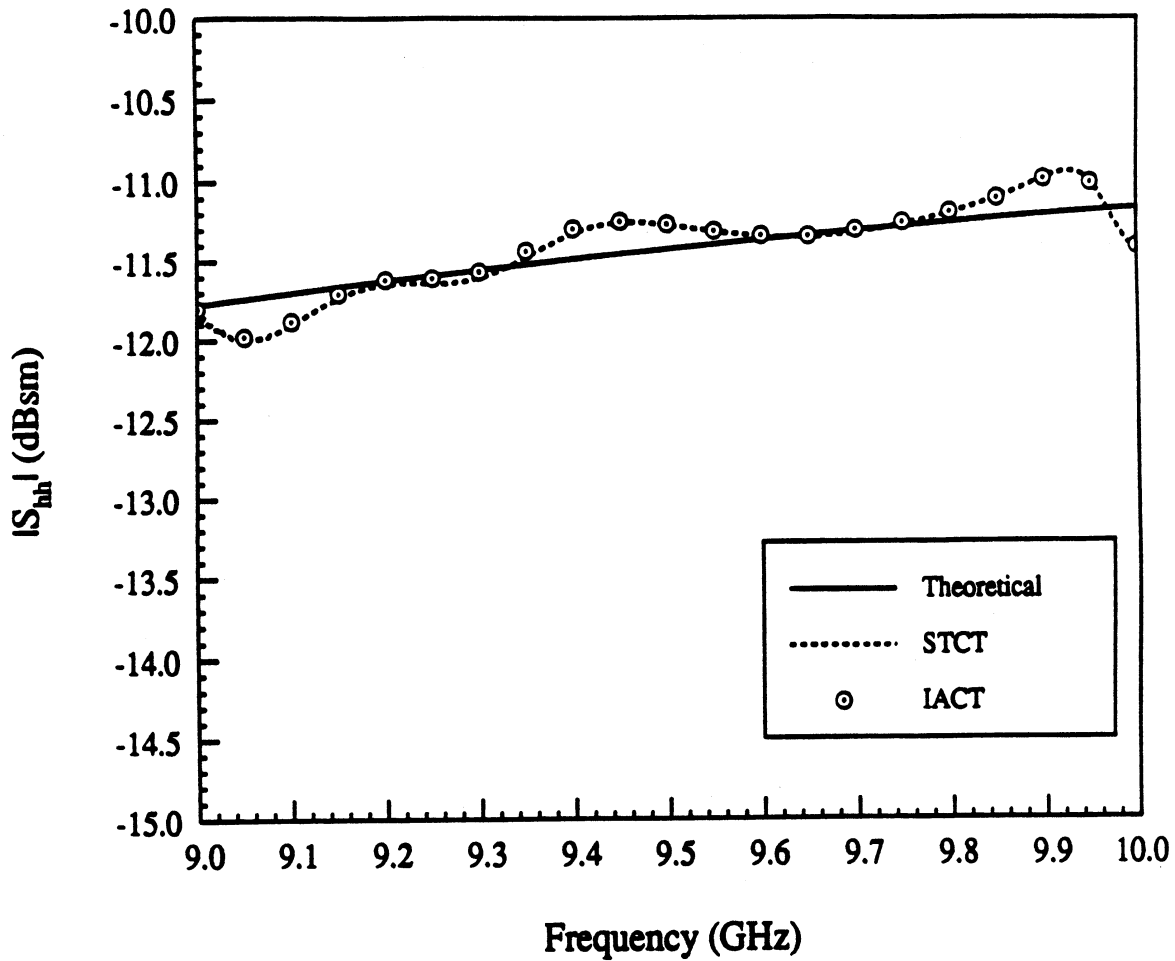


Figure A-8: Magnitude of the diagonal element (s_{hh}) of the scattering matrix of the vertical cylinder.

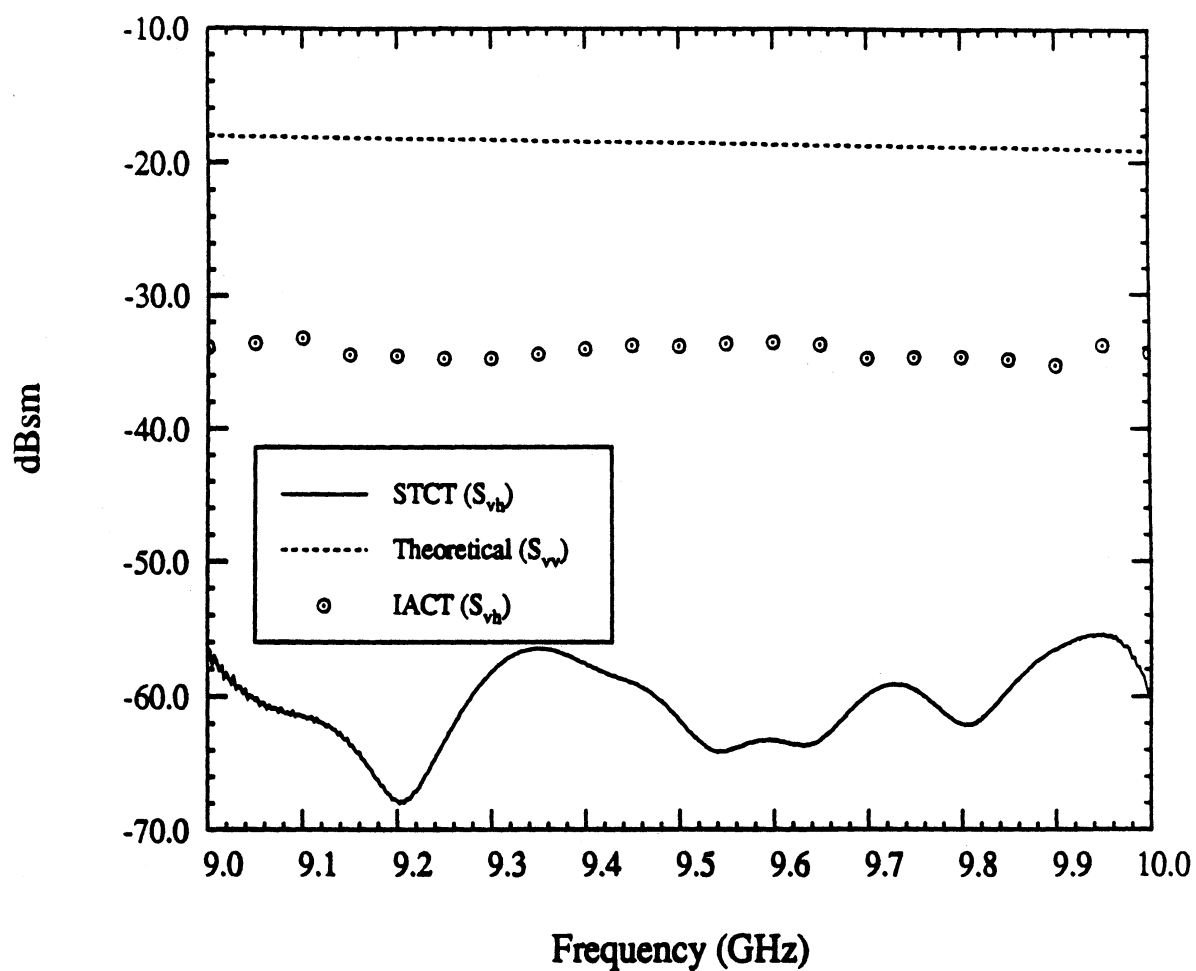


Figure A-9: Magnitude of the off-diagonal element (s_{vh}) of the scattering matrix of the vertical cylinder compared with s_{vv} .

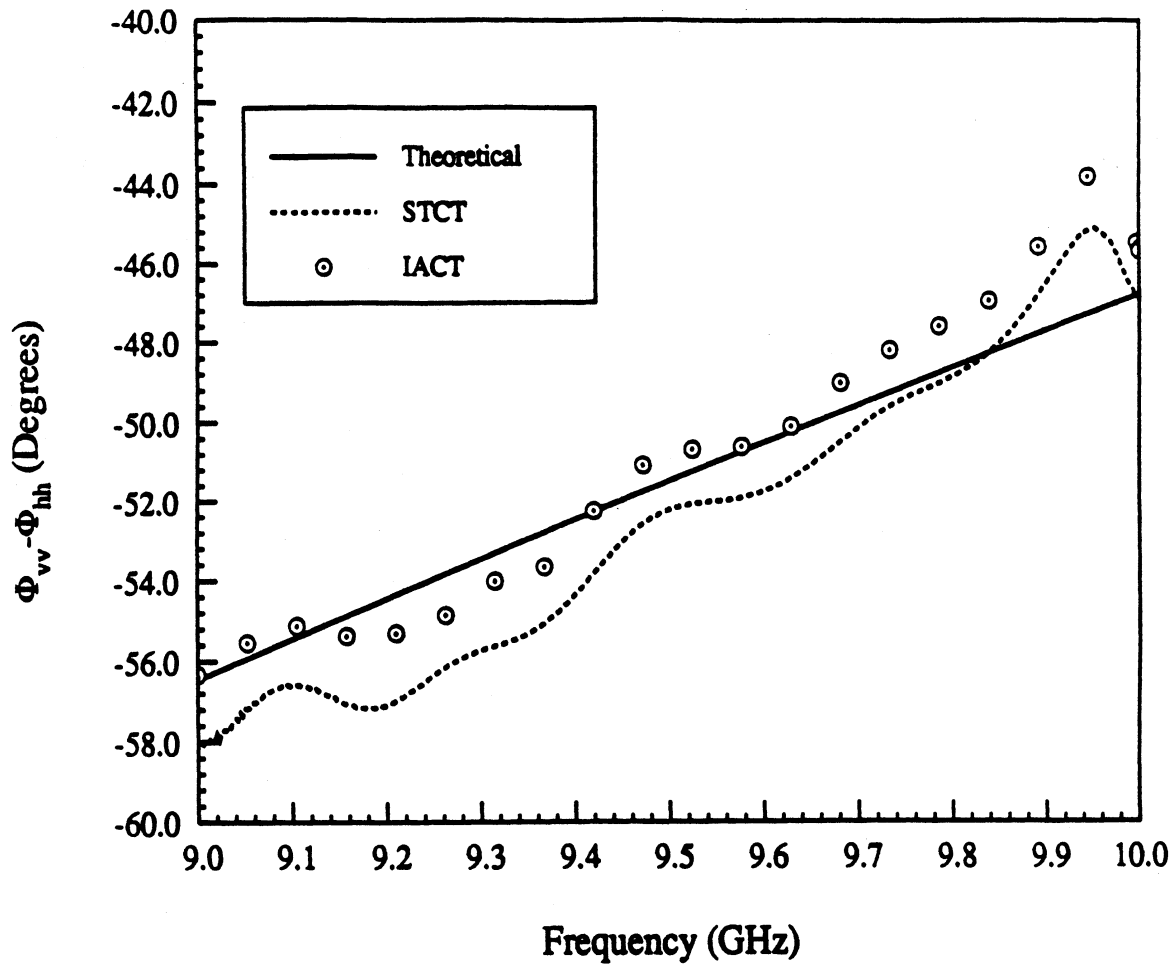


Figure A-10: Phase difference between the diagonal elements of the scattering matrix of the vertical cylinder.

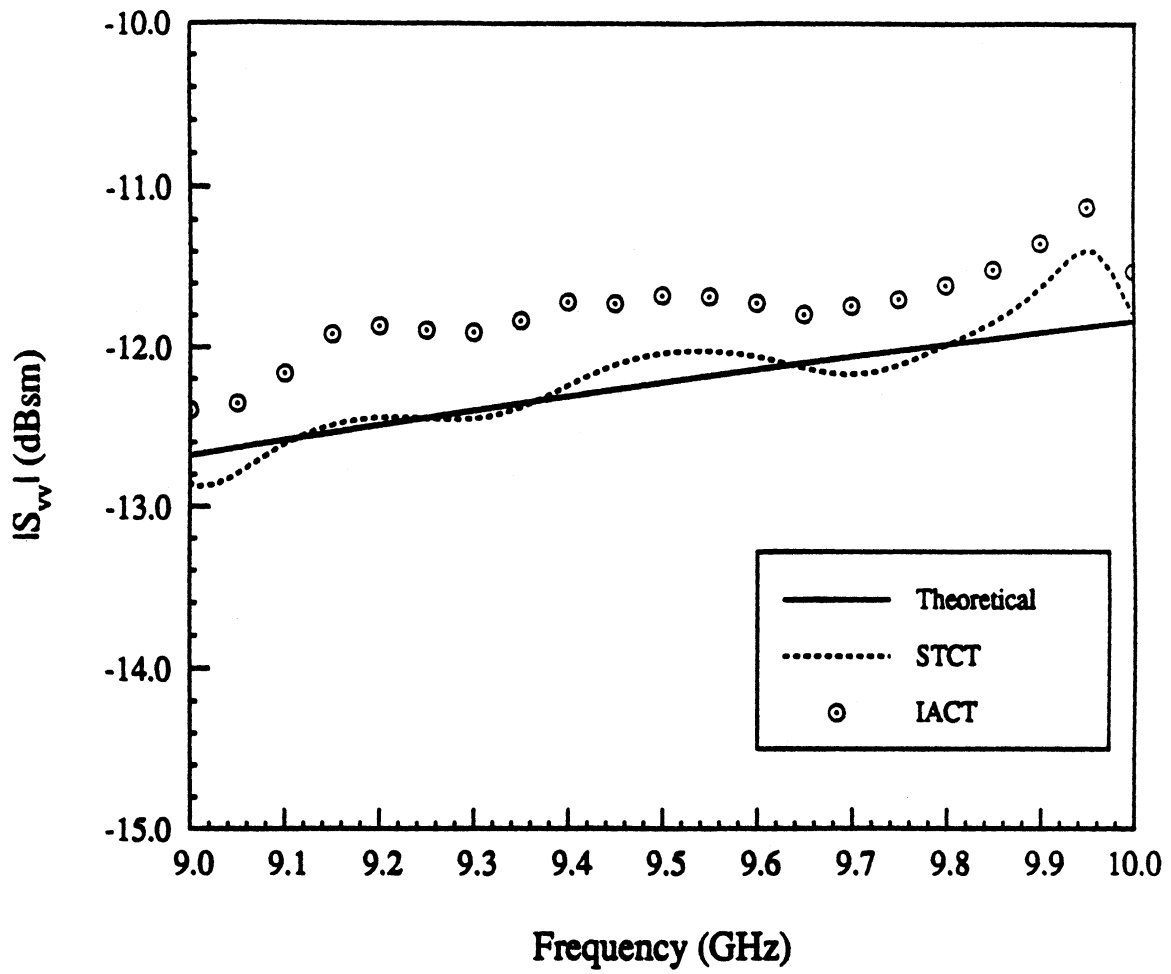


Figure A-11: Magnitude of the diagonal element (s_{vv}) of the scattering matrix of the 45° tilted cylinder.

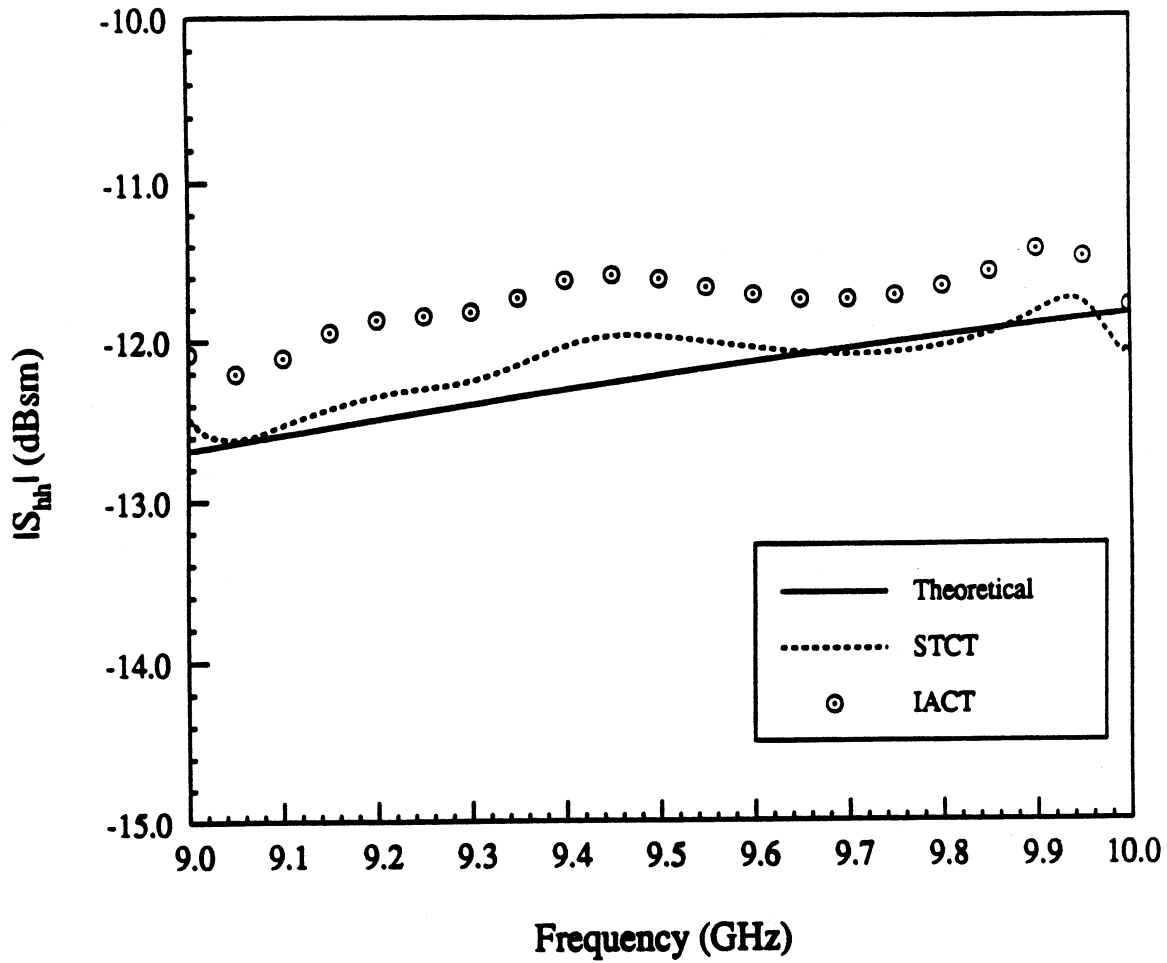


Figure A-12: Magnitude of the diagonal element (s_{hh}) of the scattering matrix of the 45° tilted cylinder.

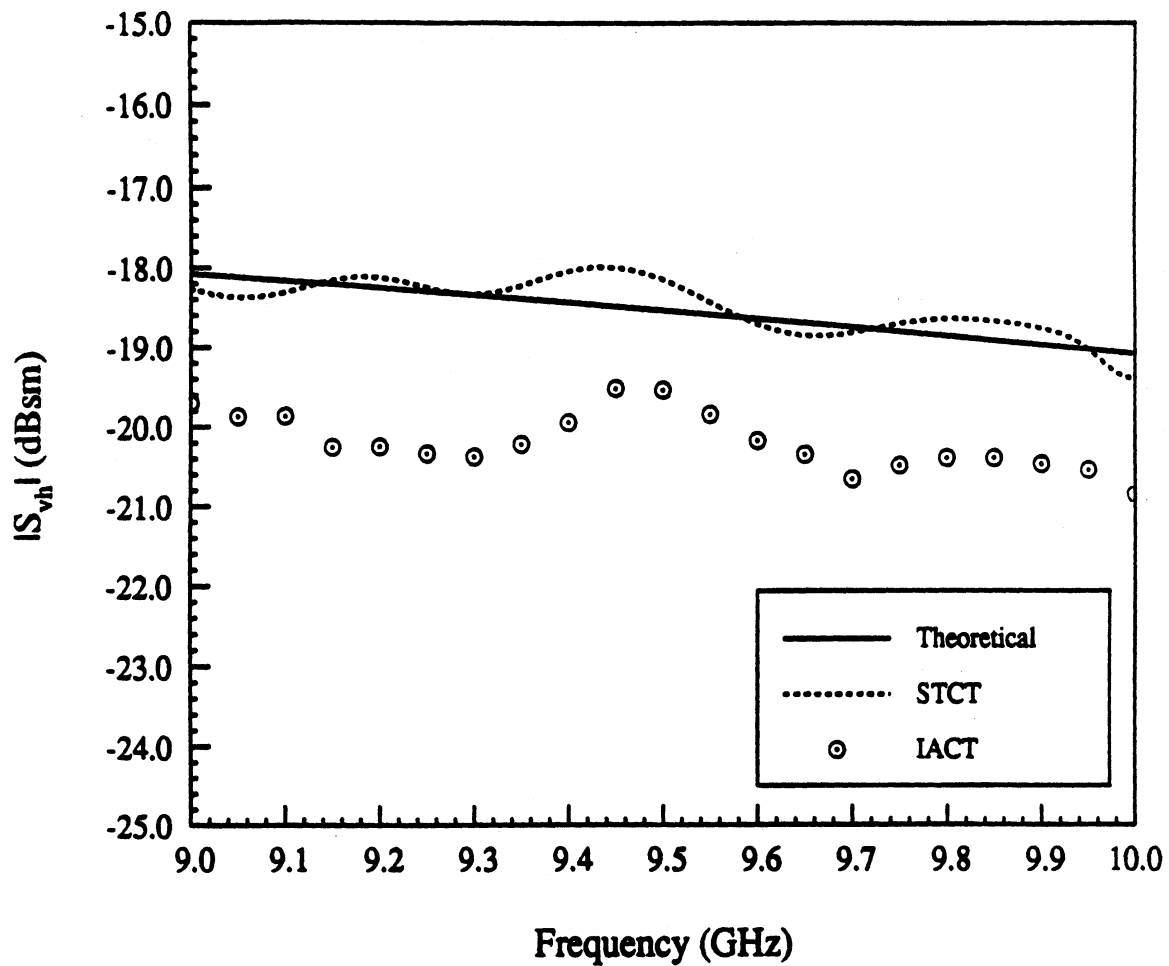


Figure A-13: Magnitude of the off-diagonal element (s_{vh}) of the scattering matrix of the 45° tilted cylinder.

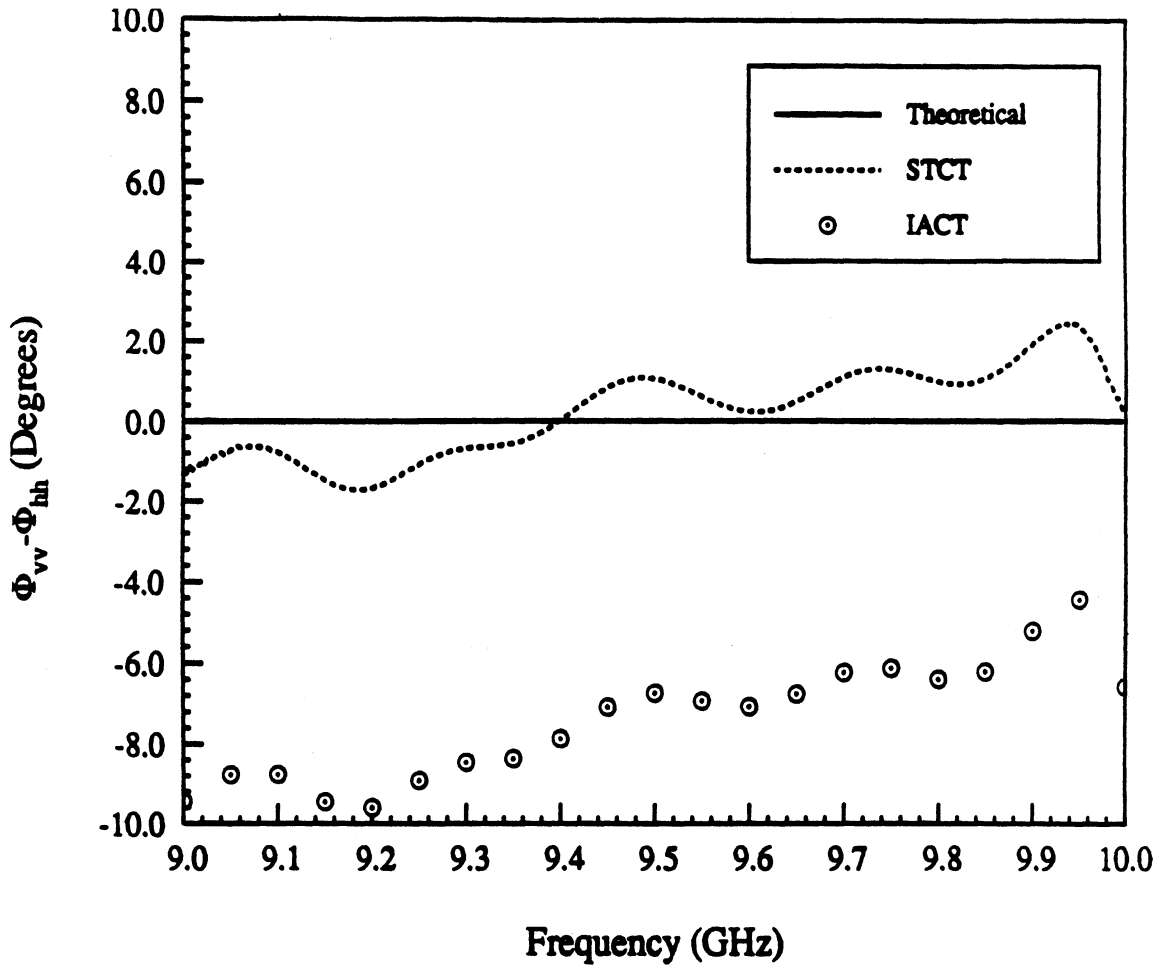


Figure A-14: Phase difference between the diagonal elements of the scattering matrix of the 45° tilted cylinder.

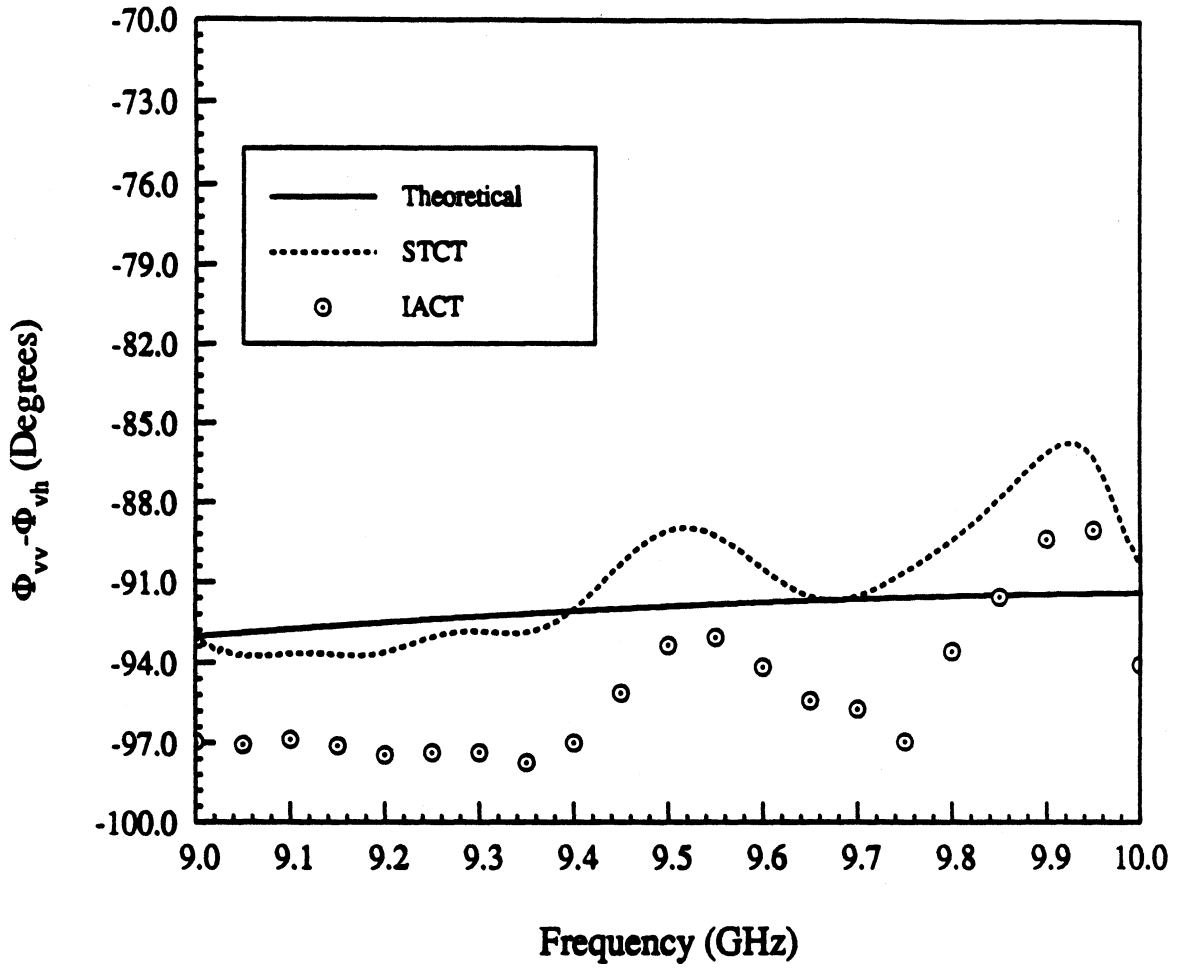


Figure A-15: Phase difference between the diagonal and the off-diagonal elements of the scattering matrix of the 45° tilted cylinder.

APPENDIX B

POLARIMETRIC MEASUREMENT PROGRAM

This program is written in HP BASIC to control the network analyzer based scatterometer (see Fig. (1)) and a stepping motor controller to acquire polarimetric measurement data of the PARCs automatically.

The main program is comprised of two major functions for calibration: 1) target measurement and 2) reference target measurement. For each measurement, the frequency band for L- and C-band is selected and the time gating parameters, i.e., the gate center and the gate span, are selected by the operator prior to data acquisition. The target measurement starts with selection of target rotation parameters such as start angle, increment angle, stop angle and rotation speed by calling the subroutine "Rotate_target". The subroutine "Rotate_target" calls the other subroutine 'comm' to communicate with a stepping motor controller through a serial port of an HP 9000 series computer. The Fourier transform of the time gated data (in magnitude and phase) of the four polarizations (VV, HH, HV, VH) are collected for each incidence angle and for the selected frequency points. The reference measurement for calibration utilizes the averaging technique to reduce the effect of thermal noise and background subtraction to increase S/N.

At the end of the measurement, the main program calls a subroutine "Store_file" to transfer the collected data to a disk drive with a title of the data file des-

ignated by a subroutine "Set_clock" and "Store_file". The subroutines of "Config_and_poll", "Hp_bus_init" and "Series_init" are to configure the system including a HP network analyzer and the HP-IB bus connections. The subroutines of "Freq_set", "Freq_sw" and "Pol_sw" are to control the network analyzer and the relay actuator for the selection of frequencies and polarizations.

```

10  !*****
20  !   L/C/X POLARIMETER MEASUREMENT PROGRAM
30  !       FILE: PARC_1
40  !*****
50  !   LAST EDIT: May 21, 1990       JPL PARC measurements
60  !
70  !*****
80  OPTION BASE 1
90  COM /Paths/ @Nwa,@Nwa_data1,@Nwa_data2,Netwrk_analyzer,@Hplib,@Relay
100 COM /Constants/ Vel,Zero(3),Exec_key$(2)
110 COM /System_config/ INTEGER Printer_flag,Debug_flag,Version$(12),Mode$(10)
      ,Out_type$(10),Sound$(3),Bell$(1),Target$(30),Ref_target$(30)
120 COM /Sys_1/ Freq$(3)[1],Freq_cent(3),Freq_span(3),Gate_cent(3),Gate_span(3)
130 COM /Sys_2/ Pol$(4)[2],Polsw$(3,4)[8]
140 COM /Sys_3/ INTEGER F_disp,P_disp
150 COM /Sys_4/ Drive_a$(15),Drive_b$(15),Drive_c$(15),INTEGER Preamble,Bytes
160 COM /Sys_5/ INTEGER Nskip,Ndata
170 COM /Sys_6/ Ref_angle,Angle,Angle$(10),Beam(3),INTEGER Npts,Ntrace
      ,Average_factor
180 COM /Sys_7/ INTEGER Meas_flag(3)
190 COM /Com4/ INTEGER Rotation_state,REAL Inc_angle,Current_angle,Start_angle,
      Stop_angle,Old_home_angle,INTEGER Sets_per_pos
200 COM /Status/ INTEGER Sc,Connect_flg,E_flg,Debug_flg,Response$(80)
210 !
220 !
230 INTEGER F,I,J,P,T,Meas_flag_old(3),Exit_flag,Nt,Nst,Nskh,Npt
240 DIM Sky_cal_file$(3)[14],Old_target_name$(30)
250 DATA "L","C","X"                ! FREQUENCY
260 DATA "VV","HH","HV","VH"       ! POLARIZATION
270 DATA 1.25,1.25,1.5              ! FREQ_CENT
280 DATA .3,.5,.5                   ! FREQ_SPAN
290 DATA 12.5,9.0,6.2              ! BEAMWIDTH
300 DATA "?*B3456","?*A56B34","?*A6B345","?*A5B346" ! L
310 DATA "?*B3456","?*A34B56","?*A4B356","?*A3B456" ! C
320 DATA "?*A34B56","?*B3456","?*A3B456","?*A4B356" ! X
330 DATA ":",700,0",":,700,1",":MEMORY,0,7"         ! DRIVE_A,B,C
340 DATA 400E-9,400E-9,400E-9      ! GATE CENTERS
350 DATA 10E-9,10E-9,10E-9         ! GATE SPANS
360 READ Freq$(*)
370 READ Pol$(*)
380 READ Freq_cent(*)
390 READ Freq_span(*)
400 READ Beam(*)
410 READ Polsw$(*)
420 READ Drive_a$,Drive_b$,Drive_c$
430 READ Gate_cent(*),Gate_span(*)
440 !
450 ! Set up_error handling routine.
460 !
470 LOAD KEY "NOKEY:MEMORY,0,1"
480 MASS STORAGE IS ":",700,1"
490 !
500 !
510 ! Initialize important parameters.
520 !
530 DEG
540 Rotation_state=-1
550 Current_angle=0.
560 MAT Meas_flag= (1)
570 Mode$="FAST ACQ"
580 F_disp=1

```

```

590 P_disp=1
600 Printer_flag=0
610 !
620 Hp_bus_init
630 IF Printer_flag=1 THEN Out_type$="PRINT/DISC"
640 !
650 Vel=2.99792458E+8
660 Ntrace=1
670 Npts=401
680 Nskip=40
690 Ndata=10
700 Average_factor=1
710 Angle$="0"
720 Angle=0
730 Ref_angle=0
740 Target$=""
750 Sound$="ON "
760 Debug_flag=0
770 Bell$=CHR$(7)
780 Exec_key$=CHR$(255)&CHR$(88)
790 Version$="Version 8.0 "
800 Exit_flag=0
810 Print_banner1
820 !
830 System_memory=VAL(SYSTEM$("AVAILABLE MEMORY"))
840 IF FNASK("INITIALIZE RAM DISK?") THEN
850     INITIALIZE Drive_c$,0
860     INITIALIZE Drive_c$,INT((System_memory)/512)
870 ELSE
880     ASSIGN @Is_it_there TO Drive_c$;RETURN Outcome
890     IF Outcome=0 THEN
900         CAT Drive_c$;NO HEADER,COUNT Entries
910         IF Entries=0 THEN INITIALIZE Drive_c$,0
920     END IF
930     ASSIGN @Is_it_there TO *
940 END IF
950 !
960 Config_and_poll
970 OUTPUT @Relay;"?*B1256A34"
980 Series_init
990 !
1000 !
1010 Start_loop: !
1020     !
1030     Print_banner4
1040         ON KEY 0 LABEL " SKY CAL ",FNTrap_level GOSUB Sky_calibrate
1050         ON KEY 1 LABEL " REFERENCE CAL",FNTrap_level GOSUB Ref_target
1060         ON KEY 2 LABEL " TARGET RUN ",FNTrap_level GOSUB Acq_target
1070         ON KEY 3 LABEL " SET FREQUENCY",FNTrap_level GOSUB Freq_set
1080         ON KEY 4 LABEL " ANGLE ",FNTrap_level GOSUB Set_angle
1090         ON KEY 5 LABEL " TARGET NAME ",FNTrap_level GOSUB Set_target
1100         ON KEY 6 LABEL " # OF TRACES ",FNTrap_level GOSUB Set_traces
1110         ON KEY 7 LABEL " # OF POINTS ",FNTrap_level GOSUB Set_points
1120         ON KEY 8 LABEL " # OF AVERAGES ",FNTrap_level GOSUB Set_average
1130         ON KEY 9 LABEL " QUIT ",FNTrap_level GOTO Quit_fast_acq
1140     GOSUB Allocate_matrix
1150     LOOP
1160     EXIT IF Exit_flag=1
1170     END LOOP
1180     GOSUB Deallocate_mtx
1190     Exit_flag=0

```

```

1200     GOTO Start_loop
1210     !
1220 Null: RETURN
1230     !
1240     !-----
1250     !
1260 Sky_calibrate: ! Acquire a sky calibration data set.
1270     !
1280     OFF KEY
1290     Clear_crt
1300     OUTPUT @Nwa;"TIMDTRANON; LOGM; GATEOFF;"
1310     OUTPUT @Nwa;"STAR ONS; STOP 400NS;"
1320     PRINT TABXY(1,10);"Please point scatterometer assembly to free space."
1330     PRINT TABXY(1,12);"Press CONTINUE when ready..."
1340     PAUSE
1350     Clear_crt
1360     PRINT TABXY(1,14);"Sky data is being collected ... "
1370     OUTPUT @Nwa;"TIMDTRANOFF; POLA; AVERFACT";VAL$(Average_factor);";"
1380     OUTPUT @Nwa;"GATEOFF; AVEROON;"
1390     !
1400     ! Get a free space (sky) measurement.
1410     !
1420     FOR T=1 TO Ntrace
1430     FOR F=1 TO 3
1440         IF Meas_flag(F)=1 THEN
1450             IF F=1 THEN
1460                 OUTPUT @Nwa;"POWE20;"
1470             ELSE
1480                 OUTPUT @Nwa;"POWE10;"
1490             END IF
1500             Freq_set(F)
1510             Freq_sw(F)
1520                 OUTPUT @Nwa;"GATECENT";VAL$(Gate_cent(F));"S;"
1530                 OUTPUT @Nwa;"GATESPAN";VAL$(Gate_span(F));"S;"
1540                 FOR P=1 TO 4
1550                     Pol_sw(F,P)
1560                     OUTPUT @Nwa;"FORM3; NUMG";VAL$(Average_factor+1);"; WAIT;"
1570                     OUTPUT @Nwa;"WAIT; GATEOON; WAIT; OUTPFORM;"
1580                     ENTER @Nwa_data1;Preamble,Bytes,Trace(*)
1590                     MAT Target_response(F,P,*)= Trace
1600                     OUTPUT @Nwa;" GATEOFF; WAIT;"
1610                 NEXT P
1620                     Nskh=Nskip+1
1630                 FOR P=1 TO 4
1640                     FOR Nt=Nskh TO Npts STEP Nskip
1650                         Nst=1+INT(Nt/Nskip)
1660                         Target_data(F,T,P,Nst)=Target_response(F,P,Nt)
1670                     NEXT Nt
1680                 NEXT P
1690             END IF
1700         NEXT F
1710     NEXT T
1720     Old_target_name$=Target$
1730     Ref_target$="SKY CALIBRATION"
1740     Store_file(Target_data(*),"SKY",FNTime_stamp$,F)
1750     Ref_target$=Old_target_name$
1760     Pol_sw(F_disp,P_disp)
1770     DISP "Sky calibration data saved."
1780     Exit_flag=1
1790 RETURN
1800 !

```

```

1810 !-----
1820 !
1830 Ref_target: ! Acquire a reference target data set.
1840 !
1850     OFF KEY
1860     Clear_crt
1870     OUTPUT @Nwa;"TIMDTRANON; LOGM; GATEOFF;"
1880     OUTPUT @Nwa;"LOGM; AUTO; STAR ONS; STOP 400NS;"
1890     PRINT TABXY(1,10);"Please point scatterometer assembly to reference targ
1900     PRINT TABXY(1,12);"Press CONTINUE when ready..."
1910     PAUSE
1920     GOSUB Set_gates
1930     OUTPUT @Nwa;"TIMDTRANOFF; POLA; AVERFACT";VAL$(Average_factor);";"
1940     OUTPUT @Nwa;"GATEOFF; AVEROON"
1950     INPUT "Enter the reference target angle: ",Ref_angle
1960     !
1970     ! Get the reference target response.
1980     !
1990     FOR T=1 TO Ntrace
2000     FOR F=1 TO 3
2010         IF Meas_flag(F) THEN
2020             IF F=1 THEN
2030                 OUTPUT @Nwa;"POWE20;"
2040             ELSE
2050                 OUTPUT @Nwa;"POWE10;"
2060             END IF
2070             Freq_set(F)
2080             Freq_sw(F)
2090                 OUTPUT @Nwa;"GATECENT";VAL$(Gate_cent(F));"S;"
2100                 OUTPUT @Nwa;"GATESPAN";VAL$(Gate_span(F));"S;"
2110             FOR P=1 TO 4
2120                 Pol_sw(F,P)
2130                 OUTPUT @Nwa;"FORM3;NUMG";VAL$(Average_factor+1);";WAIT"
2140                 OUTPUT @Nwa;"WAIT; GATEOON; WAIT; OUTPFORM;"
2150                 ENTER @Nwa_data1;Preamble,Bytes,Trace(*)
2160                 MAT Target_response(F,P,*)= Trace
2170                 OUTPUT @Nwa;"GATEOFF;"
2180             NEXT P
2190                 Nskh=Nskip+1
2200             FOR P=1 TO 4
2210                 FOR Nt=Nskh TO Npts STEP Nskip
2220                     Nst=INT(Nt/Nskip)
2230                     Target_data(F,T,P,Nst)=Target_response(F,P,Nt)
2240                 NEXT Nt
2250             NEXT P
2260         END IF
2270     NEXT F
2280     NEXT T
2290     Store_file(Target_data(*),"REF",FNTime_stamp$,F)
2300     !
2310     ! Get the reference target mount response.
2320     !
2330     BEEP
2340     PRINT TABXY(1,10);"Please remove the reference target from its mount."
2350     PRINT TABXY(1,12);"Press CONTINUE when ready..."
2360     PAUSE
2370     Clear_crt
2380     PRINT TABXY(1,14);"Data for the mount is being collected .... "
2390     FOR T=1 TO Ntrace
2400     FOR F=1 TO 3
2410         IF Meas_flag(F) THEN

```



```

2420         IF F=1 THEN
2430             OUTPUT @Nwa;"POWE20;"
2440         ELSE
2450             OUTPUT @Nwa;"POWE10;"
2460         END IF
2470         Freq_set(F)
2480         Freq_sw(F)
2490             OUTPUT @Nwa;"GATECENT";VAL$(Gate_cent(F));"S;"
2500             OUTPUT @Nwa;"GATESPAN";VAL$(Gate_span(F));"S;"
2510         FOR P=1 TO 4
2520             Pol_sw(F,P)
2530             OUTPUT @Nwa;"FORM3;NUMG";VAL$(Average_factor+1);";WAIT"
2540             OUTPUT @Nwa;"WAIT; GATEOON; WAIT; OUTPFORM;"
2550             ENTER @Nwa_data1;Preamble,Bytes,Trace(*)
2560             MAT Target_response(F,P,*)= Trace
2570             OUTPUT @Nwa;"GATEOFF;"
2580         NEXT P
2590             Nskh=Nskip+1
2600         FOR P=1 TO 4
2610             FOR Nt=Nskh TO Npts STEP Nskip
2620                 Nst=INT(Nt/Nskip)
2630                 Target_data(F,T,P,Nst)=Target_response(F,P,Nt)
2640             NEXT Nt
2650         NEXT P
2660     END IF
2670 NEXT F
2680 NEXT T
2690 Store_file(Target_data(*),"MNT",FNTime_stamp$,F)
2700 Pol_sw(F_disp,P_disp)
2710 DISP "Reference target mount response saved."
2720 Exit_flag=1
2730 RETURN
2740 Acq_target: !
2750     !
2760     OFF KEY
2770     Clear_crt
2780     OUTPUT @Nwa;"TIMDTRANON; LOGM; GATEOFF; AVEROFF;"
2790     OUTPUT @Nwa;"STAR 0NS; STOP 400NS;"
2800     PRINT TABXY(1,10);"Please point scatterometer assembly at surface target."
2810     PRINT TABXY(1,12);"Press CONTINUE when ready..."
2820     PAUSE
2830     GOSUB Set_gates
2840     OUTPUT @Nwa;"TIMDTRANOFF; WAIT; GATEOON; POLA;"
2850     !
2860     ! Get the target response.
2870     !
2880     FOR T=1 TO Ntrace
2890         !
2900         ! Get angles
2910         !
2920         IF T=1 THEN
2930             Rotation_state=-1
2940         ELSE
2950             Rotation_state=2
2960         END IF
2970         SELECT Rotation_state
2980         CASE =0
2990             Clear_crt
3000             PRINT TABXY(1,4);"When ready for measurement, press CONTINUE."
3010             BEEP
3020             PAUSE

```

```

3030         Clear_crt(3,16)
3040         PRINT TABXY(1,4);"Collecting data... "
3050     CASE ELSE
3060         PRINT TABXY(1,4);"Current angle is ";Current_angle;" degrees."
3070         Rotate_target      !
3080         WAIT 1
3090         Clear_crt(3,16)
3100         PRINT TABXY(1,4);"Collecting data ..."
3110     END SELECT
3120     FOR F=1 TO 3
3130         IF Meas_flag(F) THEN
3140             IF F=1 THEN
3150                 OUTPUT @Nwa;"POWE20;"
3160             ELSE
3170                 OUTPUT @Nwa;"POWE10;"
3180             END IF
3190             Freq_set(F)
3200             Freq_sw(F)
3210                 OUTPUT @Nwa;"GATECENT";VAL$(Gate_cent(F));"S;"
3220                 OUTPUT @Nwa;"GATESPAN";VAL$(Gate_span(F));"S;"
3230                 OUTPUT @Nwa;"GATEOON; WAIT;"
3240             FOR P=1 TO 4
3250                 Pol_sw(F,P)
3260                 OUTPUT @Nwa;"SING; FORM3; OUTPFORM;"
3270                 ENTER @Nwa_data1;Preamble,Bytes,Trace(*)
3280                 MAT Target_response(F,P,*)= Trace
3290             NEXT P
3300                 Nskh=Nskip+1
3310             FOR P=1 TO 4
3320                 FOR Nt=Nskh TO Npts STEP Nskip
3330                     Nst=INT(Nt/Nskip)
3340                     Target_data(F,T,P,Nst)=Target_response(F,P,Nt)
3350                 NEXT Nt
3360             NEXT P
3370         END IF
3380     NEXT F
3390     PRINT "# OF TRACES LEFT=",Ntrace-T
3400     NEXT T
3410     Store_file(Target_data(*),"GND",FNTime_stamp$,F)
3420     DISP "Surface target data saved."
3430     BEEP
3440     WAIT .1
3450     BEEP
3460     OUTPUT @Nwa;"CONT;"
3470     Exit_flag=1
3480     RETURN
3490     !
3500     !-----
3510     !
3520     Freq_set: GOSUB Deallocate_mtrx
3530         OFF KEY
3540         MAT Meas_flag_old= Meas_flag
3550         MAT Meas_flag= (0)
3560         Exit_flag=0
3570         ON KEY 0 LABEL "      L BAND      ",FNTrap_level GOSUB Set_l
3580         ON KEY 1 LABEL "      C BAND      ",FNTrap_level GOSUB Set_c
3590         ON KEY 2 LABEL "      X BAND      ",FNTrap_level GOSUB Set_x
3600         ON KEY 4 LABEL "      STORE      ",FNTrap_level GOTO Store_band
3610         ON KEY 5 LABEL "      ",FNTrap_level GOSUB Null
3620         ON KEY 6 LABEL "      ",FNTrap_level GOSUB Null
3630         ON KEY 7 LABEL "      ",FNTrap_level GOSUB Null

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```

3640         ON KEY 8 LABEL "           ",FNTrap_level GOSUB Null
3650         ON KEY 9 LABEL "   CANCEL   ",FNTrap_level GOTO Cancel_band
3660     LOOP
3670         EXIT IF Exit_flag=1
3680     END LOOP
3690 RETURN
3700 Set_l:   OFF KEY 0
3710         Meas_flag(1)=1
3720         F_disp=1
3730 RETURN
3740 Set_c:   OFF KEY 1
3750         Meas_flag(2)=1
3760         F_disp=2
3770 RETURN
3780 Set_x:   OFF KEY 2
3790         Meas_flag(3)=1
3800         F_disp=3
3810 RETURN
3820 Store_band: Print_banner4
3830         Exit_flag=1
3840         GOSUB Allocate_matrix
3850 RETURN
3860 Cancel_band: !
3870         MAT Meas_flag= Meas_flag_old
3880         Exit_flag=1
3890         GOSUB Allocate_matrix
3900 RETURN
3910 !
3920 !-----
3930 !
3940 Set_angle: !
3950         INPUT "Enter measurement angle: ",Angle
3960         Angle$=VAL$(Angle)&CHR$(179)&" " ! Degree sign.
3970         Print_banner4
3980 RETURN
3990 !
4000 !-----
4010 !
4020 Set_target: !
4030         LINPUT "Enter target type or name: ",Target$
4040         Target$=TRIM$(Target$)
4050         Target$=Target$&RPT$(" ",30-LEN(Target$))
4060         Print_banner4
4070 RETURN
4080 !
4090 !-----
4100 !
4110 Set_traces: !
4120         INPUT "Enter the number of traces (or angles) desired(>=3 ): ",Ntrace
4130         GOSUB Deallocate_mtrx
4140         GOSUB Allocate_matrix
4150         Print_banner4
4160 RETURN
4170 !
4180 !-----
4190 Set_points: !
4200         INPUT "Enter the number of sample points (Npts,201): ",Npts
4210         OUTPUT @Nwa;"POIN "&VAL$(Npts)&";"
4220         INPUT "Enter the data points to be stored (Ndata,10):",Ndata
4230         Nskip=INT(Npts/Ndata)
4240         Bytes=16*Ndata

```

```

4250          Print_banner4
4260          GOSUB Deallocate_mtrx
4270          GOSUB Allocate_matrix
4280 RETURN
4290 !
4300 !-----
4310 !
4320 Set_average: !
4330          INPUT "Enter averaging factor: ",Average_factor
4340          Print_banner4
4350 RETURN
4360 !
4370 !-----
4380 !
4390 Allocate_matrix: ! Allocate storage space for data.
4400          !
4410          System_memory=VAL(SYSTEM$("AVAILABLE MEMORY"))
4420          Avail_traces=MIN(Ntrace,INT(System_memory-50000-3*4*16.*Npts)/(3*4*16.*Npts))
4430          IF Avail_traces<Ntrace THEN
4440              BEEP
4450              PRINT TABXY(1,16);"Memory has capacity for only ";Avail_traces;" traces"
4460              PRINT "Press CONTINUE key to continue"
4470              PAUSE
4480              Ntrace=Avail_traces
4490          END IF
4500          ALLOCATE COMPLEX Trace(Npts),Target_response(3,4,Npts)
4510          ALLOCATE COMPLEX Target_data(3,Ntrace,4,Ndata)
4520 RETURN
4530 Deallocate_mtrx: ! Return to main program.
4540          !
4550          DEALLOCATE Target_response(*),Trace(*)
4560          DEALLOCATE Target_data(*)
4570 RETURN
4580 !
4590 !-----
4600 !
4610 Set_gates: ! Set gate centers and spans.
4620          !
4630          FOR F=1 TO 3
4640              IF Meas_flag(F) THEN
4650                  IF F=1 THEN
4660                      OUTPUT @Nwa;"POWE20;"
4670                  ELSE
4680                      OUTPUT @Nwa;"POWE10;"
4690                  END IF
4700                  Freq_set(F)
4710                  Freq_sw(F)
4720                  P=1
4730                  Pol_sw(F,P)
4740                  OUTPUT @Nwa;"TIMDTRANON; LOGM;"
4750                  OUTPUT @Nwa;"STAR 0NS; STOP 400NS; WAIT;"
4760                  OUTPUT @Nwa;"MARK1; MARKMAXI;"
4770                  OUTPUT @Nwa;"FORM3; OUTPACTI;"
4780                  ENTER @Nwa;Gate_cent(F)
4790                  OUTPUT @Nwa;"MARKOFF;"
4800                  OUTPUT @Nwa;"CONT;"
4810                  OUTPUT @Nwa;"GATESPAN";VAL$(Gate_span(F));"S;"
4820                  OUTPUT @Nwa;"GATECENT";VAL$(Gate_cent(F));"S;"
4830                  OUTPUT @Nwa;"KEY41; KEY59; KEY58; KEY59;"
4840                  LOCAL @Nwa
4850                  DISP "Adjust gate center to suit, and press CONTINUE."

```

```

4860         PAUSE
4870         OUTPUT @Nwa;"OUTPACTI;"
4880         ENTER @Nwa;Gate_cent(F)
4890         OUTPUT @Nwa;"GATESPAN";VAL$(Gate_span(F));";"
4900         OUTPUT @Nwa;"KEY41; KEY59; KEY58; KEY4;"
4910         LOCAL @Nwa
4920         DISP "Adjust gate span to suit, and press CONTINUE."
4930         PAUSE
4940         OUTPUT @Nwa;"OUTPACTI;"
4950         ENTER @Nwa;Gate_span(F)
4960     END IF
4970 NEXT F
4980 RETURN
4990 !
5000 !-----
5010 !
5020 Quit_fast_acq: ! End of program
5030 DISP "PROGRAM EXIT"
5040 GOSUB Deallocate_mtrx
5050 LOAD KEY "EDITKEY:MEMORY,0,1"
5060 STOP
5070 END
5080 !
5090 !*****
5100 !
5110 DEF FNAsk(Prompt$)
5120     OFF KEY
5130     DISP Prompt$;
5140     INPUT "",Yn$
5150     Yn$=UPC$(Yn$[1,1])
5160     SELECT Yn$
5170         CASE ="Y"
5180             RETURN 1
5190         CASE ="N",=""
5200             RETURN 0
5210         CASE ELSE
5220             RETURN 0
5230     END SELECT
5240 FNEND
5250 !
5260 !*****
5270 !
5280 DEF FNFileloc$(File$,Dir$)
5290     INTEGER C ! for the location of the ':' in Dir$ (minus 1)
5300     LET C=POS(Dir$,":")-1
5310     IF C<=0 THEN
5320         RETURN TRIM$(File$&Dir$)
5330     ELSE
5340         RETURN Dir$[1,C]&RPT$("/",Dir$[C,C]<>"/")&File$&Dir$[C+1,LEN(Dir$)]
5350     END IF
5360 FNEND ! Fileloc
5370 !
5380 !*****
5390 !
5400 DEF FNTime_stamp$(OPTIONAL Time_format)
5410     !
5420     DIM Time_digits$(4),Year_digits$(6)
5430     DIM Machine_time$(8),Machine_date$[11]
5440     REAL Timedate_now
5450     !
5460     Timedate_now=TIMEDATE

```

```

5470 Machine_date$=DATES$(Timedate_now)
5480 Machine_time$=TIMES$(Timedate_now)
5490 Time_digits$=Machine_time$[1,2]&Machine_time$[4,5]
5500 Year_digits$[1,2]=Machine_date$[10,11]
5510 IF Machine_date$[1,1]=" " THEN Machine_date$[1,1]="0"
5520 !
5530 SELECT Machine_date$[4,6]
5540 CASE ="Jan"
5550     Year_digits$[3,4]="01"
5560 CASE ="Feb"
5570     Year_digits$[3,4]="02"
5580 CASE ="Mar"
5590     Year_digits$[3,4]="03"
5600 CASE ="Apr"
5610     Year_digits$[3,4]="04"
5620 CASE ="May"
5630     Year_digits$[3,4]="05"
5640 CASE ="Jun"
5650     Year_digits$[3,4]="06"
5660 CASE ="Jul"
5670     Year_digits$[3,4]="07"
5680 CASE ="Aug"
5690     Year_digits$[3,4]="08"
5700 CASE ="Sep"
5710     Year_digits$[3,4]="09"
5720 CASE ="Oct"
5730     Year_digits$[3,4]="10"
5740 CASE ="Nov"
5750     Year_digits$[3,4]="11"
5760 CASE ="Dec"
5770     Year_digits$[3,4]="12"
5780 END SELECT
5790 !
5800 Year_digits$[5,6]=Machine_date$[1,2]
5810 SELECT NPAR
5820 CASE =0
5830     RETURN Year_digits$[5,6]&Time_digits$
5840 CASE =1
5850     IF Time_format=1 THEN
5860         RETURN Year_digits$&Time_digits$
5870     END IF
5880     IF Time_format=2 THEN
5890         RETURN Year_digits$[3,6]&Time_digits$
5900     END IF
5910 END SELECT
5920 FNEND
5930 !
5940 !*****
5950 !
5960 DEF FNTrap_level
5970     RETURN VAL(SYSTEM$("SYSTEM PRIORITY"))+1
5980 FNEND
5990 !
6000 !*****
6010 !
6020 SUB Config_and_poll
6030     COM /Paths/ @Nwa,@Nwa_data1,@Nwa_data2,Netwrk_analyzer,@Hpib,@Relay
6040     COM /System/ System_memory
6050 !
6060 ! Find out what's out there.
6070 !

```

```

6080     ALLOCATE Device_list$(0:31)[20]
6090     ALPHA PEN 4
6100     KBD LINE PEN 3
6110     KEY LABELS PEN 5
6120     Clear_crt
6130     Netwrk_analyzer=0
6140     ALLOCATE Na_ident$(80)
6150     System_memory=VAL(SYSTEM$("AVAILABLE MEMORY")) ! How much memory for RAM-DI
6160     PRINT "AVAILABLE MEMORY: ";System_memory;" BYTES"
6170     ON TIMEOUT 7,4 GOTO No_na      ! In case there is no network analyzer
6180 Is_na:  OUTPUT @Nwa;"FORM4; OUTPIDEN;"
6190         ENTER @Nwa_data2;Na_ident$
6200         IF POS(Na_ident$,"8510A") THEN Netwrk_analyzer=1
6210         IF POS(Na_ident$,"8510B") THEN Netwrk_analyzer=2
6220         IF POS(Na_ident$,"8720A") THEN Netwrk_analyzer=3
6230         IF POS(Na_ident$,"8720B") THEN Netwrk_analyzer=4
6240         IF POS(Na_ident$,"8753A") THEN Netwrk_analyzer=5
6250         IF POS(Na_ident$,"8753B") THEN Netwrk_analyzer=6
6260         LOCAL @Nwa
6270         PRINT
6280         PRINT Na_ident$
6290         PRINT Netwrk_analyzer
6300     ! Clear_crt
6310     PRINT
6320     PRINT
6330         IF Netwrk_analyzer=0 THEN
6340             !
6350             !
6360 No_na:  BEEP
6370         OFF CYCLE
6380         PRINT TABXY(1,5);"There is no active network analyzer on the HPiB bus
6390         PRINT TABXY(1,6);"Please check connections, and press the RUN key."
6400         PRINT
6410         PRINT TABXY(1,7);"If you DO NOT want to use a network analyzer,
        press the CONTINUE key."
6420         PAUSE
6430     END IF
6440     !
6450     !
6460 Check_hpib: ! Check the rest of the bus
6470     ON TIMEOUT 7,.01 GOTO Nothing
6480     !
6490     FOR Device=700 TO 731
6500         DISP "Checking for device at address: ";Device
6510         Device_list$(Device-700)="NOTHING"
6520         ASSIGN @What_is_it TO Device
6530         Outcome=SPOLL(@What_is_it)
6540         Device_list$(Device-700)="SOMETHING"
6550         PRINT Device,"SOMETHING HERE", "spoll: ";Outcome
6560         ASSIGN @What_is_it TO *
6570 Nothing: ! Skip to next device
6580     NEXT Device
6590     !
6600     OFF TIMEOUT 7
6610     ASSIGN @What_is_it TO *
6620     IF Device_list$(1)="SOMETHING" THEN
6630         DISP "Position the printer to Top-Of-Form and press CONTINUE..."
6640         PAUSE
6650         PRINTER IS PRT
6660         PRINT CHR$(27)&"&11L"; ! Set Page Breaks
6670         Printer_flag=1

```

```

6680         PRINTER IS CRT
6690         END IF
6700         DEALLOCATE Na_ident$
6710         DEALLOCATE Device_list$(*)
6720         ABORT @Hpib
6730 SUBEXIT
6740 SUBEND
6750 !
6760 ! *****
6770 !
6780 SUB Hp_bus_init
6790     COM /Paths/ @Nwa,@Nwa_data1,@Nwa_data2,Netwrk_analyzer,@Hpib,@Relay
6800     COM /Sys_1/ Freq$(*),Freq_cent(*),Freq_span(*),Gate_cent(*),Gate_span(*)
6810     COM /Sys_2/ Pol$(*),Polsw$(*)
6820     COM /System_config/ INTEGER Printer_flag,Debug_flag,Version$,
        Mode$,Out_type$,Sound$,Bell$,Target$,Ref_target$
6830     !
6840     ! This subroutine configures the HP-IB bus and presets the HP8510.
6850     !
6860     ASSIGN @Hpib TO 7
6870     ASSIGN @Nwa TO 716
6880     ASSIGN @Nwa_data1 TO 716;FORMAT OFF
6890     ASSIGN @Nwa_data2 TO 716;FORMAT ON
6900     ASSIGN @Relay TO 710
6910     REMOTE @Hpib
6920     ABORT @Hpib
6930     CLEAR @Nwa
6940     IF Debug_flag=1 THEN OUTPUT @Nwa;"DEBUON;"
6950     IF Debug_flag=0 THEN
6960         OUTPUT @Nwa;"DEBUOFF;"
6970         OUTPUT @Nwa;"TITL ""&Freq$(2)&" BAND   ""
6980     END IF
6990 SUBEND
7000 !
7010 !*****
7020 !
7030 SUB Series_init
7040 COM /System_config/ INTEGER Printer_flag,Debug_flag,Version$,Mode$,Out_type$,
        Sound$,Bell$,Target$,Ref_target$
7050     DIM Input$(80)
7060     !
7070     ! This subroutine prints a header for the printout and sets the system
7080     ! date and time.
7090     !
7100     IF Printer_flag=1 THEN PRINTER IS PRT
7110     PRINT CHR$(12)
7120     Set_clock
7130     LINPUT "ENTER MEASUREMENT SERIES TITLE",Input$
7140     Preface$=""&RPT$(" ",9)
7150     PRINT RPT$("*",70)
7160     PRINT Preface$&Input$
7170     LINPUT "ENTER OPERATOR NAME",Input$
7180     PRINT Preface$&Input$
7190     PRINTER IS CRT
7200     PRINT
7210     PRINT
7220     PRINT Preface$&"MEASUREMENT SERIES STARTED AT "&TIME$(TIMEDATE)
7230     PRINTER IS CRT
7240 SUBEND
7250 !
7260 !*****

```



```

7270 !
7280 SUB Set_clock
7290     OPTION BASE 1
7300     INTEGER I
7310     DIM Chrono$(12),Month$(12) [3]
7320     Exec_key$=CHR$(255)&CHR$(88)
7330     READ Month$(*)
7340     DATA "JAN","FEB","MAR","APR","MAY","JUN","JUL","AUG","SEP","OCT"
           ,"NOV","DEC"
7350     OUTPUT KBD;"SCRATCH KEY "&Exec_key$;
7360     Clear_crt
7370     PRINT "                               Current system date: ";DATE$(TIMEDATE)
7380     PRINT "                               Current system time: ";TIME$(TIMEDATE)
7390     PRINT
7400 Ask: LINPUT "Enter date and time (YMMDDHHMMss) : ",Chrono$
7410     IF Chrono$="" AND DATE$(TIMEDATE)<>" 1 Mar 1900" THEN
7420         Clear_crt
7430         SUBEXIT
7440     END IF
7450     Year$=VAL$(1900+VAL(Chrono$[1,2]))
7460     IF (VAL(Chrono$[3,4])<=0 OR VAL(Chrono$[3,4])>12) THEN
7470         BEEP
7480         PRINT "Incorrect month value."
7490         GOTO Ask
7500     END IF
7510     Year$=Month$(VAL(Chrono$[3,4]))&" "&Year$
7520     Year$=Chrono$[5,6]&" "&Year$
7530     SET TIMEDATE (DATE(Year$))
7540     IF (VAL(Chrono$[7,8]))>23 THEN
7550         BEEP
7560         PRINT "Incorrect hour value."
7570         GOTO Ask
7580     END IF
7590     Day$=Chrono$[7,8]&" ":"
7600     IF VAL(Chrono$[9,10])>59 THEN
7610         BEEP
7620         PRINT "Incorrect minute value."
7630         GOTO Ask
7640     END IF
7650     Day$=Day$&Chrono$[9,10]&" ":"
7660     IF (LEN(Chrono$)>10 AND LEN(Chrono$)=12) THEN
7670         IF VAL(Chrono$[11,12])>59 THEN
7680             BEEP
7690             PRINT "Incorrect seconds value."
7700             GOTO Ask
7710         END IF
7720         Day$=Day$&Chrono$[11,12]
7730     ELSE
7740         Day$=Day$&"00"
7750     END IF
7760     SET TIME TIME(Day$)
7770     Clear_crt
7780     SUBEXIT
7790     SUBEND
7800 !
7810 !*****
7820 !
7830 SUB Fix_error
7840 SELECT ERRN
7850 CASE ELSE
7860     PRINTER IS CRT

```

```

7870         PRINT "ERROR ";ERRN
7880         PRINT ERRM$
7890         PRINT " PROGRAM IS PAUSED.  FIX ERROR, IF POSSIBLE, AND CONTINUE."
7900         PAUSE
7910     END SELECT
7920     SUBEND
7930     !
7940     !*****
7950     !
7960     SUB Clear_crt(OPTIONAL INTEGER Start_line,Num_of_lines)
7970         !
7980         INTEGER I
7990         DIM Clear_line$(80)
8000         Clear_line$=""
8010         IF NPAR=0 THEN
8020             OUTPUT KBD;CHR$(255)&CHR$(75);
8030         ELSE
8040             PRINT TABXY(1,Start_line);"";RPT$(Clear_line$,Num_of_lines)
8050             PRINT TABXY(1,Start_line);"";
8060             SUBEXIT
8070         END IF
8080     SUBEND
8090     !
8100     !*****
8110     !
8120     SUB Print_banner1
8130         Clear_crt
8140         PRINT
8150         PRINT
8160         PRINT TABXY(3,16);"*****"
8170         PRINT TABXY(4,16);"*"
8180         PRINT TABXY(5,16);"*"
8190         PRINT TABXY(6,16);"*"
8200         PRINT TABXY(7,16);"*"
8210         PRINT TABXY(8,16);"*"
8220         PRINT TABXY(9,16);"*"
8230         PRINT TABXY(10,16);"*"
8240         PRINT TABXY(11,16);"*"
8250         PRINT TABXY(12,16);"*****"
8260     SUBEXIT
8270     SUBEND
8280     !
8290     !*****
8300     !
8310     SUB Print_banner2
8320     PRINT "Don't use Print_banner2."
8330     SUBEND
8340     !
8350     !*****
8360     !
8370     SUB Print_banner3
8380     PRINT "Don't use Print_banner3."
8390     SUBEND
8400     !
8410     !*****
8420     !
8430     SUB Print_banner4
8440     COM /Paths/ @Nwa,@Nwa_data1,@Nwa_data2,Netwrk_analyzer,@Hplib,@Relay
8450     COM /Constants/ Vel,Zero(*),Exec_key$
8460     COM /System_config/ INTEGER Printer_flag,Debug_flag,Version$,Mode$,
                        Out_type$,Sound$,Bell$,Target$,Ref_target$

```

```

8470 COM /Sys_1/ Freq$(*),Freq_cent(*),Freq_span(*),Gate_cent(*),Gate_span(*)
8480 COM /Sys_2/ Pol$(*),Polsw$(*)
8490 COM /Sys_3/ INTEGER F_disp,P_disp
8500 COM /Sys_4/ Drive_a$,Drive_b$,Drive_c$,INTEGER Preamble,Bytes
8510 COM /Sys_5/ INTEGER Nskip,Ndata
8520 COM /Sys_6/ Ref_angle,Angle,Angle$,Beam(*),INTEGER Npts,Ntrace,
      Average_factor
8530 COM /Sys_7/ INTEGER Meas_flag(*)
8540 !
8550 !
8560 OFF KEY
8570 Clear_crt
8580 PRINT
8590 PRINT
8600 PRINT "          PARAMETER          CURRENT VALUE"
8610 PRINT
8620 PRINT "          FREQUENCY          ";
8630 FOR F=1 TO 3
8640     IF Meas_flag(F) THEN PRINT Freq$(F)&" ";
8650 NEXT F
8660 PRINT "          "
8670 PRINT "          "
8680 PRINT "          ANTENNA ANGLE          "&Angle$
8690 PRINT "          TARGET TYPE          "&Target$
8700 PRINT "          MEASUREMENT MODE          "&Mode$&"          "
8710 ! PRINT "          CURRENT DISPLAY          "&Freq$(F_disp)&" "&Pol$(
8720 PRINT "          "
8730 PRINT "          # OF TRACES/SET          ";Ntrace
8740 PRINT "          # OF SAMPLE POINTS          ";Npts
8750 PRINT "          # OF DATA POINTS          ";Ndata
8760 PRINT "          (to be stored)          "
8770 PRINT "          # OF AVERAGES          ";Average_factor
8780 SUBEXIT
8790 SUBEND
8800 !
8810 !*****
8820 !
8830 SUB Store_file(COMPLEX Matrix(*),File_type$,Filename$,INTEGER F)
8840 !
8850     COM /Sys_1/ Freq$(*),Freq_cent(*),Freq_span(*),Gate_cent(*),Gate_span(*)
8860     COM /Sys_2/ Pol$(*),Polsw$(*)
8870     COM /Sys_5/ INTEGER Nskip,Ndata
8880     COM /Sys_6/ Ref_angle,Angle,Angle$,Beam(*),INTEGER Npts,Ntrace,
      Average_factor
8890     COM /Sys_7/ INTEGER Meas_flag(*)
8900     COM /System_config/ INTEGER Printer_flag,Debug_flag,Version$,Mode$,
      Out_type$,Sound$,Bell$,Target$,Ref_target$
8910 !
8920 !
8930     INTEGER Records_per_set,T
8940     REAL Bytes_per_set
8950     DIM Suffix$(2)
8960     ALLOCATE COMPLEX Trace(Ndata)
8970 !
8980 !
8990     DISP "Saving file."
9000     SELECT File_type$
9010     CASE ="SKY" ! Sky data.
9020         Bytes_per_set=16*Ndata
9030         Records_per_set=4*SUM(Meas_flag)*Ntrace
9040         IF SUM(Meas_flag)=3 THEN

```

```

9050         Suffix$="SA"
9060     ELSE
9070         FOR F=1 TO 3
9080             IF Meas_flag(F)=1 THEN
9090                 Mf=F
9100             END IF
9110         NEXT F
9120         Suffix$="S"&Freq$(Mf)
9130     END IF
9140         GOSUB Save_traces
9150     !
9160     !
9170 CASE ="REF"
9180     Bytes_per_set=16*Ndata
9190     Records_per_set=4*SUM(Meas_flag)*Ntrace
9200     IF SUM(Meas_flag)=3 THEN
9210         Suffix$="RA"
9220     ELSE
9230         FOR F=1 TO 3
9240             IF Meas_flag(F)=1 THEN
9250                 Mf=F
9260             END IF
9270         NEXT F
9280         Suffix$="R"&Freq$(Mf)
9290     END IF
9300         GOSUB Save_traces
9310     !
9320     !
9330 CASE ="MNT"
9340     Bytes_per_set=16*Ndata
9350     Records_per_set=4*SUM(Meas_flag)*Ntrace
9360     IF SUM(Meas_flag)=3 THEN
9370         Suffix$="MA"
9380     ELSE
9390         FOR F=1 TO 3
9400             IF Meas_flag(F)=1 THEN
9410                 Mf=F
9420             END IF
9430         NEXT F
9440         Suffix$="M"&Freq$(Mf)
9450     END IF
9460         GOSUB Save_traces
9470     !
9480     !
9490 CASE ="GND"
9500     Bytes_per_set=16*Ndata
9510     Records_per_set=Ntrace*4*SUM(Meas_flag)
9520     IF SUM(Meas_flag)=3 THEN
9530         Suffix$="GA"
9540     ELSE
9550         FOR F=1 TO 3
9560             IF Meas_flag(F)=1 THEN
9570                 Mf=F
9580             END IF
9590         NEXT F
9600         Suffix$="G"&Freq$(Mf)
9610     END IF
9620         GOSUB Save_traces
9630 END SELECT
9640 DEALLOCATE Trace(*)
9650 SUBEXIT

```

```

9660      !
9670      !
9680 Save_averaged: !
9690      ! Save the reference data file.
9700      !
9710      IF NOT Debug_flag THEN
9720          CREATE BDAT Filename$&Suffix$&Drive_c$,Records_per_set,Bytes_per_set
9730      END IF
9740      Base_record=0
9750      FOR F=1 TO 3
9760          IF Meas_flag(F)=1 THEN
9770              IF Debug_flag THEN
9780                  ASSIGN @Disc TO PRT
9790                  OUTPUT @Disc;"FILE: ",Filename$,Suffix$
9800                  OUTPUT @Disc USING Image_1;Version$,Freq_cent(F),Freq_span(F)
9810                  OUTPUT @Disc USING Image_2;Ndata,Average_factor
9820                  OUTPUT @Disc USING Image_3;Ref_target$,T
9830                  FOR P=1 TO 4
9840                      OUTPUT @Disc USING Image_4;Pol$(P),Gate_cent(F),Gate_span(F)
9850                      MAT Trace= Matrix(F,1,P,*)
9860                      OUTPUT @Disc;Trace(*)
9870                  NEXT P
9880              ELSE
9890                  ASSIGN @Disc TO Filename$&Suffix$&Drive_c$;FORMAT OFF
9900                  OUTPUT @Disc,Base_record+1;Version$,Freq_cent(F),Freq_span(F)
9910                  OUTPUT @Disc,Base_record+1;Ndata,Average_factor
9920                  OUTPUT @Disc,Base_record+1;Ref_target$,T
9930                  FOR P=1 TO 4
9940                      OUTPUT @Disc,Base_record+P;Pol$(P),Gate_cent(F),Gate_span(F)
9950                      MAT Trace= Matrix(F,1,P,*)
9960                      OUTPUT @Disc,Base_record+P;Trace(*)
9970                  NEXT P
9980                  Base_record=Base_record+4
9990              END IF
10000      END IF
10010      NEXT F
10020      ASSIGN @Disc TO *
10030      RETURN
10040      !
10050      !-----
10060      !
10070 Save_traces: !
10080      ! Save the ground target data file.
10090      !
10100      IF NOT Debug_flag THEN
10110          CREATE BDAT Filename$&Suffix$&Drive_c$,Records_per_set,Bytes_per_set
10120          Base_record=0
10130      END IF
10140      IF Debug_flag THEN
10150          ASSIGN @Disc TO PRT
10160          OUTPUT @Disc;"FILE: ";Filename$;Suffix$
10170          OUTPUT @Disc USING Image_5;Ndata,Ntrace
10180          OUTPUT @Disc USING Image_3;Target$
10190          FOR T=1 TO Ntrace
10200              FOR F=1 TO 3
10210                  IF Meas_flag(F)=1 THEN
10220                      OUTPUT @Disc USING Image_1;Version$,Freq_cent(F),Freq_span(F)
10230                      FOR P=1 TO 4
10240                          OUTPUT @Disc USING Image_4;Pol$(P),Gate_cent(F),Gate_span(F).
10250                          MAT Trace= Matrix(F,T,P,*)
10260                          OUTPUT @Disc;Trace(*)

```

```

10270         NEXT P
10280         END IF
10290         NEXT F
10300         NEXT T
10310     ELSE
10320         ASSIGN @Disc TO Filename$&Suffix$&Drive_c$;FORMAT OFF
10330 !         OUTPUT @Disc,1;Ndata,Ntrace
10340 !         OUTPUT @Disc,1;Target$
10350         FOR T=1 TO Ntrace
10360             FOR F=1 TO 3
10370                 IF Meas_flag(F)=1 THEN
10380 !                     OUTPUT @Disc,Base_record+1;Version$,Freq_cent(F),Freq_span(F)
10390                     FOR P=1 TO 4
10400 !                         OUTPUT @Disc,Base_record+P;Pol$(P),Gate_cent(F),Gate_span(F),T
10410                             MAT Trace= Matrix(F,T,P,*)
10420                             OUTPUT @Disc,Base_record+P;Trace(*)
10430                         NEXT P
10440                     Base_record=Base_record+4
10450                 END IF
10460             NEXT F
10470         NEXT T
10480     END IF
10490     ASSIGN @Disc TO *
10500 RETURN
10510 !
10520 !-----
10530 !
10540 Image_1:IMAGE (1X,12A,5X,"FREQ CENTER: ",2D.4D,5X,"FREQ SPAN: ",2D.4D)
10550 Image_2:IMAGE ("NUMBER OF POINTS: ",5D,5X,"NUMBER OF AVERAGES: ",5D)
10560 Image_3:IMAGE ("TARGET: ",30A,"GATING TARGET TYPE: ",2D)
10570 Image_4:IMAGE ("POLARIZATION: ",2A,5X,"GATE CENTER: ",SD.14DE,/,5X,"GATE SPAN
10580 Image_5:IMAGE ("NUMBER OF POINTS: ",5D,5X,"NUMBER OF TRACES: ",5D)
10590 Image_6:IMAGE (5X,SD.14DE,5X,SD.14DE)
10600 SUBEND
10610 !
10620 !*****
10630 !
10640 SUB Freq_set(INTEGER Ifreq)
10650 COM /Paths/ @Nwa,@Nwa_data1,@Nwa_data2,Netwrk_analyzer,@Hpib,@Relay
10660 COM /Sys_1/ Freq$(*),Freq_cent(*),Freq_span(*),Gate_cent(*),Gate_span(*)
10670 !
10680 ! This subroutine sets the transmit frequency for the HP8753.
10690 !
10700     SELECT Netwrk_analyzer
10710     CASE =3,=4,=5,=6
10720         OUTPUT @Nwa;"TIMDTRANOFF;"
10730     CASE =1,=2
10740         OUTPUT @Nwa;"FREQ;"
10750     END SELECT
10760     OUTPUT @Nwa;"CENT "&VAL$(Freq_cent(Ifreq))&" GHZ;"
10770     OUTPUT @Nwa;"SPAN "&VAL$(Freq_span(Ifreq))&" GHZ;"
10780 SUBEND
10790 !
10800 !*****
10810 !
10820 SUB Freq_sw(INTEGER Ifreq)
10830 COM /Paths/ @Nwa,@Nwa_data1,@Nwa_data2,Netwrk_analyzer,@Hpib,@Relay
10840     SELECT Ifreq
10850     CASE 1
10860         OUTPUT @Relay;"?*A1B2"
10870     CASE 2

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10880     OUTPUT @Relay;"?*A2B1"
10890     CASE 3
10900     OUTPUT @Relay;"?*B12"
10910     END SELECT
10920     WAIT .1
10930 SUBEND
10940 !
10950 !*****
10960 !
10970 SUB Pol_sw(INTEGER Ifreq,Ipol)
10980     COM /Paths/ @Nwa,@Nwa_data1,@Nwa_data2,Netwrk_analyzer,@Hpib,@Relay
10990     COM /Sys_1/ Freq$(*),Freq_cent(*),Freq_span(*),Gate_cent(*),Gate_span(*)
11000     COM /Sys_2/ Pol$(*),Polsw$(*)
11010     !
11020     ! This subroutine sets the transmit and receive polarization by
11030     ! sending the proper command over the HPiB to the polarization
11040     ! relays.
11050     !
11060     OUTPUT @Relay;Polsw$(Ifreq,Ipol)
11070     OUTPUT @Nwa;"TITL "" "&Freq$(Ifreq)&" BAND - "&Pol$(Ipol)&""""
11080     WAIT .1
11090 SUBEND
11100 !
11110 !*****
11120 !
11130 SUB Rotate_target
11140 OPTION BASE 1
11150 COM /Com4/ INTEGER Rotation_state,REAL Inc_angle,Current_angle,Start_angle,
           Stop_angle,Old_home_angle,INTEGER Sets_per_pos
11160 COM /Status/ INTEGER Sc,Connect_flg,E_flg,Debug_flg,Response${80}
11170 INTEGER Fs_flag,Ss_flag,Speed,Imc_status,Confirm_answer
11180     !
11190     !
11200     Confirm_answer=1
11210     Imc_status=0
11220     Debug_flg=0
11230     Fs_flag=-1
11240     Ss_flag=-1
11250     Clear_crt(3,16)
11260     !
11270     !
11280     SELECT Rotation_state
11290     CASE ==-1
11300         IF FNask("Do you wish to use the rotator?") THEN
11310             Connect_flg=0
11320             GOSUB Init_imc
11330             GOSUB Init_graph_pos
11340             GOSUB Manual_loop
11350             PRINT "Set Auto Mode Please....."
11360         ELSE
11370             Rotation_state=0
11380             GCLEAR
11390             GRAPHICS OFF
11400         END IF
11410     CASE =0
11420         SUBEXIT
11430     CASE =1
11440         GOSUB Check_position
11450         GOSUB Print_angles
11460         GOSUB Manual_loop
11470     CASE =2

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11480         GOSUB Check_position
11490         GOSUB Auto
11500     CASE =3
11510         GOSUB Check_position
11520         GOSUB Manual_loop
11530         GOSUB Auto
11540     CASE =4
11550         GOSUB Check_position
11560         GOSUB Go_home
11570     CASE =5
11580         GOSUB Check_position
11590         Rotation_state=1 ! Switch to manual mode.
11600     END SELECT
11610     SUBEXIT
11620     !
11630     !
11640 Init_imc: ! Initialize the IMC unit.
11650     GOSUB Check_4_fault
11660     PRINT TABXY(1,3);"INITIALIZING IMC"
11670     Clear_crt(4,15)
11680     Comm("4WB")           ! Set warm boot (clear flags).
11690     PRINT TABXY(1,4);"WB"
11700     Comm("4EB")           ! Clear IMC buffer.
11710     PRINT TABXY(1,4);"EB"
11720     Encoder_ratio=4096 ! 32000
11730     Comm("4ER"&VAL$(Encoder_ratio)) ! Load encoder ratio.
11740     PRINT TABXY(1,4);"ER"&VAL$(Encoder_ratio)
11750     IF FNask("Do you wish to set home at the current position?") THEN
11760         Comm("4RS",Confirm_answer)
11770         ENTER Response$;Old_home_angle
11780         Old_home_angle=Old_home_angle/93.3
11790         Comm("4PIZ0")           ! Set IMC at 0.
11800         PRINT TABXY(1,4);"PIZ"&RPT$(" ",LEN(VAL$(Encoder_ratio)))
11810         Comm("4PIA0")           ! Set IMC at 0.
11820         PRINT TABXY(1,4);"PIA"
11830         Current_angle=0
11840     END IF
11850     Comm("4SP100")           ! Set speed to (50pps).
11860     PRINT TABXY(1,4);"SP "&RPT$(" ",LEN(VAL$(Encoder_ratio)))
11870     Comm("4AC500")           ! Set acceleration (500pps^2).
11880     PRINT TABXY(1,4);"AC "
11890     Comm("4DC500")           ! Set deceleration (500pps^2).
11900     PRINT TABXY(1,4);"DC "
11910     GOSUB Check_position
11920     Rotation_state=1
11930     Clear_crt
11940     !
11950     !
11960     PRINT TABXY(1,4);"DONE INITIALIZING IMC"
11970     PRINT TABXY(1,5);"Turntable currently in manual mode."
11980     PRINT TABXY(1,6)
11990 Print_angles:
12000     PRINT TABXY(1,7);"Current angle is: ";Current_angle;" degrees."
12010     PRINT TABXY(1,8);"Starting angle is: ";Start_angle;" degrees."
12020     PRINT TABXY(1,9);"Stopping angle is: ";Stop_angle;" degrees."
12030     RETURN
12040     !
12050     !
12060 Manual_loop:! Main activation loop.
12070     LOOP
12080     ON KEY 0 LABEL "FAST SLEW CW ",FNTrap_level GOSUB Fs_cw

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12090     ON KEY 1 LABEL "FAST SLEW CCW ",FNTrap_level GOSUB Fs_ccw
12100     ON KEY 5 LABEL "SLOW SLEW CW  ",FNTrap_level GOSUB Ss_cw
12110     ON KEY 6 LABEL "SLOW SLEW CCW ",FNTrap_level GOSUB Ss_ccw
12120     ON KEY 2 LABEL "MANUAL CONTROL",FNTrap_level GOSUB Manual
12130     ON KEY 3 LABEL "TARGET GO HOME",FNTrap_level GOSUB Go_home
12140     ON KEY 4 LABEL "STOP ROTATION ",FNTrap_level GOSUB Stop_turn
12150     ON KEY 7 LABEL "SET AUTO MODE ",FNTrap_level GOSUB Set_auto
12160     ON KEY 8 LABEL "SET TARGET HOME",FNTrap_level GOSUB Set_position
12170     ON KEY 9 LABEL "RETURN          ",FNTrap_level GOTO Quit
12180     GOSUB Check_position
12190     END LOOP
12200 !
12210 !-----
12220 !
12230 Fs_cw:! Fast slew clockwise.
12240     IF Fs_flag<0 THEN
12250         Comm("4SP500")
12260         Comm("4SFN")
12270         Fs_flag=-1*Fs_flag
12280         Clear_crt(3,15)
12290         PRINT TABXY(1,15);"ROTATING CW (FAST)"
12300     ELSE
12310         Comm("4ST")
12320         Fs_flag=-1*Fs_flag
12330         Clear_crt(3,15)
12340         PRINT TABXY(1,15);"ROTATION STOPPED"
12350         GOSUB Check_position
12360     END IF
12370 RETURN
12380 !
12390 !-----
12400 !
12410 Fs_ccw:! Fast slew counterclockwise.
12420     IF Fs_flag<0 THEN
12430         Comm("4ST")
12440         Comm("4SP500")
12450         Comm("4SRN")
12460         Fs_flag=-1*Fs_flag
12470         Clear_crt(3,10)
12480         PRINT TABXY(1,15);"ROTATING CCW (FAST)"
12490     ELSE
12500         Comm("4ST")
12510         Fs_flag=-1*Fs_flag
12520         Clear_crt(3,15)
12530         PRINT TABXY(1,15);"ROTATION STOPPED"
12540         GOSUB Check_position
12550     END IF
12560 RETURN
12570 !
12580 !-----
12590 !
12600 Ss_cw:! Slow slew clockwise.
12610     IF Ss_flag<0 THEN
12620         Comm("4ST")
12630         INPUT "Speed?",Sp
12640         Comm("4SP"&VAL$(INT(Sp)))
12650         Comm("4SFN")
12660         Ss_flag=-1*Ss_flag
12670         Clear_crt(3,15)
12680         PRINT TABXY(1,15);"ROTATING CW (SLOW)"
12690     ELSE

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```

12700         Comm("4ST")
12710         Ss_flag=-1*Ss_flag
12720         Clear_crt(3,15)
12730         PRINT TABXY(1,15);"ROTATION STOPPED"
12740         GOSUB Check_position
12750     END IF
12760 RETURN
12770 !
12780 !-----
12790 !
12800 Ss_ccw:! Slow slew counterclockwise.
12810     IF Ss_flag<0 THEN
12820         INPUT "Speed?",Sp
12830         Comm("4SP"&VAL$(INT(Sp)))
12840         Comm("4SRN")
12850         Ss_flag=-1*Ss_flag
12860         Clear_crt(3,15)
12870         PRINT TABXY(1,15);"ROTATING CCW (SLOW)"
12880     ELSE
12890         Comm("4ST")
12900         Ss_flag=-1*Ss_flag
12910         Clear_crt(3,15)
12920         PRINT TABXY(1,15);"ROTATION STOPPED"
12930         GOSUB Check_position
12940     END IF
12950 RETURN
12960 !
12970 !-----
12980 !
12990 Manual: INPUT "ANGLE (IN DEGREES)=?",Inc_angle
13000         INPUT "SPEED? (~100--500 RECOMMENDED)",Speed
13010         Comm("4SP"&VAL$(Speed))
13020 Auto:   SELECT Rotation_state
13030         CASE =4
13040             GOSUB Go_home
13050             Rotation_state=2
13060             GOTO Auto
13070         CASE ELSE
13080             Angl2=Inc_angle*93.3
13090             Angl1=INT(Angl2)
13100             IF Angl2-Angl1>=.5 THEN Angl1=Angl1+1
13110             ! Current_angle=Current_angle+Inc_angle
13120             Inc_angle$=VAL$(Angl1)
13130             Comm("4IM"&Inc_angle$)
13140             Comm("4RFI")
13150         END SELECT
13160         Imc_status=0
13170         Clear_crt(3,7)
13180         PRINT TABXY(1,14);"ROTATING TARGET, PLEASE WAIT."
13190         !
13200         !
13210         WHILE NOT BIT(Imc_status,0)             ! Wait for motor to stop.
13220             Comm("4RS",Confirm_answer)
13230             ENTER Response$;Imc_status
13240             PRINT TABXY(1,15);DVAL$(Imc_status,2)
13250             GOSUB Check_position
13260             WAIT 1
13270         END WHILE
13280         Imc_status=0
13290         !
13300         !

```

```

13310      Clear_crt(3,16)
13320      PRINT TABXY(1,16);"CURRENT TARGET POSITION IS ";Current_angle;" DEGREES
13330      WAIT 2 ! Wait for target settling.
13340 RETURN
13350 !
13360 !-----
13370 !
13380 Stop_turn:Comm("4ST")
13390      WHILE NOT BIT(Imc_status,0)          ! Wait for motor to stop.
13400          Comm("4RS",Confirm_answer)
13410          ENTER Response$;Imc_status
13420          WAIT .1
13430      END WHILE
13440      Clear_crt(3,16)
13450      PRINT TABXY(1,15);"ROTATION STOPPED"
13460      GOSUB Check_position
13470      Imc_status=0
13480 RETURN
13490 !
13500 !-----
13510 !
13520 Set_auto: Comm("4SP500")
13530      GOSUB Check_position
13540      Clear_crt(3,16)
13550      PRINT TABXY(1,3);"Current starting angle: ";Start_angle;" degrees"
13560      PRINT TABXY(1,4);"Current increment angle: ";Inc_angle;" degrees"
13570      PRINT TABXY(1,5);"Current stopping angle: ";Stop_angle;" degrees"
13580      PRINT TABXY(1,6);"Current rotation speed: ";Speed
13590      PRINT TABXY(1,7);RPT$(" ",80)
13600      PRINT TABXY(1,8);"Rotator positioned at: ";Current_angle;" degrees"
13610      INPUT "Enter starting angle value (degrees): ",Start_angle
13620      INPUT "Enter increment angle (degrees): ",Inc_angle
13630      INPUT "Enter stopping angle (degrees): ",Stop_angle
13640      INPUT "Enter rotation speed of target (~500 recommended): ",Speed
13650      Speed=INT(Speed)
13660      Comm("4SP"&VAL$(Speed))
13670      IF ABS(Start_angle-Current_angle)>.1 THEN
13680          PRINT TABXY(1,9);RPT$(" ",80)
13690          PRINT TABXY(1,10);"Rotating target to starting angle..."
13700          Temp_angle=Inc_angle
13710          Inc_angle=Start_angle-Current_angle
13720          GOSUB Auto
13730          Inc_angle=Temp_angle
13740      END IF
13750      Rotation_state=2
13760      Clear_crt
13770      PRINT TABXY(1,20);"Turntable is in automatic mode. (press the RETURN
13780 RETURN
13790 !
13800 !-----
13810 !
13820 Set_position:INPUT "LOCK IN CURRENT TARGET POSITION AS REFERENCE POSITION?",Yn$
13830      IF Yn$="Y" OR Yn$="y" THEN
13840          Comm("4RS",Confirm_answer)
13850          ENTER Response$;Old_home_angle
13860          Old_home_angle=Old_home_angle/93.3
13870          Comm("4PIA0") ! Set absolute position to zero.
13880          Comm("4PIZ0") ! Set incremental position to zero.
13890          Current_angle=0
13900      ELSE
13910          PRINT "POSITION WAS NOT SET."

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```

13920             END IF
13930 RETURN
13940 !
13950 !-----
13960 !
13970 Go_home: IF Speed<200 THEN Speed=200
13980             Comm("4SP"&VAL$(Speed))
13990             Comm("4AM0") ! Move to zero absolute position.
14000             Comm("4RAN") ! Initiate movement.
14010             Comm("4MW") ! Make sure the move is completed.
14020             Imc_status=0
14030             Clear_crt(3,15)
14040             PRINT TABXY(1,14);"ROTATING TARGET TO HOME POSITION, PLEASE WAIT."
14050             WHILE NOT (BIT(Imc_status,0) AND BIT(Imc_status,5))
14060                 GOSUB Check_status
14070                 PRINT TABXY(1,15);"CURRENT STATUS: ";DVAL$(Imc_status,2)
14080                 GOSUB Check_position
14090                 WAIT .1
14100             END WHILE
14110             Clear_crt(3,16)
14120             PRINT TABXY(1,15);"TARGET AT HOME POSITION."
14130             GOSUB Check_position
14140             Imc_status=0
14150 RETURN
14160 !
14170 !-----
14180 !
14190 Check_status:! Keep an eye on the Whedco controller status.
14200             Comm("4RS",Confirm_answer)
14210             ENTER Response$;Imc_status
14220 RETURN
14230 !
14240 !-----
14250 !
14260 Check_position:! Get the current turnstile position in degrees.
14270             Comm("4RP",Confirm_answer)
14280             ENTER Response$;Motor_position
14290             Current_angle=Motor_position/93.3
14300             ! Current_angle=Current_angle+Inc_angle
14310             PRINT TABXY(1,16);"CURRENT TARGET POSITION IS ";Current_angle;" DEGREE
14320             GOSUB Draw_positions
14330 RETURN
14340 !
14350 !-----
14360 !
14370 Check_4_fault: ! Check the IMC for a fault condition and correct or
14380                 ! notify the user if necessary.
14390                 !
14400                 Comm("4FC",Confirm_answer)
14410                 ENTER Response$;Fault$
14420                 SELECT Fault$
14430                 CASE ="Power failure" ! Loss of power
14440                     RETURN
14450                 CASE ="Force DAC" ! Force DAC command was given
14460                     BEEP
14470                     PRINT "Force DAC command was given..."
14480                     DISP "Press CONTINUE to resume..."
14490                     PAUSE
14500                 RETURN
14510                 CASE ="Over-current" ! Over-current condition exists.
14520                     BEEP

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14530         PRINT "An over-current condition has been detected on the IMC."
14540         PRINT
14550         PRINT "Cycle the power to the IMC until the OV-CUR LED goes out"
14560         DISP "Press CONTINUE to reinitialize the IMC"
14570         PAUSE
14580         GOSUB Init_imc
14590         RETURN
14600         END SELECT
14610         RETURN
14620 !
14630 !-----
14640 !
14650 Init_graph_pos: ! Creates a graphical depiction of where the target is.
14660         !
14670         GINIT
14680         GCLEAR
14690         GRAPHICS ON
14700         SHOW 0,100,0,100
14710         PENUP
14720         MOVE 90,70
14730         PEN 1      ! Draw circle
14740         POLYGON 12,360,360
14750         PENUP
14760         MOVE 90,70 ! Draw old home orientation.
14770         PEN 2
14780         DRAW 90+11*COS(Old_home_angle),70-11*SIN(Old_home_angle)
14790         PENUP
14800         MOVE 90,70 ! Draw current home orientation.
14810         PEN 4
14820         DRAW 90,58
14830         PENUP
14840         MOVE 90,70 ! Draw current target orientation.
14850         PEN 3
14860         X_pos=90+11*COS(Current_angle)
14870         Y_pos=70-11*SIN(Current_angle)
14880         DRAW X_pos,Y_pos
14890 RETURN
14900 !
14910 !-----
14920 !
14930 Draw_positions: ! Draws out the angular orientations.
14940         MOVE 90,70 ! Draw old home orientation.
14950         PEN 2
14960         DRAW 90-11*SIN(Old_home_angle),70-11*COS(Old_home_angle)
14970         PENUP
14980         MOVE 90,70 ! Draw current home orientation.
14990         PEN 4
15000         DRAW 90,58
15010         PENUP
15020         DISABLE
15030         MOVE 90,70 ! Draw current target orientation.
15040         PEN -3
15050         DRAW X_pos,Y_pos
15060         MOVE 90,70
15070         PEN 3
15080         X_pos=90-11*SIN(Current_angle)
15090         Y_pos=70-11*COS(Current_angle)
15100         DRAW X_pos,Y_pos
15110         PENUP
15120         ENABLE
15130 RETURN

```

```

15140 !
15150 !-----
15160 !
15170 Quit: !
15180     SUBEXIT
15190 SUBEND
15200 !
15210 !*****
15220 !
15230 SUB Comm(C$,OPTIONAL INTEGER Confirm_answer)
15240 !
15250 ! PROGRAM MODULE:  Comm
15260 !
15270 ! PURPOSE:          Modified version of the Comm module to be used
15280 !                  for direct two way communication with the WHEDCO
15290 !                  IMC stepping motor controller.
15300 !
15310 ! UPDATE:  3.0      Version 3.0 checks to see if the card being used
15320 !                  is the HP98628A (Datacomm) or the HP98626A (Serial).
15330 !                  Depending on which card is used, the appropriate
15340 !                  registers are selected.
15350 !
15360     OPTION BASE 1
15370     COM /Status/ INTEGER Sc,Connect_flg,E_flg,Debug_flg,Response$
15380     INTEGER Baud_rate,B,Num_chars,Response_flg,Index1
15390     DIM Input$[256],Term$[256],In$[256] BUFFER,From_232$[256]
15400     DIM Num_chars$[6],Num_ltrs$[6],Out$[256] BUFFER
15410     DIM White_print$[1],Crlf$[2]
15420     IF Debug_flg THEN PRINT TABXY(1,1);"ENTERING Comm "
15430     ON ERROR GOSUB Error
15440     !
15450     !
15460     !
15470     IF Connect_flg THEN After_init
15480     Sc=20
15490     ASSIGN @Find_it TO Sc;RETURN Outcome
15500     IF Outcome=0 THEN
15510         ASSIGN @Find_it TO *
15520         CONTROL Sc,0;1           ! Reset RS-232 interface.
15530         CONTROL Sc,3;1           ! Async link protocol.
15540         CONTROL Sc,0;1           ! Set Async toggle.
15550         CONTROL Sc,8;1+2         ! Set RTS and DTR lines.
15560         CONTROL Sc,16;0          ! Disable connection timeout.
15570         CONTROL Sc,17;0          ! Disable no activity timeout.
15580         CONTROL Sc,18;0          ! Disable NO CARRIER timeout.
15590         CONTROL Sc,19;0          ! Disable transmit timeout.
15600         CONTROL Sc,20;14         ! TX baud speed = 9600
15610         CONTROL Sc,21;14         ! RX baud speed = 9600
15620         CONTROL Sc,22;0          ! No handshake with Whedco.
15630         CONTROL Sc,23;0          ! No hardwired handshake.
15640         CONTROL Sc,34;2          ! 7 bits/character.
15650         CONTROL Sc,35;0          ! 1 stop bit.
15660         CONTROL Sc,36;1          ! ODD parity.
15670         Connect_flg=1
15680     ELSE
15690         Sc=8
15700         ASSIGN @Find_it TO *
15710         ASSIGN @Find_it TO Sc;RETURN Outcome
15720         IF Outcome<>0 THEN
15730             PRINT "RS-232 card not installed. Please install and reboot."
15740             ASSIGN @Find_it TO *

```

```

15750         STOP
15760         END IF
15770         ASSIGN @Find_it TO *
15780         RESET Sc
15790         CONTROL Sc,0;1           ! Reset the RS-232 interface.
15800         CONTROL Sc,3;Baud_rate   ! Set the baud rate.
15810         CONTROL Sc,4;8+2         ! UART 8 bits/char.  ODD parity.
15820         CONTROL Sc,5;3           ! UART DTR line active.
15830         CONTROL Sc,12;128+32+16 ! Disable CD,DSR,CTS
15840         STATUS Sc,3;B            ! Confirm speed to user.
15850         Connect_flg=1
15860     END IF
15870 After_init:
15880     White_print$=CHR$(136)
15890     Crlf$=CHR$(13)&CHR$(10)
15900     PRINT CHR$(128)&CHR$(136);    ! Set up the screen.
15910     ASSIGN @Screen TO CRT
15920     ASSIGN @Kbd TO KBD
15930     ASSIGN @Rx TO BUFFER In$
15940     ASSIGN @Tx TO BUFFER Out$
15950     ASSIGN @Uart_out TO Sc
15960     ASSIGN @Uart_in TO Sc
15970     Response_flg=0                ! Reset command acknowledge flag.
15980     Response$=""                  ! Null out response string.
15990     !
16000     !
16010     ENABLE INTR Sc                ! Enable interrupt on card.
16020     TRANSFER @Tx TO @Uart_out;CONT ! Enable transfer buffers.
16030     TRANSFER @Uart_in TO @Rx
16040     ON INTR Sc,FNTrap_level GOSUB Read_loop ! Process card interrupts.
16050     IF C$<>" " THEN
16060         GOSUB Send_com             ! Send command out to controller.
16070     ELSE
16080         GOTO Quit                 ! If null command, exit quick.
16090     END IF
16100     !
16110     !
16120     !
16130 Wait_for_it:WHILE NOT Response_flg ! Waiting for acknowledgement.
16140     GOSUB Read_loop
16150     IF NPAR=2 THEN                ! We are waiting for data to be
16160     ! sent by the Whedco controller.
16170         LOOP
16180             GOSUB Read_loop
16190             IF (POS(Response$,"*")) THEN
16200                 Response$=Response$[POS(Response$,"*"),LEN(Response$)]
16210                 Response_flg=1
16220             END IF
16230             EXIT IF ((Response_flg=1) AND (POS(Response$,Crlf$)))
16240             END LOOP
16250         ELSE
16260             WHILE NOT ((POS(Response$,"*")) OR (POS(Response$,"?")))
16270                 GOSUB Read_loop
16280             END WHILE
16290             Index1=POS(Response$,"*")
16300             IF Index1=0 THEN ! Must be a "?" (Whedco command error)
16310                 ! Must be a "?" (Whedco command error).
16320                 E_flg=1          ! Notify via error flag.
16330                 Response_flg=1
16340             ELSE
16350                 ! Normal command interpretation.

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16360             E_flg=0
16370             Response_flg=1
16380             END IF
16390             END IF
16400             END WHILE
16410             GOTO Quit
16420             !
16430             !
16440 Read_loop:! Read in serial data from Whedco.
16450             !
16460             STATUS @Rx,4;Num_chars             ! Number of characters to
16470             IF Num_chars=0 THEN RETURN         ! receive, if 0 try again.
16480             Num_chars$="#",&VAL$(Num_chars)&"A" ! Set up the IMAGE for ENTER.
16490             ENTER @Rx USING Num_chars$;From_232$ ! Transfer contents.
16500             Response$=Response$&From_232$     ! Build up dialogue.
16510             RETURN                             ! Update pointers.
16520             !
16530             !
16540 Send_com:Term$=Crlf$[1,1]&C$&Crlf$
16550             Num_ltrs$="#",&VAL$(LEN(Term$))&"A"
16560             OUTPUT @Tx USING Num_ltrs$;Term$
16570             Term$=""
16580             RETURN
16590             !
16600             !
16610 Quit: OFF ERROR
16620             STATUS @Tx,10;Stat
16630             STATUS @Rx,4;Num_bytes
16640             ABORTIO @Uart_out
16650             ASSIGN @Tx TO *
16660             CONTROL @Rx,8;0
16670             STATUS @Rx,10;Stat
16680             STATUS @Rx,4;Num_bytes
16690             ABORTIO @Uart_in
16700             ASSIGN @Rx TO *
16710             DISABLE INTR Sc
16720             SUBEXIT
16730             !
16740             !
16750 Error:PRINT "HANDLING Comm ERROR"
16760             IF ERRN<>167 THEN Other_error
16770             IF Sc=8 THEN ! Process the simple card.
16780                 STATUS Sc,10;Uart_error
16790                 IF BIT(Uart_error,1) THEN Overrun
16800                 IF BIT(Uart_error,2) THEN Parity
16810                 IF BIT(Uart_error,4) THEN Break1
16820                 IF BIT(Uart_error,3) THEN Framing
16830                 E_flg=1
16840                 PAUSE
16850                 RETURN
16860             ELSE
16870                 PRINT ERRM$
16880                 E_flg=1
16890                 PAUSE
16900                 RETURN
16910             END IF
16920             !
16930             !
16940 Other: PRINT "UART error status: ";Uart_error
16950             E_flg=1
16960             RETURN

```



```
16970 !
16980 !
16990 Overrun:PRINT "Overrun"
17000     E_flg=1
17010     RETURN
17020 !
17030 !
17040 Parity: PRINT "Parity"
17050     E_flg=1
17060     RETURN
17070 !
17080 !
17090 Break1: PRINT "Break"
17100     E_flg=1
17110     RETURN
17120 !
17130 !
17140 Framing:PRINT "Framing"
17150     E_flg=1
17160     RETURN
17170 !
17180 !
17190 Other_error:PRINT "Error message: ";ERRM$
17200     PAUSE
17210     E_flg=1
17220     SUBEXIT
17230 !
17240 !
17250 SUBEND
17260 !
17270 !*****
17280 !
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