

A gridded retarding pressure sensor for ion and neutral particle analysis in flowing plasmas

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A Gridded Retarding Pressure Sensor for Ion and Neutral Particle Analysis in Flowing Plasmas

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Abstract

Although classical electrostatic plasma probes can provide detailed information regarding electrons and ions, these devices rely on charge-carrying particles both for energy filtering and detection. Neutral particles are transparent to such diagnostics. In light of this a new probe design was developed to provide a measurement of the neutral particle flux component of a flowing plasma. The simple design adopted for this probe utilizes a set of electrostatic retarding grids to remove the charged particles from the flow. The neutral particles are sensed within the tube of an off-the-shelf hot cathode ionization gauge. Additionally, by varying the potential on the retarding grids, this same probe configuration can be used to measure the ion energy distribution. Preliminary data from this probe was taken in the exhaust plume of a Hall-current accelerator designed for space propulsion. This probe was found to be a very simple, accurate diagnostic tool for such use. By using this probe in combination with inter-related diagnostics the spatial distribution of energetic charge exchange neutrals was quantified in the plume of an SPT-100 thruster.

Nomenclature

A_{col}	= Area of RPA current collector (m^2)
e	= Elementary charge (coul)
F_{esc}	= Escaping particle velocity dist.
f	= Ion energy distribution function
I	= Collected current (A)
k	= Boltzmann constant
m_i	= Mass of ion (kg)
n_i	= Density of ions (m^{-3})
n_n	= Density of neutrals (m^{-3})
n_m	= Measured density (m^{-3})
P_m	= Measured pressure (Pa)
T_m	= Measured temperature (K)
u_n	= Velocity of neutrals (m/s)
u_m	= Measured particle velocity (m/s)
V	= Ion retarding voltage (V)
$\langle x \rangle$	= Average value of quantity "x"

Introduction

Electrostatic probes have widespread application in plasma diagnostics. These probes, such as Langmuir probes and Retarding Potential Analyzers (RPA's), rely on the electric charge of ions and electrons for both energy filtering and particle detection. Neutral particles, however, are unaffected by electric or magnetic fields and are difficult to detect.

Although neutral particles are very elusive, they play an important role in many plasma processes. In electrostatic and electromagnetic space propulsion devices, for example, neutral propellant is ionized and accelerated by electromagnetic fields. Any propellant that is not ionized will therefore not be accelerated and will be unavailable for thrust. Optimizing the efficiency of such

thrusters requires knowledge of the neutral propellant distribution. In addition to thruster optimization, neutral particle analysis is required to determine the exhaust plume distortion due to ground-based test facilities.¹ In any ground-based vacuum facility there will exist a low density background neutral gas that is due to pump limitations. These neutral particles may collide with energetic plume particles through charge exchange collisions; such a collision produces an energetic neutral and a slow ion.

Neutral particle transport analysis can also be valuable to nuclear fusion research. Although most hot plasmas are almost completely ionized, there are neutral atoms continually being formed in the plasma interior. Because these particles travel straight across any confining magnetic field, significant numbers can escape from the plasma. These particles will carry information out of the plasma concerning the state of the inner region.²

The only directly detectable properties carried by neutral particles are momentum and energy. The momentum of the particles, represented by the particle flux, can be detected with a highly sensitive vacuum pressure gauge. However, the plasma flow of interest may have a very large ratio of charged-to-neutral particles. These charged particles also carry momentum and energy in addition to charge. The ions must therefore be filtered out by an electrostatic field prior to particle detection.

The objective of this investigation was to design, test, and evaluate a new probe capable of detecting neutral particle properties. In addition, the probe was designed to perform a dual function as an energy analyzer for energetic ions.

Description of Probe

The Neutral Particle Flux probe, or NPF, is essentially a hybrid between a vacuum pressure sensor and a traditional gridded RPA. An off-the-shelf MKS tubulated hot

cathode ionization gauge was used as the detector for this probe. Affixed to the inlet of the gauge tube was a set of four grids. By varying the potential on these grids particles could be selectively admitted or denied access to the tube collector. A schematic of the probe is shown in Figure 1.

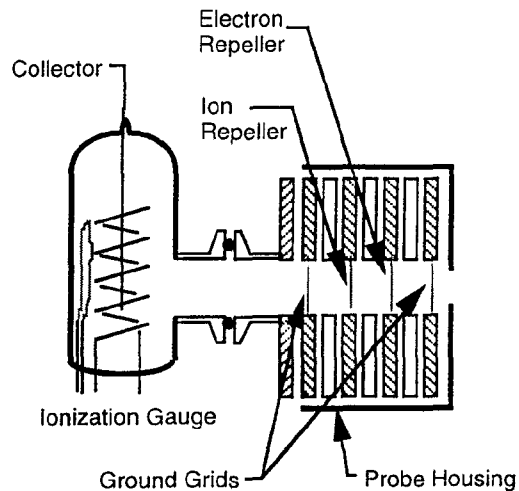


Figure 1. Schematic of NPF probe construction.

The probe body was constructed of stainless steel; the inlet aperture was 1.3cm in diameter. Each of the four grid pieces were formed by sandwiching 30 line/cm electroformed nickel mesh between two 0.8mm thick stainless steel washers using electrically conducting epoxy. The insulating spacers between the grid frames were constructed out of 6.4mm thick Teflon.

The first grid and probe body were grounded; this provided EMI shielding for the ionization gauge from the plasma. The second grid was biased approximately -40V. This was sufficient to repel all plasma electrons from the sensing volume. The third grid was connected to a variable high voltage power supply, capable of 0 to 500V. By varying the potential of this grid, selective energy filtering of positive ions was achieved as in a traditional RPA. By setting this ion retarding grid to the maximum positive potential all ions were prevented from entering the gauge tube. In this mode, the probe output was sensitive only to neutral particles. The fourth and final grid was grounded. This prevented any stray electric

fields induced by the ion retarding grid from interfering with normal hot cathode gauge performance and calibration.

Diagnostic Theory

The probe designed for this study was capable of measuring both neutral particle flux and ion energy distribution. The theoretical relations necessary for data reduction are presented in this section.

Ion Energy Distribution. The technique of using electrostatic grids to selectively retard ions is well known in plasma diagnostics.² The NPF probe functioned as a typical RPA albeit with a non-typical current collector. Instead of using a flat surface to directly collect ion current as is typical in traditional RPA analysis, the NPF probe collected ion current along the length of the existing hot cathode gauge collector wire.

The relationship between collected current and ion retarding potential for an RPA is given as

$$I(V) = \frac{e^2 n_i A_{\text{probe}}}{m_i} \int_V^{\infty} f(V') dV' \quad [1]$$

where V is the voltage of the ion retarding grid and f is the ion energy distribution function. In equation (1) A_{coll} is the area of the ion current collector. In traditional RPA's a planar collector is used providing a clear definition of this area. For the NPF probe, this area is not well defined; the narrow collector wire of the ion gauge tube most likely does not absorb current over its entire surface area. However, this has little impact on our desired measurement. By differentiating equation (1) it is easily shown that

$$-\frac{dI}{dV} = \frac{e^2 n_i A_{\text{probe}}}{m_i} f(V). \quad [2]$$

Since we are only interested in determining the shape of $f(V)$, $-dI/dV$ will be directly proportional to our desired function regardless of the value of A_{coll} .

Neutral Particle Flux. By dialing the ion retarding grid to a suitably large positive voltage all ions will be repelled from the sensing volume. In this mode the NPF probe becomes a detector for neutral particle flux.

As an output, the NPF probe measures the pressure within the ionization gauge tube. In order to glean useful information from this a relation must be derived relating the measured pressure to the free stream neutral particle flux. This can be accomplished by using free molecular theory to establish a flux balance.

At equilibrium conditions, the total flux of neutral particles entering the ionization gauge tube must be equal to the total flux of particles exiting the tube. Specifically,

$$n_n \langle u_n \rangle_{\text{in}} = n_m \langle u_m \rangle_{\text{out}} \quad [3]$$

The flux of particles leaving the tube can be theoretically evaluated. In order to do this we assume the ionization gauge tube can be modeled as a gas reservoir containing a macroscopically stagnant equilibrium gas. Additionally we neglect the effects of the finite length entrance to the tube and model it as an ideal orifice. If the mean free path of the gas within the tube is larger than the dimensions of the inlet orifice, then intermolecular collisions can be neglected. This is an accurate approximation: at a measured pressure of 1×10^{-4} torr within the tube the mean free path is greater than 10cm while the inlet orifice has a diameter of 1cm. The escaping molecules, therefore, move through the entering molecules without interaction, and the quantities can be evaluated separately.

Denoting the positive z direction exiting the orifice and assuming the gas in the ion gauge tube to be Maxwellian, the velocity distribution of molecules escaping from the ideal orifice is

$$F_{\text{esc}} = n_m \left(\frac{m}{2\pi k T_m} \right)^{3/2} e^{-\frac{m(v_x^2 + v_y^2 + v_z^2)}{2kT_m}} \quad [4]$$

for $v_z > 0$, and $F_{esc} = 0$ otherwise. The escape flux of particles is then integrated as

$$\begin{aligned} n_m \langle u_m \rangle &= \int_{-\infty}^{\infty} dv_x \int_{-\infty}^{\infty} dv_y \int_0^{\infty} dv_z v_z F_{esc}(v) \\ &= n_m \sqrt{\frac{kT_m}{2\pi m}} \end{aligned}$$

Substituting this into equation (3) yields an expression relating the neutral particle flux to measured quantities:

$$n_n \langle u_n \rangle_{in} = n_m \sqrt{\frac{kT_m}{2\pi m}} \quad [5]$$

By using the ideal gas law to relate n_m to the measured pressure, P_m , we can close the relationship between NPF sensor output and neutral particle flux:

$$n_n \langle u_n \rangle_{in} = \frac{P_m}{\sqrt{2\pi m k T_m}} \quad [6]$$

Experimental Procedure

This section presents a description of the experimental set up, test procedure, and parameters used in the investigation.

Facilities. All tests reported here were performed in the Plasmadynamics and Electric Propulsion Laboratory (PEPL) at the University of Michigan. The centerpiece of this laboratory is a large 6m dia. by 9m long space simulation vacuum chamber capable of base pressures near 2×10^{-5} torr. A detailed description of this facility has been published elsewhere.³ All data was acquired using National Instruments signal conditioning equipment in conjunction with a computer controlled A/D system.

The flowing plasma for these tests was provided by two similar devices. Both of these devices were closed drift Hall current accelerators used for electric propulsion. Extensive diagnostics were performed on a Fakel-built SPT-100 flight model thruster

powered by a Space Systems/Loral Power Processing Unit (PPU). This device has been extensively documented in literature.^{1,4,5,6} Nominal operating conditions utilized for this device consisted of 300V discharge voltage at 4.5A. Propellant was Xe at 56 SCCM total flow rate with a 10% cathode split. The second device used to produce a plasma was a laboratory model of the SPT-100 built by the Moscow Aviation Institute. Nearly identical in design, the use of this laboratory model allowed adjustable settings of discharge voltage, current, and propellant flow rates.

Set-up and Procedure. In order to troubleshoot initial design iterations of the NPF probe the laboratory model thruster was used. For these tests, the NPF probe was mounted to a remote positioning system within the vacuum chamber (described in ref. 3) and placed at various locations within the thruster plume. Aspects such as cable design, EMI minimization, and thruster thermal impacts were investigated in these tests.

Once the operational difficulties with the probe had been corrected, a comprehensive suite of diagnostics was performed using the NPF and other probes on the flight model SPT-100; the results of this investigation are also presented at this conference.¹ For these investigations plume surveys of the SPT-100 were taken at 0.5m radial distance from the thruster exit plane as well as 1.0m.

Probe Operation. The hot cathode ionization gauge sensor on the NPF was controlled using a standard MKS Model 919 controller. The 919 was calibrated for Xe in units of Torr. Both ion and electron retarding potentials were provided by laboratory power supplies.

During operation of the NPF in neutral flux mode the 919 controller required no modifications. However, some difficulties were encountered during operation in RPA mode, which will be described later in the *Results* section. These difficulties required that the ionization gauge grid and filament be de-activated, while still monitoring the

current from the collector. This was accomplished through factory consultation by simply re-positioning three internal dip switches within the controller.

Results and Discussion

Ionization Gauge Verification. In order to allow remote positioning of the NPF probe within the volume of the vacuum chamber a custom set of power cables were constructed. These cables, fed through the vacuum chamber wall, had an overall length of approximately 20m. The NPF probe ionization gauge operation was verified by simply comparing the pressure measured at vacuum (no plasma flow) to the pressure measured by existing ionization gauges on the tank. During this initial verification it was discovered that the NPF probe ionization gauge indicated a measured pressure of around 3×10^{-7} torr while the tank pressure was actually at 4×10^{-5} torr. The cause of this was identified as signal power loss in the cables. This problem was remedied by shortening the cables to approximately 10m. After this configuration change the NPF ionization gauge confirmed the pressure measured by the tank gauges. Pressure measured by the ionization gauge showed absolutely no change when voltage was applied to either ion or electron retarding grids.

RPA Mode. The ability of the NPF sensor to measure the ion energy distribution function was tested by operating the sensor in RPA mode. This testing was performed in the plume of the laboratory model Hall thruster.

The electron retarding grid was set to a potential of -30V; this value was suitable to repel all plasma electrons in this flow.⁷ By adjusting the potential of the ion retarding grid ions were permitted to enter the ionization gauge tube according to equation (2). The 919 controller converted the current measured by the ionization gauge collector wire into a proportional equivalent pressure in units of torr. Initial tests revealed an interesting complication due to the ionization gauge. The data from this RPA-type trace of

pressure (current) vs. ion retarding potential is shown as Figure 2.

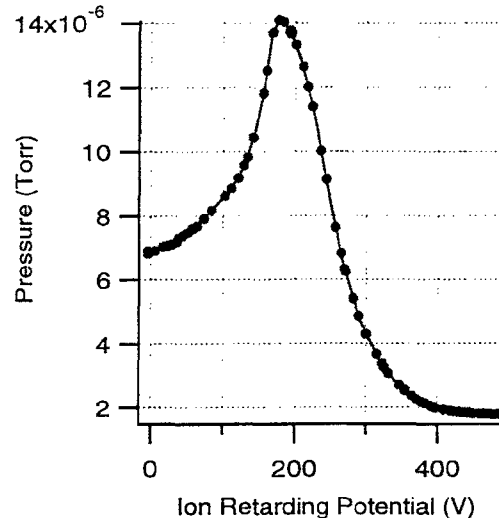


Figure 2. NPF Sensor output in RPA mode showing effect of ionization gauge grid bias voltage. Laboratory thruster used with $V_d=200V$, $I_d=4.2A$. Probe 15 deg off thruster centerline 73cm radius from exit plane.

The behavior of the data in Figure 2 is due to the field configuration within the ionization gauge tube. The ionization gauge tube senses pressure (usually due to neutral particles) by ionizing a fraction of the particles through electron impact ionization, then sensing the ion current with the collector. The collector is encircled by a helical grid which is used to collect electrons. In order to ensure efficient ion current collection the gauge grid voltage is maintained at 180V while the collector is held at ground. This causes the collector to have the lowest potential. When the ionization gauge tube is exposed to an incident flow of plume ions in addition to the tube-formed ions, the interaction of the gauge grid voltage with the NPF ion retarding grid causes the peak shown in Figure 2. This can be understood by examining Figure 3.

When the ion retarding grid has a voltage less than 180V it has no effect on the incident plume ions that are capable of reaching the collector; the ion gauge grid voltage repels all ions with energies less than 180V from direct impingement on the

collector. The ion retarding grid does, however have an effect on the measured pressure due to its interaction with ions formed within the tube due to electron impact ionization. Some ions formed immediately outside of the gauge grid will have sufficient energy to overcome the ion retarding grid field and will exit the tube undetected. However, some ions formed outside of the gauge grid will not have enough energy to overcome this barrier. These ions will be reflected back in to the tube. This will effectively increase the density and hence the measured pressure within the tube. At an ion retarding potential of 180V, no ions formed within the tube will be capable of exiting the tube and hence the measured pressure will be a maximum. As the potential of the ion retarding grid is raised above 180V the number of plume ions capable of entering the gauge will be reduced as in a traditional RPA, and the measured pressure will again decrease.

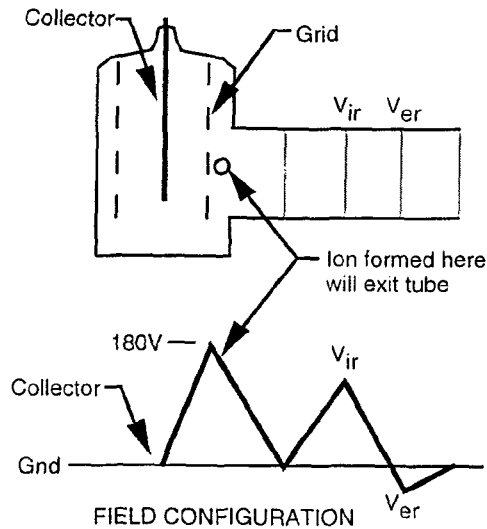


Figure 3. Schematic showing interaction of electric fields in ionization gauge with the NPF grids.

The sensor, then, operates like a "two way" RPA due to the interaction of the fields. The portion of the curve in Figure 2 below 180V represents an RPA characteristic for ions formed within the tube, while the portion of the curve above 180V represents an RPA characteristic for the plume ions.

In order to eliminate this adverse field interaction it was necessary to de-activate the filament and grid mechanisms within the ionization gauge tube during operation of the NPF in RPA mode. According to manufacturer consultation, this was accomplished by re-positioning three dip switches within the 919 controller. The controller, thus configured, simply operated as a high resolution ammeter monitoring the ion current incident on the collector wire. The results of an RPA trace taken in the plume of the flight model SPT with this configuration is shown as Figure 4.

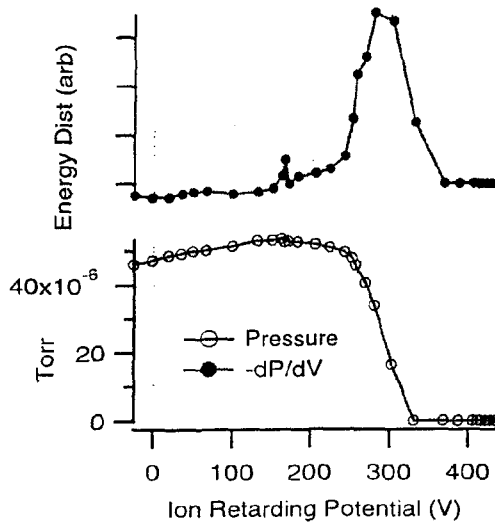


Figure 4. An RPA mode trace taken with the ionization gauge grid and filament deactivated at 0.5m radius from SPT-100 15 deg off thruster centerline.

Neutral Flux Mode. By analyzing the data shown in Figure 4 it was apparent that an ion retarding voltage of 500V was sufficient to repel all plume ions from entering the ionization gauge tube. In this mode of operation the ionization gauge filament and grid were activated as per normal operation of the tube, and the pressure indicated by the 919 controller was due entirely to neutral particles entering the tube volume.

Extensive data was taken in neutral flux mode in the plume of the flight model SPT-

100. This data is detailed in reference (1) and will be summarized here.

At a radial position of 0.5m from thruster exit plane the sensor was rotated from thruster centerline through 60 deg off centerline in 2 deg increments. The sensor output is shown as Figure 5.

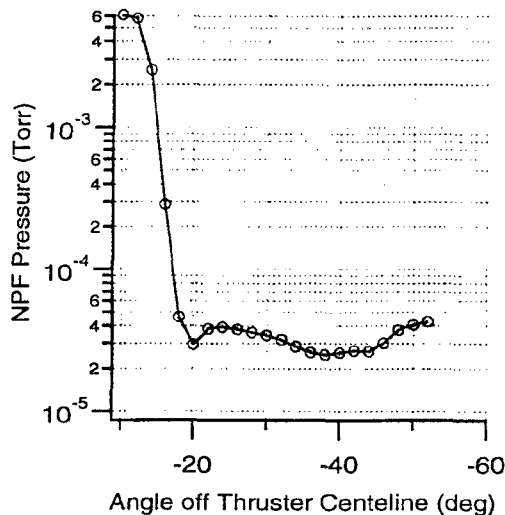


Figure 5. NPF sensor output in neutral flux mode at 0.5m radius from SPT-100 exit plane.

In order to prevent destruction of the ionization gauge filament, the 919 controller utilizes an overpressure protect set point. Above this pressure the gauge tube is deactivated. Although the set point for this function was increased to the maximum pressure permitted by the controller, the ionization gauge still exceeded this pressure for points within 10 deg of thruster centerline. Therefore data in this regime was not obtained.

By using equation (6), the raw output of the NPF sensor can be converted to units of neutral particle flux. In order to verify this data a separate analysis of neutral particle properties was used as a comparison. By combining measurements of convective heat flux due to both ions and neutrals with convective heat flux due only to ions it was possible to calculate the convective heat flux due to the neutral particles only.¹ This neutral convective heat flux is proportional

to the third moment of the neutral particle velocity distribution function while the NPF-measured flux is proportional to the first moment. A comparison between these two quantities is shown as Figure 6.

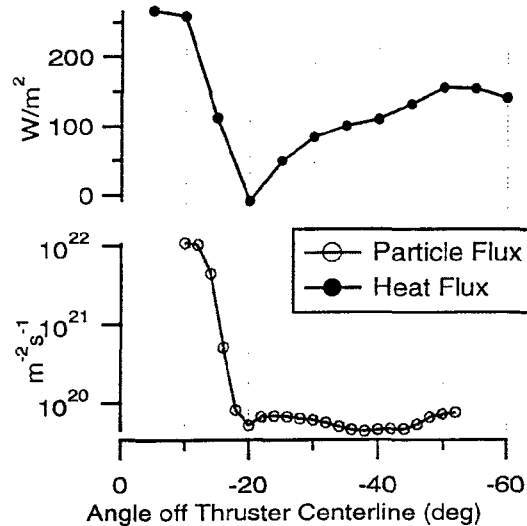


Figure 6. Comparison between NPF measured neutral flux and calculated neutral particle convective heat flux at 0.5m radius from the exit plane of the SPT-100.

Both the NPF probe data and the calculated neutral particle heating show almost identical trends. This suggests that the NPF probe-measured neutral particle flux is an accurate physical diagnostic.

Conclusion

Once the complication due to the ionization gauge grid voltage was identified and corrected the pressure (current) vs voltage trace obtained was typical of a traditional RPA. The trace obtained as Figure 3 is very comparable to RPA data obtained with a proven conventional RPA in parallel investigations.¹ For this preliminary probe design the measured ion energy distribution agrees well with the accepted form for this thruster. However, for future design iterations of this probe one improvement would be beneficial: currently the probe body and first grid are grounded. This imposes a disturbance to the flowing plasma, whose

potential may be several volts above tank ground. It would be advantageous to allow the probe body and first grid to float. This would minimize the plasma disturbance and reduce the uncertainty in the ion energy distribution function.

By comparing the NPF measured neutral particle flux to calculated values of neutral particle heating it is apparent that the trends are nearly identical. This agreement between separate diagnostic techniques strongly suggests that the NPF sensor output reflects true physical quantities. Although an accurate absolute calibration of the sensor was not performed, a simple free molecular analysis can be used to provide useful quantitative data.

The NPF probe measurements of neutral particle flux showed great sensitivity, however the range of the ionization gauge sensor proved too small for the plasma flows investigated in this study. For points within 10 degrees of thruster centerline at 0.5m radius the tube pressure induced by the neutral particle flow was higher than the maximum range of the ionization gauge. This caused a region of unobtainable data. Future versions of this probe technique will employ a vacuum pressure sensor with a measurement range more adequate for these type of diagnostics.

Acknowledgments

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